ATTACHMENT H

Kevin Weight

From: marilyn milum <marilynmilum@yahoo.com>
Sent: Wednesday, December 6, 2023 7:03 AM

To: Kevin Weight; Helana Ruter

Subject: Fw: 333 N 7th Ave.

Hi Kevin,

The letter below is from one of our brokers we have been using for the last few years representing the property at 333 N 7th Ave.

Please include this for our file concerning the hardship meeting. Thank you.

Sincerely, Marilyn Milum

Sent from Yahoo Mail for iPhone [mail.onelink.me]

Begin forwarded message:

On Tuesday, December 5, 2023, 1:25 PM, Justin Horwitz < justin.horwitz@svn.com> wrote:

Craig/Marilyn,

Please let this email serve as my insight on the value of the property and particularly how the value has been impacted by the existing structures over the course of 3+ years of attempting to sell your property. Generally speaking, the majority of developers that are willing to pay market pricing for development property are not structured for nor interested in pursuing sites that require historic preservation as part of a planned development. We are finding that most of the development community is interested solely in the land so that they can more freely plan a development with a clearer path to entitlements. We are currently asking \$9.2mm for the 2.39 AC site. That is ±\$88 PSF on land value which I believe is right in line with the market and I do believe the site would have sold long ago if it weren't for the complexities created by the push for historic preservation. It's hard to specifically gauge how much loss in value will occur if a developer is to incorporate these structures, but at this moment and certainly for the foreseeable future, we are finding that there is not any interested parties at any price.

Justin Horwitz, SIOR | Senior Advisor

SVN Desert Commercial Advisors | AZ O/I CRE Sales Team

5343 N. 16th St., Suite 100 | Phoenix, AZ 85016

Phone 480.425.5518 | Mobile 480.220.2674

justin.horwitz@svn.com | www.svndesertcommercial.com [svndesertcommercial.com]

AZ O/I LinkedIn [linkedin.com]

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From: marilyn milum <marilynmilum@yahoo.com>
Sent: Thursday, December 7, 2023 10:30 AM

To: Kevin Weight; Helana Ruter

Subject: Another break-in

The police were there again this morning. Homeless people sleeping in the building. More wasted resources of Phoenix PD

The police have to clear the property each time and make sure no one is inside, that is a big job. And a dangerous job. Swat units, canine units and the use of many officers was not meant to be used in this way.

Marilyn

<u>Sent from Yahoo Mail for iPhone [mail.onelink.me]</u>

From: marilyn milum <marilynmilum@yahoo.com>
Sent: Wednesday, December 6, 2023 11:14 PM

To: Kevin Weight; Helana Ruter

Subject: Fw: 333 N 7th Ave.

Hi Kevin,
Please add this letter of opinion from one of the primary brokers who has had it listed since 2019.
Than you,
Marilyn Milum

Sent from Yahoo Mail for iPhone [mail.onelink.me]

Begin forwarded message:

On Wednesday, December 6, 2023, 9:35 PM, Paul Borgesen <paul.borgesen@transwestern.com> wrote:

Marilyn,

It is my opinion that potential HP restrictions have kept multiple groups from making an offer on the property as it is not financially feasible to bring the current structure up to code as well as incorporate it into a new development. Most developers are not willing to take on the city or HP try and deal with this potential hurdle. Most groups hear that there may be an interest in the property from HP and that is the end of the conversation about the project. The property is zoned to allow apartments and is surrounded by new apartment development and this in my opinion would be the highest and best use for the land this would also bring you as the seller the highest value.

Paul Borgesen, SIOR

Senior Vice President

Capital Markets | Investment Sales

TRANSWESTERN

2501 E. Camelback Rd, Suite 1

Phoenix, Arizona 85016

Direct: 602.296.6377

Cell: 602.214.9033

transwestern.com [transwestern.com]

From: marilyn milum <marilynmilum@yahoo.com>

Sent: Tuesday, December 5, 2023 1:44 PM

To: Paul Borgesen <paul.borgesen@transwestern.com>

Subject: Fw: 333 N 7th Ave.

Hi Paul,

Please write us a similar letter and also state we missed that window of opportunities where Justin also told me earlier there may have well been multiple bidders, bidding war if HP buildings did not need to stay and interests rates and building rates were lower, etc

Thank you 🙏

P S this is being used in our hardship hearing and they wanted a statement of this sort for

An argument in addition to what you had provided previously.

Sent from Yahoo Mail for iPhone [mail.onelink.me]

Begin forwarded message:

On Tuesday, December 5, 2023, 1:25 PM, Justin Horwitz < <u>justin.horwitz@svn.com</u>> wrote:

Craig/Marilyn,

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Justin Horwitz, SIOR | Senior Advisor SVN Desert Commercial Advisors | AZ O/I CRE Sales Team 5343 N. 16th St., Suite 100 | Phoenix, AZ 85016
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From: marilyn milum <marilynmilum@yahoo.com>

Sent: Thursday, December 7, 2023 1:26 AM

To: Kevin Weight; Helana Ruter

Subject: 333 N 7th ave

Kevin,

You may wonder why two different brokers letters.

The two brokers have been working since 2019 on trying to sell our property on &th ave. Justin is still at SVN and Paul has chosen to change companies but they are still colisting since the two had it listed at the one compant when they were associates. You are possibly wondering why I am up so late my husband just left to check on the property on 7th since we are have had tresspassers coming in at night sleepng, and making messes, very hazadous.

After multiple breakends we secured the building further and he needs to check if the barriers we used are working or weather they are down, meaning they got in again. Marilyn

From:marilyn milum <marilynmilum@yahoo.com>Sent:Thursday, December 7, 2023 9:08 PMTo:Kevin Weight; Helana Ruter; marilyn milum

Subject: Invoice for one year

Please note that this is just for one year in which we extended it it for as long as we were under contract with the developer which was in the purchase agreement.

We have a different carrier now and at this moment I cannot locate our invoice.

From: marilyn milum <marilynmilum@yahoo.com>

Sent: Thursday, December 7, 2023 9:15 PM

To: Kevin Weight; Helana Ruter

Subject: insurance and taxes

I have been trying to download our tax amounts we have paid for the last two years. The site has been down.

It is public knowledge so I will say when I looked up a few days ago it was a little over \$40,000.00 and has been that amout approx., for the last two years.

Burns & Wilcox

150 Burns & Wilcox Center 14631 N. Scottsdale Road Scottsdale, AZ 85254

Insurance Quote

Date: Monday, June 13, 2022

Agency: NEATE DUPEY INSURANCE GROUP

Attn: ANDY DUPEY

Insured: MILUM TEXTILE SERVICES, INC

Application / Policy: APP80562253

We are pleased to submit our **QUOTE** for the above captioned insured. Please review this **QUOTE** carefully as coverage offered may be **DIFFERENT** than the coverage requested.

Proposed Policy Period: 6/14/2022 - 6/14/2023

Insurance Carrier: MT VERNON SPECIALTY INSURANCE COMPANY

Line of Business: PACKAGE

Price Breakout:

Premium: \$ 16,687.00

Carrier Policy Fee:

Carrier Inspection Fee:

 Brokerage Fee:
 \$ 1,700.00

 State Tax:
 \$ 551.61

 Stamping Fee:
 \$ 36.77

 Total Due:
 \$ 18,975.38

Agency Commission: 15.00%

Additional Subjectivities Required for Binding:

**FEES ARE FULLY EARNED

We appreciate the opportunity and look forward from hearing from you. Please call or e-mail us if you have any questions.

Melinda Lampson Burns & Wilcox

^{*}BROKER FEE WILL BE ADDED TO ANY A/P ENDORSEMENT OR AUDIT

Sent from Yahoo Mail for iPhone [mail.onelink.me]

From: marilyn milum <marilynmilum@yahoo.com>
Sent: Thursday, December 7, 2023 11:11 PM

To: Kevin Weight; Helana Ruter **Subject:** comments about 333N 7th ave

To:marilyn milum
Thu, Dec 7 at 11:06 PM
Kevin,

Please include this in the files. Thank you.

In case you are wondering why there are two different companies with our brokers, Justin and Paul were associates at the same firm before Paul went to work for a different firm. Both of these gentlemen have worked very hard to represent us and are still working on the listing. They have reported to us during the last several years their obstacles in selling our property that have been mainly the "Historical Preservation" ("HP") problem we have with the City that prevents successful sales efforts. Non one wants to buy such a property, which has been confirmed repeatedly by our brokers' many sales contacts.

Both have told us repeatedly that buyers are not interested in dealing with HP. We have also have had extensive feedback that it would be cost prohibitive to even try to save these structures.

We can no longer maintain them. It has caused a huge burden financially on us not to mention what is has done to us mentally and physically and our quality of life. We are septuagenarians that want to retire and the property is our retirement fund. My husband is ill and this is not equitable for us to bear the burden and expense of this property. It has been debilitating. We can no longer deal with these costs after four years of determined sales efforts. To impose such a mandate on two individuals is criminal or at least unconstitutional. We feel like someone has stolen our property and we have to bear the burden of paying a ransom for it as well as in the interim maintaining the property for the thieves.

Property taxes, Insurance, utilities, and to maintain such as broken windows, kicked in doors, trash, feces, graffiti, and our precious time.

Prop 207 was a clear indication that the citizens in Arizona do not want this abuse by government officials.

I hate to be so blunt, but that is now how we are feeling. We have earnestly tried to work with the City, we are in the fifth year of this tyrany and we are tired of all the red tape and emotional, physical, financial abuse we have been dealt by the city and it is truly time for the City to release this terrible burden. We feel the City has gone too far.

We are asking for fairness and justice. We also think there are political schemes behind this to stop more contemporary development rather than just to save a "priceless" building. There is no significant historic value to preserve, it is simply a manipulation and political effort by primarily a very small number of people who want to limit the density.

We have been damaged. These are dilapidated buildings that have outlived their use.

We believe this mandate has enough severe impact to our rights that it warrants compensation. The whole idea of "historic" is so subjective. The City should bear the cost and pay for it if they want a museum. Instead the City wants to give rich developers, taxpayers money at their whim and when the taxpayer will probably never see the inside of these buildings they want to keep. I s that fair and equitable? The City is on record telling us over and over do not pursue a demo permit, it will be turned down and told us they would not let the buildings go.

These are decaying buildings that need to be torn down for useful housing.

Since it has gotten cold now, the homeless are trespassing causing the SWAT teams, the canine teams and multiple officers (a dozen or more, yesterday), more today. Every time a break in occurs, we call the police they have to search the property and clear it. What a horrible use of our police resources. This is inviting criminal activity downtown. These officers could lose their lives going into the dilapidated buildings to search nooks and corners, closets, all room by room. These intruders are scared inside the building and could react with violence towards our City's finest.

Our freedom has been taken from us.

All of this has occurred because a very small number of people have a whim for saving these junky, old buildings with no modern times commercial, viable use.

Please help resolve these serious matters in the near future well before October by when these issues would be five years with out resolution.

A solution will also help our efforts to sell the property which has been substantially slowed by other substantially more complex matters than HP considerations for a building that does not seem to meet any realistic HP concerns compared to other HP properties.

We have reviewed the check lists requested and feel like most of these requests i.e., getting itemized construction costs to restore the 100 year old property are burdensome and are not applicable to the site. We never plan on using the property for another commercial laundry and to get an itemized costs would be so expensive and unrealistic it assumed these request would be for much smaller projects. To do what you are requesting would be a hardship and speaking with a contractor undoable.

It would be 10's of thousands of dollars and a waste of the contractors time and ours.

The contractors would not take us seriously.	

Thank you.

Sincerely,

Marilyn Milum

From: marilyn milum <marilynmilum@yahoo.com>

Sent: Friday, December 8, 2023 8:50 AM **To:** Kevin Weight; Helana Ruter

Subject: Property Taxes, Utilities, maintanence, insurance

Good morning Kevin,

TO add to file please

WE have calclated between \$ in excess of 100,000 a year saving the property for PHOENIX

Multiple insurance companies turned us down for insuranc

Insuring an empty building is risky and to keeping this place up is simply unsastainable for us

In the last couple of weeks we have turned off utilities

Aps we beleive has left one meter on by mistake.

We need to call them to turn off the last meter



NEWS

Historic riding ring collapses in Ashland

Rob Haneisen/Daily News staff

Published 11:01 p.m. ET Jan. 29, 2011 | Updated 1:50 a.m. ET Jan. 30, 2011

A massive outdoor riding ring, one of only four of its kind in the country and a local historic landmark, collapsed yesterday afternoon after years of decay and weighed down by several feet of recent snow.

The building off Olive Street was brought to the site in 1975 by Bill Sibson, 59, who disassembled the 60- by 120-foot building at the old Waseeka Farm on Chestnut Street with the help of family and friends. In 1979, he put it back together at his mother's Gleanmoura Farm, which means Mary's Glen in Gaelic.

The building had rare German lamella roof architecture, which gave the appearance of criss-crossed arches 25 feet above the riding floor. That structure held the roof up without needing any poles or beams in the center of the floor, which made it a perfect riding ring for horse lessons.

The building was first constructed around 1920 as a birthday gift for a daughter in the Powers family at Waseeka Farm, Sibson said.

Elegant in appearance at the height of its use, what remained yesterday was a pancaked heap of timbers and boards.

"I was walking my dog, and I heard a loud crack, and I saw it collapse," said Rory Warren, who lives on Clinton Street in Hopkinton and was one of the people who helped Sibson assemble the ring decades ago. "It's in seven sections, and it just came down like dominoes."

Warren and Sibson said snow stacked on the roof was definitely the reason for yesterday's collapse around 2:15 p.m., although the structure was in rough shape and had already started to lean before this winter.

Ashland Fire Lt. David Iarussi said neighbors heard the collapse and called police and fire departments. No one was in or near the building when it fell, and a huge cloud of dust flew up.

"It's the loss of a historic structure," Iarussi said.

Warren recalled the intricacy of the diamond-patterned roof and the simplicity of its white pine-board design, which allowed for interchangeable parts.

"Now it's gone - just a pile of wood on the ground," Warren said.

When the family disassembled the old ring in 1975, every bolt, nut and shingle, plus the lamella planks, were stored in garages and barns - trucked over from Chestnut Street in the family station wagon - until they could be painstakingly put back together.

"We had to cat's-paw every nail out of it," Sibson said.

Sibson said he thinks the other three lamella buildings in the country were made into aircraft hangars.

"I knew that it was going to go ... but I didn't think it would crush so flat," Sibson said.

Sibson said he hopes the town will let him salvage some of the lamella boards this spring so he can one day build a small cabin with the historic pieces.

(Rob Haneisen can be reached at rhaneis@cnc.com or 508-626-3882.)

From: marilyn milum <marilynmilum@yahoo.com>
Sent: Thursday, December 21, 2023 12:30 AM

To: Helana Ruter; Kevin Weight
Subject: A little more complicated Lamella

https://www.google.com/gasearch?q=lamella%20roof%20collapses&tbm=&shem=rime&source=sh/x/gs/m2/5#fpstate=ive&vld=cid:2426b60c,vid:YsJqJKtrwlk,st:0 [google.com]

Sent from Yahoo Mail for iPhone [mail.onelink.me]

From: marilyn milum <marilynmilum@yahoo.com>
Sent: Thursday, December 21, 2023 12:40 AM

To: Helana Ruter; Kevin Weight

Subject: Complicated

Politically I'm not sure the Lamella enthusiast

Would be as supportive if they knew Zollinger was part of the Nazi party . Is the public going to be accepting of the Nazi link with the Nazi example of superior engineering...?

https://www.youtube.com/watch?v=YsJqJKtrwlk [youtube.com]

Sent from Yahoo Mail for iPhone [mail.onelink.me]

From: marilyn milum <marilynmilum@yahoo.com>
Sent: Thursday, December 28, 2023 3:48 PM

To: Helana Ruter; Kevin Weight

Subject: Roof collapse

Not sure if I sent this one

3:47



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ancy, so likely not too critical unless a whole series of unese are breaking.

Is this in snow country? Maybe some unbalanced load caused flex failures.

LaminatedTimber (Materials)

22 Nov 19 16:07

Wooden lamella roofs were known for not handling unbalanced roof loadings and there were many, many failures and collapses. Some lamella roofs are doing fine. I know of at least five in Wisconsin that have been performing fine for 80 or so years. A school gym was built in Fargo, ND in the 1960s with solid wooden lamella and it collapse within months of completion. The guy who sat at my desk before me bid it as a glued laminated timber radial rib dome and when told it would go to lamella he "warned" the general contractor about problems. Hate to say "I told you so" but that's what happened. Luckily there were no injuries in the collapse.

I cannot help with your design and repair method.

Lamella is a real groovy looking type of material. The historical and iconic Brown Derby in LA has a lamella roof.

Andreas

McSEplic (Structural)

24 Dec 19 00:18

The lamella roof was developed and patented by Friedrich Zollinger after WWI as a way to address the housing shortage (it uses about 46% of the lumber compared to the previous wood construction techniques used in Germany.

infolinks

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eng-tips.com — Private

Sent from Yahoo Mail for iPhone [mail.onelink.me]

Justin Horwitz - SVN Paul Borgesen - Transwestern 5343 N. 16th St. #100 Phoenix, AZ 85016

Helena Ruter City of Phoenix Historic Preservation Officer 200 W. Washington St. Phoenix, AZ 85003

Dear Ms. Ruter.

On behalf of Paul Borgesen, Senior Vice President with Transwestern, and myself, Justin Horwitz, Senior Advisor with SVN, please accept this letter in relation to the Milum Textile property located at 333 N 7th Ave, Phoenix, AZ 85007.

Paul and I are commercial real estate agents with substantial experience selling development properties particularly in Downtown Phoenix. In April 2020, we began actively listing the subject property for sale and to this point, we have been unsuccessful in solidifying a buyer for the property. Throughout the course of our listing, the subject property has received good interest from prospective buyers. However, following initial conversation with various zoning attorneys, the overwhelming majority of prospective buyers do not pursue the purchase of the property due to concerns over multiple City of Phoenix interests in historical preservation of several major structures. This has presented a number of challenges, but a few of the main issues are as follows:

- 1. The process is relatively more complex. Incorporating historical structures on any site adds multiple layers of processes to the design, planning, and zoning stages that eliminates a number of quality developers. The majority of developers we have presented the site to ultimately are not equipped to handle an abnormal development process or do not have an interest in taking on the risk given the amount of unpredictable expenses in the pre-development and construction phases. Simply put, our experience has been that most developers want a "cookie cutter" site that allows them to repeat their typical planning, zoning, design, and construction processes. This site does not allow for that with historical structures in place.
- Historical structures in their current location dramatically hinder design capabilities and limit a
 developers ability to maximize density in its planned development. This directly impacts the
 ultimate price they are willing to pay for the property.
- 3. Retaining the structures creates liability that adds significant costs to a project making it infeasible. The existing structures are quite old and have had years of industrial wear and tear placed on them. Again placing more unpredictability and liability into a project than any prospective buyer has been willing to take on.
- 4. Items 1-3 listed above are primarily addressing the items of contention solely from a redevelopment perspective. We have also spent countless hours over these last few years attempting to identify end users that have an interest in retaining and using the existing structures. While we have had groups acknowledge the unique elements of the structures and have a vision for an end use, the estimated costs of renovations steer groups away from pursuing a purchase of the property. To be more specific, we had a licensed general contractor walk the property and while we could not get a specific bid, we were provided with a rough estimate upwards of \$10MM to simply bring the building up to code. This was purely contemplating the

costs to bring the building up to current code (i.e. remove and replace the existing complex utility system, replace the electrical system, treat any asbestos due to the age of the structure, sure up the roof system that requires significant inspection to even understand its current condition, redesign and replace the entire HVAC system, and address general ADA items just to name a few). Again, this is only to bring the building to code in a "vanilla shell" condition and does not include the cost to customize the interior layout for an end user.

The main purpose of this letter is to attempt to identify how much the property is worth as raw land with all structures demolished as opposed to its value with various structures historically preserved. This proves to be a rather difficult task. While we have contemplated comparable sales for land sites in the immediate area (please see Exhibit "A" - Comparable Sales enclosed), it's virtually impossible to identify a value for the property with structures in place. As mentioned above, in over three years of tireless efforts to find a buyer, we have come up empty handed. One could argue that there is no buyer in the foreseeable future for this property at any price given the significant cost of improvements due to the issues listed above. Alternatively, as it pertains to the potential value of the land with all structures demolished, we have identified seven comparable sites based on location, land size, and/or intended use for the property. The sales comparables range from \$111 PSF to \$316 PSF on land value only. The average of the seven comparable sales is \$201 PSF. Relative to the subject property, one could argue that without any historically preserved structures, the land's value is upwards of \$21MM for the 2.39 AC of land. Our current asking price for the property is \$9.2MM with no qualified parties pursuing at this price. We do however have a number of groups that have indicated a high level of interest in the property if the owner of the property can deliver the property with either a demo permit for the entirety of the site or with all structures fully demolished.

In closing and as mentioned above, without any prospective buyers to currently reference, it is difficult for Paul or I to determine the value of the property with historically preserved structures in place. However, it is safe to assume that the loss in value to the property would be significant relative to the comparable sales in the area.

Please feel free to reach out should you have any questions.

Sincerely,

Justin Horwitz

Just Hout

Paul Borgesen

Exhibit "A" - Comparable Sales

<u>Site</u>	Land Size	Sale Price/ Land PSF	Sale Date	<u>Notes</u>
520 S. 5th St. Phoenix, AZ 85004	2.56 AC	\$17,300,000 \$155 PSF	12/8/23	Existing parking lots; Covered land purchase.
840 N. Central Ave. Phoenix, AZ 85004	1.11 AC	\$10,500,000 \$217 PSF	12/8/23	Part of assemblage.
343 E. Lincoln St. Phoenix, AZ 85004	1.00 AC	\$8,643,000 \$198 PSF	10/2/23	Future use for Phoenix Suns/Mercury.
114 E. Portland St. Phoenix, AZ 85004	0.64 AC	\$8,820,000 \$316 PSF	2/2023	Future development site.
510 E. Lincoln St. Phoenix, AZ 85004	1.60 AC	\$9,500,000 \$136 PSF	1/5/23	Future development site.
601 N. Central Ave. Phoenix, AZ 85004	1.83 AC	\$22,000,000 \$275 PSF	3/2/22	Future development site.
362 N. 3rd Ave. Phoenix, AZ 85003	0.76 AC	\$3,700,000 \$111 PSF	12/29/21	Future development site
AVERAGES		\$201 PSF		

From: marilyn milum <marilynmilum@yahoo.com>

Sent:Monday, January 8, 2024 2:22 PMTo:Kevin Weight; Helana RuterSubject:Important information

Please add this to our HP file and please make available to HP commission and city council members.

We feel like the city of Phoenix has not done their due diligence in insisting on keeping structures when they know virtually nothing about their safety.

This is very risky. Sincerely, Marilyn Milum

Sent from Yahoo Mail for iPhone [mail.onelink.me]

From: marilyn milum <marilynmilum@yahoo.com>

Sent: Monday, January 8, 2024 2:17 PM

To: Kevin Weight; Helana Ruter; Roger Strassburg

Subject: Sensitivity analysis of Kiewitt-Lamella reticulated domes due to member loss -

ScienceDirect

https://urldefense.com/v3/__https://www.sciencedirect.com/science/article/abs/pii/S0143974X21004983__;!!Lkj WUF49MRd51_ry!YS_y5Q2hnymJZQY8-OEQ-SbJIQ36tP5gb5x5whpMIF5Upyv_9NY1x9eMw_Z-NMfaAnWPo1FVyLmapJpS4ssrj66u9Lqs-Q\$

Sent from my iPhone



Journal of Constructional Steel Research

Volume 188, January 2022, 107016

Sensitivity analysis of Kiewitt-Lamella reticulated domes due to member loss

Zubin Zhang, Ruiyi Gu, Haiqin Wang 🙎 🖂

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https://doi.org/10.1016/j.jcsr.2021.107016
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Abstract

In this paper, the sensitivity analysis of Kiewitt-Lamella (K-L) reticulated domes with different parameters is carried out. The sensitivity of the dome due to gradual and sudden member loss is analyzed in both static and dynamic aspects, which will clarify the distribution rule of the member sensitivity and provide a reference for future study on the K-L reticulated dome. The results show that the member sensitivity of the K-L reticulated dome with different design parameters has similar regularities. For the members in the same ring, the sensitivity of the latitudinal members is larger than that of the diagonal members. For the members in different rings, the sensitivity of the inner ring members is larger than that of the outer ring members. In addition, the static analysis shows that the latitudinal members closer to the radial members are more sensitive than those apart from the radial members, and the diagonal members paralleling to the radial members are more sensitive than the unparallel ones. The dynamic analysis shows that the K-L reticulated dome will experience a local internal force redistribution after a member's sudden loss and finally reach a stable equilibrium state on the new load transfer path without overall progressive collapse. According to the research results, a better scheme for strengthening the structure is proposed: increasing the cross-section of sensitive members is more effective in improving the structural stability than important components.

Introduction

The single-layer reticulated dome is usually used as the supporting structure for the roof of large cylindrical storage tanks in the petrochemical industry. With the development of the storage tank roof to a large span, the single-form reticulated dome used in realistic projects has been restricted. For example, a great many members of the Lamella reticulated dome are gathered at the apex, and the structure of the node is complicated, so some members must be properly removed, which changes the force transfer path and then adversely affects the overall structural strength. In addition, for the Kiewitt reticulated dome, the number of nodes in the outermost ring increases as the span increases, and the difficulty of construction also increases. Combining these two types of reticulated domes to form a Kiewitt-

Lamella (K-L for short) composite single-layer reticulated dome can effectively solve the above-mentioned problem for the large-span single reticulated dome [1,2].

At present, scholars worldwide have conducted many kinds of research on Kiewitt reticulated domes and Lamella reticulated domes, and there are also many studies on structure sensitivity. Gao *et al.* [3] discussed the problems of redundancy related to the Alternate path method, and the sensitivity of the Kiewitt single-layer reticulated dome was explored. Han *et al.* [4] evaluated the redundancy and progressive collapse performance of the large-span Lamella single-layer and double-layer domes based on the ultimate bearing capacities in both the original and damaged status. Sebastian [5] and Chen *et al.* [6] proposed the sensitivity index based on the internal force responses of members to identify sensitive members, which plays an important role in evaluating structural safety and reliability. However, the research on K-L reticulated domes needs to be strengthened, especially the sensitivity analysis of them. It plays an important role in determining the context of the system, optimizing algorithms, reliability evaluation of system performance, and structural redundancy research [7,8]. In fact, sensitivity analysis is a major prerequisite in the establishment of structural optimization, reliability evaluation, and parameter identification [9]. However, there are many members in single-layer reticulated shells, and the effects of different members on the elastoplastic stability of the structure are often different, and it has been proven that the failure of some critical members may lead to the progressive collapse of the space structure [[10], [11], [12]]. Therefore, the elastoplastic stability of the K-L reticulated dome on the member sensitivity is worth studying.

In realistic construction, collapse accidents of space structures have occurred around the world. For example, in 2004, due to the perforation of the ceiling in the terminal of the Charles de Lego Airport, the critical metal connecting members could not continue to bear the weight, and eventually collapsed. In 2014, South Korea continued to snow for many days, and the final snow load reached $0.9 \, \text{kN/m}^2$, which far exceeded the design load value. A space structure that does not consider this effect may collapse suddenly due to partial damage caused by the failure of a member without significant deformation in advance. If a member loses stability, it will inevitably affect other members connected to it. Therefore, the stability of a specific member cannot be analyzed in isolation, and the interaction of other members should be comprehensively considered and determined from the overall structural analysis [13]. Pandey et al. [8] proposed a redundancy assessment method based on sensitivity analysis. In this method, the response of the structure under design load is used as the research object, the member loss is used as the analysis parameter, and the member sensitivity and the structural redundancy are quantified theoretically with a numerical method. Subsequently, on this basis, the Japanese Society of Steel Construction considered the buckling of a single member, and made this redundancy assessment method further suitable for large-span space structures [14]. In recent years, Shekasheband et al. [15] divided the member loss into gradual and sudden loss, and carried out a numerical investigation into the static and dynamic response of tensegrity systems in the event of gradual and sudden member loss.

Therefore, in this paper, the sensitivity analysis of large-span reticulated dome due to member loss refers to the above method. One member is removed each time, and the static and dynamic response of the domes in the event of gradual or sudden member loss is investigated. The response and characteristics of the studied structures include the load-deflection response in static analysis and displacement-time history of the structures in the dynamic analyses. In addition, an effective measure for improving structural stability is discussed, which will provide a reference scheme for the designers.

Section snippets

Analysis model

Take one of the K-L reticulated domes as an example to illustrate the analysis model. As shown in Fig. 1(a), the span is 60m, the rise-to-span ratio is 1/4, the symmetrical sectors are 8, and the frequencies are 12 (Kiewitt: Lamella is 9:3).

For rings from the inside to the outside, they are marked as the first to the twelfth ring. Since the members are directly in contact with the top skin and need to bear the bending moment, it is appropriate to use I-beams but not steel circular pipes [16]. ...

Sensitivity analyses of the K-L reticulated dome due to gradual member loss

A unique K-L reticulated dome can be determined when the span, the rise-to-span ratio, the number of rings, the symmetrical sectors, and the frequency ratio are determined. Currently, most of the large-span K-L reticulated domes are 12 rings. Therefore, the sensitivity analysis of the K-L reticulated dome with 12 rings is carried out, and both geometric and material nonlinearities analyses are performed to obtain the ultimate bearing capacity. Considering the influence of latitudinal members...

Sensitivity analyses of the K-L reticulated dome due to sudden member loss

The previous part discussed the static load-bearing capacity of the dome in the event of gradual member loss. Practically, when losing a member in the structure which is under load, energy stored in this member is released, and this induces a state of transient vibration in the structure. In order to compare the response of the damaged dome caused by the sudden member loss, this section introduces the results of the dynamic analysis. The K-L reticulated dome, which with 60m spans, 1/4...

Distribution and influence of sensitive members and important members

According to the previous analysis, the sensitivity of the members in different areas is different. In this paper, the sensitive member is defined as the member with a sensitivity larger than 5%, and the important member is defined as the member with a negative sensitivity. The important member loss can contain or block continuous structural damage. The sensitive members and the important members are shown in red and blue, respectively, as shown in Fig. 18.

It verifies that the sensitive members ...

Conclusion and discussion

In this paper, a numerical investigation into the static and dynamic response of the K-L reticulated domes in the event of gradual and sudden member loss is carried out. The results of this study are used to obtain certain conclusions regarding the sensitivity of the K-L reticulated domes to member loss. In addition, the distribution of sensitive members and important members is distinguished, and a more economical structural reinforcement scheme is proposed and verified according to this rule, ...

Funding

This research did not receive any specific grant from funding agencies in the public, commercial, or not-for-profit sectors....

Declaration of Competing Interest

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper....

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Static stability analysis of Kiewitt-Lamella single-layer reticulated shells

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Sensitivity analysis on load carrying capacity of K6 single-layer reticulated domes to member failure Architect. Struct. Design (2009)

Q.H. Han et al.

Progressive collapse analysis of large-span reticulated domes

Int. J. Steel. Struct. (2014)



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Sensitivity of post-buckling behaviour of single layer reticulated shells to loading and member imperfections ¬

2022, Structural Stability Research Council Conference 2022, Held in conjunction with NASCC: The Steel Conference

Application of Sensitivity Analysis to Progressive Collapse Resistance of Planar Truss Structures 7

2022, Applied Sciences (Switzerland)

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Reading A Building: More Roof Size-Up

When reading a building, do you include the roof in your size-up, and if so, what are you thinking about? To assist with this question, let's consider some important factors that are worthy of your consideration.

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By John W. Mittendorf

When reading a building, do you include the roof in your size-up, and if so, what are you thinking about? To assist with this question, let's consider some important factors that are worthy of your consideration. Obviously, some factors will be dependent on the type of roof construction in your particular area, however, West Coast roofs and East Coast roofs have a lot in common in both construction methods and styles.

Open Web Bar Joist

Open web bar joist (or metal deck) roofs are the commercial roof of choice in the Midwest and Eastern portions of the country and are primarily steel truss construction underneath a metal decking (Q decking). The metal decking is covered by built-up layers of insulation material, tar, and composition. As steel loses it's strength around 1,000 degrees, such roofs have a quick failure rate with minimal warning, and suppression personnel should be aware of these hazards. However, another more subtle hazard is that fire can propagate between

the metal base and the composition covering, enhancing the spread of fire with minimal visible warning signs.

Older Truss Roofs

These roofs are found anywhere in the country and on various types and sizes of commercial buildings primarily constructed during the 1800s until the 1950s – until the introduction of the flat roof with its numerous variations. The older truss roofs were normally constructed with a "large" size of wooden truss members, 1 x 6-inch sheathing as a roof base/covering, and can be found in numerous styles as follows:

Bridge Truss: This type is recognizable by it's characteristic sloping sides, ends, and flat top.

Gable Truss: This type is also identified by it's gable or peaked roof design. Parallel Chord Truss: This roof looks similar to other types of flat roofs but can be found on older buildings and is constructed from a "large" size of truss members (compared to newer lightweight truss members).

Lamella: Although this roof can be similar in external appearance to other types of arch roofs, it is significantly different as it was constructed in an egg crate – geometric or diamond-patterned – design. This roof can be found on gymnasiums, recreational buildings, large supermarkets, etc.

Tied Truss: This arched roof uses metal tie rods to give lateral support to the walls of the building. Tie rods with turnbuckles are used below each arch member (as there is no bottom chord) to ensure that the arches do not push the exterior walls outward. With this mind, it is easy to see if fire exposes metal tie rods in this type of roof, a collapse of the building is more than a possibility. Hint: If you are ever inside a building and observe this type of roof construction, make a mental note for future reference as it may save your life!

Bowstring Truss: Most firefighters are familiar with the "bowstring truss" roof as numerous fire service writers have appropriately written on the hazards of this common roof. Interestingly, whether you are a firefighter on the East or West coast (or anywhere in between), you will likely have this roof in your municipality. It is constructed of "large-size" wooden members (Note: most wooden members used in these older truss roofs were "rough-cut" or full size lumber and used steel plates and bolts for connectors) with 1 x 6-inch sheathing roof decking. Multiple firefighter deaths attributed to this specific roof have cautioned firefighters to assume a defensive position if a working fire is encountered.

John W. Mittendorf joined the Los Angeles City (CA) Fire Department (LAFD) in 1963, rising to the rank of captain II, task force commander. In 1981, he was promoted to battalion chief and in the year following became the commander of the In-Service Training Section. In 1993, he retired from LAFD after 30 years of service. Mittendorf has been a member of the National Fire Protection Research Foundation on Engineered Lightweight Construction Technical Advisory Committee. He has provided training programs for the National Fire

Academy in Emmitsburg, Maryland; the University of California at Los Angeles; and the British Fire Academy at Morton-in-Marsh, England. He is a member of the editorial advisory board of Fire Engineering and author of the books Truck Company Operations (Fire Engineering, 1998) and Facing the Promotional Interview (Fire Engineering, 2003).

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A lot of the integrity is no longer there, not up to US safety standards.

11:13



roofs [1]. In 1925, the idea spread to America as well [3].

1.2 Previous Roof Failures

Due to the curve of the lamella roof, these structures are susceptible to failure from high wind loads. In 1926, hurricane winds caused the destruction of two lamella buildings in

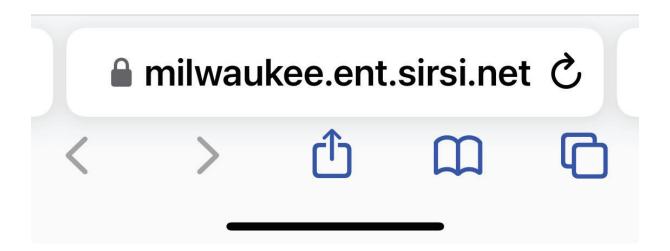
19

Florida with one roof being torn off completely and deposited upside-down a few hundred feet away [1, 7].

Lamella roof construction was principally in use from its introduction by Zollinger up until the 1940s, with construction mostly halted because of wind failures. Engineers at the time used a wind load of 10 psf on the vertical projection for normal wind areas and 37.5 psf for high-wind regions. The latter wind pressure correlated with a 130 mph wind speed, the highest measured in that era [1].

In modern times, the wind loads on a curved roof are better known thanks to modern wind tunnel testing and computer simulations. It is now known that wind flowing over a curved roof creates unlift (similar to an aircraft wing), not simply a uniform horizontal.

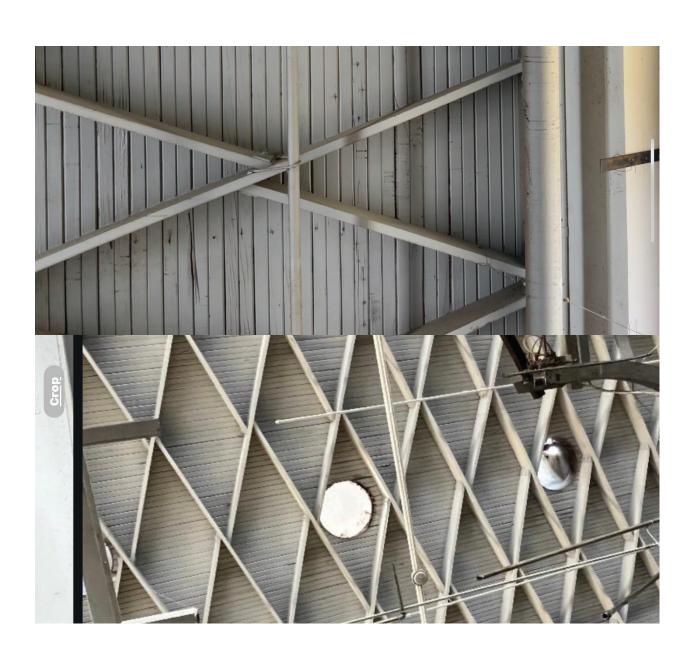
load on the vertical projection. This creates a very different loading condition than the horizontal load which could potentially explain the failures of some lamella roofs in the first half of the 1900s.

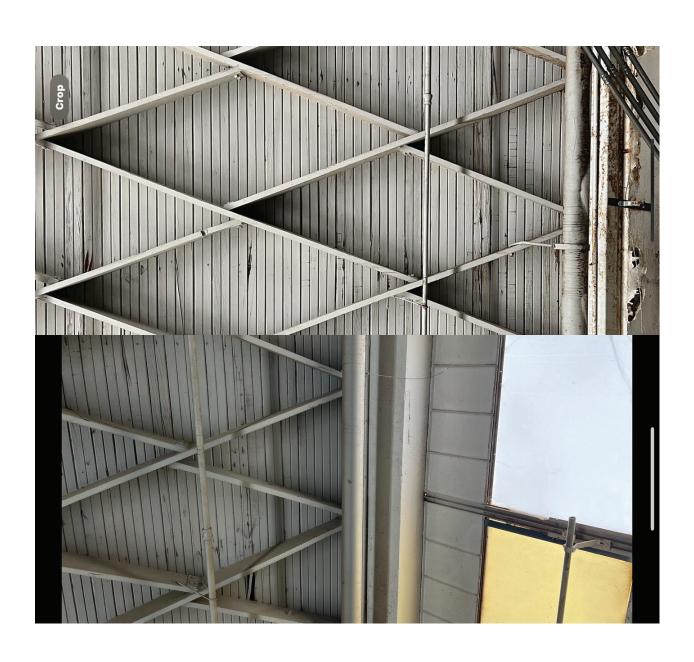


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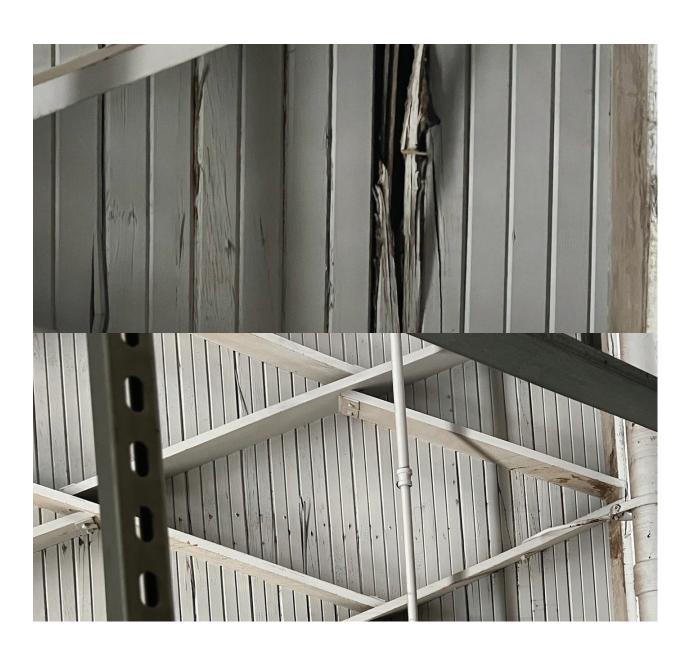


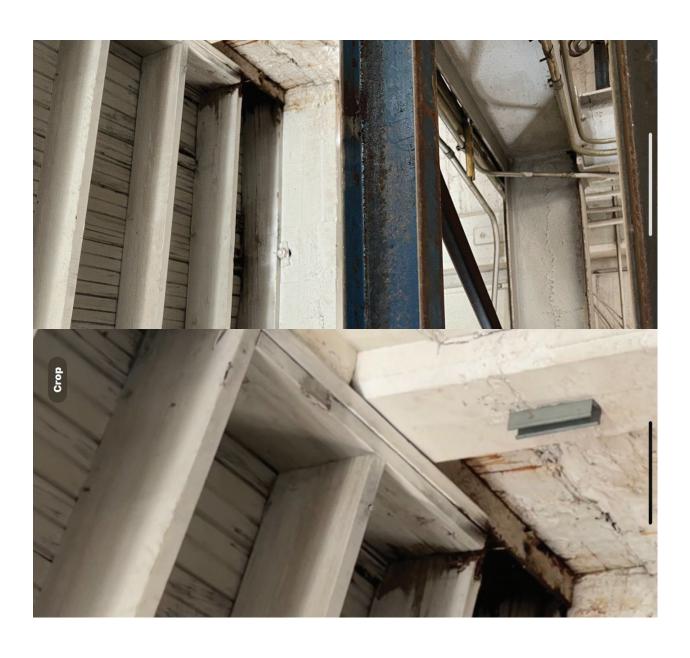
















Engineering Structures

Volume 188, 1 June 2019, Pages 111-120

Identification of critical members for progressive collapse analysis of single-layer latticed domes

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Highlights

- Distribution of critical member in different single-layer dome types is investigated.
- Method is proposed to identify critical members based on collapse mechanism of domes.
- Proposed identification method performs excellent compared to the other two methods.
- Methods are proposed to increase progressive collapse resistance of single-layer dome.

Abstract

This paper presents a method to identify the critical member in a single-layer latticed dome, which in the context of progressive collapse is defined as the member whose removal causes the most severe damage. The distribution of critical members in four typical types of single-layer latticed domes, including the Kiewit dome, the Ribbed dome, the Schwedler dome and the Lamella dome, is investigated through a comprehensive Alternate Path analysis scheme composed of hundreds of individual <u>dynamic nonlinear</u> analyses. The Alternate Path analyses also confirm the progressive collapse mechanism of single-layer latticed domes, i.e., the nodal snap-through buckling at either end of the initially removed member. On this basis, a critical member identification method is established, using an index that implicitly estimates the relative vulnerability to node buckling following the removal of a member to determine the criticality of this member. This method along with two other methods, using either static <u>axial force</u> or <u>free vibration</u>

response, are evaluated via comparison against the nonlinear dynamic Alternate Path analysis results, and this proposed method shows a beyond-compare accuracy. Furthermore, based on the established understanding of the progressive collapse mechanism and the factors influencing the node buckling resistance, three methods for increasing the progressive collapse resistance of single-layer latticed domes are presented.

Introduction

Structural domes have a long history in the built environment as an important design feature of many famous structures around the world. Since the days of Ancient Rome, permanent structures roofed with domes made from natural stones, bricks and concrete (found in ancient buildings in several countries) have been constructed [1], and continual improvements have been made in constructing and analysing these types of domes, even in the new millennium [2], [3], [4]. Starting from the last century, various forms of single-layer latticed domes built with steel become popular because they are capable of achieving larger span, and have been widely used for buildings with large-span roofing, such as sports stadiums and exhibition centres. For their efficient and safe design, extensive studies have been conducted to investigate the static and dynamic stability [5], [6], the earthquake resistance [7], [8] and the optimum topological design [9], [10] of dome structures. However, the progressive collapse of single-layer latticed domes remains an area in need of research. Since the collapse of the World Trade Centre towers in 2001, a substantial body of research has been carried out on the progressive collapse of frame structures [11], [12] and, more recently, certain types of roof structures, mainly truss-type roofs [13], [14], [15], [16], while research on single-layer latticed domes is very limited. Zhao et al. [17] conducted progressive collapse tests on two single-layer latticed Kiewitt domes, showing the collapse of a dome can be caused merely by the loss of a single critical member. This demonstrates a need for further investigation of current analysis and design methods for single-layer latticed domes.

The Alternate Path (AP) Method, in which a load-carrying element is removed to evaluate the structural capacity in resisting local damage, is arguably the most appropriate and widely accepted method for studying the structural progressive collapse performance. The previously mentioned tests by Zhao et al. [17] and the tests by Xu et al. [18] on two single-layer latticed domes, a Kiewit-Lamella dome and a geodesic dome, were carried out following the concept of the AP method. Han et al. [19] numerically employed the AP method to investigate the collapse resistance of a Schwedler Monoclinal dome subject to the loss of several members or nodes. When applying AP method to a structure constructed with many members, an important issue to be addressed is how to identify the critical members (sometimes referred to as the sensitive members), i.e., the members whose removal cause the most severe damage. Accurately identifying the critical members ensures the most dangerous collapse scenarios are taken into account, and also helps to reduce the computational cost. The current guidelines for progressive collapse design of frame structures [20], [21] stipulate that, critical members of a building include the columns at corners, as well as columns located at the middle of both the long side and the short side of the building.

However, thus far there is no such codified recommendation for single-layer latticed domes, and therefore an efficient method for identifying the critical members in single-layer latticed domes, the types of which are numerous and the geometry of which can be complicated, is of significant interest. Some researchers recommended FE-based methods [18], [19], [22], in which, to evaluate the criticality of a dome member, the structural response of the remaining structure following the removal of this member is obtained through static linear or nonlinear FE analysis, and is compared with that of the intact structure. Analysis is performed on each member in the dome, and the critical members are identified as those whose initial failure cause the most severe reduction of structural performance. Such methods are accurate in terms of finding the critical members, but cannot be considered as an efficient strategy for selection of removed member in the AP method because they are, in actuality, static AP method themselves. An attractive method for identifying the critical members should require no or just one simple analysis of the structure. Xu [23] studied several single-layer Kiewit domes, and suggested that the critical members were in lines with members having the most pronounced response, with respect to the fundamental vibration mode, in an eigenvalue vibration

analysis of the intact dome. This selection criterion, although very simple, is not established on account of the failure mechanism under progressive collapse and thus, as will be shown later in this study, has a poor accuracy.

This paper presents a method to identify the most critical members in a single-layer latticed dome on the basis of the progressive collapse mechanism. In the tests by Zhao et al. [17], nodal snap-through buckling at either end of the initially removed member, referred to as "node buckling" in this present study, was regarded as the collapse mechanism for the tested Kiewit dome models. This conclusion is first examined for other types of single-layer latticed domes as well as Kiewit domes with different geometries through an extensive nonlinear dynamic AP analysis, which also extends the pool of the experimental results, providing an overview of the critical member distribution in different types of single-layer latticed domes. The critical member identification method is then proposed, using an index that implicitly estimates the relative vulnerability to node buckling following the removal of a member to determine the criticality of this member. This method along with two other methods, using either static axial force or free vibration response as identification index, are evaluated via comparison against the nonlinear dynamic AP analysis results. Furthermore, based on the established understanding of the progressive collapse mechanism and the factors influencing the node buckling resistance, several methods for increasing the progressive collapse resistance of single-layer latticed domes are suggested.

Section snippets

Prototype structure

In order to gain a comprehensive understanding of the distribution of the critical members, four single-layer latticed domes being the most common types, i.e., a Kiewit dome, a Ribbed dome, a Schwedler dome and a Lamella dome, are investigated by means of nonlinear dynamic AP analysis.

The Kiewit dome as shown in Fig. 1a has a constant span of 40 m, which is a moderate span for single-layer latticed domes. The rise-span ratio affects the stability characteristic of the dome and thus can have...

Criticality index based on node buckling

On the basis of the progressive collapse mechanism, a method for identifying the most critical members in a single-layer latticed dome is proposed in this section. Fig. 5 illustrates a schematic diagram of a node undergoing snap-through buckling. Subjected to a vertical point load, the end node of the removed member remains connected to the adjoining members. For an intact latticed dome, its in-plane and out-of-plane stiffness can be approximated by converting the latticed dome into an...

Application of criticality index in determining critical members

For the prototype Kiewit dome, the criticality indices of all members are determined using Eqs. (6), (7), (8), (9), (10). Table 1 shows the results. It is observed that among all 44 members, those ranking in the top 10% in terms of criticality index are all the most critical members determined by the nonlinear dynamic AP analysis, and those ranking in the top 20% almost cover all the members in the first two criticality grades. Therefore, the proposed criticality index shows an appreciable...

Methods for increasing the progressive collapse resistance of single-layer latticed domes

The progressive collapse resistance of a single-layer latticed dome is limited by the collapse load of the most critical members. Therefore, by reducing the criticality indices of these members, the progressive collapse resistance of the

dome can be improved. This is achievable through the following methods.

The first is to enhance dome members. Once the most critical members are identified, a higher overall progressive collapse resistance of the dome can be achieved by enhancing these members...

Conclusion

This paper presents a method to identify the most critical members for progressive collapse analysis and design of single-layer latticed domes. A comprehensive finite-element Alternate Path analysis is performed on four typical types of single-layer latticed domes, i.e., the Kiewit dome, the Ribbed dome, the Schwedler dome and the Lamella dome, demonstrating that the nodal snap-through buckling at either end of the initially removed member is the major progressive collapse mechanism. On this...

Acknowledgement

The work presented in this paper was funded by the National Natural Science Foundation of China (Grands No. 51678432 and No. 51708417)....

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Further information on research data
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System-reliability-based disaster resilience analysis: Framework and applications to structural systems 2022, Structural Safety

Citation Excerpt:

...However, when the structural system becomes highly sophisticated and local damage or an element failure plays a critical role in the structural collapse, cascading failure mechanisms need to be considered. On the other hand, various redundancy indices considering the effects of cascading failures have also been developed [63–69]. The methods perform dynamic analyses during which damaged elements are removed to simulate load redistributions....



Elasto-plastic buckling behaviour of aluminium alloy single-layer cylindrical reticulated shells with gusset joints

2021, Engineering Structures

Citation Excerpt:

...Therefore, the elasto-plastic buckling behaviour of reticulated shells should be investigated exhaustively. Existing research observations indicate that the main parameters that influence the elasto-plastic buckling behaviour of reticulated shells include the shell type, span, rise-to-span ratio, load distribution, member section, support condition, member curvature, initial geometric imperfection, material property, and joint stiffness [3,13–21]. The influence of other parameters on the buckling capacity can be understood intuitively....

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Experimental and numerical study on cable breakage equivalent force in cable-stayed structures consisting of low-relaxation seven-wire steel strands

2020, Structures

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...However, a data acquisition frequency of 0.5 kHz was used and the location of data acquisition was also close to the strand breakage location. In the published results [5,9,10,12–16], there is a discrepancy between the assumed CBEF(t) related to different cable-stayed structures and also its duration. On the other hand, none of these simple assumptions have been verified....

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2019, Engineering Structures

Citation Excerpt:

...The last several decades have witnessed plenty of progressive collapse incidents of building structures, leading to a growing interest in this disproportional failure phenomenon among academic and engineering communities. As a result, a great number of studies have been conducted to investigate the progressive collapse resistance of multi-storey frame structures [1–3] and, more recently, roof structures [4–6]. Among the various types of roof structures, trusses have received the most attention in the research of progressive collapse....

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Michala Steel lalliella Locio ny Llugo Julinelo

A lightweight structure from the 1920s

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Author(s): Joram Tutsch, Rainer Barthel

Presented at IABSE Symposium: Engineering the Future, Vancouver, Canada, 21-23 September 2017, published in Engineering the Future, pp. 623-630

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In the mid-1920s, the German engineer Hugo Junkers (1859-1935) designed an innovative roof construction that is regarded a milestone in the development of lightweight structures. A rhomboid framewo...

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Author(s): Joram Tutsch (Technical University of Munich, Munich, Germany)

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Year: 2017

DOI: 10.2749/vancouver.2017.0623

Abstract: In the mid-1920s, the German engineer Hugo Junkers (1859-1935) designed an innovative roof construction that

is regarded a milestone in the development of lightweight structures. A rhomboid framework of slender steel elements forms a barrel vault that covers a span of up to 50 meters. More than 200 of these roofs – and associated patents – have been commercialized and built all over the world. Unfortunately, most of them do not

exist anymore or are in bad condition.

Figure 1. Inner view of hangar with Junkers roof in northern Munich (Tutsch, 2013)

This paper describes the historical steps of the technical development of the construction (in Chapter 1). The framework is designed along strict geometric rules, which in turn have a large influence on the load-bearing behaviour. Both geometry and structure are systematically analysed (in Chapter 2 and 3).

Finally, an example of the investigations and the analysis of a hangar from 1934 (Fig. 1) in northern Munich is

presented (in Chapter 4).

Keywords:

steel listed building lightweight structures lamella roof modular

barrel vault load-bearing kmtaoinTtaebnlaenocfe.Contents

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Article

The Geometry of Timber Lamella Vaults: Prototype Analysis

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Abstract: This paper presents timber lamella structures applied to the circular cylinder surface when all lamellae axes intersect at the nodes. To achieve the uniformity of all elements in this structure, the geometry of the structure must be carefully designed. The main methods for the research are graphical and numerical methods for geometric design and a prototype construction for a specific geometric pattern. The methods are discussed for their ease of replication, as well as the possibility of reinterpretation on other surfaces, while the prototype design and construction give insight into the process from design to execution. The combination of these methods allows for a thorough analysis of the geometry for lamella structures. The analysis shows that geometrical design must begin from the whole to the lamella, and that the number of element types in the structure depends on the disposition of the elements and the angle of the pattern. The discussion shows the advantages and limitations of the proposed methods, while the conclusions give the guidelines for the implementation of lamella structures into new design projects.

Keywords: right circular cylinder; parametric equations; graphical method; timber structures



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1. Introduction

Lamella structures are spatial structures in a diamond pattern formed by ribs called lamellae [1]. They are usually classified as braced structures—vaults and domes [1,2]. This paper will present timber lamella vaults when the diamond pattern of lamellae is applied to a circular cylinder surface. Contemporary tendencies in architecture, following the sustainable development trend, have led architects to think about the return to natural materials and the reduction of pollution created by the construction industry. The advantages of historical timber structures are being examined for possible modification and application in contemporary architectural practice. Lamella structures have stood out because of their aesthetics, economy and ease of construction.

1.1. Literature Review

The design of the Zollinger roof structure made an impact on the construction industry after World War I. The *roof of modernism* [3] was designed by the architect Friedrich Zollinger and patented in 1921 [4]. When invited to the City Council meeting for the rebuilding of Merseburg, Germany at the end of 1918, the architect Zollinger had an idea of how to design a simple construction model for new houses. The loadbearing elements of the house would be made out of cast-in-place concrete, and the innovative roof structure would be constructed out of timber lamellae, easily prefabricated and assembled even by untrained workers. The diamond pattern of the structure, reinforced with decking, required no additional structural elements, making it cost-efficient compared to traditional roofs. The analysis of material consumption showed that traditional roofs require twice as much material per square meter of the floor plan as the Zollinger roof. The section of this timber lamella structure shows that the roof shape is a segmental arch consisting of two circular segments. This form provides additional volume, so two floors could have been placed under the roof as shown in Figure 1 [5].

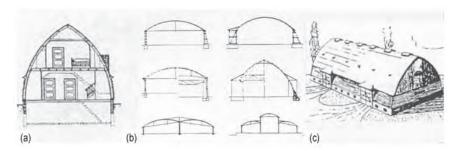


Figure 1. Lamella roofs for (a) housing, (b) halls and (c) barns [5].

The roof is constructed out of timber lamellae with variable cross-section and the upper edge was shaped to follow the arch of the roof. Lamellae were all uniform in shape and size. Two types of lamellae were applied, based on the roof span. The dimensions of the first type were width/height/length = b/h/L = 2.5/15/190 cm and the second were b/h/L = 5/30/150 cm (Figure 2) [6]. When the need for production halls with large spans increased, so did the cross-section of the lamellae, which showed great deflections right after the construction [7]. Other architects started experimenting with the change of disposition and the doubling of the lamellae [7,8], but soon new types of lamella structures were designed, using steel elements and purlins as reinforcement [7,9].

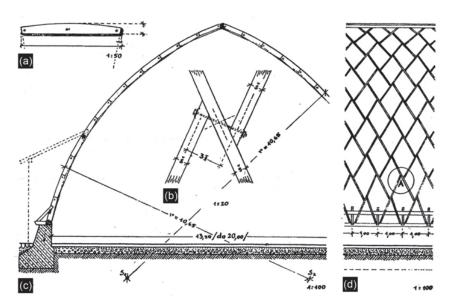


Figure 2. Zollinger lamella roof design: (a) lamella detail with dimensions, (b) joints of lamellae, (c) transverse section and (d) longitudinal section of the roof for housing [6].

The geometry of the first lamella roofs was half of a circular cylinder surface or its segment, in the span to rise ratio between 1:2 (semicircle) to 1:8 (flat arch) [10]. Later, the diamond pattern was applied to the spherical surface for dome structures and to this day, examples on free-form geometries can be found. Lamella structures were built all over the world, from timber to concrete, all following the geometry of a cylinder [7,9–11]. Other types of geometries were too complex to calculate without a computer. If the geometry is symmetrical on both axes, the number of equations is smaller, and the calculation is simpler [10]. With the use of computer software, new lamella structures on free-form geometries were erected.

The aesthetics and expressiveness of the diamond pattern have made lamella structures the primary choice for large-span objects where the structure remains visible. The advantage of lamella structures is the uniformity of the elements—the lamellae and their joints, which

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lead to the ease of production and assembly, the speed of erection and the minimised cost of the overall structure regarding the volume it covers. In order to preserve its advantages, it is necessary to find a suitable geometrical pattern for the lamellae axes to be applied to a circular cylinder surface. Throughout the years, several solutions were designed in timber and steel. The original structure, the Zollinger roof, was made out of timber planks placed vertically to the floor. Each lamella is twice the size of the diamond, and they are connected interchangeably, one in the middle of the other [1]. Three lamellae intersect at the node, with one central and two connecting lamellae shown in Figure 2. They are spaced apart for three widths of the lamella to mount the bolts [12]. This spacing also allows for the lamellae to be placed vertically and to follow the curve of the vault. The length of lamellae in steel lamella structures by engineers Emil M. Hünnebeck and Hugo Junkers is the size of the diamond, which allows them to put the connecting lamellae closer and to still follow the vaulted surface [13]. In these structures, the lamellae are rotated or translated in the horizontal plane to have all uniform elements and to follow the envelope of the cylinder, as presented in Figure 3. This creates an eccentricity at the node, resulting in the moment around the vertical lamella axis for the dominant axial forces in the structure.

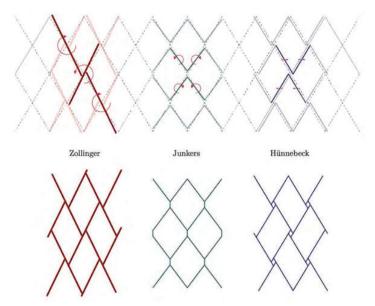


Figure 3. Diagrams showing three types of lamella vaults and the rotation/translation of the lamellae **(up)** with different types of nodes **(down)** [13].

Recent developments in lamella structures have shown the possibility to apply the diamond pattern on a number of forms using contemporary tools. Authors research regularities in different geometries trying to find the best structural pattern and the construction strategy for timber structures [14–16]. In recent years, a development in lamella structures was presented through workshops, experiments and built objects such as TIJ Bird Observatory [17–19].

1.2. The Aim of the Study

This paper discusses the geometry of timber lamella vaults. The design and position of the lamellae on the cylindrical surface have to be precisely defined in order to maintain the diamond pattern and the uniformity of the elements. The focus of this research is the lamella structure where all lamellae axes intersect at the node to avoid eccentricity (Figure 4). This will create a problem of rotation of lamellae in relation to the cylindrical surface, which is analysed and presented in this paper. The aim of this study is to better understand the geometry of lamella structures to be easily modified and adapted for use

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in contemporary structures. The idea is to comprehend the regularities of the geometrical design for cylindrical surfaces for the purpose of interpretation on other surfaces.



Figure 4. Diagram showing the node with one central lamella and two connecting lamellae when lamellae axes intersect.

The methods applied in this paper are the graphical method, the numerical method and prototype design. The graphical method presented in this paper is a novel approach, not found in the literature. The authors used different software to find the best possible solution for the geometric design of the lamellae axes. To expand the analysis, and to precisely define the geometry of the axes, a numerical method was applied. The authors presented a new method for defining the geometry of the axes and compared it to the method presented by Tutsch [13]. The prototype design was derived from result comparison of the graphical and numerical method. This prototype shows the level of uniformity of the elements and the time needed for prefabrication and construction. The erection of the prototype followed the instructions presented by Hosseinzadeh [10] since no other authors describe the method of erection.

The discussion includes all three approaches for the geometrical analysis and presentation of timber lamella vaults: (1) the graphical method, (2) the numerical method and (3) the physical model. The conclusions of this research affirm the aim of the study and open new questions for further research.

2. The Geometrical Design Methods

To obtain the precise geometry of the lamellae, the research was carried out using graphical and numerical methods. The main criterion is that the uniformity of the elements needs to be preserved since this is one of the main advantages of lamella structures.

The chosen geometry for the lamella vault is a cylinder surface. The cylinder type is a right circular cylinder, consisting of two of the same parallel bases the shape of a circle. The envelope of a cylinder is a perpendicular surface with all the same and parallel lines equal to the height of the cylinder, which is the vertical distance between the two bases.

The original lamella structure, the Zollinger roof, was designed as two circular cylinder surface segments of the same radius that meet along the ridge. Cylinder surface segments were also used for other types of buildings, such as halls and barns [5,7,9,11].

2.1. The Graphical Method

2.1.1. Connecting of the Arched Lamellae

The first iteration for the geometrical design of the lamella structure using the graphical method was based on the analysis of the lamella joint. The observed joint is a modification of the original joint for a Zollinger roof. In this joint, the axes of the lamellae intersect at the node, reducing the eccentricity. The three lamellae at the node are connected using steel plates bolted to the lamellae [20]. The research conducted by engineers Scheer and Purnomo at TU Berlin has shown a layout of the lamella structure, with a span of 21.5 m, a length of 21 m, an arch rise of 6.2 m and arch segments for the angle 120° [21]. The presented layout was used to design one lamella as a starting point for the geometry of the structure. Lamellae were connected one to another, forming an arch in one direction. The other direction of the lamellae was obtained by the rotation of the arch for 120° . The idea was for all lamellae to be vertical to the floor plane, that is, for the arches to move translationally and to form the vaulted structure.

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This design process turned out to be wrong because the lamellae cannot be placed vertically and intersect at the node at the same time. When all the arches made from lamellae are in place, it can be observed that the node of the lamellae is not where it should be placed—each lamella should be connected to the middle of the lamella from the other direction. Figure 5 shows the details A, B and C with respect to the structure. Detail A shows the only position where it is possible to place a lamella vertically to the floor plane and that is the ridge of the vault. Detail B shows the slight distance of the lamella from the middle of the other one, at 1/4 of the arch, while detail C shows the greatest deviation of one lamella to the middle of the other, observed at the point of support of the structure.

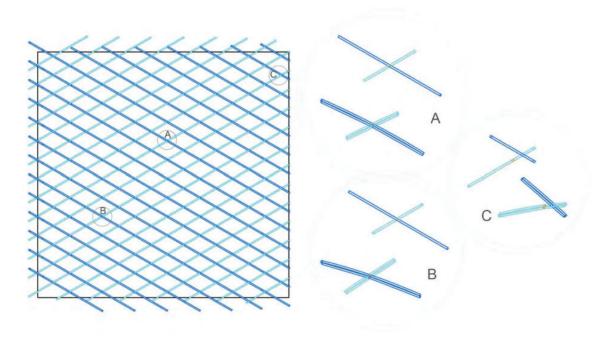


Figure 5. The plan and details of the lamella vault for the graphical method of connecting the lamellae in an arch with details A, B and C showing the misplacement of the connecting lamellae in the node.

The conclusion is that lamella structures cannot be designed starting from an individual element to the whole assembly because the ends of connecting lamellae do not meet at the middle of the central lamella. It is necessary to start with the whole to obtain a more accurate geometry of the lamellae. Vertical sections through the circular cylinder give an ellipse in the section, which cannot give uniform lamellae.

2.1.2. Projection of the Pattern to the Cylinder Surface

The second iteration was led by the idea that the fastest and simplest way of obtaining the diamond pattern structure on a cylinder surface is to project the pattern to the cylinder surface in software for 3D design, such as Rhino [22]. The half-radius of the base circle for the cylinder was r=12.4 m and the length of the cylinder was l=21 m. The arch segment had a span of a=21.5 m and a rise of f=6.2 m, giving the length of the arch $a_1=26$ m. The network was made with angles of 60° and 120° , the length of the cylinder surface l=21 m and the width equal to the length of the arch segment of the cylinder $a_1=26$ m. The proportions of the cylinder were obtained from the layout by Scheer and Purnomo [21]. When the network is projected onto the cylinder the disposition of lamellae is obtained. This process is shown in Figure 6, which shows the detail of the structure with different lengths of lamellae from support to the ridge.

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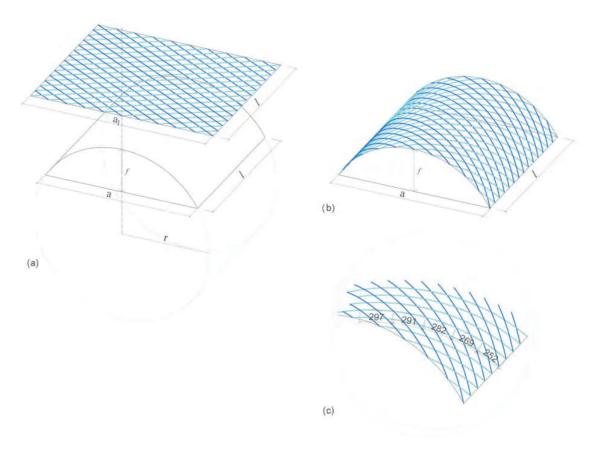


Figure 6. The axonometric view and detail of the lamella vault for the graphical method of projection of the diamond pattern to the cylinder surface: (a) the projection plane and the cylinder surface for projection, (b) the axis of the lamellae lying on the cylinder surface, (c) detail of the lamellae axes showing their different lengths.

This process of geometrical design has many advantages. It is easily understandable, so it is easy to replicate and apply to any surface. It is not time-consuming, nor it is necessary to always apply the same diamond pattern with angles of 60° and 120°, allowing more design freedom. The lamellae are vertical to the floor plane and intersect at the node, creating a continuous surface for placement of any roof tiling. The only problem is the different lengths of the lamellae, which is why this design does not fulfil the main criteria of the uniform elements. On the other hand, each horizontal segment of the vault has the same lamellae with the same joints, thus making sets of uniform elements. From the ridge to the supports, the length of the lamellae decreases and the angle of the bevelling increases. This structure could be easily prefabricated using a CNC machine for the shaping of the lamellae, in order to decrease the time for their production. If steel plates are used for the joints, a large number of different sets would not be economical to make. However, there are lamella structures constructed like this, such as the ice rink structure in Toronto from 2019 with T-section joints [23].

2.1.3. Division of Cylinder Surface to Equal Parts

The third iteration for the geometric design was also led starting from the whole to the elements with the aim for the lamellae of the same geometric characteristics to have uniform elements and to fulfil the main criteria. Based on the layout presented by Scheer and Purnomo [21], a segment of the cylinder surface was divided into equal parts, radially into 20 segments and longitudinally at every 0.75 m to obtain all the nodes of the lamellae. Lamellae rest on supports every 1.5 m and the nodes are placed interchangeably as each

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lamella connects to the middle of the one from the other direction (Figure 7a). The nodes were connected with lines passing two lengths of the diamond to obtain the desired length of the lamellae. Two types of lamellae were obtained, the ones 3 m in length and the ones on the perimeter with a length of 1.5 m. These lamellae axes do not intersect at the nodes, so the connection was simulated by a short line, which presented the joint (Figure 7b). Straight axis lamellae create a structure similar to a folded plate, which was not the idea behind the design. The lamellae needed to have the arched axis that lies on the cylinder surface in order to have all the same lamellae and a uniform surface of the structure.

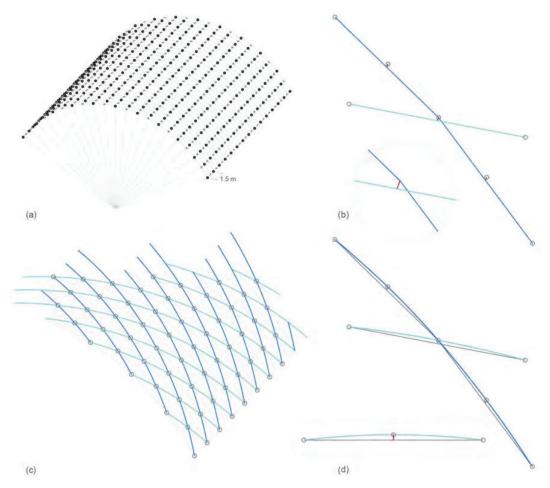


Figure 7. The process of division of cylinder surface to equal parts: (a) axonometric view of the lamellae vault with nodes of the lamellae spaced 1.5 m apart, (b) detail of each lamellae span and the connections at the nodes, (c) segment of a lamellae vault with all arched axes of the lamellae intersecting in the node and (d) detail of the arched lamellae defined by the span and rise lines.

The arched axis of the lamellae was designed using the two lines, which defined the plane for each lamella in the structure. The ends of the line connecting the nodes and the top of the line presenting the connection define the arch span and rise (Figure 7d). The most precise geometry is derived this way and the geometrical model fulfils the main criteria. All lamellae have the same geometry and uniform joints, making the production of the elements easy for mass prefabrication.

2.2. The Numerical Method

The geometrical shape that connects all the nodes and divides the cylindrical surface into uniform segments is a helix.

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Starting with the parametric equation of a circle [13]

$$x_{k} = \begin{pmatrix} y \\ z \end{pmatrix} = \begin{pmatrix} R\cos\varphi \\ R\sin\varphi \end{pmatrix} \tag{1}$$

from which the parametric equation for a circular cylinder is obtained

$$x_{kz} = \begin{pmatrix} x \\ y \\ z \end{pmatrix} = \begin{pmatrix} x \\ R\cos\varphi \\ R\sin\varphi \end{pmatrix}$$
 (2)

the parametric equation of the helix can be derived

$$x_{s} = \begin{pmatrix} x \\ y \\ z \end{pmatrix} = R \begin{pmatrix} (\varphi - \varphi_{0}) \tan \beta_{s} \\ \cos \varphi \\ \sin \varphi \end{pmatrix}$$
 (3)

with pitch

$$h_{s} = 2\pi R tan \beta_{s}. \tag{4}$$

The angle formed by the lamellae is constant and can be derived from the parameters, i.e., the length of the roof—L, the length of the arch—B, the number of cylinder divisions in the X-direction—m and the number of cylinder divisions in the Y-direction—n, as shown in Figure 8a, with its equation given as follows:

$$\tan \beta_{\rm s} = \frac{n \cdot L}{m \cdot B} \tag{5}$$

$$\beta_{\rm s} = \arctan \frac{n \cdot L}{m \cdot B} \tag{6}$$

The radius of curvature of the helix is

$$R_{\rm s} = \frac{R}{\cos^2 \beta_{\rm s}} \tag{7}$$

and its arch length is

$$B_{s} = \frac{B}{\cos \beta_{s}} \tag{8}$$

deriving the abstract angle of the opening of the helix

$$\alpha_{\rm s} = \frac{B_{\rm s}}{R_{\rm c}} = \frac{B \cdot \cos \beta_{\rm s}}{R} = \alpha \cdot \cos \beta_{\rm s} \tag{9}$$

Based on the elements of the lamella roof structures, as presented in Figure 8b, the authors of this paper derive the following parametric equations for the two helixes that form the basic geometry of the lamella roof:

$$x_{s_1} = \begin{pmatrix} x \\ y \\ z \end{pmatrix} = \begin{pmatrix} \frac{h}{\alpha} \cdot \varphi \\ R\cos\left(\varphi + \frac{k_1}{2} \cdot \alpha\right) \\ R\sin\left(\varphi + \frac{k_1}{2} \cdot \alpha\right) \end{pmatrix}$$
(10)

$$x_{s_2} = \begin{pmatrix} x \\ y \\ z \end{pmatrix} = \begin{pmatrix} \frac{\frac{h}{\alpha} \cdot \varphi}{R\cos\left(\varphi + \frac{k_2}{2} \cdot \alpha\right)} \\ -R\sin\left(\varphi + \frac{k_2}{2} \cdot \alpha\right) \end{pmatrix}$$
(11)

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-h is the length of the helix for one lamella,

$$h = \frac{L}{m} \tag{12}$$

 $-\alpha$ is the angle of the helix needed for one lamella,

$$\alpha = \frac{B}{n} \tag{13}$$

 $-\varphi$ is a variable that defines the segment of the helix (the length of the lamella axis is the angle of 24°);

 $-k_1$ is a coefficient that is an even number;

-k₂ is a coefficient that is an odd number.

Coefficients k_1 and k_2 define the movement of the helixes relative to one another for half of the length of a lamella to get the right geometry for each lamella to connect to the middle of the one from the other direction.

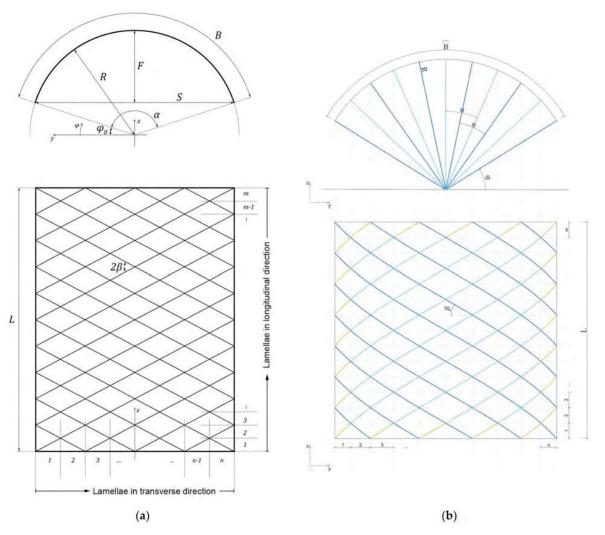


Figure 8. Floor plan and section of the lamella vault for geometrical analysis (a) by Tutsch [13]; (b) by the authors.

In comparison to the parametric equation of the helix by Tutsch [13], the parametric equations provided by the authors define each lamella axis, taking into account the mutual

relation of lamellae. The helix equation by Tutsch defines the helix that follows the segment of the cylinder envelope, not taking into account that the helix from the other direction has to be translated for half of the length of the lamella. The authors define the length of a lamella as a segment of the helix with the variable ϕ , while the coefficients k_1 and k_2 enable the connection of the lamellae in the middle of the central lamellae. The graphic output of the equations by the authors was developed in Wolfram Mathematica and is presented in Figure 9.

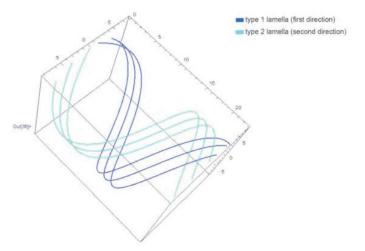


Figure 9. The graphic presentation of the parametric equations for the helixes developed in Wolfram Mathematica. The blue graph shows the helix from one direction and the green one shows the helix from the other, translated for half of the lamella length.

When applying the numerical method for the geometrical design, the conclusion is that even the infinitely small segment of a helix is a spatial curve. This results in lamellae torqued around their longitudinal axes, which complicates the manufacture, see Figure 10a. For lamellae to be manufactured, an idealisation is needed. Each segment of a helix needs to be converted to an arch, as it was shown in the graphical method, in order to define a planar curve for the lamellae manufacture. This leads to a slight rotation of the connecting lamellae in the node, as presented in Figure 10b.

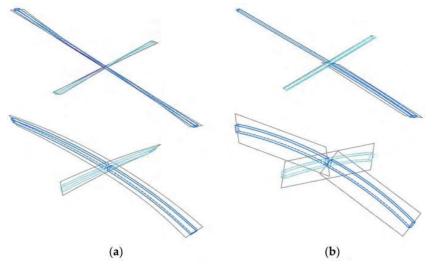


Figure 10. The axonometric view of the intersection of the lamellae at the node (a) showing the lamellae axes following the helix curve obtained by the numerical method, and (b) showing vertical axial planes of the lamellae in order to present the rotation at the node obtained by the graphical method.

3. The Physical Model of a Lamella Vault

In architecture, physical models help to solve problems during the design process, working in parallel with drawings, 3D models and construction with materials corresponding to the designed structure [24]. During this process, different aspects of the design can be modified or changed due to the design process on various scales and with a variety of tools. Design problems can be resolved from the level of the node to the structure as a whole. This practice was common in historical constructions when knowledge was acquired by model design and construction and their analysis. This process of constant iterations and relations between designing on a computer and designing a physical model is called complex modelling in contemporary architecture [25]. The hypothesis is that it helps with better observation and learning about the design.

Following the conclusions of the geometry analysis, the prototype was designed from the lamellae with axes as planar arches to be easily manufactured. The axes of the lamellae intersect at the node, eliminating the eccentricity that appeared at the original joint, making this prototype an improvement of the historical lamella structure.

3.1. The Design of the 3D Model

The first step towards the design of a physical model of a timber lamella vault was the design of a 3D model with all the necessary details of the lamellae and their joints. The model was based on the arched lamellae axes obtained by the graphical method presented in Figure 7, since the geometry of the axes provided by the numerical method results in torqued lamellae, see Figure 10a,b. The cross-section was first assigned to the lamella placed vertically to the floor plane and their connecting lamellae in the middle. The ends of the lamellae were bevelled following the vertical axis planes of the lamellae so that the whole cross-section of the connecting lamellae was pressed onto the middle of the central one. The lamellae were then rotated around the axis of the cylinder in order to obtain the whole structure. Thus, all lamellae are the same and all lamellae axes lie in the envelope of the cylinder. Arches along the gables were designed as three-hinged arches. Lamellae pressed onto the gable were cut obliquely by following the vertical plane of the three-hinged arch.

The joints for the lamellae were designed with steel plates bolted to the lamellae. The inspiration was a T-section joint presented in the Timber Construction Manual [26]. This joint is designed using two steel plates welded to each other to form a T-section. The difference between this joint and the applied one is that, in this design, two steel plates were placed on the outside edges of the lamellae and welded to the central steel plate. The T-section joint is placed inside the lamellae and requires additional shaping, as opposed to the applied joint. The supports were designed as point supports following the same design logic as the joints.

The final design is presented in Figures 11 and 12. The 3D model of the structure can be observed in Figure 11, while Figure 12 presents floor plans and sections of the structure, providing information about its dimensions.

3.2. Elements for the Physical Model

The designed structure has a span of 10.75 m, it is 3.1 m high and requires 81 lamellae. Based on the position of the lamellae in the structure, six types can be distinguished. All lamellae have the same radius of curvature because they all lie on the cylinder surface. The length of most lamellae is approximately 3 m, except the ones along the perimeter, which are 1.5 m long (Table 1). Type 1 has a span of 289 cm and it is the most used type in the structure. Type 3 shows the lamellae next to the supports, and type 4 are the lamellae lying on the gable arch. Two special types are types 5 and 6, which lie on the arch and the supports at the same time. The differences between the lamella types are created by the length and the different angles of the bevelling of the ends. The disposition of the lamellae in the diamond pattern with angles 60° and 120° requires this number of types, and it cannot be reduced. The cross-section of the lamellae is width/height = b/h = 6/16 cm.

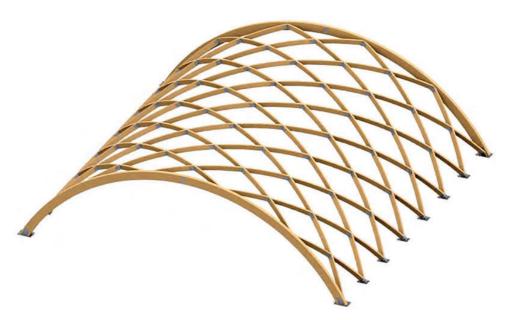


Figure 11. Three-dimensional model of the designed lamella vault.

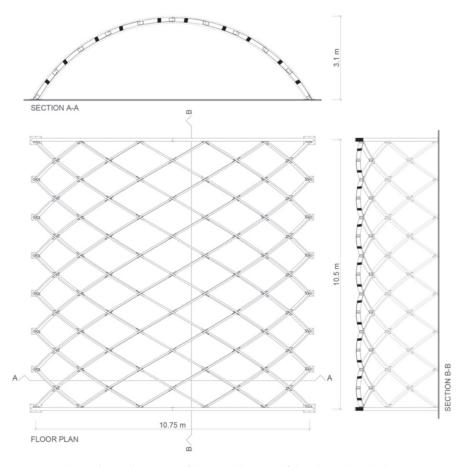


Figure 12. Floor plan and sections of the lamella vault of the physical model.

Table 1. Specification of timber lamellae.

Туре	Span of a Lamella [cm]	Number of Lamellae	Total Volume for the Type [m ³]
1	289	33	1.007
2	289	24	0.732
3	149.5	12	0.189
4	153	8	0.129
5	292	2	0.062
6	148	2	0.031
		Total:	2.15

The structure has six types of joints based on their position inside the structure: two types of lamellae joints, the arch and the lamellae joints, the support joints and two types of arch and lamella support joints. The dimensions of the steel plates depended on the position of the node and its geometry, as well as the position of the bolts according to technical regulations (Table 2). The width of the steel plates was 3 mm for all of the joints, except for the supports made from 5 mm thick steel plates. The used bolts were M12, class 5.6.

The majority of the lamellae belong to types 1 and 2 (Table 1) where the bevelling of the lamellae shows that they are mirrored one in reference to the other. Other types of lamellae are derived from types 1 and 2. The same goes for the joints.

3.3. Construction of the Physical Model

The prefabrication of the elements preceded the construction of the designed timber lamella vault. The base for lamellae was made from an arched glued laminated timber beam, with an arch radius of 844 cm and outer edge length of 630 cm. In order to have 81 lamellae, 35 base arches needed to be made. The gable three-hinged arches were made from four equal arched glued laminated timber beams, with an arch radius of 635 cm and an outer edge length of 680 cm. Steel plate joints were prefabricated in a workshop according to the design, out of 3 mm and 5 mm steel plates with mechanically predrilled holes for bolts. The anchor plates were made from 10 mm thick steel plates.

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Table 2. Specification of steel joints.

Туре		Number of Joints	Total Volume for the Type [m ³]	Total Weight for the Type [kg]
1	POS1 POS2 POS3	70	0.0132	103.620
2	POS3 POS3	48	0.00904	70.964
3	POSSa POSSa POSSa POS7a POS7b	8	0.00249	19.547
4	POS10a POS10b POS9b POS8b	12	0.00391	30.694
5	POS13a POS12		0.000855	6.712
6	\(\sqrt{\sq}\sqrt{\sq}}\sqrt{\sq}}}}}}}}}}}}}}}}}}}}}}}}}}}}}}}}}}}}	OS12a 2	0.000855	6.712
			Total:	238.25

The construction of the lamella vault started with the placement and levelling of the anchor plates, anchored to the ground with M16 anchor bolts. Support joints were welded to anchor plates at the designed positions to provide a good starting point for mounting timber elements. The shaping and placement of three-hinged arches was the next step. The gable arches were measured and shaped on the ground, connected with steel plates at the hinge, and then lifted and placed into the supports. The positions of the joints for the lamella and the arch were measured and marked. The joints were then mounted to the

three-hinged arch. To achieve the stability of the gable arch, the first lamellae needed to be placed near the arch supports, as presented in Figure 13. The construction layout dictated the sequence of the lamellae assembly, starting from one gable to the next, forming one bay at a time in order to check the dimensions and the positions of the lamellae and the joints. The described process of bay-by-bay construction was presented as the best manner of construction for a lamella vault [10].

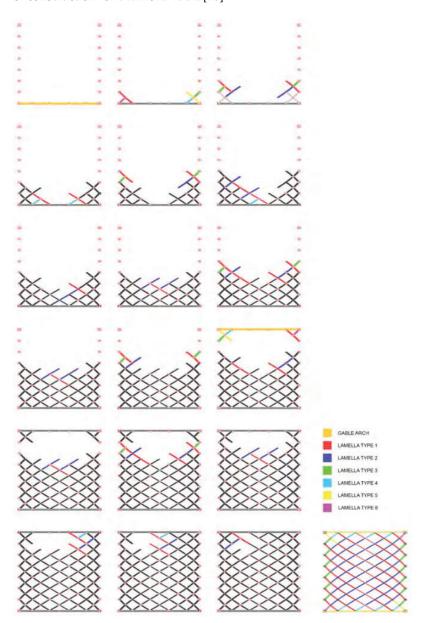


Figure 13. A diagram of the construction process of the physical model.

The base arches for the lamellae were delivered to the building site where they were measured and bevelled according to the specifications. During the construction, it was concluded that the base arches tended to elongate because of high temperatures, so the position of the joints had to be measured according to the triangle between the edge joints and the middle one. The joints were mounted onto the middle of each lamella on the ground. The lamellae would be then placed at the designed position in the structure and controlled by the position of the stings marking the height of the nodes. The lamellae

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would be temporarily secured with screws until the whole bay was positioned, after which the holes for the bolts would be drilled and the bolts mounted.

At the beginning of the construction, there was a need for additional supports, since the structure was very unstable. With the increase of the bays, the structure began to adapt to the cylinder shape. The larger number of lamellae showed that every other lamella reinforced the previous one and set its position in the structure. This was observed as a successive relief in the construction process right after the construction of the first bay, and it was confirmed after half of the structure was constructed.

The construction experience contributed to a better understanding of the timber lamella vault. Conclusions were drawn regarding the method of assembly and the preparation of the structural elements. This experience also opened questions related to the modification of the structure.

The construction process and the physical model are shown in Figure 14.



Figure 14. Photo of the construction process and the physical model in detail.

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4. Discussion

Lamella structures are a specific type of spatial structure primarily because of their diamond pattern. They have the advantage of the uniformity of elements, leading to an economical structure that is easily assembled. This pattern creates an unstable system if no additional structural elements are applied. One of the ways to solve this problem is to form moment connections between lamellae. In order to design a lamella structure, the geometry must be precisely defined.

The original joint has a large moment of eccentricity compared to the other types of joints and the load capacity of the bolts connecting the three lamellae at the node is much smaller [12,26]. Throughout the years, engineers have suggested a modification of the original joint and have designed a joint with all three lamellae axes intersecting at the node, thus eliminating the eccentricity [20,21,26]. The proposed joints are usually designed with steel plates, having a greater loading capacity than the original one. The geometrical design and the prototype presented in this paper are for the lamella structure where all lamellae axes intersect at the node, and the eccentricity is eliminated.

The chosen geometry of the lamella structure in this paper is a lamella vault. The diamond pattern is applied to the envelope of the right circular cylinder. The material of the lamellae is timber, and the joints are formed out of steel plates bolted to the lamellae.

The discussion in this paper is led by the following criteria:

- 1. The geometry of the structure must provide uniformity of all structural elements.
- 2. The lamellae must intersect at the nodes to reduce the eccentricity of the joints.
- 3. The construction must be simple and performed in a short period.
- 4. The designed structure must be economical.

The criteria are derived from the advantages of historical lamella structures, which must not be damaged by the modification of the structure.

The geometrical design of the lamella vault was approached using the graphical method and the numerical method. The numerical method for geometrical design opens the possibility of easy modification of set parameters. The diamond pattern of the lamellae can be applied to any type of surface by following the methodology shown in Section 2.2. The authors' numerical method presents a further observation of the specific pattern of lamellae and gives the possibility of adaptation, which would include the interchangeability of the original connection—one lamella connects to the middle of the next one from the other direction. The presented parametric equations can also be used for 3D modelling in different software plug-ins, such as Grasshopper for Rhino. This enables the fast and precise design of the geometrical model [15,16,19]. For the physical model, the axis curves of the lamellae would have to be optimised. The parametric definition of the helix, even for an infinitesimal segment, gives a spatial curve, so it is necessary to modify it into a planar curve—an arch that will define the axis of the lamella for the construction. One of the graphical methods has shown this modification. The presented graphical methods have shown two possible approaches to geometric design: (1) from lamella to the whole structure and (2) from the whole to the lamella. The analysis has shown that the right process of design is the second one and both graphical methods that followed this process have proven successful.

The method of pattern projection to the cylinder surface creates a reasonable structure with all vertical lamellae that intersect at the nodes. This geometry does not fulfil the first criteria since there are numerous sets of uniform lamellae, depending on the density of the structural pattern. This could be overcome by the production of lamellae on a CNC machine, thus reducing the prefabrication time. The number of joint sets would be the same as the number of lamellae sets, so a simple joint must be designed to be easily modified for different angles in the structure. If the elements were to be mass-produced, this structure would have complied with all the criteria except the first one.

The method of division of the cylinder surface into equal parts was applied to the design of the physical model of the lamella vault. This method gives a uniform structure with six types of lamellae and the corresponding joints, no matter the density of the pattern

since the types of the elements depend on their position in the structure. The differences among lamellae are created because of different angles for bevelling, which also influences the angles in the joints. Types 1 and 2 are mirrored elements, which are the consequence of the diamond pattern and the angles of 60° and 120° . The number of types could be reduced for one if the pattern was created with 90° angles. This proves that the structure fulfils the first two criteria. The only problem with this structure is the rotation of lamellae at the nodes because the axes of the lamellae intersect at the nodes.

In historical lamella structures, the rotation/translation of the lamellae was applied in the horizontal plane to have all lamellae vertical to the floor [13]. This resulted in a variety of joints that had large moments of eccentricity, since the lamellae do not intersect at the nodes, but the criteria for uniform elements was fulfilled. The advantage of Junkers' structure, over the ones of Zollinger and Hünnebeck, was that all the joint elements were the same. In comparison to these structures, the designed joint for the presented physical model has reduced the eccentricity in the node, leaving the axes of lamellae to intersect. On the other hand, the rotation of the lamellae appears in the vertical plane, making a torsional movement around the axis, so they are not vertical in relation to the floor. The rotation of the lamellae at the node is the consequence of the approximation of the arched axis of the lamella corresponding to the helix curve, as presented in Sections 2.1.3 and 2.2. This rotation of the lamellae demands further shaping after the construction is finished, to provide a continuous surface, as it would be for the vertically placed lamellae.

The construction of the physical model for the timber lamella vault with a 10.75 m span and a length of 10.5 m lasted seven days with only three workers. The hypothesis is that five workers would finish the construction in a smaller amount of time, thus also fulfilling the third criterion. The number of workers and the period of construction affect the economy of the structure [27], i.e., the cost of construction is reduced for a small number of workers and the short construction time. In comparison to standardised timber vaults, this structure is not economical because all the elements are specially designed only for this structure, while standardised vaults use mass-produced elements.

The discussion and analysis of the presented geometry of timber lamella vaults still leave an open question for choosing the best way to design a lamella structure, thus giving the designer the possibility to adapt the structure to its needs.

5. Conclusions

The presented research shows the problems of the geometrical design of timber lamella vaults. The diamond pattern of the lamellae is applied to the right circular cylinder envelope with the idea to explore different methodologies for geometrical design that could be replicated on any type of surface. The physical model of the structure has presented problems that emerge during the construction, contributing to the thorough analysis from design to execution.

The conclusions about the geometry of timber lamella vaults are drawn as follows:

- The graphical geometrical design method needs to follow the process of design from the whole to the lamella to obtain the correct geometry with as many possible uniform elements.
- The graphical method following the process of projection of the pattern to the cylinder surface gives various sets of uniform elements—lamellae and the corresponding joints—leaving them vertical to the floor plan. This process is easily replicated and the lamellae pattern is easily modified to meet designers' needs.
- The graphical method of the division of the cylinder surface into equal parts results in the most uniform elements. The lamellae are rotated around their longitudinal axis, so they are not vertical to the floor plan.
- The smallest possible number of element types is five for timber lamella vaults where the axes of lamellae intersect at the nodes. This can be achieved only for the 90° angle between the lamellae, that is, for the square pattern of lamellae.

The geometrical design approach using the numerical method gives parametric equations that are easily modified in 3D modelling software to meet designers' needs.

The presented geometrical analysis and physical model of a timber lamella vault have shown the adaptability of lamella structures and the possibility to use them in different contemporary architectural projects.

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Design and Analysis of

Timber Lamella Segmental Arches

by

Glenn Frazee

A Report Submitted to the Faculty of the

Milwaukee School of Engineering

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Abstract

A lamella roof offers a unique architectural feature in its interwoven network of timbers. As a roof system, the stiffness created by the interlocking members results in a curved roof that uses less material than a traditional rafter and purlin design. The goal of this paper is for the reader to be able to create a preliminary design of a lamella roof that will be strong enough to withstand the loads stipulated by the most current ASCE 7-10 Minimum Design Loads for Buildings and Other Structures. This design is facilitated by load tables developed by the author using the finite element method and connection tables in compliance with the National Design Specification for Wood Construction 2005 Edition using the Allowable Stress Design (ASD) procedure. In reality, the values used for this preliminary design will give a conservative design that could most likely be lightened with a more in-depth structural analysis. Testing on a steel lamella model shows inconclusive results when compared to those predicted by the load table program developed by the author and should be investigated further.

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Nomenclature

```
Symbols
a = one-half span of arch (von Kármán method only)
A = area
A = \text{vertical reaction} (Scofield method only)
b = breadth or thickness of lumber section
B = \text{vertical reaction } #2 \text{ (Scofield method only, used if reactions are unbalanced)}
C_e = exposure factor
C_s = slope factor
C_t = thermal factor
d = \text{dead load (Scofield Method only)}
d = depth of lumber section
D = axial thrust in lamella arch
E = Young's modulus (modulus of elasticity)
f = \text{rise of arch (von Kármán method only)}
\tilde{f} = beam element forces vector
\tilde{F} = combined forces vector
I = moment of inertia about the X-X axis
\tilde{k} = beam element stiffness matrix
```

 \tilde{K} = combined stiffness matrix

 ℓ = length of lamella between top bolt centers

 ℓ_{c-c} = center-to-center length of lamella

 L_r = construction live load

n = number of lamellas in the span of an arch

p = live load per unit length of horizontal projection (von Kármán method only)

 p_g = ground snow load

 p_f = flat roof snow load

q = dead load per unit length of arc (von Kármán method only)

r =Rise-to-Span ratio (T/S)

R = radius of curvature of lamella arch

s = snow load (Scofield method only)

s =shift of lamella connection

S = span of lamella arch

 S_b = balanced snow load

 S_u = unbalanced snow load

 S_{xx} = section modulus about the X-X axis

T = rise of lamella arch

 \tilde{u} = beam element displacement matrix

 \tilde{U} = combined displacement matrix

W =wind load (Scofield method only)

x = distance measured from arch line of symmetry, distance from origin

 θ = skew angle (or angle of inclination) of transverse lamella arches

Abbreviations

AISC American Institute of Steel Construction

ASCE American Society of Civil Engineers

DL Dead Load (Gravity Load)

FEA finite element analysis

LL Live Load (Gravity Load)

mph miles per hour

NDS National Design Specification

plf pounds per lineal foot

psf pounds per square foot

SL Snow Load (Gravity Load)

WL Wind Load

Glossary

Rise – Height of curved roof from springing points to apex

Span – Clear distance covered by a roof

Springing Point – Hinging point in a two-pinned arch

Thrust – Force on a lamella parallel to its long dimension

1 Introduction

A lamella roof is made up of a series of intersecting skewed arches, each arch made up of smaller individual pieces called lamellas. These skewed arches come together to form a curved roof profile. J. S. Allen puts it well:

The timber arched roof was made up of relatively short timbers referred to as 'lamellas' varying in thickness and depth depending upon the span but identical for any given span. These lamellas are curved on their top edges and beveled at the ends which are radial to the curvature and are bolted together on edge with the curved side uppermost, to form a rhomboid network of framing timbers. In this manner the external surface of the roof takes up the arched form [1].

Figure 1 displays the recently completed Hale County Animal Shelter, a project designed and constructed by the Rural Studio of Auburn University. Easily visible are the individual lamella pieces and the rhomboid patterns they create. The tops are cut to fit the curved profile of the roof. Connection details will be discussed later.



Figure 1 - Hale County Animal Shelter [2].

Figure 2 shows four different configurations for a lamella roof. This paper will focus on the segmental arch, where the profile of the roof follows a segment of a circle rather than a parabola or a gothic arch [3].

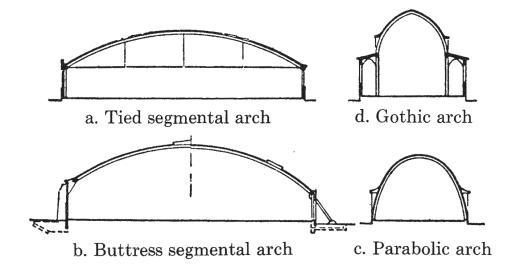


Figure 2 - Types of Lamella Roofs [3].

End support conditions, such as the tied arch or the buttressed arch, account for the resulting horizontal thrust in the springing ends of the arch [4]. While such supports should be taken into consideration in the roof design, it is beyond the scope of this project to delve into the different design calculations pertaining to each.

1.1 History of Lamella Construction

Lamella construction originated from the German architect Friedrich Zollinger (Figure 3) around 1920. Zollinger was appointed Town Building Advisor at Merseburg/Saale in 1918 at a time when Merseburg was experiencing a housing crunch [1].



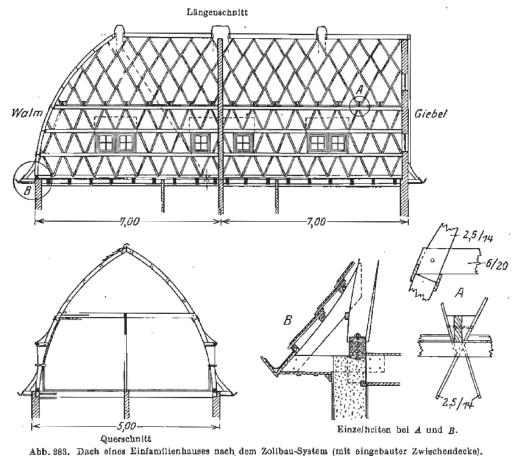
Figure 3 - Friedrich Zollinger [5].

Because of the new ammonia factories and coal mines, thousands of workers moved to the city for work. Unsurprisingly, no new houses were built during World War I and there was a housing shortage for the new workers [1].

To solve this problem, architects of the time improved upon existing ideas or created new building techniques [5]. Zollinger created the "Zollbau Lamellen Dach" system, which utilized precast concrete panels and gothic arched roofs to create dwellings. He created the Merseburg Building Company which then went on to build over 1,250 apartments. Interestingly, the Zollbau method also encouraged the tenants of these flats to help out with construction and, given the assembly-line nature of the method, this was easy to achieve [1]. The Merseburg Building Company acquired material and land for the "self-help settlers" and also looked after the planning and organization of construction projects [5].

Zollinger applied for and received patents in Germany (1921), Australia (1922), and in the United Kingdom (1923). His patent documents show roofs using gothic arches and

"a number of similar curved or straight wood, iron, or reinforced concrete units, bars, or battens" [1]. Figure 4 shows a drawing of a typical house built with the Zollbau method.



n. 283. Dach eines Einammennames nach dem Zondau-System (int eingebauter Zwischendecke

Figure 4 - Lamella Roof Using the Zollbau Method [6].

Over time Zollinger refined his Zollbau method for larger spans, such as for churches, schools, and large halls. The idea caught on in Europe and was used widely for arched roofs [1]. In 1925, the idea spread to America as well [3].

1.2 Previous Roof Failures

Due to the curve of the lamella roof, these structures are susceptible to failure from high wind loads. In 1926, hurricane winds caused the destruction of two lamella buildings in

Florida with one roof being torn off completely and deposited upside-down a few hundred feet away [1, 7].

Lamella roof construction was principally in use from its introduction by Zollinger up until the 1940s, with construction mostly halted because of wind failures. Engineers at the time used a wind load of 10 psf on the vertical projection for normal wind areas and 37.5 psf for high-wind regions. The latter wind pressure correlated with a 130 mph wind speed, the highest measured in that era [1].

In modern times, the wind loads on a curved roof are better known thanks to modern wind tunnel testing and computer simulations. It is now known that wind flowing over a curved roof creates uplift (similar to an aircraft wing), not simply a uniform horizontal load on the vertical projection. This creates a very different loading condition than the horizontal load which could potentially explain the failures of some lamella roofs in the first half of the 1900s.

2 Fabrication of Lamella Pieces

The advantage of the circular segmental lamella arch is that a lamella cut to fit the curve of the arch will fit anywhere on the arch. Because of this, if one creates a template for a lamella on the arch, this same template can be used for every lamella. The only difference is due to the right and left skew of the intersecting arches. Depending on the skew, the bevels on the lamella ends will have to be cut one way or the other. The left-and right-hand lamellas are mirror copies of each other, however. Figure 5 illustrates the difference in the left- and right-handed lamellas in that the bevel angles change direction based on the direction of skew.

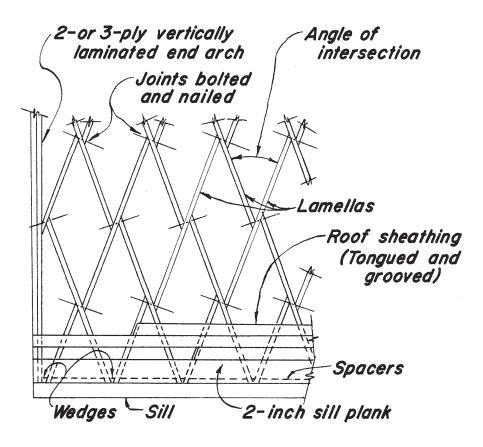


Figure 5 - Lamella Roof Plan View [3].

The designer most likely will know two properties of the arch: its span, S, and its rise, T. From this information, one can find the radius, R [8]:

$$R = \frac{4T^2 + S^2}{8T}. (1)$$

Since the roof arch is circular, skewing the arch results in a lamella arch that follows an elliptical curve [4]. If the radius of the circular roof arch is given by R and the skew of the lamella arches is given by θ , the minor axis of the elliptical path the lamella arches follow has a length of 2R and the length of the major axis would be given by

$$\ell_{major} = \frac{2R}{\cos \theta}.\tag{2}$$

The length of the individual lamella planks is a function of the load capacity of the plank, the curvature of the roof, and the general aesthetics of the roof design. Depending on the loading conditions of the roof, lamella sizes may need to be chosen based off of the load resistance capacity of the board cross-section.

A smaller radius of curvature of the roof limits the length that a lamella plank can reach depending on its depth. A board with a shallower depth will need to be shorter so that cutting out the curvature of the roof on the top of the plank still leaves enough depth on the ends for adequate connection detailing. Figure 6 depicts this relationship.

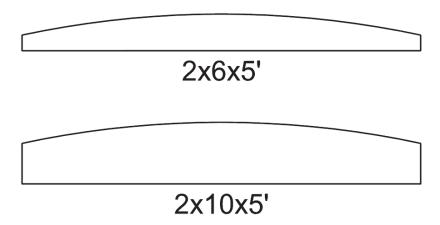


Figure 6 - Lamella Planks with a Radius of Curvature of 12 Feet.

From Figure 6, one can see that the 2x6 plank would not have adequate space on the ends for proper connection detailing while the 2x10 example with the same top radius of curvature would.

Several maximum length tables were developed by the author based on connection detailing considerations. These tables can be found in Appendix B pages 133-142. Section 2.1.1 delves into the connection considerations in more detail.

In designing for aesthetics, having too few boards making up the arch of the lamella roof would appear clunky, boxy, and awkward. Figure 7 illustrates this situation. The inside of the roof appears more angular and harsh and the lamellas themselves are hulking and ungainly. However, this roof uses less lamellas, requiring fewer connections and less labor to install. Also, since the spacing between lamellas increases, the load that each lamella takes on increases, necessitating an increase in size.

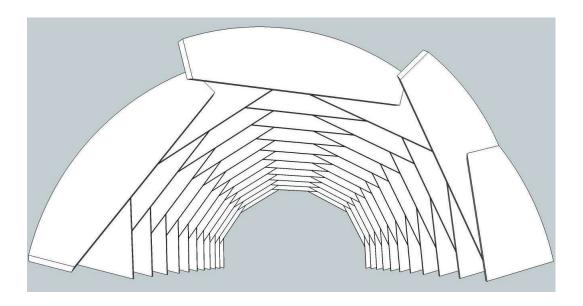


Figure 7 - Three Lamellas Per Arch.

Increasing the number of lamellas per arch makes for a more aesthetically pleasing roof structure. Figure 8 is an image of a lamella roof with nine lamellas per arch. Instead of the roof feeling boxy, the curves are more flowing and the lamellas themselves are more elegant and lithe. Less ceiling space is wasted with the extra depth of the deeper members from Figure 7, resulting in an eye-pleasing ceiling.

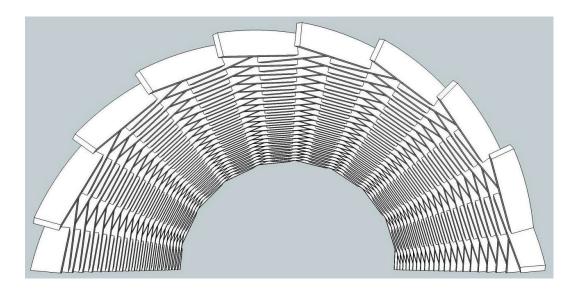


Figure 8 - Nine Lamellas Per Arch.

The number of lamellas per arch is up to the designer, though there is an upward bound on the number of lamellas that can fit into one arch. While having more planks per curve would reduce the spacing between them, resulting in lower loads per lamella which could reduce the necessary cross-section, this trade off may not be cost effective. The trick is finding the right balance between aesthetics and constructability.

2.1 Template Creation

Since the lamellas are essentially modular and can be used anywhere on the roof, creating a cut template is the most efficient means of mass-producing the lamellas. The following sections will further explain the parameters that go into the template creation.

2.1.1 Connection Requirements

Connections in lamella structures are generally handled by bolts or nails or some combination thereof. Depending on the size of the members, specially-made connection plates can also be used [3]. Figure 9 shows the two connection types.

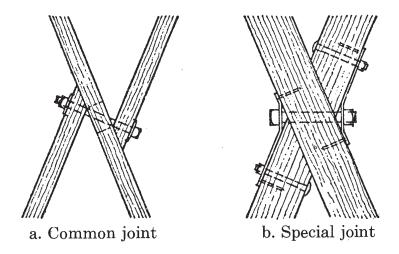


Figure 9 - Example Lamella Connections [3].

Generally, the connection detail labeled "Special joint" in Figure 9 is used for material thicknesses greater than three inches nominal [1]. These allow for the load paths in the lamellas to follow a concentric path which reduces the forces in the connections and the lamellas themselves, as opposed to the eccentric connection of the "Common joint." Having the connection detail of the "Special joint" simplifies the connection to a simple compression connection [9]. Due to the fact that these types of connections need to be specially fabricated and engineered for each project, their design is beyond the scope of this project.

The National Design Specification for Wood Construction (NDS) specifies certain conditions that must be met for wood connections. The direction of the load path through the connection dictates the edge and end distances as well as bolt spacing. These conditions are tabulated in Tables 11.5.1A, 11.5.1B, 11.5.1C, and 11.5.1D of the NDS 2005 Specification, which are displayed in the Appendix A page 130-131 as well as in the rest of the section. These tables give the distances in a multiple of the connector dowel diameter, *D*.

Table 11.5.1A, shown in Figure 10, dictates the edge distance requirements. Though the primary load path is axial compression, there is still a bit of shear perpendicular to grain that must be accounted for.

Table 11.5.1A	Edge Distance		
	Requirements ^{1,2}		
Direction of Loading	Minimum Edge Distance		
Parallel to Grain:			
when $\ell/D \le 6$	1.5D		
when $\ell/D > 6$	$1.5D$ or $\frac{1}{2}$ the spacing		
	between rows, whichever is		
	greater		
Perpendicular to Grain: ²			
loaded edge	4D		
unloaded edge	1.5D		

- 1. The ℓ/D ratio used to determine the minimum edge distance shall be the
 - (a) length of fastener in wood main member/D = ℓ_m/D (b) total length of fastener in wood side member(s)/ $D = \ell_s/D$
- 2. Heavy or medium concentrated loads shall not be suspended below the neutral axis of a single sawn lumber or structural glued laminated timber beam except where mechanical or equivalent reinforcement is provided to resist tension stresses perpendicular to grain (see 3.8.2 and

Figure 10 - Connection Edge Distance Requirements [10].

The loaded edge (top edge) of the lamella must have an edge distance of 4D while the bottom edge must have 1.5D. The second part of the parallel to grain consideration does not apply since the ℓ/D ratio will never be greater than six. A 2x member would need a bolt smaller than $\frac{1}{4}$ for the ℓ/D ratio to exceed six; however, anything smaller than that would not be used in construction.

Tables 11.5.1B (Figure 11) and 11.5.1C (Figure 12) have two columns for the connection parameters. Choosing a distance from one of the columns instead of the others will affect the Geometry Factor C_{Δ} , which is a reduction factor used in determining dowel fastener connection strength. In order to make C_{Δ} equal to one, the minimum edge distances and fastener spacings must all be met.

Table 11.5.1B	End Distance				
	Requiremen	Requirements			
	End Dis	End Distances			
	Minimum	Minimum end			
	end distance				
	for $C_{\Delta} = 0.5$	= 1.0			
Direction of Loading					
Perpendicular to Grain	2D	4D			
Parallel to Grain,					
Compression:					
(fastener bearing away					
from member end)	2D	4D			
Parallel to Grain,					
Tension:					
(fastener bearing to-					
wards member end)					
for softwoods	3.5D	7D			
for hardwoods	2.5D	5D			

Figure 11 - Connection End Distance Requirements [10].

Figure 11 displays the minimum end distances to the cut end of the board. Since the primary load on the lamella connections is a perpendicular to grain load through shear and compression parallel to grain from the axial load, the top two rows of the table in Figure 11 govern. While under wind loading there may be some tension developed due to uplift of the roof, this tension force is so much smaller than the compressive force that the connection, properly designed for the compressive load, will most likely be able to resist it anyway.

Table 11.5.1C	Spacing Requirements				
	rs in a Row				
	_	Spacing			
Direction of Loading	Minimum spacing	Minimum spacing for $C_{\Delta} = 1.0$			
Parallel to Grain	3D	4D			
Perpendicular to		Required spacing for			
Grain	3D	attached members			

Figure 12 - Connection Spacing for Fasteners in a Row [10].

Understanding what "fasteners in a row" means is seen in Figure 13.

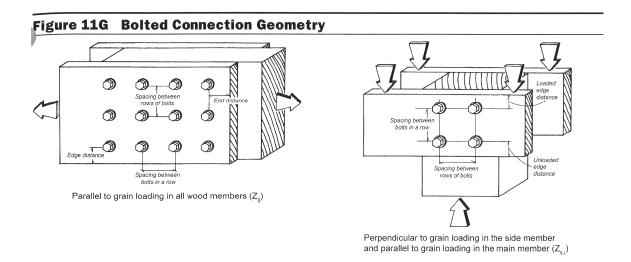


Figure 13 - Diagram of Bolt Spacing [10].

Determining "fasteners in a row" depends on the direction of load. Figure 13 shows how the direction of load changes the designation of spacing between rows and between bolts in a row. Figure 14 then shows the minimum distances for spacing between rows.

Table 11.5.1D	Spacing Requirements Between Rows ^{1,2}			
Direction of Loading	Minimum Edge Distance			
Parallel to Grain:	1.5D			
Perpendicular to Grain:				
when $\ell/D \le 2$	2.5D			
when $2 < \ell/D < 6$	$(5\ell + 10D) / 8$			
when $\ell/D \ge 6$	5D			

- 1. The ℓ/D ratio used to determine the minimum edge distance shall be the lesser of:
 - (a) length of fastener in wood main member/D = ℓ_m /D (b) total length of fastener in wood side member(s)/D = ℓ_s /D
- 2. The spacing between outer rows of fasteners paralleling the member on a single splife plate shall not exceed 5" (see Figure 11H).

Figure 14 - Connection Spacing Between Rows [10].

Since the orientation of the row changes depending on the load path, one must take the greater of the two spacing conditions. Since the ratio of ℓ/D will never exceed six and the

next highest spacing is 4D from Figure 12, the spacing between bolts in the lamella connection will, at most, be four times the bolt diameter. Whether or not the $2 < \ell/D < 6$ condition from Figure 14 will apply depends on the member thickness and bolt diameter and whether or not its associated minimum spacing will be greater than 4D. For 2x lumber, this only happens with $\frac{1}{4}$ " bolts; because of this, using $\frac{1}{4}$ " bolts will result in having a C_d value of 1.0 for all cases.

Combining all of these requirements results in the following two connection details, shown in Figure 15 and Figure 16. One should note that the bolt connection line is at an angle to the perpendicular due to the geometry of the connection.

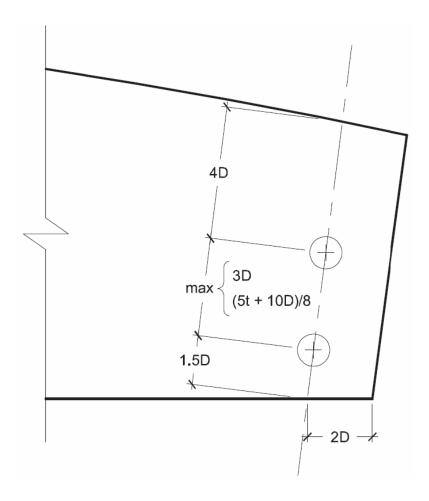


Figure 15 - Connection Detail for $C_{\Delta} = 0.5$.

In most cases, the end distance requirement of 2D is already met or exceeded due to the bevel at the end of the lamella. Even with a bevel cut of 45° , a 2x member will still have about 2'' end distance, taking bolt diameter into consideration, adequate for even 1'' diameter bolts.

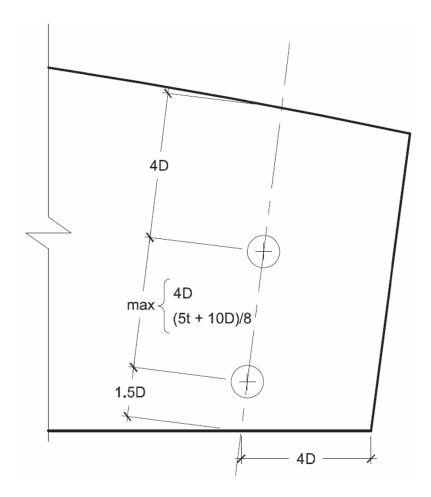


Figure 16 - Connection Detail for $C_{\Delta} = 1.0$.

One may notice that the end distance requirement of 4D changes the geometry of the connection significantly. In order to keep the edge of the bolt on line with the end of the bevel (see Figure 17) and still comply with Figure 16, a 2x member will need bolts with a diameter equal to or smaller than one-half inch. A 3x member could have bolts as large as $\frac{7}{8}$ " in diameter and still comply; however, 3x members would use the "Special joint"

found in Figure 9 so the spacing found in Figure 15 and Figure 16 do not apply. Because of this, only connections using bolts smaller than those just listed can use a C_{Δ} value of 1.0.

One should note that although there are only two bolts shown in the connection on Figure 15 and Figure 16, having more than two bolts is perfectly acceptable so long as the minimum spacing and end distances are met.

Also, complying with the connection detail such that $C_{\Delta} = 1.0$ necessitates increasing the member depth to accommodate the increased spacing or using a shorter lamella while keeping the same radius of curvature on the top.

In order to connect the side lamellas through the continuous lamella, the middle of each lamella must have slotted holes.

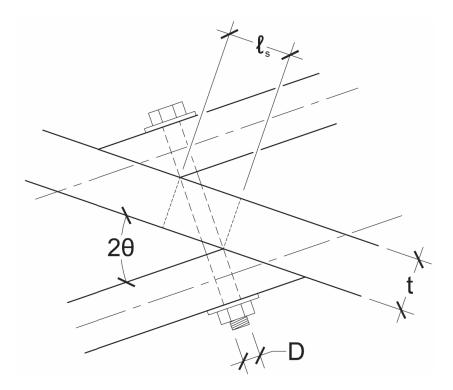


Figure 17 - Connection Slot Plan View.

The slots are located with the same spacing as the bolts. The slot length ℓ_s is

$$\ell_s = t \tan 2\theta + \frac{D}{\cos 2\theta} + 0.25''. \tag{3}$$

In the author's opinion, adding an extra quarter inch to the slot length will allow a little tolerance for fabrication error and make for easier construction.

Since the bolts on the ends can be in two configurations depending on the C_{Δ} value, so too can the slots. Figure 18 and Figure 19 show both configurations. The same spacing and end distances used on the end bolt connection should be used on the slots.

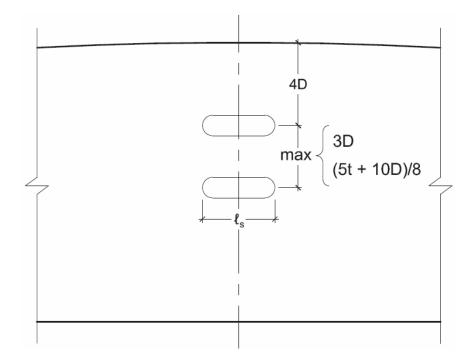


Figure 18 - Connection Slots Elevation View for $C_{\Delta} = 0.5$.

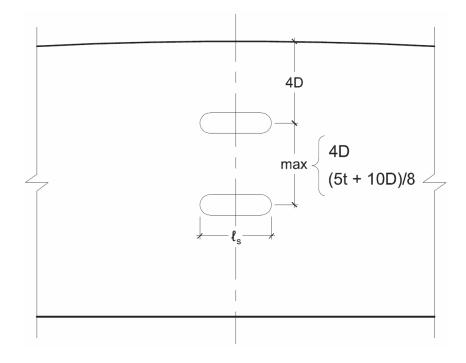


Figure 19 - Connection Slots Elevation View for $C_{\Delta} = 1.0$.

2.1.2 Actual Lamella Length

The arc of the lamella roof is in itself a chord of a larger circle. Using simple trigonometry, one can find the angle β that this big arc subtends of the circle:

$$\beta = 2\arccos\left(\frac{R-T}{R}\right). \tag{4}$$

Figure 20 depicts this layout.

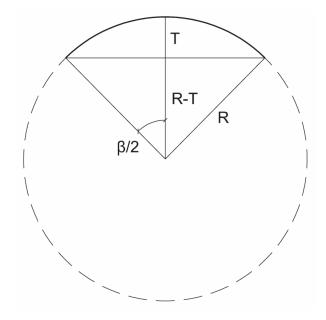


Figure 20 - Roof Arch as a Portion of a Circle.

From there, finding the length of the individual lamellas begins by choosing the number of lamellas, n, that the span of the roof arch. After doing so, one then divides the arch into a series of chords. The secant line between the ends of these chords is the center-to-center length of the lamella ℓ_{c-c} , as shown in Figure 21.

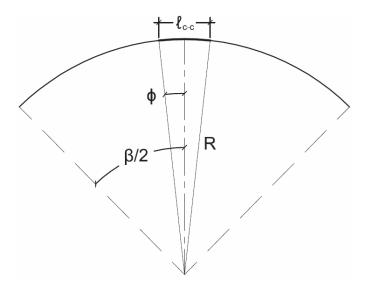


Figure 21 - Lamella as a Portion of the Roof Arch.

This secant line subtends a portion of the arc of the roof where the angle that it subtends is 2ϕ , found by

$$2\phi = \frac{\beta}{n} = \frac{2}{n}\arccos\left(\frac{R-T}{R}\right). \tag{5}$$

The center-to-center length is then found as

$$\ell_{c-c} = 2R\sin\phi. \tag{6}$$

From there, the spacing is simply

Spacing =
$$\ell_{c-c} \tan \theta$$
. (7)

Then, the length of the lamella between bolt centerlines is

$$\ell = \frac{\ell_{c-c}}{\cos \theta} = \frac{2R\sin \phi}{\cos \theta}.$$
 (8)

This length represents the length of the lamella from where its centerline crosses the centerline of the bolts. Combining Equations (7) and (8) results in Table 1.

Table 1 - Spacing of Lamellas with a Given Skew Angle.

Spacing of Lamellas with a Given Skew Angle								
Length	Skew Angle of Lamella Arch [θ] (deg)							
[ℓ] (ft)	19°	19.5°	20°	20.5°	21°	21.5°	22°	22.5°
3.0	0.98	1.00	1.03	1.05	1.08	1.10	1.12	1.15
3.5	1.14	1.17	1.20	1.23	1.25	1.28	1.31	1.34
4.0	1.30	1.34	1.37	1.40	1.43	1.47	1.50	1.53
4.5	1.47	1.50	1.54	1.58	1.61	1.65	1.69	1.72
5.0	1.63	1.67	1.71	1.75	1.79	1.83	1.87	1.91
5.5	1.79	1.84	1.88	1.93	1.97	2.02	2.06	2.10
6.0	1.95	2.00	2.05	2.10	2.15	2.20	2.25	2.30
6.5	2.12	2.17	2.22	2.28	2.33	2.38	2.43	2.49
7.0	2.28	2.34	2.39	2.45	2.51	2.57	2.62	2.68
7.5	2.44	2.50	2.57	2.63	2.69	2.75	2.81	2.87
8.0	2.60	2.67	2.74	2.80	2.87	2.93	3.00	3.06
8.5	2.77	2.84	2.91	2.98	3.05	3.12	3.18	3.25
9.0	2.93	3.00	3.08	3.15	3.23	3.30	3.37	3.44
9.5	3.09	3.17	3.25	3.33	3.40	3.48	3.56	3.64
10.0	3.26	3.34	3.42	3.50	3.58	3.67	3.75	3.83
10.5	3.42	3.50	3.59	3.68	3.76	3.85	3.93	4.02
11.0	3.58	3.67	3.76	3.85	3.94	4.03	4.12	4.21
11.5	3.74	3.84	3.93	4.03	4.12	4.21	4.31	4.40
12.0	3.91	4.01	4.10	4.20	4.30	4.40	4.50	4.59
12.5	4.07	4.17	4.28	4.38	4.48	4.58	4.68	4.78
13.0	4.23	4.34	4.45	4.55	4.66	4.76	4.87	4.97
13.5	4.40	4.51	4.62	4.73	4.84	4.95	5.06	5.17
14.0	4.56	4.67	4.79	4.90	5.02	5.13	5.24	5.36
14.5	4.72	4.84	4.96	5.08	5.20	5.31	5.43	5.55
15.0	4.88	5.01	5.13	5.25	5.38	5.50	5.62	5.74
15.5	5.05	5.17	5.30	5.43	5.55	5.68	5.81	5.93
16.0	5.21	5.34	5.47	5.60	5.73	5.86	5.99	6.12
16.5	5.37	5.51	5.64	5.78	5.91	6.05	6.18	6.31
17.0	5.53	5.67	5.81	5.95	6.09	6.23	6.37	6.51
17.5	5.70	5.84	5.99	6.13	6.27	6.41	6.56	6.70
18.0	5.86	6.01	6.16	6.30	6.45	6.60	6.74	6.89
18.5	6.02	6.18	6.33	6.48	6.63	6.78	6.93	7.08
19.0	6.19	6.34	6.50	6.65	6.81	6.96	7.12	7.27
19.5	6.35	6.51	6.67	6.83	6.99	7.15	7.30	7.46
20.0	6.51	6.68	6.84	7.00	7.17	7.33	7.49	7.65

Figure 22 shows a plan view of this situation while Figure 23 shows a detailed view.

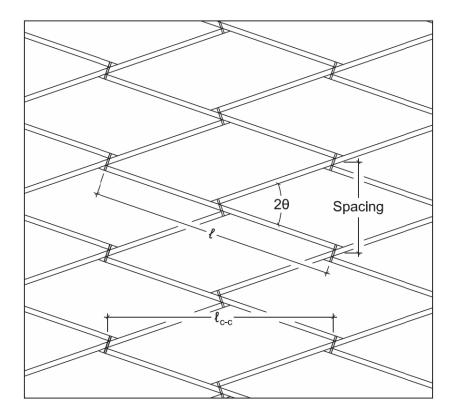


Figure 22 - Lamella Length and Spacing.

Since the lamellas are connected eccentrically, their length must be adjusted to take the eccentricity into account. The center-to-center length of the lamella is found by the designer by using Equation (6) and is used for calculations (see Section 5.2). The additional length is a function of the skew of the lamella arches, the diameter of the bolts, and the thickness of the lamellas themselves.

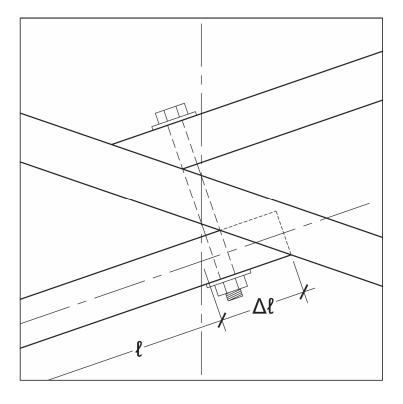


Figure 23 - Additional Length Due to Eccentricity.

This additional length, $\Delta \ell$, can be found through simple trigonometry, though the derivation is somewhat lengthy and thus omitted:

$$\Delta \ell = \frac{1}{2} \left[\frac{t + 2D \tan 2\theta}{2 \sin \theta \cos \theta} + \frac{t}{\tan 2\theta} \right]. \tag{9}$$

Then, since this $\Delta \ell$ is added on each end of the lamella, the total lamella length becomes

$$\ell_T = \ell + 2\Delta\ell. \tag{10}$$

Substituting Equations (8) and (9) into Equation (10) yields

$$\ell_T = \frac{2R\sin\phi}{\cos\theta} + \frac{t + 2D\tan 2\theta}{2\sin\theta\cos\theta} + \frac{t}{\tan 2\theta}.$$
 (11)

Now the lamella subtends an angle $2\phi_T$ in its own skewed plane, similar to Equation (5), where according to Warner [11],

$$\phi_T = \arcsin\left(\frac{\ell_T}{2R}\right). \tag{12}$$

After this, one must find the bevel angles on the ends of the lamellas. There are two that must be found – the radial bevel and the skew bevel. Warner goes through a derivation in his monograph that shows that for typical lamella roofs, where there are sufficient lamellas per arch such that ϕ_T is around or less than 10°, the radial bevel can be approximated to ϕ_T and the skew bevel to 20 [11]. It should be noted that the bolt holes are also skewed to approximately the same ϕ_T angle. These two bevels are illustrated in Figure 24.

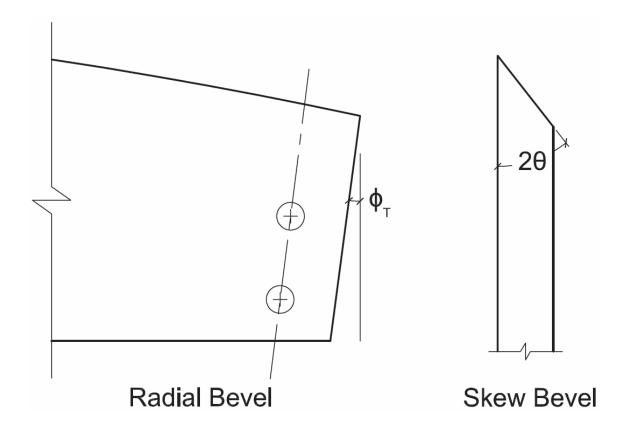


Figure 24 - Lamella End Bevels.

While one could firm down a more exact value for the bevel angles, expecting typical construction power tools to cut an angle to anything more precise than a whole number is impractical. The same holds true for the bolt line skew.

Another factor for constructability considerations is the "shift" of the connection, as illustrated in Figure 25.

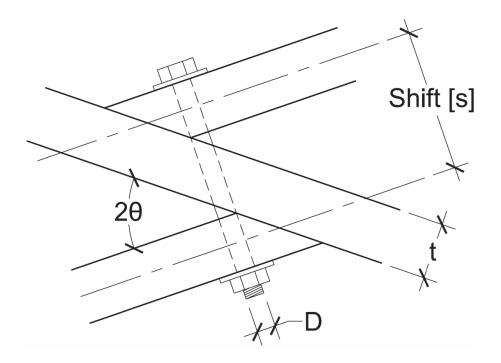


Figure 25 - "Shift" of the Lamella Connection.

The shift is determined by the thickness of the lamella, the skew angle of the lamella arch, and the bolt diameter. From trigonometry, the shift can be found by

$$s = D \tan 2\theta + \frac{t}{\cos 2\theta} + t, \tag{13}$$

or reduced, according to Masani [4], as

$$s = t(1 + \sec 2\theta) + D\tan 2\theta. \tag{14}$$

This information is easily tabulated as shown in Table 2. Note that this table only applies to 2x lumber with an actual thickness of 1.5 inches.

Table 2 - Lamella Connection Shift.

	Shift of Lamellas with a Given Skew Angle [s] (in)							
Bolt			Skew Ar	ngle of Lam	nella Arch [θ] (deg)		
Diameter	19°	19.5°	20°	20.5°	21°	21.5°	22°	22.5°
1/4"	3.60	3.63	3.67	3.70	3.74	3.78	3.83	3.87
5/16"	3.65	3.68	3.72	3.76	3.80	3.84	3.89	3.93
3/8"	3.70	3.73	3.77	3.81	3.86	3.90	3.95	4.00
1/2"	3.79	3.84	3.88	3.92	3.97	4.02	4.07	4.12
5/8"	3.89	3.94	3.98	4.03	4.08	4.13	4.19	4.25
3/4"	3.99	4.04	4.09	4.14	4.19	4.25	4.31	4.37
7/8"	4.09	4.14	4.19	4.25	4.31	4.37	4.43	4.50
1"	4.18	4.24	4.30	4.36	4.42	4.48	4.55	4.62

From here one can determine the length of bolt needed for the connection by adding a thickness of lamella and extra for the nut and washers. An inch to an inch and half extra should suffice. Thus,

$$\ell_{bolt} \ge t \left(2 + \sec 2\theta \right) + D \tan 2\theta + \left(\text{extra} \right) = s + t + \left(\text{extra} \right). \tag{15}$$

Obviously the builder would want to choose a length of bolt commonly available by manufacturers.

2.1.3 Top Curve Cut

When looking at a section view of the roof arch, the top curve of the lamella follows the same circular curve as the entire roof. However, since the lamellas themselves are skewed, the curvature on the top is elliptic.

Masani states that the elliptic curve on the top of the lamella can be approximated by a simple circular arc with a radius slightly larger than that of the roof itself [4]. This is probably due to the fact that since the lamella is so short in comparison to the entire curvature of the roof, the minute differences between the elliptic curve and the circular curve will be indistinguishable. In fact, when the author attempted to draw an illustration depicting the difference between the elliptic curve and the circular curve, the difference was so minute that unless he zoomed in very close, it was impossible to differentiate between the two.

This arc would have a span of ℓ_T and a rise of

$$T' = R - \frac{\ell_T}{2\tan\phi_T},\tag{16}$$

along with a radius of

$$R' = \frac{4(T')^{2} + \ell_{T}^{2}}{8(T')} = \frac{4\left(R - \frac{\ell_{T}}{2\tan\phi_{T}}\right)^{2} + \ell_{T}^{2}}{8\left(R - \frac{\ell_{T}}{2\tan\phi_{T}}\right)}.$$
(17)

This arc would have a point of tangency at the midpoint of the lamella at the very top of the plank. The detail for the top curve is shown in Figure 26 (*d* is the depth of the lamella).

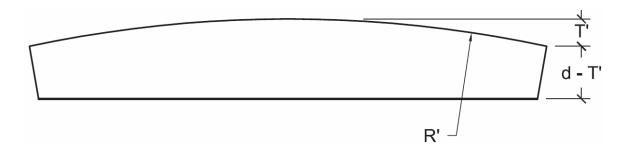


Figure 26 - Top Curvature Cut Detail.

3 Analysis of the Lamella Arch

Analysis of the lamella roof is carried out assuming that it acts like a two-hinged arch [1, 4]. Unfortunately, there exists no closed-form analytical solution for the moments, thrusts, and horizontal reactions of such an arch.

3.1 Arch Approximation Methods

Before the advent of calculators and computerized structural analysis packages, several approximate analytical methods were developed to solve for the forces in a two-hinged arch under a given loading condition. Two of those methods were the von Kármán Method and the Scofield Method. The author has also conducted a computer analysis of the arch using a finite element analysis method, which will also be discussed.

3.1.1 von Kármán Method

Sometime in the late 1930's Theodore von Kármán developed an approximate analysis for the two-hinged arch while working at the California Institute of Technology. His approximation assumes the arch follows a parabolic curve instead of a circular to simplify the derivations. As von Kármán developed this method while in California, it is perhaps not surprising that snow loading is not included; however, he includes radial loads from the structure weight, uniform vertical loads from live loads, and uniform horizontal loads from wind loads [12]. In the following sections, Equations (18) through (52) are taken from or derived from von Kármán's paper [12].

3.1.1.1 Live Load (Uniform Vertical Load)

For live loads, von Kármán replaces the uniform vertical load with a uniform perpendicular load (perpendicular to the curve of the arch along its entire length) and a uniform horizontal load along the vertical section, as depicted in Figure 27.

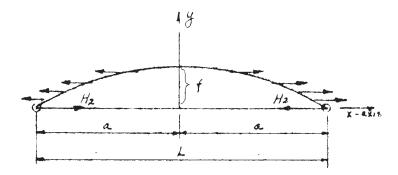


Figure 27 - Live Load Replacement [12].

The vertical reaction from the perpendicular load is

$$V_1 = pR\sin\phi_0 \tag{18}$$

where

$$\sin \phi_0 = \frac{a}{R} \tag{19}$$

SO

$$V_1 = pR\left(\frac{a}{R}\right). \tag{20}$$

Since a is half the span, Equation (20) can be rewritten as

$$V_{\ell} = \frac{pS}{2},\tag{21}$$

which should come as no surprise as it is the same as one would find through elementary statics from a uniform vertical load on the projected member length.

The horizontal reaction is a combination of the reaction due to the perpendicular load and the uniform vertical load. For this von Kármán writes

$$H_{\ell} = H_1 + H_2 = pR\cos\phi_0 + \frac{3}{7}pf. \tag{22}$$

Substituting

$$\cos \phi_0 = \frac{R - f}{R},\tag{23}$$

the total horizontal reaction due to live load becomes

$$H_{\ell} = pR - \frac{4}{7} pf. \tag{24}$$

After this, the thrust at the springing points can then be approximately found as

$$T_{\ell} = p\left(R + \frac{3}{7}f\right). \tag{25}$$

The moment equation for the arch is

$$M_{\ell} = \frac{pf^2}{14} \left(1 - 8\frac{x^2}{a^2} + 7\frac{x^4}{a^4} \right), \tag{26}$$

but since a is half the arch span,

$$M_{\ell} = \frac{pf^2}{14} \left(1 - 32 \frac{x^2}{S^2} + 112 \frac{x^4}{S^4} \right). \tag{27}$$

At the center point of the arch the positive moment will be greatest. This is also the spot where x is equal to zero, which simplifies Equation (27) to

$$M_{\ell} = \frac{pf^2}{14}.\tag{28}$$

3.1.1.2 Dead Load (Radial Load)

For this analysis, the dead load q is defined as the load per unit length of the arc. Since the curvature of the arc changes with its distance from the centerpoint, the load on the horizontal projection of the arch also changes. Coincidentally, the dead load can be considered to have a load q plus an additional variably distributed load. This load increases as it gets closer to the springing points of the arch.

Because of this, the total horizontal reaction is the sum of the reaction from the uniform load and the reaction of the variable load:

$$H_d = H_3 + H_4, (29)$$

where

$$H_3 = qR - \frac{4}{7}qf \tag{30}$$

and

$$H_4 = \frac{4}{21} qf. {31}$$

Combining Equations (30) and (31), the equation for the dead load horizontal reaction becomes

$$H_d = qR - \frac{8}{21}qf. {(32)}$$

Since the additional variably distributed load changes with distance from the center of the arch, the vertical component of the force in the arch also changes. The equation for the vertical component can be expressed as

$$V_{xd} = qx + q \frac{f}{R} \frac{x^3}{3a^2}.$$
 (33)

At the arch springing points, x equals a, which makes the vertical reaction

$$V_d = qa\left(1 + \frac{f}{3R}\right). \tag{34}$$

Substituting half of the span for a yields

$$V_d = \frac{qS}{2} \left(1 + \frac{f}{3R} \right). \tag{35}$$

The thrust at any point in the arch can be found by

$$T_d = \sqrt{H_d^2 + V_{xd}^2}. (36)$$

One can substitute Equations (32) and (34) for H_d and V_{xd} in the above equations, then solve for the sill thrust by substituting a for x:

$$T_d = q\sqrt{\left(R - \frac{8}{21}f\right)^2 + a^2\left(1 - \frac{f}{3R}\right)^2},$$
 (37)

which can be approximated as

$$T_d = q \left(R + \frac{13}{21} f \right). \tag{38}$$

Like the horizontal reaction, the moment due to dead load is also a combination of a uniform load and the variably distributed load:

$$M_d = M_3 + M_4, (39)$$

where

$$M_3 = \frac{qf^2}{14} \left(1 - 8\frac{x^2}{a^2} + 7\frac{x^4}{a^4} \right) \tag{40}$$

and

$$M_4 = \frac{-qf^2}{42} \left(1 - 8\frac{x^2}{a^2} + 7\frac{x^4}{a^4} \right). \tag{41}$$

Combining these results in the dead load moment equation, we have

$$M_d = \frac{qf^2}{21} \left(1 - 8\frac{x^2}{a^2} + 7\frac{x^4}{a^4} \right). \tag{42}$$

Or, since a is half of the arch span,

$$M_d = \frac{qf^2}{21} \left(1 - 32 \frac{x^2}{S^2} + 112 \frac{x^4}{S^4} \right). \tag{43}$$

Also, the positive moment will be the greatest in the middle of the arch which is where *x* is equal to zero. At this point, the maximum positive moment is given by:

$$M_d = \frac{qf^2}{21}. (44)$$

Here von Kármán comments that the dead load moment is 2/3 the live load moment with the same load magnitude.

3.1.1.3 Wind Load (Uniform Vertical Load)

The wind load acting on the arch is assumed to be a uniformly distributed vertical load w acting on the vertical projection of the arch, as shown in Figure 28.

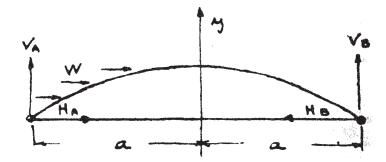


Figure 28 - Wind Load on the Arch [12].

The vertical reactions are equal and opposite and can be found through simple statics.

They are:

$$V_{A} = -V_{b} = \frac{-wf^{2}}{4a}.$$
 (45)

Since a is half of the span, Equation (45) becomes

$$V_A = -V_b = \frac{-wf^2}{2S}. (46)$$

From here, the two horizontal reactions are found to be:

$$H_A = -\frac{5}{7} wf \tag{47}$$

and

$$H_B = \frac{2}{7} wf, (48)$$

with the direction of each horizontal reaction being opposite that of the direction of the wind load, as expected.

The moment equation for the arch changes depending on which side of the arch is being examined. On the windward side of the arch, the moment formula is

$$M_{w} = \frac{wf^{2}}{28} \left[-1 - 7\frac{x}{a} + 8\left(\frac{x}{a}\right)^{2} - 14\left(\frac{x}{a}\right)^{4} \right]. \tag{49}$$

If one substitutes half of the span for a, it becomes

$$M_{w} = \frac{wf^{2}}{28} \left[-1 - 14\frac{x}{S} + 32\left(\frac{x}{S}\right)^{2} - 224\left(\frac{x}{S}\right)^{4} \right].$$
 (50)

On the leeward side of the arch, the moment formula is:

$$M_{w} = \frac{wf^{2}}{28} \left[-1 - 7\frac{x}{a} + 8\left(\frac{x}{a}\right)^{2} \right]$$
 (51)

or

$$M_{w} = \frac{wf^{2}}{28} \left[-1 - 14\frac{x}{S} + 32\left(\frac{x}{S}\right)^{2} \right].$$
 (52)

3.1.2 Scofield Method

This method comes from the book *Modern Timber Engineering*, 5th ed. published in 1963 [3]. Scofield appears to partially base his design calculations on the von Kármán method. In this analysis, four primary load patterns are considered: radial loads from the dead weight of the structure, uniform vertical loads from live load, uniform horizontal

loads from wind load, and uniform vertical loads on half of the structure from snow drift loads [3]. In the following sections, Equations (53) through (71) are from Scofield [3].

3.1.2.1 Dead Load (Radial Load)

The dead load on an arch acts upon its entire curved length, not just the projected horizontal length. The loading diagram appears in Figure 29.

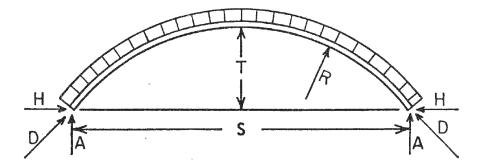


Figure 29 - Dead Load [3].

Scofield lists the following equations to solve for the arch forces:

$$A = 0.5 dS \sqrt{1 + \frac{16}{3} \left(\frac{T}{S}\right)^2},\tag{53}$$

$$H = \frac{AS}{2T} - dR, (54)$$

$$D = H\left(\frac{R - T}{R}\right) + \frac{AS}{2R},\tag{55}$$

and

Maximum
$$M = 0.068 dT^2$$
. (56)

3.1.2.2 Construction Live Load (Uniform Vertical Load)

The loading for a construction live load, acts on the horizontal projection of the arch, making the vertical reactions easily solved by elementary statics. The loading diagram is shown in Figure 30.

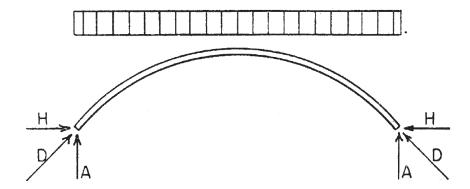


Figure 30 - Construction Live Load [3].

Equations (57) through (59) can be employed to calculate the roof live load forces:

$$A = 0.5L_rS, (57)$$

$$H = L_r (R - 0.57356T), (58)$$

and

Maximum
$$M = -0.09092L_rT^2$$
. (59)

The thrust, D, is the same as Equation (55) in the radial load case.

3.1.2.3 Snow Drift Load (Uniform Vertical Load on Half of Structure)

Snow is assumed to accumulate on the leeward face of the lamella roof with a uniform weight distribution, as seen in Figure 31.

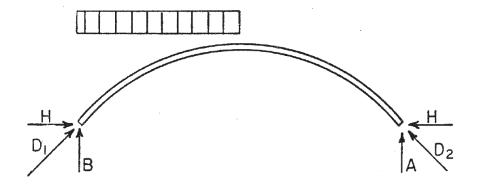


Figure 31 - Snow Drift Load [3].

The unbalanced loading creates unbalanced vertical support conditions, necessitating the addition of reaction B. This loading also creates two different thrusts, D_1 and D_2 .

Equations (60) through (65) are employed to determine the forces for the snow drift load:

$$A = \frac{sS}{8},\tag{60}$$

$$B = \frac{3sS}{8},\tag{61}$$

$$H = \frac{s}{2} (R - 0.57356T), \tag{62}$$

$$D_{1} = H\left(\frac{R - T}{R}\right) + \frac{BS}{2R},\tag{63}$$

$$D_2 = H\left(\frac{R - T}{R}\right) + \frac{AS}{2R},\tag{64}$$

and

Maximum M =
$$\frac{AS}{2} - HT - R(\sqrt{A^2 + H^2} - H)$$
. (65)

3.1.2.4 Wind Load (Uniform Horizontal Load)

Wind is assumed to be uniformly distributed over the rise of the arch with the load projected on the height of the arch, as seen in Figure 32.



Figure 32 - Wind Load [3].

The horizontal load adds different vertical, horizontal, and thrust reactions at each springing point. These forces are:

$$A = -B = \frac{WT^2}{2S},\tag{66}$$

$$H_1 = \frac{19WT}{64},\tag{67}$$

$$H_2 = \frac{45WT}{64},\tag{68}$$

$$D_{1} = \frac{WT}{64} \left(13 - \frac{3T}{R} \right), \tag{69}$$

$$D_2 = \frac{WT}{64} \left(45 - \frac{29T}{R} \right),\tag{70}$$

and

Maximum
$$M = 0.154WT^2$$
. (71)

3.1.3 Finite Element Method

As stated in Section 4.1, there is no closed-form analytical solution for the forces and reactions in an arch. The only way to truly "solve" for the forces and reactions is to perform a finite element analysis.

Beam elements are the finite elements of choice for this model. They are made up of individual linear elements with end reactions on the local x- and y-axes as well as moment reactions on either end, giving six degrees of freedom per element. Figure 33 depicts this layout.

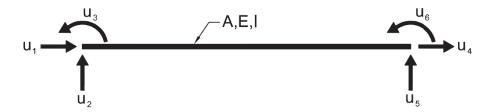


Figure 33 - Beam Element.

The author chose to approximate the lamella arch with 40 beam elements of the same length ℓ , cross-sectional area A, moment of inertia I, and modulus of elasticity E. The beam elements follow the curve of the arch with the nodes falling on the line of the circular arch, creating a series of secant lines. Obviously, the more beam elements used, the closer the analysis will be to the "exact" solution. However, 40 beam elements give a good enough approximation for design purposes.

Gravity loads are placed at the nodes between beam elements. To solve for the end reactions in the arch is a multi-step process. First, one must solve for the displacements of the nodal points [13]:

$$\tilde{U} = \tilde{K}^{-1}\tilde{F}.\tag{72}$$

These matrices have already been reduced to include only the unrestrained degrees of freedom (i.e., the end support conditions were removed). Once the nodal displacements are found, the end support conditions are added back into the \tilde{U} matrix and the \tilde{K} matrix is expanded to include the end stiffnesses. The nodal reactions at the springing ends are then found by [13]:

$$\tilde{F} = \tilde{K}\tilde{U}. \tag{73}$$

However, in order to do this process, one must begin with both the combined stiffness matrix \tilde{K} and the combined force vector \tilde{F} . The \tilde{K} matrix is made up of all the stiffness matrices from all the beam elements. The beam element stiffness matrix, \tilde{k} , is as follows [14]:

$$\tilde{k} = \begin{bmatrix}
\frac{EA}{\ell} & 0 & 0 & \frac{-EA}{\ell} & 0 & 0 \\
0 & \frac{12EI}{\ell^3} & \frac{6EI}{\ell^2} & 0 & \frac{-12EI}{\ell^3} & \frac{6EI}{\ell^2} \\
0 & \frac{6EI}{\ell^2} & \frac{4EI}{\ell} & 0 & \frac{-6EI}{\ell^2} & \frac{2EI}{\ell} \\
\frac{-EA}{\ell} & 0 & 0 & \frac{EA}{\ell} & 0 & 0 \\
0 & \frac{-12EI}{\ell^3} & \frac{-6EI}{\ell^2} & 0 & \frac{12EI}{\ell^3} & \frac{-6EI}{\ell^2} \\
0 & \frac{6EI}{\ell^2} & \frac{2EI}{\ell} & 0 & \frac{-6EI}{\ell^2} & \frac{4EI}{\ell}
\end{bmatrix}.$$
(74)

This stiffness matrix only applies when the beam element is oriented so that it runs parallel to the horizon. However, the 40 beam elements that approximate the arch are all

rotated to different angles. A generalized image of the rotated beam elements is shown in Figure 34.

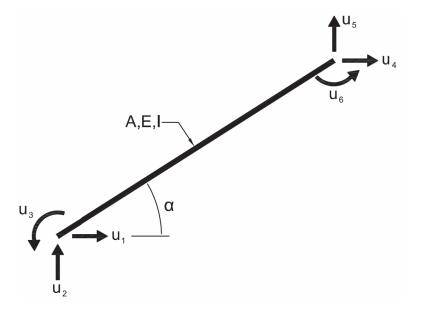


Figure 34 - Rotated Beam Element.

Since the 40 beam elements are all rotated to some degree, the stiffness matrix for each must be changed. To do this, the beam stiffness matrix must be multiplied by a transformation matrix as such [13]:

$$\tilde{k}_{rot} = \tilde{T}^T \tilde{k} \tilde{T}, \tag{75}$$

where

$$T = \begin{bmatrix} \cos \alpha & \sin \alpha & 0 & 0 & 0 & 0 \\ -\sin \alpha & \cos \alpha & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & \cos \alpha & \sin \alpha & 0 \\ 0 & 0 & 0 & -\sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix},$$
(76)

which gives the stiffness matrix for a beam element rotated to an angle α . This process must be carried out individually for all 40 beam elements in the arch. After that, all of the individual \tilde{k}_{rot} matrices must be combined by aligning corresponding degrees of freedom [13]. This results in a \tilde{K} matrix of size 123 x 123. Obviously, this matrix math is far too cumbersome to do manually, so computer aid is required.

The next part of the process is assembling the forces vector, \tilde{F} . As stated before, the uniform loads are converted to nodal loads at each of the nodes between beam elements. The loads are developed for the various load cases as described in the following subsections.

3.1.3.1 Dead Loads

The dead load is a function of the length of the beam elements adjacent to the corresponding node. Simply stated, the dead load acting on the node is the weight per unit length of the beam element (and other structure load assumed to be included with dead load) multiplied by half of the beam element length on either side of the node. If all beam elements are the same length, the nodal loads at every node besides the nodes at the springing points of the arch should be exactly the same. The loads at the springing point nodes should be exactly half of the loads on the rest of the nodes.

3.1.3.2 Live Load

The live load is a function of the horizontal projection of the beam elements around a node. As each beam element has some rotation of angle α , the horizontal component of the beam element is the beam element length multiplied by the cosine of the angle, or

$$\ell_{r} = \ell \cos \alpha. \tag{77}$$

The live load on a node will then be the product of the uniform vertical load and the sum of half of the horizontal components of the adjacent beam elements. These values will vary depending on the curvature of the arch. If the arch was flat, all of the nodal loads would be the same since the beam element horizontal components would be the same as their lengths.

3.1.3.3 Snow Loads

The loads due to snow come in two varieties: balanced snow loads (S_b) and unbalanced snow loads (S_u) or drift loads. Since the lamella roof is curved, this complicates things slightly in finding the balanced and unbalanced snow loads. To find these loads, section 7.4.3 of the ASCE 7-10 code was used. The code specifies that the loading diagrams for the different curvature cases should be based on Figure 7-3 of the ASCE 7-10 code, which can be found in Appendix C, page 144. The flat roof snow load, p_f , is used to find the sloped roof snow loads and can be found by

$$p_f = 0.7C_e C_t I_s p_g, (78)$$

where

 C_e = Exposure Factor,

 C_t = Temperature Factor,

 I_s = Importance Factor,

 p_g = Ground Snow Load.

To simplify calculations and the generation of load tables, some assumptions were made. The thermal factor C_t is assumed to be 1.2 based on ASCE 7-10, Table 7-3. The

assumption is that the lamella roof will be covering an unheated space or one that is open to the air. This may not be true for all cases but will give the worst case for a conservative design. From this thermal factor, the slope factor C_s can be found from the graphs in Figure 35.

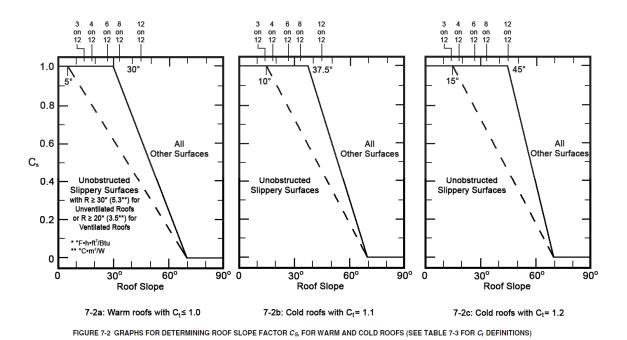


Figure 35 - Graphs for Determining Roof Slope Factor C_s [15].

Since the thermal factor is 1.2, the far right graph in Figure 35 must be used. Also, the roofing surface is assumed to not be an unobstructed slippery surface, demarked by the solid line.

ASCE Figure 7-3 (see Appendix C, page 144) has variables C_s^* and C_s^{**} used for calculations. For Case 1, C_s^* will be equal to 1.0, found by reading the rightmost chart above at an eave slope of 30°. From here, it is easy to see that C_s^{**} for all cases will also be 1.0, since that value is taken at a 30° slope, too. It is also assumed that the lamella

roof will not be abutting any other structure so the alternate distribution for Case 2 and Case 3 in ASCE Figure 7-3 need not be used.

ASCE Figure 7-3 also necessitates finding the exposure factor C_e for the roof. This factor is listed in Table 7-2 of the ASCE 7-10 and ranges from 0.7 to 1.2. Here it is assumed that the lamella roof falls under Exposure Category C and that the structure is "Fully Exposed," giving a C_e value of 0.9 (see Appendix C, page 143 for Table 7-2). For the majority of buildings, this C_e value will be a conservative design value.

Using these assumptions, ASCE Figure 7-3 was adjusted by the author to become the Simplified Figure 7-3 found in Appendix C, page 145. The loading patterns from that table are used for snow load calculations.

The Importance Factor I_s for the roof is assumed to be 1.10, which correlates to a building that falls under Risk Category III.

Section 7.3.4 of the ASCE 7-10 also stipulates a Minimum Snow Load for Low-Slope Roofs, p_m . For curved roofs, this occurs when the angle between the springing end and the apex of the roof is less than ten degrees [15]. In order to ignore this case, loads were only calculated for roofs where that angle exceeds ten degrees. It should also be noted that for ground snow loads greater than 20 psf,

$$p_{m} = (20 \text{ psf})I_{s}, \tag{79}$$

which would always be less than the p_f loads for anything over 30 psf.

3.1.3.4 Wind Loads

It should be noted that the wind loads stipulated in ASCE 7-10 are very generalized and most likely far greater than anything the building structure will ever experience. In many cases, lower wind loads can be found by doing wind tunnel testing with a scale model of the building and surrounding area, including other buildings and topological configurations. Since it is unrealistic to perform this analysis for every possible arch configuration, the approximate method from ASCE 7-10 is used.

Wind loads on the lamella roof are based on the Directional Procedure from ASCE 7-10, Chapter 27 [15]. First, one must find the velocity pressure:

$$q_z = 0.00256K_zK_{zt}K_dV^2 \text{ (lb/ft}^2).$$
 (80)

From this, the design wind pressures can be calculated:

$$p = qGC_p - q_iGC_{pi}. (81)$$

The wind directionality factor, K_d , is given as 0.85 for an arched roof according to Table 26.6-1 in ASCE 7-10 [15]. It is assumed that the lamella structure is on flat ground with no topographic irregularities, so the topographic factor K_{zt} can be set equal to 1.0, as shown in Section 26.8.2 in ASCE 7-10 [15].

For ease of calculation, the building is assumed to be in Exposure Category C, the second-windiest Category. This means that the building is assumed to be in an area of flat, open country or flatlands. The next-windiest is Category D, which assumes conditions like open water and/or similar for over 5,000 feet upwind.

The Exposure Category affects the calculation of K_z , the velocity pressure exposure coefficient, as found by

$$K_{z} = \max \left[2.01 \left(\frac{z}{z_{g}} \right)^{\left(\frac{2}{\alpha} \right)}, 2.01 \left(\frac{15}{z_{g}} \right)^{\left(\frac{2}{\alpha} \right)} \right], \tag{82}$$

where z is the height above ground where the pressure is taken and z_g and α are two coefficients found in Table 26.9-1 of ASCE 7-10. For Exposure Category C, they are 900 and 9.5, respectively [15].

The Directional Procedure also specifies finding a Pressure Coefficient C_p for the structure. It specifies two different scenarios for an arched roof: one with the roof springing from an elevated wall and one with the roof springing from ground level. It is assumed that the lamella arch is part of a roof and thus springs from an elevated wall.

The arch acts kind of like the wing of an airplane in that the windward side receives downward pressure while the middle and leeward parts receive uplift. This loading scenario is presented in Figure 36, which graphically depicts that which is shown in ASCE 7-10 Figure 27.4-3. Areas that receive uplift have a negative value for C_p value. The windward and middle portions of the arch have a C_p value that is dependent on r, the ratio of the rise, R to the span, S [15].

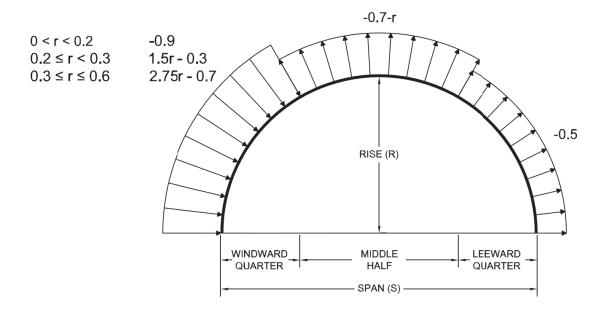


Figure 36 - Pressure Coefficients C_p for Arched Roof [15].

One can easily see the effect of uplift on the middle and leeward portions of the arched roof. It should also be noted that the C_p coefficient for the windward portion changes depending on the Rise-to-Span ratio.

The final necessary piece is the internal pressure coefficient GC_{pi} . It is assumed that the roof will be enclosed as per Section 26.10 in the ASCE 7-10. This gives a GC_{pi} value of ± 0.18 according to Table 26.11-1.

It should also be noted that this procedure is only valid for buildings classified as "low-rise," meaning buildings under 60 feet in height. Because of this, the load tables developed by the author do not list rises above this 60-foot limit.

3.2 Comparison of Analysis Methods

The three analysis methods discussed previously will now be compared to see if the von Kármán method or the Scofield method can be used for a close approximation instead of a complicated computer analysis.

The methods were tested using an arch with the following characteristics:

- Risk Category III structure
- 40 ft span
- 10 psf dead load
- 120 mph wind zone (equates to 10 psf for von Kármán/Scofield)
- 30 psf ground snow
- 20 psf construction live load

The loads on the arch were found in intervals for rises between two and twenty feet. Since, in the finite element model, the moment capacity of the arch is dependent on the stiffness, two tables for each loading case were developed. One reflects a flexible arch with the ratio of the moment of inertia to the area (I/A) equal to one, and the other a stiff arch with the I/A ratio equal to one hundred.

For a comparison to real lumber shapes, a 24x24 sawn lumber member has the greatest I/A ratio at about 46. A 2x3 has the smallest I/A ratio at about 0.52 but is not a deep enough member to use for lamella construction. Any lumber with at least a 4 inch nominal depth has an I/A of at least 1.

In Table 3 through Table 10, the highlighted light grey cells feature values found through the von Kármán or Scofield methods, which are within ten percent of the FEA model values. Also note that entries with a positive percent difference have values higher than those from the FEA model and thus are conservative design values. These values are highlighted in dark gray. Essentially, any highlighted entries would be suitable approximations for the given Rise-to-Span ratio. Dashed entries are greater than 1000% difference.

3.2.1 Dead Loads

The values given by von Kármán and Scofield for the end reactions and arch thrusts are, for the most part, close to those found through the finite element analysis.

Table 3 - Flexible Arch Analyses Comparison for Dead Load.

Dead Load - Flexible Arch										
Rise/Span von Kármán Method						Scofield Method				
[r] (-)	Vertical Reaction	Horizontal Reaction	Thrust	Moment	Vertical Reaction	Horizontal Reaction	Thrust	Moment		
0.050	0.0%	0.3%	0.4%	310.1%	0.0%	0.4%	0.5%	485.6%		
0.075	0.0%	0.2%	0.3%	-2.6%	0.0%	0.4%	0.5%	39.1%		
0.100	-0.1%	0.1%	0.4%	-18.3%	0.0%	0.4%	0.7%	16.7%		
0.125	-0.2%	0.2%	0.6%	-22.9%	0.0%	0.6%	1.0%	10.1%		
0.150	-0.4%	0.3%	0.8%	-25.3%	-0.1%	0.8%	1.2%	6.7%		
0.175	-0.6%	0.5%	1.0%	-27.1%	-0.1%	1.0%	1.5%	4.1%		
0.200	-1.0%	0.8%	1.3%	-28.7%	-0.2%	1.3%	1.6%	1.7%		
0.225	-1.6%	1.2%	1.5%	-30.4%	-0.3%	1.5%	1.8%	-0.6%		
0.250	-2.2%	1.8%	1.8%	-32.1%	-0.4%	1.8%	1.8%	-3.0%		
0.275	-3.0%	2.5%	2.1%	-33.8%	-0.5%	2.0%	1.8%	-5.4%		
0.300	-3.9%	3.5%	2.4%	-35.4%	-0.7%	2.3%	1.6%	-7.8%		
0.325	-5.0%	4.7%	2.6%	-37.1%	-0.9%	2.6%	1.5%	-10.2%		
0.350	-6.2%	6.2%	2.9%	-38.8%	-1.1%	2.9%	1.2%	-12.6%		
0.375	-7.5%	8.1%	3.2%	-40.4%	-1.3%	3.2%	0.9%	-15.0%		
0.400	-8.9%	10.2%	3.4%	-42.1%	-1.6%	3.6%	0.5%	-17.3%		
0.425	-10.3%	12.8%	3.7%	-43.6%	-1.8%	4.0%	0.1%	-19.5%		
0.450	-11.9%	15.9%	4.0%	-45.1%	-2.1%	4.4%	-0.4%	-21.7%		
0.475	-13.5%	19.5%	4.2%	-46.6%	-2.4%	4.9%	-0.9%	-23.8%		
0.500	-15.1%	23.7%	4.5%	-48.0%	-2.7%	5.4%	-1.4%	-25.8%		

The values for the von Kármán method are fairly close except for the moment calculation. Since few, if any, arched lamella roofs have such a small r ratio, it would be safe to say that the von Kármán method fails for the moment calculation. The values for

the Scofield method match very well for all r values except for the moment entries, which are within range until the Rise is about one-quarter of the span.

Table 4 - Stiff Arch Analyses Comparison for Dead Load.

	Dead Load - Stiff Arch									
Rise/Span		von Kármái	Scofield Method							
[r] (-)	Vertical Reaction	Horizontal Reaction	Thrust	Moment	Vertical Reaction	Horizontal Reaction	Thrust	Moment		
0.050	0.0%	32.8%	31.3%	-	0.0%	32.9%	31.4%	-		
0.075	0.0%	14.6%	13.5%	-	0.0%	14.8%	13.7%	-		
0.100	-0.1%	8.3%	7.4%	-	0.0%	8.6%	7.7%	-		
0.125	-0.2%	5.4%	4.7%	-	0.0%	5.8%	5.1%	-		
0.150	-0.4%	3.9%	3.4%	564.9%	-0.1%	4.4%	3.9%	849.5%		
0.175	-0.6%	3.2%	2.8%	72.3%	-0.1%	3.7%	3.2%	146.1%		
0.200	-1.0%	2.8%	2.4%	12.3%	-0.2%	3.3%	2.8%	60.4%		
0.225	-1.6%	2.8%	2.3%	-8.9%	-0.3%	3.2%	2.6%	30.1%		
0.250	-2.2%	3.1%	2.4%	-18.9%	-0.4%	3.1%	2.4%	15.8%		
0.275	-3.0%	3.7%	2.5%	-25.7%	-0.5%	3.2%	2.2%	6.1%		
0.300	-3.9%	4.5%	2.6%	-30.3%	-0.7%	3.3%	1.9%	-0.4%		
0.325	-5.0%	5.6%	2.8%	-33.6%	-0.9%	3.4%	1.6%	-5.2%		
0.350	-6.2%	7.0%	3.0%	-36.4%	-1.1%	3.6%	1.3%	-9.2%		
0.375	-7.5%	8.7%	3.3%	-38.7%	-1.3%	3.9%	1.0%	-12.5%		
0.400	-8.9%	10.8%	3.5%	-40.8%	-1.6%	4.1%	0.5%	-15.5%		
0.425	-10.3%	13.4%	3.8%	-42.7%	-1.8%	4.5%	0.1%	-18.2%		
0.450	-11.9%	16.4%	4.0%	-44.5%	-2.1%	4.8%	-0.4%	-20.7%		
0.475	-13.5%	19.9%	4.2%	-46.1%	-2.4%	5.3%	-0.9%	-23.1%		
0.500	-15.1%	24.1%	4.5%	-47.6%	-2.7%	5.8%	-1.4%	-25.2%		

Making the arch stiffer lowers the axial thrust in the arch, which in turn lowers the horizontal reaction. For this reason, there is more highlighted in dark gray in Table 4 as compared to Table 3, as the forces are being overestimated.

3.2.2 Live Loads

Surprisingly, the values predicted by the von Kármán and Scofield methods were both very close to the FEA analysis, except for the von Kármán moment. For the flexible arch (Table 5), the values for horizontal reaction were all within 1% of the FEA model. All of the thrust values for each analysis were either within 10% of the FEA model or at least overestimated the thrust for a conservative design.

Table 5 - Flexible Arch Analyses Comparison for Live Load.

	Live Load - Flexible Arch							
Rise/Span	von K	lármán Met	hod	Scofield Method				
[r] (-)	Horizontal Reaction	Thrust	Moment	Horizontal Reaction	Thrust	Moment		
0.050	0.3%	0.4%	101.9%	0.3%	0.4%	157.0%		
0.075	0.1%	0.4%	-8.2%	0.1%	0.3%	16.8%		
0.100	0.1%	0.6%	-18.2%	0.1%	0.4%	4.1%		
0.125	0.1%	1.1%	-20.5%	0.0%	0.5%	1.2%		
0.150	0.0%	1.7%	-21.3%	0.0%	0.6%	0.2%		
0.175	0.0%	2.6%	-21.6%	0.0%	0.7%	-0.1%		
0.200	0.1%	3.8%	-21.6%	0.0%	0.7%	-0.3%		
0.225	0.1%	5.3%	-21.6%	0.0%	0.8%	-0.3%		
0.250	0.1%	7.1%	-21.6%	0.0%	0.8%	-0.2%		
0.275	0.1%	9.2%	-21.5%	0.0%	0.7%	-0.1%		
0.300	0.2%	11.6%	-21.4%	0.0%	0.6%	0.1%		
0.325	0.2%	14.3%	-21.2%	0.0%	0.5%	0.2%		
0.350	0.3%	17.3%	-21.1%	0.0%	0.3%	0.4%		
0.375	0.3%	20.6%	-20.9%	0.1%	0.1%	0.7%		
0.400	0.4%	24.1%	-20.7%	0.1%	-0.2%	0.9%		
0.425	0.5%	27.5%	-20.6%	0.2%	-0.8%	1.1%		
0.450	0.7%	30.8%	-20.4%	0.3%	-1.6%	1.4%		
0.475	0.8%	33.9%	-20.3%	0.4%	-2.9%	1.5%		
0.500	1.0%	36.8%	-20.3%	0.5%	-4.2%	1.5%		

Unfortunately, the moments predicted by the von Kármán method were almost all far too low to be acceptable approximations. However, almost all of the moments found in the Scofield method were within the 10% margin, which demonstrates that the Scofield method accurately predicts the forces in a flexible arch for live load.

Similar to the results in Table 4, Table 6 for the stiff arch shows that the forces predicted by the two approximation methods are overestimates.

Table 6 - Stiff Arch Analyses Comparison for Live Load.

	Live Load - Stiff Arch							
Rise/Span	von K	ármán Met	thod	Scofield Method				
[r] (-)	Horizontal Reaction	Thrust	Moment	Horizontal Reaction	Thrust	Moment		
0.050	32.8%	31.3%	-	32.8%	31.3%	-		
0.075	14.6%	13.6%	-	14.6%	13.5%	-		
0.100	8.2%	7.6%	-	8.2%	7.3%	-		
0.125	5.3%	5.2%	-	5.2%	4.6%	-		
0.150	3.7%	4.4%	167.2%	3.6%	3.2%	240.1%		
0.175	2.7%	4.4%	38.4%	2.7%	2.4%	76.2%		
0.200	2.1%	5.0%	7.1%	2.0%	1.9%	36.3%		
0.225	1.7%	6.1%	-4.7%	1.6%	1.6%	21.3%		
0.250	1.4%	7.7%	-11.3%	1.3%	1.3%	12.9%		
0.275	1.2%	9.6%	-14.8%	1.1%	1.1%	8.4%		
0.300	1.1%	11.9%	-16.8%	0.9%	0.9%	5.9%		
0.325	1.0%	14.5%	-18.0%	0.8%	0.7%	4.3%		
0.350	1.0%	17.5%	-18.8%	0.7%	0.4%	3.4%		
0.375	0.9%	20.7%	-19.2%	0.7%	0.2%	2.9%		
0.400	1.0%	24.2%	-19.4%	0.7%	-0.1%	2.6%		
0.425	1.0%	27.6%	-19.5%	0.7%	-0.8%	2.4%		
0.450	1.1%	30.9%	-19.6%	0.7%	-1.6%	2.4%		
0.475	1.2%	33.9%	-19.6%	0.8%	-2.8%	2.4%		
0.500	1.3%	36.9%	-19.7%	0.8%	-4.2%	2.2%		

3.2.3 Wind Load

Since the wind loading assumed by the von Kármán and Scofield methods is completely different than the loading dictated by ASCE 7-10, the values for the arch forces are nowhere near those found with the finite element analysis. There is no way one could use the approximation methods for wind loads.

Table 7 - Flexible Arch Analyses Comparison for Wind Load.

Wind Load - Flexible Arch							
Rise/Span	von K	ármán Met	thod	Scofield Method			
[r] (-)	Horizontal Reaction	Thrust	Moment	Horizontal Reaction	Thrust	Moment	
0.050	-99.5%	-99.5%	-99.6%	-99.5%	-99.5%	-98.1%	
0.075	-98.9%	-99.0%	-99.1%	-98.9%	-98.9%	-96.1%	
0.100	-98.2%	-98.2%	-98.5%	-98.1%	-98.2%	-93.7%	
0.125	-97.3%	-97.3%	-97.9%	-97.2%	-97.3%	-91.1%	
0.150	-96.2%	-96.3%	-97.3%	-96.1%	-96.3%	-88.4%	
0.175	-95.1%	-95.1%	-96.7%	-94.9%	-95.3%	-85.7%	
0.200	-94.0%	-94.0%	-96.1%	-93.7%	-94.3%	-83.1%	
0.225	-92.8%	-92.8%	-95.5%	-92.6%	-93.4%	-80.8%	
0.250	-91.3%	-90.9%	-95.3%	-91.0%	-91.9%	-79.8%	
0.275	-90.2%	-89.6%	-94.9%	-89.9%	-91.0%	-78.1%	
0.300	-89.3%	-88.4%	-94.6%	-88.9%	-90.2%	-76.7%	
0.325	-88.4%	-87.2%	-94.3%	-88.0%	-89.5%	-75.5%	
0.350	-87.7%	-86.1%	-94.1%	-87.2%	-89.0%	-74.7%	
0.375	-86.6%	-83.7%	-94.2%	-86.1%	-87.5%	-75.0%	
0.400	-86.1%	-82.8%	-94.1%	-85.6%	-87.1%	-74.7%	
0.425	-85.8%	-82.0%	-94.1%	-85.3%	-86.9%	-74.6%	
0.450	-85.6%	-81.3%	-94.1%	-85.0%	-86.9%	-74.6%	
0.475	-85.1%	-78.7%	-94.3%	-84.5%	-85.5%	-75.3%	
0.500	-85.0%	-78.2%	-94.3%	-84.4%	-85.6%	-75.6%	

Table 8 - Stiff Arch Analyses Comparison for Wind Load.

Wind Load - Stiff Arch							
Rise/Span	von K	ármán Met	thod	Scofield Method			
[r] (-)	Horizontal Reaction	Thrust	Moment	Horizontal Reaction	Thrust	Moment	
0.050	-99.4%	-99.4%	-97.6%	-99.3%	-99.3%	-89.7%	
0.075	-98.8%	-98.8%	-98.4%	-98.7%	-98.8%	-92.9%	
0.100	-98.0%	-98.1%	-98.0%	-98.0%	-98.0%	-91.5%	
0.125	-97.1%	-97.2%	-97.5%	-97.0%	-97.2%	-89.3%	
0.150	-96.1%	-96.2%	-97.0%	-96.0%	-96.2%	-86.9%	
0.175	-95.0%	-95.0%	-96.4%	-94.8%	-95.2%	-84.5%	
0.200	-93.9%	-93.9%	-95.9%	-93.6%	-94.3%	-82.2%	
0.225	-92.7%	-92.7%	-95.4%	-92.4%	-93.3%	-80.0%	
0.250	-91.2%	-90.8%	-95.2%	-90.9%	-91.8%	-79.2%	
0.275	-90.2%	-89.6%	-94.8%	-89.8%	-90.9%	-77.6%	
0.300	-89.2%	-88.3%	-94.5%	-88.8%	-90.2%	-76.3%	
0.325	-88.4%	-87.2%	-94.3%	-87.9%	-89.5%	-75.2%	
0.350	-87.6%	-86.0%	-94.1%	-87.1%	-88.9%	-74.4%	
0.375	-86.5%	-83.7%	-94.2%	-86.0%	-87.4%	-74.9%	
0.400	-86.1%	-82.7%	-94.1%	-85.6%	-87.1%	-74.5%	
0.425	-85.8%	-81.9%	-94.1%	-85.2%	-86.9%	-74.4%	
0.450	-85.5%	-81.2%	-94.1%	-85.0%	-86.8%	-74.5%	
0.475	-85.0%	-78.7%	-94.3%	-84.4%	-85.5%	-75.2%	
0.500	-85.0%	-78.2%	-94.3%	-84.4%	-85.6%	-75.6%	

3.2.4 Snow Drift Load

The loading pattern for snow drifts followed by Scofield is similar to the one stipulated in ASCE 7-10 in that they only affect one half of the arch. After that, the similarities end. ASCE takes into account snow piling and snow slipping due to the roof curvature, whereas Scofield just assumed the drift was a uniform lineal load.

Table 9 - Flexible Arch Analyses Comparison for Snow Drift Load.

Snow Drift Load - Flexible Arch								
Rise/Span	Scofield Method							
[r] (-)	Vertical Horizontal Reaction Reaction		Thrust	Moment				
0.050	8.9%	41.3%	39.6%	2.2%				
0.075	8.9%	41.0%	37.4%	3.6%				
0.100	8.9%	40.8%	34.9%	5.1%				
0.125	8.9%	40.7%	32.4%	6.9%				
0.150	2.9%	33.2%	22.7%	1.7%				
0.175	-3.1%	24.1%	12.6%	-4.3%				
0.200	-6.9%	17.6%	5.7%	-7.5%				
0.225	-6.0%	14.0%	3.3%	-7.9%				
0.250	-1.9%	12.6%	4.3%	-5.6%				
0.275	3.8%	12.7%	7.5%	-2.7%				
0.300	11.0%	13.6%	12.5%	0.7%				
0.325	19.5%	15.2%	19.1%	4.4%				
0.350	30.0%	17.7%	27.8%	7.3%				
0.375	33.6%	18.2%	30.5%	8.6%				
0.400	36.6%	18.9%	32.7%	8.5%				
0.425	38.8%	19.5%	34.3%	8.5%				
0.450	39.9%	19.8%	34.5%	8.2%				
0.475	40.7%	20.0%	34.5%	6.5%				
0.500	41.2%	20.5%	34.1%	5.2%				

For the most part the Scofield analysis overestimates the forces in the arch due to drift loading even though the load patterns are different. Unfortunately, it underestimates the moment in a couple of cases for the flexible arch and in about half of the cases for the stiff arch.

Table 10 - Stiff Arch Analyses Comparison for Snow Drift Load.

Snow Drift Load - Stiff Arch						
Rise/Span	Scofield Method					
[r] (-)	Vertical Reaction	Horizontal Reaction	Thrust	Moment		
0.050	8.9%	87.0%	80.7%	-31.3%		
0.075	8.9%	61.3%	54.0%	-17.1%		
0.100	8.9%	52.3%	43.2%	-8.6%		
0.125	8.9%	48.0%	37.0%	-3.0%		
0.150	2.9%	38.0%	25.2%	-5.0%		
0.175	-3.1%	27.4%	14.1%	-9.1%		
0.200	-6.9%	20.0%	6.6%	-11.1%		
0.225	-6.0%	15.9%	3.9%	-11.0%		
0.250	-1.9%	14.1%	4.8%	-8.4%		
0.275	3.8%	13.9%	7.8%	-5.2%		
0.300	11.0%	14.7%	12.7%	-1.6%		
0.325	19.5%	16.1%	19.3%	2.4%		
0.350	30.0%	18.5%	27.9%	5.5%		
0.375	33.6%	18.9%	30.6%	7.0%		
0.400	36.6%	19.5%	32.8%	7.1%		
0.425	38.8%	20.1%	34.4%	7.3%		
0.450	39.9%	20.3%	34.6%	7.2%		
0.475	40.7%	20.5%	34.5%	5.6%		
0.500	41.2%	20.9%	34.1%	4.4%		

3.3 Effects of Curvature on Arch Forces

Changing the amount of curvature in the arch has a profound effect on the distribution of forces in the arch. An arch with a low rise will, not surprisingly, act more like a straight beam. A half-circle arch will act much differently.

Appendix D displays twenty graphs that show the horizontal reaction, axial force, and moments plotted against the Rise-to-Span ratio for all five load types. The graphs show two lines, one representing the forces on a stiff arch (I/A = 100), and one for a flexible one (I/A = 1). This design is to show how changing the stiffness of the arch changes the force-resisting characteristics of the arch and to give an envelope of acceptable forces. For the most part, the two curves are very similar. In a few cases they differ. A few specific cases are highlighted in the following sections.

The graphs are based on the following arch characteristics:

- 40 foot span
- 10 psf dead load
- 20 psf construction load
- 115 mph wind
- 30 psf ground snow load

Graphs based on different spans and loadings would obviously have different values, but the general shape of the curves would be the same.

3.3.1 Dead Load

The curves on the dead load graphs are, for the most part, very similar. A flexible arch has slightly higher forces for the horizontal reaction, axial force, and negative moment. This difference is more apparent as the Rise-to-Span ratio decreases. However, the positive moment graph shows a large difference between a stiff and flexible arch in the low rise ranges with the stiff arch curve looking like a V, as seen in Figure 37.

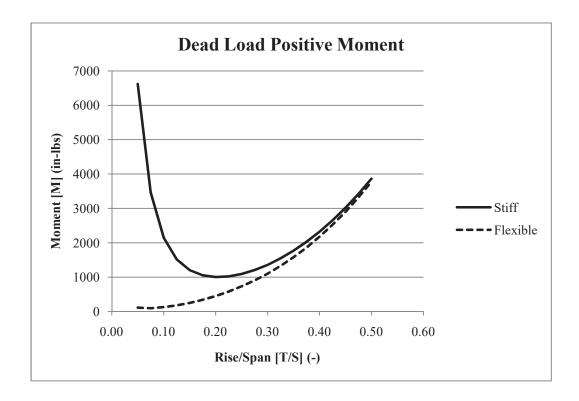


Figure 37 - Dead Load Positive Moment Graph.

The stiff arch curve gives greater values for design forces for all r values. Its greater stiffness gathers more moment than the flexible arch.

3.3.2 Construction Load

Since the construction load type is very similar to the dead load, the graphs are also similar. The same holds true here where the horizontal reaction, axial force, and negative moment graphs have the stiff and flexible curves very similar. The positive moment graph has the curves very different but the graph is similar to the dead load case, as seen in Figure 38.

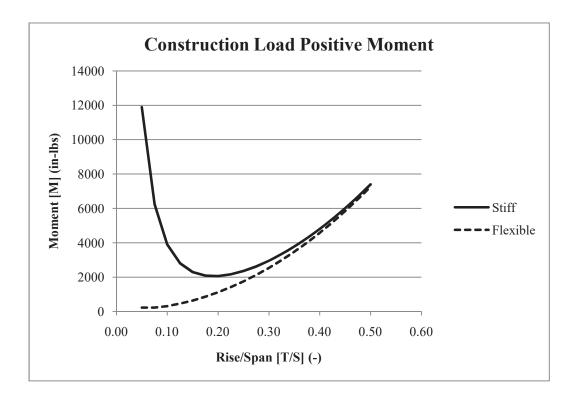


Figure 38 - Construction Load Positive Moment Graph.

Again, the same V-shaped curve appears for the stiff arch, showing that its stiffness allows it to accrue more moment than the flexible arch.

3.3.3 Wind Load

Since uplift plays a large role in the wind load, the graphs for the axial force, negative moment, and positive moment look a little different than for the other gravity loads. For starters, the axial force is in tension instead of compression, and looks like the reverse view of the dead load or construction load graphs for axial force. The horizontal reaction graph is similar to the previous two loading types. The two moment graphs are different, however, as shown in Figure 39 and Figure 40.

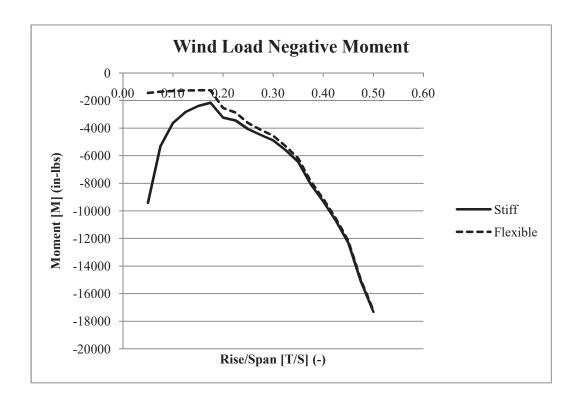


Figure 39 - Wind Load Negative Moment Graph.

The stiff arch has greater forces early on and then follows the flexible curve closely after the r ratio passes 0.17.

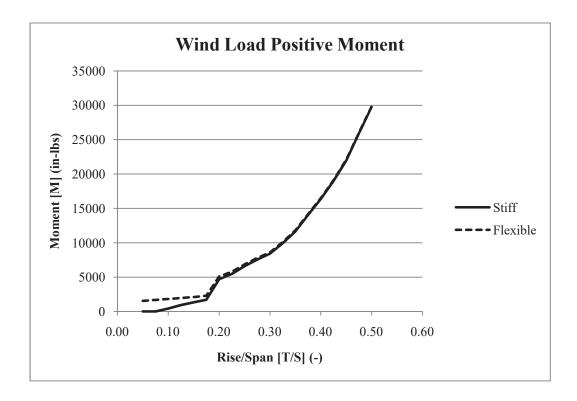


Figure 40 - Wind Load Positive Moment Graph.

In Figure 40 the curves follow closer to the dead load and construction load positive moment curve for a flexible arch. There is little difference between the stiff and flexible arch.

3.3.4 Snow Drift Load

The drift load horizontal reaction and axial force graphs are similar to the other gravity load types but have bumps due to the changing loadings as the curvature of the roof changes. The two moment graphs are nothing like the others, as seen in Figure 41 and Figure 42.

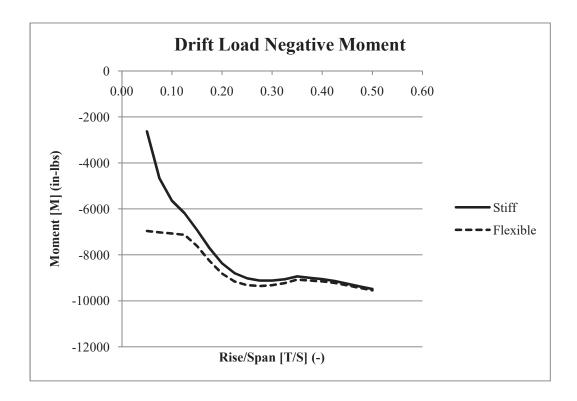


Figure 41 - Drift Load Negative Moment Graph.

Figure 41 shows that the flexible arch takes on a little more moment for lower rises but then the two curves follow each other when r is greater than 0.15. The bumps in Figure 41 are due to the changing curvature of the roof affecting the load pattern.

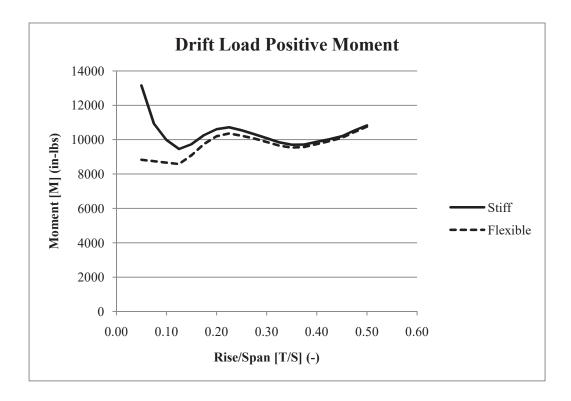


Figure 42 - Drift Load Positive Moment Graph.

The curves in Figure 42 are close, for the most part, with the stiff arch getting more moment in low rises. The bumps are again due to the changing loading patterns.

3.3.5 Balanced Snow Load

The balanced snow load is a similar loading to the construction load except where the load tapers off at the ends due to the roof slope. Because of this, the four graphs for each of the arch forces are very similar to the corresponding dead load and construction load graphs, including the V-shaped curve for the stiff arch positive moment. Figure 43 displays one of these graphs.

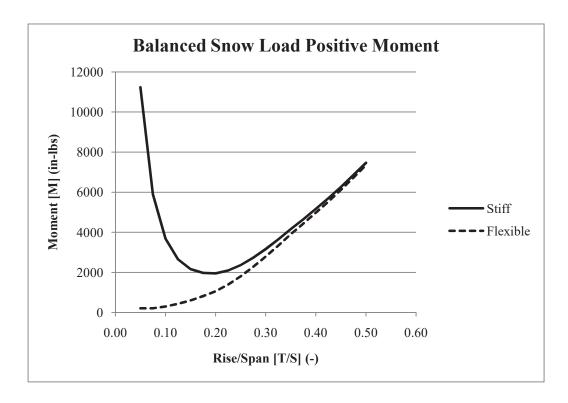


Figure 43 - Balanced Snow Load Positive Moment Graph.

Like the two other balanced gravity loads, the stiff arch takes on much higher positive moment in low rises.

3.3.6 Application for Load Tables

The load tables developed by the author will use the stiff arch curves instead of the flexible curves. There are two reasons for this choice. One is that the stiff arch gives higher moments (for the most part) and moment contributes more to the stress in the lamella than axial loads do. Second is that the roof will more likely act like a stiff arch because of the interplay in the rhomboid grid of the lamellas and the fact that the roof diaphragm over the lamellas will stiffen them as well.

It should also be noted that arched roofs are generally not designed to have r values close to 0 or 0.5, but fall somewhere in between. The most disparity between the stiff and

flexible arch curves occurs in the low rise range in which few, if any, arched roofs are built.

3.3.7 Example Moment Diagrams

As the arch starts out with a low rise, under uniform vertical loading, the majority of the moment will be positive moment, which would be analogous to the bending of a simple beam. As the rise of the arch increases, the "sides" of the arch will incur negative moment from the arch resisting outward buckling. This is illustrated in Figure 44, Figure 45, and Figure 46.

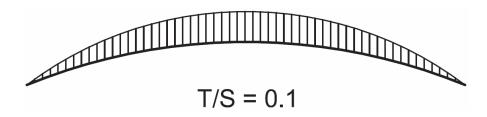


Figure 44 - Moment Diagram for Arch with Low Rise.

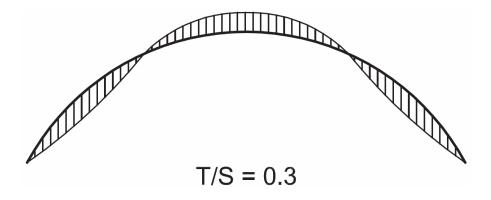


Figure 45 - Moment Diagram for Arch with Medium Rise.

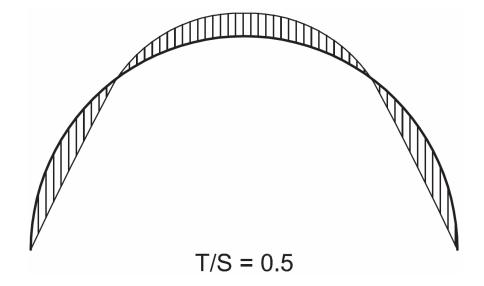


Figure 46 - Moment Diagram for Semi-Circular Arch.

The cut-off point for the beginning of negative moment depends on the stiffness of the arch. The stiffer the arch, the more moment capacity it has and the sooner it can take on negative moment as the rise increases. Generally, this point appears to be where the rise of the arch is around 10% of the span.

4 Development of Design Tables

To facilitate the design of a lamella roof, the author developed a set of design tables – one for the loads on the lamella roof (Section 4.1) and another for the connection between the lamellas (Section 4.2).

4.1 Load Tables

These tables display roof spans from 20 to 120 feet and show loads based on a changing rise. Two different sets were developed: one in the 115 mph wind zone with varying snow loads and one with zero snow loading but with varying wind speeds. The former is meant to be used in non-hurricane regions of the United States and the latter in hurricane regions, like Florida, which can expect zero annual snowfall.

The tables were created in Microsoft Excel using the finite element analysis approach discussed in Section 3.1.3 and using the ASD load combinations and loading patterns stipulated in ASCE 7-10. To aid in table generation, the author programmed a macro in which the user inputs the system parameters (span, rise, cross-sectional area, moment of inertia, and loadings) and the macro runs the various rise-to-span ratios through the FEA matrices, finds loads for each load case, and takes the worst loading from all cases for the design loads. This means that the maximum moment may come from one load case and the maximum axial force from another. The same can be said about the base reactions. Also, the point of maximum moment and maximum axial force most likely do not coincide, but designing a lumber beam-column (i.e., lamella) to resist those simultaneous maximum forces will give a conservative design.

Values in the tables are given in units per foot of arch.

Discussed earlier was the notion of the arch stiffness being a function of the moment of inertia over the cross-sectional area. In keeping with the conclusions drawn in Section 3.3.6, the author uses an *I/A* value of 100, representing a stiff arch, for the FEA calculations.

Additionally, when designing for ASD while using the NDS design specification, one must pay attention to the load duration factor C_D . Since wood has a load carrying capacity that increases when the load duration decreases, the NDS assigns different C_D values based on the load case. For example, NDS Table 2.3.2 specifies that an occupancy live load with a ten-year duration gets a C_D value of 1.0 while a wind load with a ten-minute duration gets a C_D value of 1.6. The load duration factor is used to increase (or decrease, if the dead load controls) the design values of the lumber used, essentially making the wood stronger.

Since the load duration factor changes depending on the loads used in the load combination, the various loads found through the FEA method must be normalized. Consider, for example, the load case D + 0.75(0.6W) + 0.75S. The NDS specification states that the C_D value for the shortest duration load be used for the combination, which means that the above load combination has a C_D value of 1.6. To normalize it, the loads found by the FEA spreadsheet are divided by that C_D value. Then, when designing the lamella to carry the load, the load duration factors can all be assumed to be 1.0.

The finished load tables are found in Appendix F.

4.2 Connection Tables

Connection design is based on the assumption that the lamella connection can be modeled as a double-shear connection, as shown in Figure 47.

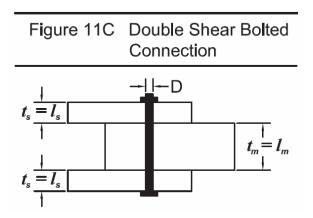


Figure 47 - Double Shear Bolted Connection [10].

As seen in Section 2.1.1, the actual geometry is a little more complicated (see Figure 17) but the approximation is close enough. Since the lamellas are all the same thickness, ℓ_s would be equal to t, the thickness of the member, and

$$\ell_m = \frac{t}{\cos 2\theta}.\tag{83}$$

These two values are used in determining the shear capacity of the connection.

The NDS stipulates finding several yield limit states for a double shear connection.

These limit states are shown in Appendix A, Figure A-1. Since the connection is double shear, only modes I_m , I_s , III_s , and IV apply, each with its own yield limit equation. They are

$$I_{m}: Z = \frac{D\ell_{m}F_{em}}{R_{d}}, \tag{84}$$

$$I_{s}: Z = \frac{D\ell_{s} F_{es}}{R_{d}}, \tag{85}$$

$$III_{s}: Z = \frac{2k_{3}D\ell_{s}F_{em}}{(2+R_{e})R_{d}},$$
(86)

and

IV:
$$Z = \frac{2D^2}{R_d} \sqrt{\frac{2F_{em}F_{yb}}{3(1+R_e)}},$$
 (87)

where

D = dowel diameter, in,

 $F_{yb} =$ dowel bending yield strength, psi,

 R_d = reduction term,

 $R_e = F_{em}/F_{es}$

 ℓ_m = main member dowel bearing length, in,

 ℓ_s = side member dowel bearing length, in,

 F_{em} = main member dowel bearing strength, psi,

 F_{es} = side member dowel bearing strength, psi,

and

$$k_{3} = -1 + \sqrt{\frac{2(1+R_{e})}{R_{e}} + \frac{2F_{yb}(2+R_{e})D^{2}}{3F_{em}\ell_{s}^{2}}}.$$
 (88)

At this point some simplifications and substitutions can be made. To start, F_{em} and F_{es} are dependent only on the wood specific gravity and the dowel diameter, making them the same value [10]:

$$F_{em} = F_{es} = F_{e\perp} = \frac{6,100G^{1.45}}{\sqrt{D}}.$$

The $F_{e\perp}$ equation is used since the shear load acts perpendicular to grain. Since F_{em} and F_{es} are the same value, R_e simplifies to one. Completing the substitutions yields:

$$k_3 = -1 + \sqrt{4 + \frac{F_{yb}D^{2.5}}{3,050t^2G^{1.45}}},$$
(89)

$$I_{m}: Z = \frac{6,100G^{1.45}t\sqrt{D}}{R_{d}\cos 2\theta}, \tag{90}$$

$$I_s: Z = \frac{12,200G^{1.45}t\sqrt{D}}{R_d},$$
 (91)

$$III_{m}: Z = \frac{12,200k_{3}G^{1.45}t\sqrt{D}}{3R_{d}},$$
(92)

and

IV:
$$Z = \frac{2D^2}{R_d} \sqrt{\frac{6,100G^{1.45}F_{yb}}{9\sqrt{D}}}$$
 (93)

The R_d value changes depending on the yield mode. Figure 48 shows how it is determined. The footnote in the table notes that for threaded fasteners with a nomimal diameter greater than or equal to $\frac{1}{4}$ " and a root diameter, D_r , less than $\frac{1}{4}$ " (i.e., $\frac{1}{4}$ " and $\frac{5}{16}$ " bolts),

$$R_d = K_D K_\theta,$$

where

$$K_D = 10D_r + 0.5,$$

and K_{θ} is the same as shown in Figure 48. In all cases, since the shear is perpendicular to grain, θ is 90° and K_{θ} becomes 1.25.

Table 11.3.1B Reduction Term, R _d					
Fastener Size	Yield Mode	Reduction Term, R _d			
$0.25'' \leq D \leq 1''$	I_{m}, I_{s} II III_{m}, III_{s}, IV	$\begin{array}{c} 4~K_{\theta} \\ 3.6~K_{\theta} \\ 3.2~K_{\theta} \end{array}$			
$D<0.25^{\prime\prime}$	$I_m,I_s,II,III_m,III_s,IV$	K_D^{-1}			
Notes: $K_{\theta} = 1 + 0.25(\theta - 90)$					
$\theta = \text{ maximum angle of load to grain } (0 \le \theta \le 90)$					
for any member in a connection					
D = diameter, in. (see 11.3.6)					
$K_D = 2.2$	for $D \le 0.17$ "	for $D \le 0.17$ "			
$K_D = 10D + 0.5$ for 0.17 " $< D < 0.25$ "					

^{1.} For threaded fasteners where nominal diameter (see Appendix L) is greater than or equal to 0.25" and root diameter is less than 0.25", $R_{\text{d}}=K_{D}K_{\text{\theta}}.$

Figure 48 - Reduction Term, R_d [10].

5 Lamella Strength and Connection Design

The following sections will display how one analyzes a lamella section for design adequacy and how one would design the connection to withstand the loads applied on it.

5.1 Lamella Strength Analysis

The lamella is designed based on the assumption that it acts like a beam-column with biaxial bending and compression. Testing done by the author (see Section 7.3.1) backs up this assumption. According to the NDS Section 3.9.2, the equation for bending and axial compression is

$$\left[\frac{f_c}{F_c'}\right]^2 + \frac{f_{b1}}{F_{b1}'\left[1 - (f_c/F_{cE1})\right]} + \frac{f_{b2}}{F_{b2}'\left[1 - (f_c/F_{cE2}) - (f_{b1}/F_{bE})^2\right]} \le 1.0$$
 (94)

where

$$f_c < F_{cE1} = \frac{0.822E'_{\min}}{\left(\frac{\ell_{e1}}{d_1}\right)^2},$$
 (95)

$$f_c < F_{cE2} = \frac{0.822E'_{\min}}{\left(\frac{\ell_{e2}}{d_2}\right)^2},$$
 (96)

$$f_{b1} < F_{bE} = \frac{1.20E'_{\min}}{\left(R_B\right)^2},\tag{97}$$

with

 f_{b1} = actual edgewise bending stress (bending load applied to narrow face of member),

 f_{b2} = actual flatwise bending stress (bending load applied to wide face of member),

 f_c = actual compressive stress from axial load,

 d_1 = wide face dimension (lamella depth),

 d_2 = narrow face dimension (lamella thickness),

 ℓ_e = effective column length (NDS Section 3.7.1.2),

and

$$F'_{b1} = F_b C_D C_M C_t C_L C_F C_{fu} C_i C_r, (98)$$

$$F'_{b2} = F_b C_D C_M C_t C_L C_F C_{fit} C_i C_r, (99)$$

$$F_c' = F_c C_D C_M C_t C_F C_i C_P \tag{100}$$

and

$$E'_{\min} = E_{\min} C_M C_t C_t C_T. \tag{101}$$

Since the lamella is continually braced in the weak direction by the roof diaphragm, Equation (96) essentially becomes infinite, which reduces Equation (94) to

$$\left[\frac{f_c}{F_c'}\right]^2 + \frac{f_{b1}}{F_{b1}'\left[1 - (f_c/F_{cE1})\right]} + \frac{f_{b2}}{F_{b2}'\left[1 - (f_{b1}/F_{bE})^2\right]} \le 1.0.$$
 (102)

As stated before, the C_D factor has already been normalized to 1.0. Also, for most of the adjustment factor C values, the values can be eliminated by setting them equal to 1.0, as well. Since the lamella roof is most likely going to be built indoors, the moisture, temperature, and incising factors - C_M , C_D and C_D are respectively - will be equal to 1.0. In strong-axis bending, the flat use factor C_{fu} drops to 1.0 since it is bending edge-wise. For the maximum positive moment, the compression edge of the lamella is continually

braced by the sheathing on top, and the beam stability factor C_L becomes 1.0, as well. For the maximum negative moment, the compression edge is only braced at the lamella ends and at the halfway point by the adjacent lamellas. Assuming the lamella is loaded with a uniformly distributed load, the effective length ℓ_e can be one of two values. If the unsupported length ℓ_u divided by the lamella depth is less than seven ($\ell_u/d < 7$), then

$$\ell_{g} = 2.06\ell_{u}.\tag{103}$$

If the ratio is above seven, then the effective length becomes

$$\ell_e = 1.63\ell_u + 3d. \tag{104}$$

The beam stability factor is then found by

$$C_{L} = \frac{1 + \left(F_{bE}/F_{b}^{*}\right)}{1.9} - \sqrt{\left[\frac{1 + \left(F_{bE}/F_{b}^{*}\right)}{1.9}\right]^{2} - \frac{F_{bE}/F_{b}^{*}}{0.95}},$$
(105)

where

$$F_b^* = F_b C_D C_M C_t C_F C_i C_r,$$

$$F_{bE} = \frac{1.20E'_{\min}}{R_B^2},\tag{106}$$

and

$$R_{\scriptscriptstyle B} = \sqrt{\frac{\ell_{\scriptscriptstyle e} d}{b^2}},\tag{107}$$

where b is the thickness of the lamella and R_B shall not exceed 50. The two different C_L values must be used in conjunction with their respective moments when using Equation (102) to check the beam-column.

In weak-axis bending, the flat-use factor C_{fu} is given by Tables 4A, 4B, 4C, and 4F of the NDS 2005 Supplement. The beam stability factor C_L is 1.0 since its depth is less than its width.

The size factor C_F depends on the size of the member and the species and will change depending on the lamella size. Unless the lamellas are spaced at 24 inches or less, the repetitive member factor C_r will be equal to 1.0, as well (it will be 1.15 otherwise). Since the lamella roof does not meet any of the specifications of NDS Section 4.4.2, the buckling stability factor C_T is also equal to 1.0.

Finding the column stability factor C_P is a little more involved. For starters, the NDS limits the slenderness ratio ℓ_e/d to 50 (75 during construction), where

$$\ell_{e} = K_{e}\ell. \tag{108}$$

The connections between the lamellas are assumed to be pinned-pinned so the effective length factor K_e is equal to 1.0. The lamellas are braced continuously in the weak direction by the roof sheathing so the Y-Y axis slenderness ratio is zero. The side lamellas, though they brace the continuous lamella at the half-points, only brace in the weak direction; thus, the effective length in the X-X axis is

$$\ell_{a} = \ell. \tag{109}$$

In reality, the lamella would have to be very long and very shallow in order for the slenderness ratio to be greater than 50. For example, a 2x8 lamella with a depth of 7.25 inches would have to be over 30 feet long for this to happen.

Equation 3.7-1 of the NDS then gives the column stability factor as

$$C_{P} = \frac{1 + \left(F_{cE}/F_{c}^{*}\right)}{2c} - \sqrt{\left[\frac{1 + \left(F_{cE}/F_{c}^{*}\right)}{2c}\right]^{2} - \frac{F_{cE}/F_{c}^{*}}{c}},$$
(110)

where

 $F_c^* = F_c C_D C_M C_t C_F C_i,$

c = 0.8 for sawn lumber,

c = 0.85 for round timber poles and piles,

c = 0.9 for structural glued laminated timber or structural composite lumber,

and F_{cE} is the same as in Equation (95).

The thrust and moments found in the Load Tables must be adjusted to take the skew of the lamella arches and the length of the lamellas into consideration. The moment in the strong direction must be multiplied by the spacing of the lamellas and divided by the cosine of the skew angle since the lamella follows the skewed arch [3]:

$$M_{x,lam} = \frac{(\text{Spacing})(M_{load\ table})}{\cos \theta}.$$
 (111)

The thrust is taken up by two lamellas, since the compressive force can go two ways in each connection node [3]:

$$F_{a,lam} = \frac{(\text{Spacing})(F_{a,load\ table})}{2\cos\theta}.$$
 (112)

In the weak direction, the moment is generated by the force couple created by the side lamellas abutting the middle lamella, which is simply the axial thrust multiplied by the shift of the connection:

$$M_{v,lam} = (F_{a,lam})(s). \tag{113}$$

With a chosen trial section one can begin the process of checking the section for adequacy. First, one finds the bending and compressive stresses by

$$f_{b1} = \frac{M_{x,lam}}{S_{xx}},\tag{114}$$

$$f_{b2} = \frac{M_{y,lam}}{S_{yy}},\tag{115}$$

and

$$f_c = \frac{F_{a,lam}}{A}. (116)$$

After these calculations, it is a simple matter of checking the calculated stresses for compliance in the interaction equation from Equation (102). It is important to note that this calculation must be done twice – once for the maximum positive moment and once for the maximum negative moment, each calculated with their respective C_L values.

5.2 Connection Design

Since the connections between lamellas are achieved using dowel-type fasteners, the equation for the reference design value is

$$Z' = ZC_D C_M C_t C_g C_\Delta C_{eg} C_{di} C_{tn}, (117)$$

where

 C_D = Load Duration Factor,

 C_M = Wet Service Factor,

 C_t = Temperature Factor,

 C_g = Group Action Factor,

 C_{Λ} = Geometry Factor,

 C_{eg} = End Grain Factor,

 C_{di} = Diaphragm Factor,

 C_{tn} = Toe-Nail Factor.

Again, the C_D factor is normalized to 1.0 as in Section 4.1. Unless the building is exposed to the elements, the C_M and C_t factors will most likely be 1.0 as well.

The calculation for the group action factor is rather long. To begin,

$$C_{g} = \left[\frac{m(1 - m^{2n})}{n \left[(1 + R_{EA}m^{n})(1 + m) - 1 + m^{2n} \right]} \left[\frac{1 + R_{EA}}{1 - m} \right], \tag{118}$$

where

n = number of fasteners in a row,

$$R_{EA} = \min \left[\frac{E_s A_s}{E_m A_m}, \frac{E_m A_m}{E_s A_s} \right],$$

 E_m = modulus of elasticity of main member, psi,

 E_S = modulus of elasticity of side member, psi,

 $A_m =$ cross-sectional area of main member, in²,

 $A_s =$ cross-sectional area of side member, in²,

$$m = u - \sqrt{u^2 - 1},$$

$$u = 1 + \gamma \frac{s}{2} \left[\frac{1}{E_m A_m} + \frac{1}{E_s A_s} \right],$$

s = center-to-center spacing between adjacent fasteners in a row, in,

 $\gamma = 180,000 D^{1.5}$ (for dowel-type fasteners in wood-to-wood connections).

However, since the side member and main members are the same material, and the side lamella area will always be less than the middle lamella due to the curvature cut,

$$E_{m} = E_{s} = E,$$

$$R_{EA} = \frac{A_{s}}{A_{m}},$$

$$u = 1 + \frac{180,000D^{1.5}s}{2E} \left[\frac{1}{A_{m}} + \frac{1}{A_{s}} \right].$$

The geometry factor, if the designer follows the diagrams in Figure 15 and Figure 16, can be equal to the value shown for those figures. If it falls somewhere between, C_{Δ} can be found through one of two ways. To quote the NDS,

When dowel-type fasteners are used and the actual end distance for parallel or perpendicular to grain loading is greater than or equal to the minimum end distance (see Figure 11) for $C_{\Delta} = 0.5$, but less than the minimum end distance for $C_{\Delta} = 1.0$, the geometry factor, C_{Δ} , shall be determined as follows:

$$C_{\Delta} = \frac{\text{actual end distance}}{\text{minimum end distance for } C_{\Delta} = 1.0}$$

and

When the actual spacing between dowel-type fasteners in a row for parallel or perpendicular to grain loading is greater than or equal to the minimum spacing (see Figure 12), but less than the minimum spacing for $C_{\Delta} = 1.0$, the geometry factor, C_{Δ} , shall be determined as follows:

$$C_{\Delta} = \frac{\text{actual spacing}}{\text{minimum spacing for } C_{\Delta} = 1.0}$$
. [10]

The end grain factor, C_{eg} , only applies when a fastener is loaded in withdrawal from the end grain of a member and can be set to 1.0. Since the lamella connection is not part of a diaphragm, the C_{di} is also 1.0. And, since the connections are not toe-nailed, C_{tm} is equal to 1.0 as well.

The connection joint between the continuous and non-continuous lamellas must be designed to handle both vertical shear perpendicular to grain and thrust parallel to grain as a result of the eccentric connection. The vertical shear can be found by [16]:

$$V_{\perp} = \frac{4M_{x,lam}}{\ell},\tag{119}$$

where $M_{x,lam}$ is the moment in the strong axis of the lamella resulting from the chosen loading combination.

The thrust parallel to grain results in tension in the bolts. According to Scofield [3], the magnitude of this tension is found by

$$T_{bolts} = \frac{2F_{a,lam}\cos\theta}{\tan 2\theta},\tag{120}$$

where $F_{a,lam}$ is the thrust in each lamella and θ is the skew angle of the lamella arch. This tension would be split evenly between the bolts in the connection. Also, the tension in

the bolt would need to be transferred to the lamella through a washer of appropriate surface area so as not to crush the surrounding wood, as discussed later in this section [16].

The tensile capacity of a bolt is found by multiplying the diameter of the bolt by its yielding stress [4]. However, since there is a reduction in area due to the threads at the end of the bolt, one must use the root diameter D_r for calculations:

$$T_{cap} = D_r F_{y,bolt}. (121)$$

The yielding stress of the bolts is usually 36,000 psi. Using this information, Table 11 was created.

Table 11 - Strength Properties for Standard Hex Bolts.

Properties for Standard Hex Bolts						
D :	Root	Root	Tensile			
Diameter	Diameter	Area [A _r]	Capacity			
[D]	[D _r] (in)	(in²)	[Z] (lbs)			
1/4"	0.189	0.028	1005			
5/16"	0.245	0.047	1695			
3/8"	0.298	0.070	2510			
1/2"	0.406	0.129	4660			
5/8"	0.514	0.207	7465			
3/4"	0.627	0.309	11115			
7/8"	0.739	0.429	15440			
1"	0.847	0.563	20280			

Washers are used to transfer the tension load from the bolts to the lamellas and must be designed so as not to crush the surrounding wood. First, one must compute the compression strength of the wood perpendicular to grain [10]:

$$F_{c\perp}' = F_{c\perp} C_M C_t C_i C_b. \tag{122}$$

If the lamella roof is enclosed such that the lamellas are indoors, the C_M , C_b and C_b factors will all drop to 1.0. The NDS has the following to say about the bearing area factor C_b :

Reference compression design values perpendicular to grain, $F_{c\perp}$, apply to bearings of any length at the ends of a member, and to all bearings 6" or more in length at any other location. For bearings less than 6" in length and not nearer than 3" to the end of a member, the reference compression design value perpendicular to grain, $F_{c\perp}$, shall be permitted to be multiplied by the following bearing area factor, C_b :

$$C_b = \frac{\ell_b + 0.375}{\ell_b} \tag{123}$$

where

 ℓ_b = bearing length measured parallel to grain, in. [10]

Since the bolts, and therefore the washers, are closer than 3" to the end of the lamella, the bearing area factor does not increase the design compression strength perpendicular to grain and can be set to 1.0, as well.

From there, the necessary washer area needed for the tension developed in the bolts is found by

$$A_{washer} \ge \frac{T_{bolts}}{F_{c\perp}'}. (124)$$

This washer area is split between the bolts in the connection and applies for the entire connection. In other words, half of the washer area is for one side of the connection and the other half for the other side.

It should be noted that due to friction between the bolts and the surrounding wood, the forces in the connection will be slightly less than those computed, yielding slightly conservative design values [16].

6 Design Example

In this section, the tables developed by the author are used to design an example lamella roof. The example is designed following the NDS 2005 Specification using the Allowable Stress Design (ASD) process. It should be noted that Load & Resistance Factor Design (LRFD) is a perfectly acceptable design approach; however, the tables developed by the author use ASD load combinations.

In this example, the combination of snow drift and dead loads likely generates the largest loads on the structure; however, due to the nature of the load tables it is impossible to tell. The following design criteria apply:

- 40 ft span (S = 40 ft)
- 10 ft rise (T = 10 ft)
- Southern Pine No.1 lumber
- Structure dead load (D) = 10 psf
- Construction live load $(L_r) = 20 \text{ psf}$
- Grounld snow load $(p_g) = 30 \text{ psf}$
- Basic wind speed (V) = 120 mph
- 10 lamellas per arch (n = 10)
- Skew angle of 19° ($\theta = 19^{\circ}$)

6.1 Lamella Strength Check

First one must find the nominal length ℓ of the lamellas. Thus,

$$R = \frac{4T^2 + S^2}{8T} = \frac{4(10')^2 + (40')^2}{8(10')} = 25',$$

$$2\phi = \frac{\beta}{n} = \frac{2}{n}\arccos\left(\frac{R - T}{R}\right) = \frac{2}{10}\arccos\left(\frac{25' - 10'}{25'}\right) = 10.62^\circ,$$

$$\ell_{c-c} = 2R\sin\phi = 2(25')\sin\left(\frac{10.62^\circ}{2}\right) = 4.63',$$

$$\ell = \frac{\ell_{c-c}}{\cos\theta} = \frac{6.604'}{\cos(19^\circ)} = \boxed{4.90'}.$$

Looking at Table 1, this length of lamella at a skew of 19° gives a spacing of about 1.59 feet. From looking at the load tables in Appendix E, the following loads (per foot section of arch) are caused by the aforementioned design criteria:

- $R_y = 545 \text{ lbs}$
- $R_x = 520 \text{ lbs}$
- $F_a = 755 \text{ lbs}$
- $M^{-} = -8800 \text{ in-lbs}$
- $M^+ = 10110$ in-lbs

The axial thrust in each lamella is then found by

$$F_{a,lam} = \frac{\left(755 \text{ lbs/ft}\right)\left(1.59'\right)}{2\cos(19^\circ)} = \boxed{636 \text{ lbs}},$$

and the moments are

$$M_{x,lam}^{-} = \frac{(1.59')(-8800 \text{ in-lbs/ft})}{\cos(19^{\circ})} = \boxed{-14837 \text{ in-lbs}},$$

$$M_{x,lam}^{+} = \frac{(1.59')(10110 \text{ in-lbs/ft})}{\cos(19^{\circ})} = \boxed{17046 \text{ in-lbs}},$$

$$M_{y,lam} = (636 \text{ lbs})(3.79'') = \boxed{2865 \text{ in-lbs}}.$$

The lamella must now be designed as a biaxial beam-column to withstand the combined thrust and moment for both positive and negative moment. For the positive moment, the compression side of the member is assumed to be braced continuously by the sheathing on the rooftop. For the negative moment, the side lamellas abut the continuous lamella at the half-points providing lateral bracing, giving an unbraced length of half of the lamella length. The weak-axis bending is assumed to be braced at the endpoints only. A 2x10 trial section will be used for the strength checks – it has the following characteristics [10]:

•
$$A = 13.88 \text{ in}^2$$

•
$$S_{xx} = 21.39 \text{ in}^3$$

•
$$S_{yy} = 3.469 \text{ in}^3$$

•
$$F_b = 1300 \text{ psi}$$

•
$$F_c = 1600 \text{ psi}$$

•
$$E_{min} = 620,000 \text{ psi}$$

Since the size of the bolts is unknown, the shift of the connection cannot be immediately found. The author assumes ½" bolts in design which, according to Table 2, gives a shift of 3.79". The forces due to the thrust and moments are as follows:

$$f_c = \frac{F_{a,lam}}{A} = \frac{636 \text{ lbs}}{13.88 \text{ in}^2} = \boxed{45.9 \text{ psi}},$$

$$f_{b1}^- = \frac{M_{x,lam}^-}{S_{xx}} = \frac{14837 \text{ in-lbs}}{21.39 \text{ in}^3} = \boxed{693.6 \text{ psi}},$$

$$f_{b1}^+ = \frac{M_{x,lam}^+}{S_{xx}} = \frac{17046 \text{ in-lbs}}{21.39 \text{ in}^3} = \boxed{796.9 \text{ psi}},$$

and

$$f_{b2} = \frac{M_{y,lam}}{S_{yy}} = \frac{2865 \text{ in-lbs}}{3.469 \text{ in}^3} = \boxed{825.8 \text{ psi}}.$$

Now the adjustment factors must be found. Since the lamellas are spaced at less than 24" on-center, the repetitive member factor C_r can be set to 1.15, which increases the bending strength of the lumber. From here, the beam stability factor C_L for the negative moment is calculated:

$$\frac{\ell_u}{d_1} = \frac{\left(4.90'_2\right)\left(12\,\text{in/ft}\right)}{9.25''} = 3.18' < 7.0,$$

$$\ell_e = 2.06\ell_u = 2.06\left(\frac{4.90'_2}{2}\right)\left(12\,\text{in/ft}\right) = 60.52'',$$

$$R_B = \sqrt{\frac{\ell_e d}{b^2}} = \sqrt{\frac{(60.52'')(9.25'')}{(1.5'')^2}} = 15.77,$$

$$E'_{\min} = E_{\min} C_M C_i C_i C_T = (620,000 \text{ psi})(1.0)(1.0)(1.0)(1.0) = 620,000 \text{ psi},$$

$$F_{bE} = \frac{1.20E'_{\min}}{R_B^2} = \frac{1.20(620,000 \text{ psi})}{(15.77)^2} = 2990 \text{ psi},$$

$$F_b^* = F_b C_D C_M C_t C_F C_i C_r = (1300 \text{ psi})(1.0)(1.0)(1.0)(1.0)(1.0)(1.15),$$

$$F_b^* = 1495 \text{ psi},$$

$$C_{L} = \frac{1 + \left(F_{bE}/F_{b}^{*}\right)}{1.9} - \sqrt{\left[\frac{1 + \left(F_{bE}/F_{b}^{*}\right)}{1.9}\right]^{2} - \frac{F_{bE}/F_{b}^{*}}{0.95}},$$

$$C_{L} = \frac{1 + \left(2990/1495\right)}{1.9} - \sqrt{\left[\frac{1 + \left(2990/1495\right)}{1.9}\right]^{2} - \frac{2990/1495}{0.95}},$$

$$C_{L} = \boxed{0.956}.$$

Now, the column stability factor C_P is determined:

$$F_{cE} = \frac{0.822E'_{\min}}{\left(\frac{\ell_e}{d}\right)^2} = \frac{0.822(620,000 \text{ psi})}{\left(\frac{(4.90')(12 \text{ in/ft})}{9.25''}\right)^2} = 12630 \text{ psi},$$

$$F_c^* = F_c C_D C_M C_t C_F C_i = (1600 \text{ psi})(1.0)(1.0)(1.0)(1.0)(1.0) = 1600 \text{ psi},$$

$$c = 0.8,$$

$$C_{P} = \frac{1 + \left(F_{cE}/F_{c}^{*}\right)}{2c} - \sqrt{\left[\frac{1 + \left(F_{cE}/F_{c}^{*}\right)}{2c}\right]^{2} - \frac{F_{cE}/F_{c}^{*}}{c}}{c}},$$

$$C_{P} = \frac{1 + \left(12630/1600\right)}{2\left(0.8\right)} - \sqrt{\left[\frac{1 + \left(12630/1600\right)}{2\left(0.8\right)}\right]^{2} - \frac{12630/1600}{0.8}},$$

$$C_{P} = \boxed{0.973}.$$

Now, the remaining three design stresses for the unity check equation follow:

$$F'_{b1} = F_b C_D C_M C_t C_L C_F C_{fu} C_i C_r = (1300 \text{ psi})(1.0)(1.0)(1.0)(0.956)(1.0)(1.0)(1.0)(1.0),$$

$$F'_{b1} = 1430 \text{ psi}.$$

The above F'_{b1} applies to the negative moment since the C_L value is for an unbraced length of 2.45 feet. For the positive moment with the compression edge continually braced, $C_L = 1.0$ and

For the compressive strength parallel to grain,

$$F'_c = F_c C_D C_M C_t C_F C_i C_P = (1600 \text{ psi})(1.0)(1.0)(1.0)(1.0)(1.0)(0.973),$$

$$F'_c = 1556 \text{ psi}.$$

From here, it is a simple matter to plug the values into the modified unity equation. For the negative moment:

$$\left[\frac{f_c}{F_c'}\right]^2 + \frac{f_{b1}}{F_{b1}'\left[1 - \left(f_c/F_{cE1}\right)\right]} + \frac{f_{b2}}{F_{b2}'\left[1 - \left(f_{b1}/F_{bE}\right)^2\right]} \le 1.0,$$

$$\left[\frac{45.9 \text{ psi}}{1556 \text{ psi}}\right]^2 + \frac{693.6 \text{ psi}}{\left(1430 \text{ psi}\right)\left[1 - \left(\frac{45.9 \text{ psi}}{12630 \text{ psi}}\right)\right]} + \frac{825.8 \text{ psi}}{\left(1794 \text{ psi}\right)\left[1 - \left(\frac{693.6 \text{ psi}}{2990 \text{ psi}}\right)^2\right]} \le 1.0,$$

$$\left[0.974 < 1.0 \text{ O.K.}\right],$$

and the positive moment is

$$\left[\frac{45.9 \text{ psi}}{1556 \text{ psi}}\right]^{2} + \frac{796.9 \text{ psi}}{\left(1495 \text{ psi}\right)\left[1 - \left(\frac{45.9 \text{ psi}}{12630 \text{ psi}}\right)\right]} + \frac{825.8 \text{ psi}}{\left(1794 \text{ psi}\right)\left[1 - \left(\frac{796.9 \text{ psi}}{2990 \text{ psi}}\right)^{2}\right]} \le 1.0,$$

$$\boxed{1.031 > 1.0}.$$

The unity equation checks out for the negative moment but is about 3% high for the positive moment. However, since the loads on the arch are generally overstated and the stiffness of the roof will increase with the addition of the roof diaphragm, this extra 3% is of small concern and can most likely be ignored. Thus, a 2x10 section is adequate for design.

It should also be noted that the end supports of the arch need to be designed to carry 520 lbs. of lateral force per foot and 545 lbs. of gravity load per foot.

6.2 Connection Design

As mentioned in Section 4, there are two load paths in the connection. First, the vertical shear through the connection is found by

$$V_{\perp} = \frac{4M_{x,lam}^{+}}{\ell} = \frac{4(17046 \text{ in-lbs})}{4.90'(12 \text{ in/ft})} = \boxed{1160 \text{ lbs}}.$$

The positive moment is used because its magnitude is greater than that of the negative moment.

The tension due to the eccentric connection is

$$T_{bolts} = \frac{2F_{a,lam}\cos\theta}{\tan 2\theta} = \frac{2(636 \text{ lbs})(\cos 19^\circ)}{\tan 38^\circ} = \boxed{1540 \text{ lbs}}.$$

From Equation (117), we know that the strength of a connection is determined by

$$Z' = ZC_{\scriptscriptstyle D}C_{\scriptscriptstyle M}C_{\scriptscriptstyle t}C_{\scriptscriptstyle g}C_{\scriptscriptstyle \Delta}C_{\scriptscriptstyle eg}C_{\scriptscriptstyle di}C_{\scriptscriptstyle tn}.$$

The duration, moisture, temperature, end grain, diaphragm action, and toe-nail factors can all be set to 1.0 as discussed in Section 5.2. For determining the group action factor, a couple properties of the lamella must be found first. From Equation (11), the total length of the lamella, assuming ½" bolts, is

$$\ell_{T} = \frac{2R\sin\phi}{\cos\theta} + \frac{t + 2D\tan 2\theta}{2\sin\theta\cos\theta} + \frac{t}{\tan 2\theta},$$

$$\ell_{T} = \frac{2(25')(12''/1')\sin(5.31^{\circ})}{\cos(19^{\circ})} + \frac{1.5'' + 2(0.5'')\tan(38^{\circ})}{2\sin(19^{\circ})\cos(19^{\circ})} + \frac{1.5''}{\tan(38^{\circ})},$$

$$\ell_{T} = 65.09'' \approx 5' - 5\frac{3}{32}''.$$

Then, the angle ϕ_T is

$$\phi_T = \arcsin\left(\frac{\ell_T}{2R}\right) = \arcsin\left(\frac{65.09''}{2(25')(12''/1')}\right) = 6.228^\circ,$$

which means that the rise of the individual lamella, according to Equation (16), is

$$T' = R - \frac{\ell_T}{2\tan\phi_T} = (25')(12''/1) - \frac{65.09''}{2\tan(6.228^\circ)} = 1.77'',$$

which must be subtracted from the depth of the lamella to find its depth at the connecting end. Since a 2x10 has a depth of 9.25", the depth at the connecting ends would be 7.48". Looking at Table B-4, the maximum length of a lamella for $\frac{1}{2}$ " bolts while still keeping the geometry factor C_{Δ} equal to 1.0 is 8.5' for three bolts and a radius of 25'. Since the total length of lamella is under that maximum, the spacing for keeping C_{Δ} equal to 1.0 should be used. According to Figure 16, this spacing is 4D, which would be 2" for $\frac{1}{2}$ " bolts.

For the group action factor,

$$E_m = E_s = E = 620,000 \text{ psi},$$

 $R_{EA} = \frac{A_s}{A_m} = \frac{(7.48'')(1.5'')}{13.88 \text{ in}^2} = 0.808.$

Then, with a spacing of 2",

$$u = 1 + \frac{180,000D^{1.5}s}{2E} \left[\frac{1}{A_m} + \frac{1}{A_s} \right],$$

$$u = 1 + \frac{180,000(0.5'')^{1.5}(2'')}{2(620,000 \text{ psi})} \left[\frac{1}{13.88 \text{ in}^2} + \frac{1}{(7.48'')(1.5'')} \right],$$

$$u = 1.107,$$

$$m = u - \sqrt{u^2 - 1} = (1.107) - \sqrt{(1.107)^2 - 1} = 0.8339.$$

Plugging these values in to solve for the geometry factor yields

$$C_{g} = \left[\frac{m(1-m^{2n})}{n\left[(1+R_{EA}m^{n})(1+m)-1+m^{2n}\right]}\right]\left[\frac{1+R_{EA}}{1-m}\right],$$

$$C_{g} = \left[\frac{(0.8339)(1-(0.8339)^{2(3)})}{(3)\left[(1+(0.808)(0.8339)^{3})(1+0.8339)-1+(0.8339)^{2(3)}\right]}\right]\left[\frac{1+0.808}{1-0.8339}\right],$$

$$C_{g} = 0.990.$$

Essentially, the design strength of the connection will only be reduced by 1% since the geometry factor is the only one not equal to 1.0.

Table F-1 shows that a $\frac{1}{2}$ " bolt can withstand 530 lbs of shear for southern pine (G = 0.55), so three bolts would have a shear capacity of

$$Z' = ZC_D C_M C_t C_g C_\Delta C_{eg} C_{di} C_{tn},$$

$$Z' = (3)(530 \text{ lbs})(1.0)(1.0)(1.0)(0.990)(1.0)(1.0)(1.0)(1.0) = 1570 \text{ lbs},$$

which is greater than the 1160 lbs required. Table 11 shows that a ½" bolt has a tensile capacity of 4460 lbs so by observation, three of them are more than sufficient for the 1540 lbs required.

The compression strength perpendicular to grain of the lamella is

$$F'_{c\perp} = F_{c\perp} C_M C_t C_i C_b = (565 \text{ psi})(1.0)(1.0)(1.0)(1.0) = \overline{(565 \text{ psi})},$$

and the required area of washers is then

$$A_{washer} \ge \frac{T_{bolts}}{F'_{c\perp}} = \frac{1540 \text{ lbs}}{565 \text{ psi}} = \boxed{2.72 \text{ in}^2}.$$

Washer size should be specified by the engineer based on availability of materials. If regular stamped washers have insufficient area, oversized square washers may need to be used.

7 Prototype Models

In order to better visualize and demonstrate the concept of the lamella roof, two models were created. They gave the author a better understanding of how the lamella roof fits together and also demonstrated the ease of assembly of the system. Also, a steel model allowed the author to perform load testing with strain gauges.

7.2 Matboard Model

A proof-of-concept model was created using matboard connected with #3 solid brass fasteners. The lamella pieces were cut using a laser cutter and assembled by hand. While assembling the model (shown in Figure 49), the author noted that as more pieces were added to the lamella arch, the arch itself became more stiff, indicating an interaction having to do with the interesting connection style used by lamella construction.



Figure 49 - Matboard Proof-of-Concept Model.

7.3 Steel Model

After the proof-of-concept model was made, a model made of sheet steel was fabricated and donated by H. Kubenik Metals of Milwaukee, WI. The model was precision-cut using a computer-controlled plasma cutter with the ends bent in a machine press (see Figure 50 and Figure 51). The steel model was approximately a two-times scale copy of the matboard model.



Figure 50 - Plasma Cutting of Steel Lamellas.



Figure 51 - Bending of Steel Lamellas.

After cutting and bending, the lamellas were assembled with machine screws, lock washers, and nuts to create a section of a lamella arch. The finished product is displayed in Figure 52.



Figure 52 - Assembled Steel Lamella Arch.

Though hard to see in Figure 52, the top of the arch had a distinct curvature in the short direction, resulting in "cupping" of the entire structure. This is most likely due to the fact that the drafting model used for fabrication was not as exact as required for a perfect fit.

The properties of the steel arch ended up being:

- Span [S] = 75''
- Rise [T] = 37''
- Arch width of 24"

- 12 ga. A36 steel (thickness [t] = 0.1084")
- 10-32 x ½" machine screws
- 8.5 lamellas per arch (n = 8.5)
- Spacing = 4.145"
- Nominal depth of lamella [d] = 2.05"

7.3.1 Load Testing

The steel model was tested to see if the resultant stresses on the model fit with those predicted by the load table program created by the author. Special bearing plates were fabricated out of 2x4 lumber to act like pinned connections at the springing ends of the arch as shown in Figure 53.



Figure 53 - Steel Model Bearing Plates.

To find the stresses in the lamellas during testing, several strain gauges were affixed to the model, as depicted in Figure 54.



Figure 54 - Strain Gauge Close-up.

The strain gauges were placed in the middle of a lamella span to reduce any affects that stress concentrations might have had on the results. They were placed in three groups at different parts of the arch, as shown in Figure 55. The gauge groups are depicted with a black rectangle.

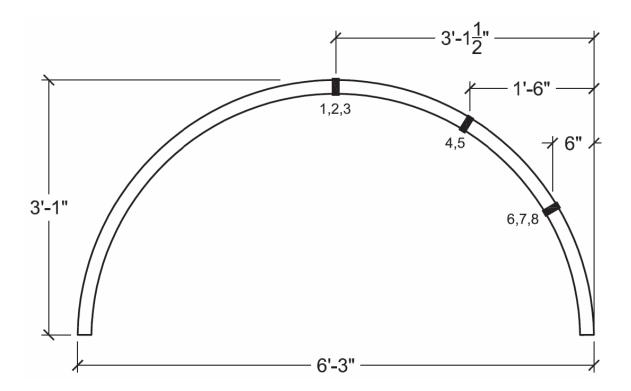


Figure 55 - Strain Gauge Locations.

The numbers next to the rectangles represent the locations of the numbered strain gauges. The gauge with the lower number is closer to the outside of the arch, i.e., SG1 is at the very top of the arch while SG3 is directly below it on the inside of the arch.

Three different loading patterns were used for testing, shown in Figure 56, Figure 57, and Figure 58, each to simulate a different real-world loading. Plastic bags, each filled with ten pounds of sand, were used to control the amount of load. The first series was to simulate a balanced snow load.



Figure 56 - Balanced Snow Load Simulation.

The second simulated a snow drift load by stacking sand bags on half of the structure. The sand bag loading does not exactly reflect the loading pattern depicted in the Simplified Figure 7-3 found in Appendix C since the loading is uniform. To counter this discrepancy, the loading pattern in the Simplified Figure 7-3 was averaged to $1.35p_f$ and the load put into the load table program was adjusted to match this value.



Figure 57 - Snow Drift Load Simulation.

The final loading stacked bags on the apex of the arch to simulate a point load. This point load would essentially be a lineal load along the length of the apex if the roof.



Figure 58 - Point Load Simulation.

Since the bags had a total bearing area of 10" by 10" (or 100 in²), the corresponding area load would be 14.4 psf. Also, the point load simulation is modeled as a lineal load on the length of the apex, which would correlate to 12 plf per bag. These corresponding loads were inputted into the load table program developed by the author, then adjusted using the process outlined in the end of Section 5.1 to find the predicted stresses on the lamellas. Since the lamellas are in biaxial bending and compression, the stresses are summed to reflect correlate to the correct combination of compression and moment.

Table 12 displays a list of data found during the different loading tests. Perhaps most interesting are the data from Strain Gauges 6-8. It appears that weak axis bending was so

large that on the face the strain gauges were attached, the effects of strong axis bending and compression were not enough to put compressive stress into the fibers.

Table 12 - Strain Gauge Testing Data.

		l a a d				Actual Stre	ss [σ] (psi))		
	ı	Load	SG1	SG2	SG3	SG4	SG5	SG6	SG7	SG8
>	1	12 bags	-348	580	319	-290	435	1769	1682	841
Snow	2	24 bags	-696	1160	1073	-377	841	3712	3219	1682
	3	36 bags	-1015	1653	1798	0	1624	5887	4901	2813
Balanced	4	48 bags	-1305	2291	2610	812	2755	8671	7308	4611
ala	5	60 bags	-1537	3045	3451	1885	4408	11687	10092	6902
8	6	72 bags	-1769	4205	5423	2465	5771	15573	13949	10527
	7	6 bags	29	522	464	-1160	203	522	551	464
≝	8	12 bags	58	870	870	2523	493	1131	1189	1044
Drift	9	18 bags	58	1218	1218	-3770	870	1769	1827	1624
Snow	10	24 bags	87	1566	1595	-4901	1305	2436	2581	2291
Sr	11	30 bags	87	1914	1943	-5829	1885	3219	3364	2987
	12	36 bags	116	2291	2407	-6728	2494	4031	4176	3654
	13	4 bags	-435	261	493	261	290	609	464	174
þ	14	8 bags	-725	725	1073	203	493	1044	725	290
Load	15	12 bags	-957	1131	1682	348	841	1624	1160	464
Point	16	16 bags	-1218	1508	2204	464	1131	2175	1566	667
Po	17	20 bags	-1537	1827	2813	638	1479	2871	2059	957
	18	24 bags	-1682	2291	3480	783	1769	3509	2523	1218

The data showing the predicted stress values are shown in Table 13.

Table 13 - Predicted Fiber Stresses.

		l a a d			Р	rogram Str	ess [σ] (ps	i)		
		Load	SG1	SG2	SG3	SG4	SG5	SG6	SG7	SG8
>	1	12 bags	-1125	-355	414	521	516	1463	670	-122
Snow	2	24 bags	-2249	-711	828	1042	1031	2926	1341	-244
d S	3	36 bags	-3374	-1066	1243	1564	1547	4389	2011	-367
Balanced	4	48 bags	-4499	-1421	1657	2085	2062	5852	2681	-489
ala	5	60 bags	-5624	-1776	2071	2606	2578	7315	3352	-612
В	6	72 bags	-6748	-2131	2486	3127	3094	8778	4022	-734
	7	6 bags	-872	-346	179	-741	1861	1219	968	717
Ħ	8	12 bags	-1744	-692	359	-1482	3721	2437	1936	1435
Drift	9	18 bags	-2615	-1038	538	-2223	5582	3655	2903	2151
Snow	10	24 bags	-3487	-1385	718	-2964	7443	4873	3871	2869
Sr	11	30 bags	-4359	-1731	897	-3705	9303	6092	4839	3586
	12	36 bags	-5231	-2077	1076	-4446	11164	7310	5807	4303
	13	4 bags	-1102	-172	759	83	406	711	264	-184
þe	14	8 bags	-2204	-343	1518	166	812	1424	527	-369
Load	15	12 bags	-3306	-515	2277	248	1218	2135	791	-553
Point	16	16 bags	-4408	-686	3036	331	1625	2846	1054	-737
Pc	17	20 bags	-5510	-858	3795	414	2031	3558	1318	-922
	18	24 bags	-6612	-1029	4554	495	2439	4270	1582	-1106

One can compare the predicted stresses to the actual stresses, which results in Table 14.

Table 14 - Percent Difference in Predicted versus Observed Stress.

		11		Percen	t Differen	ce in Predi	cted versus	s Observed	Stress	
		Load	SG1	SG2	SG3	SG4	SG5	SG6	SG7	SG8
>	1	12 bags	69%	263%	23%	156%	16%	-21%	-151%	788%
Snow	2	24 bags	69%	263%	-30%	136%	18%	-27%	-140%	788%
ठ	3	36 bags	70%	255%	-45%	100%	-5%	-34%	-144%	867%
Balance	4	48 bags	71%	261%	-57%	61%	-34%	-48%	-173%	1042%
ala	5	60 bags	73%	271%	-67%	28%	-71%	-60%	-201%	1229%
8	6	72 bags	74%	297%	-118%	21%	-87%	-77%	-247%	1534%
	7	6 bags	103%	251%	-159%	-57%	89%	57%	43%	35%
Drift	8	12 bags	103%	226%	-143%	270%	87%	54%	39%	27%
	9	18 bags	102%	217%	-126%	-70%	84%	52%	37%	25%
Snow	10	24 bags	102%	213%	-122%	-65%	82%	50%	33%	20%
s	11	30 bags	102%	211%	-117%	-57%	80%	47%	30%	17%
	12	36 bags	102%	210%	-124%	-51%	78%	45%	28%	15%
	13	4 bags	61%	252%	35%	-216%	29%	14%	-76%	194%
aq	14	8 bags	67%	311%	29%	-23%	39%	27%	-37%	179%
2	15	12 bags	71%	320%	26%	-40%	31%	24%	-47%	184%
oint	16	16 bags	72%	320%	27%	-40%	30%	24%	-49%	190%
P	17	20 bags	72%	313%	26%	-54%	27%	19%	-56%	204%
	18	24 bags	75%	323%	24%	-58%	27%	18%	-60%	210%

Unfortunately, this comparison of values appears to be inconclusive. There are enough values within the 10-40% overestimate range to make one wonder if the matrix program is predicting values correctly. Yet there are also plenty of values so far out of range that the predictions seem wildly incorrect. More testing would help clarify these inconsistencies.

Another factor that may contribute to the discrepancy in values is the stiffness of the steel arch. The exact stiffness is difficult to determine due to the nature of the lattice structure, and, as mentioned before, a stiffer structure has a tendency to take on more moment. A more flexible arch would see more axial thrust, which could help bring some of the values closer to those predicted by the matrix program.

The author also tested to see if the horizontal reaction of the arch matched that predicted by the matrix program. For testing, two tension gauges were attached to either end of the steel lamella arch to measure the horizontal reaction. One end of the arch was placed atop rollers to facilitate the stretching of the tension gauges while the other end was held in place. This setup is show in Figure 59.



Figure 59 - Horizontal Reaction Test Setup.

The steel arch was subjected to the same loading conditions as the strain gauge tests. The measurements found on the tension gauges were averaged. Since the arch was two feet wide, the average of the two reactions is directly proportionate to the horizontal reaction per foot given by the matrix program developed by the author. The actual values are compared to the predicted values in Table 15.

Table 15 - Horizontal Reaction Comparison.

		Hor	izontal R	eaction	
	l	Load	Actual (lbs)	Predicted (lbs)	% Diff
3	1	12 bags	6.25	17	63%
Balanced Snow	2	24 bags	13.5	34.1	60%
d S	3	36 bags	19.75	51.1	61%
nce	4	48 bags	27.75	68.1	59%
ala	5	60 bags	37.5	85.1	56%
-	6	72 bags	-	102.2	-
	7	6 bags	4.25	17.4	76%
Ħ	8	12 bags	8.25	34.8	76%
٥	9	18 bags	12.75	52.1	76%
Snow Drift	10	24 bags	16	69.5	77%
s	11	30 bags	22.25	86.9	74%
	12	36 bags	27.25	104.3	74%
	13	4 bags	3.5	8.7	60%
þe	14	8 bags	6.5	17.3	62%
P	15	12 bags	8.5	26	67%
Point Load	16	16 bags	14.5	34.6	58%
Pc	17	20 bags	17.25	43.3	60%
	18	24 bags	20.75	51.9	60%

In all cases, the predicted values are gross overestimates of the actual horizontal reactions. Again, this could have to do with the stiffness of the steel arch being different than that used in the matrix program. Fortunately, none of the values are underestimates and the design horizontal reaction would be a conservative design value.

8 Conclusion

The lamella structure offers a unique and aesthetically pleasing architectural roof that has the added bonus of using less material than what a "traditional" roof spanning the same distance might. Its modular nature makes fabrication a cost-effective, repetitive task, and its use of widely-available dimensional lumber makes its construction an attainable goal for many smaller projects.

Previous efforts to engineer the structure relied on approximations due to the lack of computer analysis. Modern matrix systems can be used to accurately solve for the forces in a two-pinned arch with a given stiffness and updated building codes allow the engineer the peace of mind to know that he or she is designing for a real-life loading scenario.

The load tables developed by the author, coupled with a detailed explanation of the calculations necessary to check for member and connection adequacy, should allow one to perform an introductory strength check and come up with a preliminary design for a lamella roof. However, due to the fact that the loading patterns employed by the ASCE 7-10 are generally overestimates and due to the fact that the calculations assume absolutely worst case loads, a more in-depth analysis should be undertaken to more accurately find the forces in the arched roof.

Through testing, the author was unfortunately unable to find conclusive evidence that the values predicted by his matrix program matched those found from testing. Some values were close enough to be matches while others were clearly not. More testing and refining of the matrix program should be undertaken to determine how exactly the two relate.

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Appendix A: NDS 2005 Tables and Figures

This appendix features a copy of a relevant figure (Figure A-1) and copies of relevant tables (Tables A-1 through A-5) from the American Forest and Paper Association's National Design Specification for Wood Construction.¹

Table A-1 – Edge Distance Requirements.

Table 11.5.1A	Edge Distance Requirements ^{1,2}
Direction of Loading	Minimum Edge Distance
Parallel to Grain:	
when $\ell/D \le 6$	1.5D
when $\ell/D > 6$	1.5D or ½ the spacing
	between rows, whichever is
	greater
Perpendicular to Grain:2	
loaded edge	4D
unloaded edge	1.5D

^{1.} The I/D ratio used to determine the minimum edge distance shall be the lesser of:

Table A-2 – End Distance Requirements.

Table 11.5.1B	End Distanc	е
	Requiremen	ts
	End Dis	stances
	Minimum end distance for $C_{\Delta} = 0.5$	Minimum end distance for C_{Δ} = 1.0
Direction of Loading		
Perpendicular to Grain	2D	4D
Parallel to Grain, Compression: (fastener bearing away from member end)	2D	4D
Parallel to Grain, Tension: (fastener bearing to- wards member end)		
for softwoods	3.5D	7D
for hardwoods	2.5D	5D

¹ American Forest & Paper Association. 2006. *National Design Specification for Wood Construction*, 2005 Edition. Washington D.C.: American Forest & Paper Association.

⁽a) length of fastener in wood main member/D = ℓ_m/D (b) total length of fastener in wood side member(s)/D = ℓ_s/D

^{2.} Heavy or medium concentrated loads shall not be suspended below the neutral axis of a single sawn lumber or structural glued laminated timber beam except where mechanical or equivalent reinforcement is provided to resist tension stresses perpendicular to grain (see 3.8.2 and 10.1.3).

Table A-3 – Spacing Requirements for Fasteners in a Row.

Table 11.5.1C Spacing Requirements for Fasteners in a Row

		Spacing
Direction of Loading	Minimum spacing	Minimum spacing for $C_{\Delta} = 1.0$
Parallel to Grain	3D	4D
Perpendicular to		Required spacing for
Grain	3D	attached members

Table A-4 – Spacing Requirements Between Rows.

Table 11.5.1D	Spacing Requirements Between Rows ^{1,2}
Direction of Loading	Minimum Edge Distance
Parallel to Grain:	1.5D
Perpendicular to Grain:	
when $\ell/D \le 2$	2.5D
when $2 < \ell/D < 6$	$(5\ell + 10D) / 8$
when $\ell/D \ge 6$	5D

^{1.} The ℓ/D ratio used to determine the minimum edge distance shall be the

Table A-5 – Reduction Term, R_d .

Table 11.3.1B Reduction Term, R_d

Fastener Size	Yield Mode	Reduction Term, R _d
$0.25'' \le D \le 1''$	I_m, I_s II III_m, III_s, IV	$\begin{array}{c} 4~K_\theta \\ 3.6~K_\theta \\ 3.2~K_\theta \end{array}$
$D<0.25^{\prime\prime}$	$I_m,\ I_s,\ II,\ III_m,\ III_s,\ IV$	K_D^{-1}

Notes:

$$K_{\theta} = 1 + 0.25(\theta - 90)$$

 θ = maximum angle of load to grain $(0 \le \theta \le 90)$

for any member in a connection

D= diameter, in. (see 11.3.6)

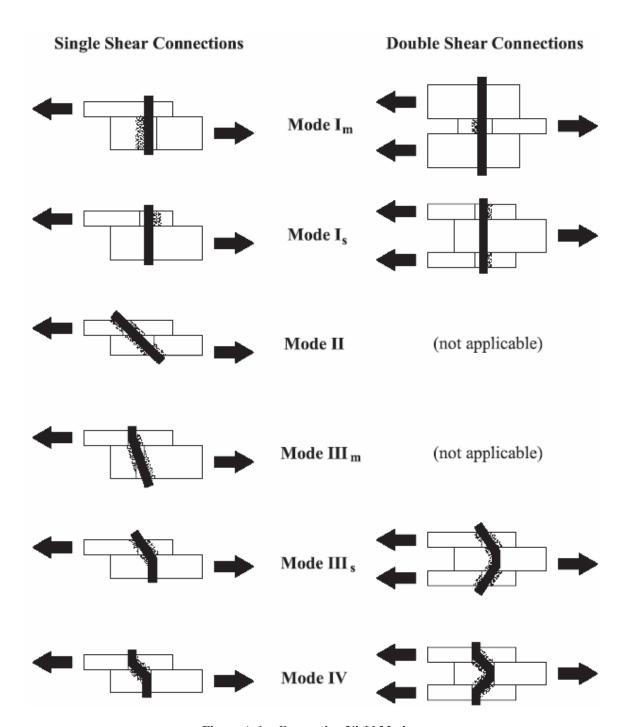
 $K_D = 2.2$ for $D \le 0.17$ "

for 0.17" < D < 0.25" $K_D = 10D + 0.5$

⁽a) length of fastener in wood main member/D = $\ell_{\rm m}/D$ (b) total length of fastener in wood side member(s)/D = $\ell_{\rm s}/D$

^{2.} The spacing between outer rows of fasteners paralleling the member on a single splife plate shall not exceed 5" (see Figure 11H).

^{1.}For threaded fasteners where nominal diameter (see Appendix L) is greater than or equal to 0.25" and root diameter is less than 0.25", $R_{\text{d}}=K_{D}K_{\text{B}}.$



 $Figure \ A-1-Connection \ Yield \ Modes.$

Appendix B: Curvature versus Length Tables

Tables B-1 through B-10 feature curvature versus lamella length data for 1/4", 5/16", 3/8", 1/2", 3/4", 7/8", and 1" bolts.

			Table B-1	- 1/4" Bolt	s - Maximur	n Lamella L	ength [ℓ] (ft))				
	$C_{\Delta}=1.0$											
Radius	us (2) Bolts per Connection					(3) Bolts per Connection			(4) Bolts per Connection			
[R] (ft)	2 x 4	2 x 6	2 x 8	2 x 10	2 x 6	2 x 8	2 x 10	2 x 6	2 x 8	2 x 10		
1000	31.0	55.9	70.8	84.7	42.1	60.5	76.3	20.4	48.1	66.9		
900	29.4	53.1	67.2	80.4	40.0	57.5	72.4	19.4	45.7	63.5		
800	27.8	50.0	63.4	75.8	37.7	54.2	68.3	18.3	43.1	59.9		
700	26.0	46.8	59.3	70.9	35.3	50.7	63.9	17.1	40.3	56.0		
600	24.1	43.4	54.9	65.7	32.7	47.0	59.2	15.9	37.4	51.9		
500	22.0	39.6	50.2	60.0	29.9	42.9	54.1	14.5	34.1	47.4		
400	19.7	35.5	44.9	53.7	26.8	38.4	48.4	13.0	30.6	42.4		
450	20.9	37.6	47.6	56.9	28.4	40.7	51.3	13.8	32.4	45.0		
400	19.7	35.5	44.9	53.7	26.8	38.4	48.4	13.0	30.6	42.4		
350	18.5	33.2	42.0	50.2	25.1	36.0	45.3	12.2	28.6	39.7		
300	17.1	30.8	38.9	46.5	23.2	33.3	41.9	11.3	26.5	36.8		
275	16.4	29.5	37.3	44.6	22.3	31.9	40.2	10.9	25.4	35.2		
250	15.7	28.1	35.6	42.5	21.2	30.4	38.3	10.4	24.2	33.6		
225	14.9	26.7	33.8	40.3	20.2	28.9	36.4	9.9	23.0	31.9		
200	14.1	25.2	31.9	38.0	19.0	27.3	34.3	9.3	21.7	30.1		
175	13.2	23.6	29.8	35.6	17.8	25.5	32.1	8.7	20.3	28.2		
150	12.2	21.9	27.6	33.0	16.5	23.7	29.8	8.1	18.8	26.1		
125	11.2	20.0	25.3	30.1	15.1	21.6	27.2	7.4	17.2	23.9		
100	10.0	17.9	22.6	27.0	13.6	19.4	24.4	6.7	15.5	21.4		
90	9.5	17.0	21.5	25.6	12.9	18.4	23.1	6.4	14.7	20.3		
80	9.0	16.1	20.3	24.2	12.2	17.4	21.8	6.0	13.9	19.2		
70	8.5	15.0	19.0	22.6	11.4	16.3	20.4	5.7	13.0	17.9		
60	7.9	14.0	17.6	21.0	10.6	15.1	18.9	5.3	12.0	16.6		
50	7.2	12.8	16.1	19.2	9.7	13.8	17.3	4.8	11.0	15.2		
40	6.5	11.5	14.4	17.2	8.7	12.4	15.5	4.4	9.9	13.6		
30	5.7	10.0	12.5	14.9	7.6	10.8	13.5	3.8	8.6	11.8		
20	4.7	8.2	10.3	12.2	6.3	8.8	11.0	3.2	7.1	9.7		
18	4.5	7.8	9.8	11.6	5.9	8.4	10.5	3.0	6.7	9.2		
16	4.2	7.4	9.2	10.9	5.6	7.9	9.9	2.9	6.4	8.7		
14	4.0	6.9	8.6	10.2	5.3	7.4	9.3	2.7	6.0	8.2		
12	3.7	6.4	8.0	9.5	4.9	6.9	8.6	2.6	5.6	7.6		
10	3.4	5.9	7.3	8.7	4.5	6.3	7.9	2.4	5.1	7.0		

Table B-2 - 5/16" Bolts - Maximum Lamella Length [\ell] (ft) $C_{\Delta} = 1.0$ Radius (2) Bolts per Connection (3) Bolts per Connection (4) Bolts per Connection [R] (ft) 2 x 4 2 x 6 2 x 10 2 x 6 2 x 8 2 x 8 2 x 8 2 x 10 2 x 10 2 x 12 1000 22.4 51.7 82.0 35.1 55.9 67.5 72.7 41.1 62.1 77.5 900 49.0 77.8 53.1 69.0 21.3 64.1 33.3 39.0 58.9 73.6 800 20.1 46.3 60.4 73.3 31.4 50.0 65.1 36.8 55.5 69.4 68.6 700 43.3 46.8 18.8 56.6 29.4 60.9 34.5 52.0 64.9 63.6 600 17.4 40.1 52.4 27.3 43.4 56.4 31.9 48.1 60.1 500 16.0 36.6 47.8 58.0 24.9 39.6 51.5 29.2 44.0 54.9 400 14.3 32.8 42.8 52.0 22.3 39.4 35.5 46.1 26.1 49.1 450 15.2 34.8 45.4 55.1 23.7 37.6 48.9 27.7 41.7 52.1 400 14.3 32.8 42.8 52.0 22.3 35.5 46.1 26.1 39.4 49.1 350 30.7 40.1 48.6 20.9 33.2 43.1 24.5 36.9 46.0 13.4 300 28.5 45.0 30.8 40.0 42.6 12.4 37.1 19.4 22.7 34.1 11.9 27.3 43.1 29.5 21.7 40.8 275 35.6 18.6 38.3 32.7 250 11.4 26.0 33.9 41.1 17.7 28.1 36.5 20.7 31.2 38.9 225 24.7 39.0 26.7 19.7 29.6 36.9 10.8 32.2 16.8 34.7 200 10.2 23.3 30.4 36.8 15.9 25.2 32.7 18.6 27.9 34.8 175 9.6 21.8 28.4 34.5 14.9 23.6 30.6 17.4 26.2 32.6 20.2 150 8.9 26.4 31.9 13.8 21.9 28.4 16.1 24.2 30.2 125 18.5 29.2 12.6 20.0 25.9 22.2 8.2 24.1 14.8 27.6 100 7.3 16.6 21.6 26.1 11.3 17.9 23.2 13.2 19.9 24.7 90 7.0 15.7 20.5 24.8 10.8 17.0 22.0 12.6 18.9 23.5 80 6.6 14.9 19.3 23.4 10.2 16.1 20.8 11.9 17.8 22.2 21.9 70 6.2 13.9 18.1 9.5 15.0 19.5 11.1 16.7 20.7 12.9 20.3 14.0 19.2 60 5.8 16.8 8.9 18.0 10.3 15.5 50 5.3 11.8 15.4 18.6 8.1 12.8 16.5 9.5 14.1 17.6 40 4.8 10.6 13.8 16.6 7.3 11.5 14.8 8.5 12.7 15.7 30 4.2 9.2 12.0 14.4 10.0 12.8 7.4 11.0 13.7 6.4 20 3.5 7.6 9.8 11.8 5.3 8.2 10.5 6.1 9.1 11.2 18 3.3 7.2 9.3 11.2 5.0 7.8 10.0 5.8 8.6 10.6

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8.3

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Table B-3 - 3/8" Bolts - Maximum Lamella Length [l] (ft) $C_{\Delta} = 1.0$ Radius (2) Bolts per Connection (3) Bolts per Connection (4) Bolts per Connection [R] (ft) 2 x 6 2 x 6 2 x 10 2 x 8 2 x 8 2 x 10 2 x 8 2 x 10 2 x 12 1000 22.0 46.0 63.3 78.5 48.8 67.4 27.5 54.1 71.3 900 20.9 64.0 51.3 43.6 60.0 74.5 46.3 26.1 67.6 800 41.1 70.2 19.7 43.7 60.3 24.7 48.4 63.8 56.6 700 38.5 53.0 65.7 18.5 40.9 56.4 23.1 45.3 59.7 55.3 600 35.7 49.1 60.9 17.1 37.9 52.3 21.4 42.0 500 32.6 44.8 55.6 15.7 47.8 19.6 38.3 50.5 34.6 400 14.1 42.8 34.3 45.2 29.2 40.1 49.8 31.0 17.5 450 30.9 42.6 14.9 32.9 45.3 18.6 36.4 47.9 52.8 400 29.2 40.1 49.8 14.1 31.0 42.8 17.5 34.3 45.2 350 13.2 40.0 32.1 42.3 27.3 37.6 46.6 29.0 16.4 300 12.2 39.2 25.3 34.8 43.1 26.9 37.1 15.2 29.8 275 11.7 35.5 28.5 37.5 24.3 33.3 41.3 25.8 14.6 250 23.2 31.8 39.4 11.2 24.6 33.9 13.9 27.2 35.8 225 25.8 22.0 30.2 37.4 10.6 23.3 32.1 13.2 34.0 200 20.7 28.5 10.0 30.3 12.5 24.4 32.1 35.3 22.0 175 19.4 26.7 33.0 9.4 20.6 28.4 11.7 22.8 30.0 150 18.0 24.7 30.6 8.7 19.1 26.3 10.9 21.2 27.8 125 24.0 10.0 19.3 25.4 16.5 22.6 28.0 8.0 17.5 100 14.8 20.2 25.0 7.2 15.7 21.5 8.9 17.3 22.8 90 14.0 19.2 23.8 6.9 14.9 20.5 8.5 16.5 21.6 80 13.2 18.1 22.4 6.5 14.1 19.3 8.0 15.5 20.4 70 12.4 17.0 21.0 6.1 13.2 18.1 7.5 14.6 19.1 5.7 16.8 17.7 60 11.5 15.7 19.5 12.2 7.0 13.5 50 10.5 14.4 17.8 5.2 11.2 15.3 6.4 12.4 16.2 40 9.5 12.9 15.9 4.7 10.0 13.7 5.8 11.1 14.5 30 11.2 4.1 8.7 11.9 9.6 12.6 8.2 13.8 5.1 20 9.2 11.3 3.4 7.2 9.8 4.2 7.9 10.3 6.8 18 6.5 8.8 10.8 3.3 6.8 9.3 4.0 7.5 9.8 3.1 7.1 16 6.1 8.3 10.2 6.5 8.8 9.3 2.9 8.2 14 5.7 7.8 9.5 6.1 3.6 6.7 8.7

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8.1

7.4

Table B-4 - 1/2" Bolts - Maximum Lamella Length [l] (ft) $C_{\Delta} = 1.0$ Radius (2) Bolts per Connection (3) Bolts per Connection (4) Bolts per Connection [R] (ft) 2 x 6 2 x 8 2 x 10 2 x 12 2 x 8 2 x 10 2 x 10 2 x 12 2 x 12 2 x 14 1000 28.7 83.9 23.5 69.9 52.2 69.9 52.2 69.9 23.5 52.2 900 49.5 79.6 22.3 49.5 66.3 49.5 27.3 66.3 22.3 66.3 800 25.7 46.7 62.5 75.1 21.1 46.7 62.5 21.1 46.7 62.5 700 43.7 43.7 24.1 58.5 70.2 19.7 58.5 19.7 43.7 58.5 65.0 600 22.3 40.5 54.2 18.3 40.5 54.2 18.3 40.5 54.2 500 20.4 37.0 49.5 59.4 16.7 37.0 49.5 16.7 37.0 49.5 400 18.3 33.1 44.3 53.2 15.0 44.3 33.1 44.3 15.0 33.1 450 19.4 35.1 47.0 56.4 15.9 35.1 47.0 15.9 35.1 47.0 400 18.3 33.1 44.3 53.2 15.0 33.1 44.3 15.0 33.1 44.3 350 41.5 49.8 14.1 31.0 41.5 31.0 41.5 17.1 31.0 14.1 300 15.9 28.7 46.1 28.7 28.7 38.4 38.4 13.0 38.4 13.0 27.5 44.1 12.5 12.5 36.8 275 15.2 36.8 27.5 36.8 27.5 250 14.5 26.3 35.1 42.1 11.9 26.3 35.1 11.9 26.3 35.1 225 24.9 40.0 24.9 13.8 33.3 11.3 33.3 11.3 24.9 33.3 200 13.0 23.5 31.4 37.7 10.7 23.5 31.4 10.7 23.5 31.4 175 12.2 22.0 29.4 35.3 10.0 22.0 29.4 10.0 22.0 29.4 20.4 20.4 150 11.3 27.3 32.7 9.3 27.3 9.3 20.4 27.3 18.7 29.9 18.7 24.9 125 10.4 24.9 8.5 8.5 18.7 24.9 100 9.3 16.7 22.3 26.7 7.7 16.7 22.3 7.7 16.7 22.3 90 8.9 15.9 21.2 25.4 7.3 15.9 21.2 7.3 15.9 21.2 15.0 80 8.4 15.0 20.0 23.9 6.9 20.0 6.9 15.0 20.0 70 7.9 14.1 18.7 22.4 6.5 14.1 18.7 6.5 14.1 18.7 13.0 17.4 20.8 17.4 17.4 60 7.3 6.0 13.0 6.0 13.0 50 6.7 11.9 15.9 19.0 5.5 11.9 15.9 5.5 11.9 15.9 40 6.0 10.7 14.2 17.0 5.0 10.7 14.2 5.0 10.7 14.2 30 9.3 12.4 14.8 4.4 9.3 12.4 4.4 9.3 12.4 5.3 20 4.4 7.7 10.1 12.1 3.6 7.7 10.1 3.6 7.7 10.1

18

16

14

12

10

4.2

3.9

3.7

3.5

3.2

7.3

6.9

6.5

6.0

5.5

9.6

9.1

8.5

7.9

7.2

11.5

10.8

10.1

9.4

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3.5

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6.0

5.5

9.6

9.1

8.5

7.9

7.2

Table B-5 - 3/8" Bolts - Maximum Lamella Length [ℓ] (ft)

 $C_{\Delta} = 0.5$ Radius (2) Bolts per Connection (3) Bolts per Connection (4) Bolts per Connection [R] (ft) 2 x 4 2 x 10 2 x 6 2 x 8 2 x 6 2 x 8 2 x 8 2 x 10 2 x 10 2 x 12 1000 79.1 6.1 47.1 64.1 26.3 50.8 68.9 32.6 56.8 73.4 900 44.7 75.1 48.3 65.4 5.9 60.8 24.9 31.0 53.9 69.6 800 5.5 42.1 57.3 70.8 23.5 45.5 29.2 50.8 65.7 61.6 700 5.2 39.4 53.7 66.3 22.0 42.6 57.7 27.3 47.6 61.5 39.5 600 4.8 36.5 49.7 61.4 20.4 53.4 25.3 44.1 56.9 500 4.5 33.4 45.4 56.1 18.7 36.1 48.8 23.2 40.3 52.0 400 29.9 40.6 50.2 32.3 4.0 16.7 43.7 20.8 36.1 46.5 450 4.2 31.7 43.1 53.2 17.7 34.2 46.3 22.0 38.2 49.3 400 4.0 29.9 40.6 50.2 16.7 32.3 43.7 20.8 36.1 46.5 350 28.0 38.0 46.9 15.7 30.2 40.9 19.4 43.6 3.8 33.7 25.9 37.9 40.3 300 3.5 35.2 43.5 14.5 28.0 18.0 31.3 24.8 41.7 13.9 38.6 275 3.4 33.8 26.8 36.3 17.3 30.0 250 3.3 23.7 32.2 39.7 13.3 25.6 34.6 16.5 28.6 36.9 225 22.5 3.1 30.6 37.7 12.6 24.3 32.9 15.7 27.1 35.0 200 2.9 21.2 28.8 35.6 11.9 22.9 31.0 25.6 14.8 33.0 175 2.8 19.9 27.0 33.3 11.2 21.5 29.0 13.8 24.0 30.9 150 2.6 18.4 25.0 30.8 10.4 19.9 26.9 12.8 22.2 28.6 125 16.9 28.2 9.5 18.2 24.6 20.3 2.4 22.9 11.8 26.2 100 2.2 15.1 20.5 25.2 8.5 16.3 22.0 10.6 18.2 23.4 90 2.1 14.4 19.5 24.0 15.5 20.9 10.0 17.3 22.2 8.1 7.7 80 2.0 13.6 18.4 22.6 14.6 19.7 9.5 16.3 21.0 70 1.9 12.7 17.2 21.2 7.2 13.7 18.5 8.9 15.3 19.7 11.8 15.9 19.6 60 1.8 6.7 12.7 17.1 8.3 14.2 18.2 50 1.7 10.8 14.6 17.9 6.1 11.6 15.7 7.6 13.0 16.7 40 1.5 9.7 13.1 16.1 5.5 10.4 14.0 6.8 11.6 14.9 30 8.4 11.4 13.9 9.1 12.2 5.9 10.1 13.0 1.4 4.8 20 1.2 6.9 9.3 11.4 4.0 7.5 10.0 4.9 8.3 10.6 18 1.1 6.6 8.9 10.9 3.8 7.1 9.5 4.7 7.9 10.1 6.2 10.2 16 1.1 8.4 3.6 6.7 9.0 4.4 7.5 9.5 1.0 5.9 7.9 6.3 14 9.6 3.4 8.4 4.2 7.0 8.9 12 1.0 5.5 7.3 8.9 3.2 5.9 7.8 3.9 6.5 8.3 5.0 10 0.9 6.7 8.1 2.9 5.4 7.1 3.6 6.0 7.6

Table B-6 - 1/2" Bolts - Maximum Lamella Length [ℓ] (ft)

 $C_{\Delta} = 0.5$

	$C_{\Delta} = 0.5$									
Radius	(2) Bolts per	Connection	on	(3) Bol	ts per Conr	nection	(4) Bol	ts per Coni	nection
[R] (ft)	2 x 6	2 x 8	2 x 10	2 x 12	2 x 8	2 x 10	2 x 12	2 x 10	2 x 12	2 x 14
1000	36.1	56.5	73.2	86.6	38.8	60.5	76.3	44.5	64.3	79.3
900	34.2	53.6	69.4	82.2	36.8	57.5	72.4	42.2	61.0	75.3
800	32.3	50.6	65.5	77.5	34.7	54.2	68.3	39.8	57.6	71.0
700	30.2	47.3	61.3	72.5	32.5	50.7	63.9	37.3	53.9	66.4
600	28.0	43.9	56.7	67.2	30.1	47.0	59.2	34.5	49.9	61.5
500	25.6	40.1	51.8	61.4	27.5	42.9	54.1	31.5	45.6	56.2
400	22.9	35.9	46.4	54.9	24.7	38.4	48.4	28.3	40.8	50.3
450	24.3	38.0	49.2	58.2	26.1	40.7	51.3	29.9	43.3	53.3
400	22.9	35.9	46.4	54.9	24.7	38.4	48.4	28.3	40.8	50.3
350	21.5	33.6	43.4	51.4	23.1	36.0	45.3	26.5	38.2	47.1
300	19.9	31.1	40.2	47.6	21.4	33.3	41.9	24.5	35.4	43.6
275	19.1	29.8	38.5	45.6	20.5	31.9	40.2	23.5	33.9	41.8
250	18.2	28.4	36.7	43.5	19.6	30.4	38.3	22.4	32.3	39.8
225	17.3	27.0	34.9	41.3	18.6	28.9	36.4	21.3	30.7	37.8
200	16.3	25.5	32.9	38.9	17.5	27.3	34.3	20.1	29.0	35.7
175	15.3	23.8	30.8	36.4	16.4	25.5	32.1	18.8	27.1	33.4
150	14.2	22.1	28.5	33.7	15.2	23.7	29.8	17.4	25.1	30.9
125	13.0	20.2	26.1	30.8	13.9	21.6	27.2	15.9	23.0	28.3
100	11.6	18.1	23.4	27.6	12.5	19.4	24.4	14.3	20.6	25.3
90	11.1	17.2	22.2	26.2	11.9	18.4	23.1	13.6	19.5	24.0
80	10.5	16.2	20.9	24.7	11.2	17.4	21.8	12.8	18.4	22.7
70	9.8	15.2	19.6	23.1	10.5	16.3	20.4	12.0	17.3	21.2
60	9.1	14.1	18.2	21.4	9.8	15.1	18.9	11.2	16.0	19.7
50	8.3	12.9	16.6	19.6	8.9	13.8	17.3	10.2	14.6	18.0
40	7.5	11.6	14.9	17.6	8.0	12.4	15.5	9.2	13.1	16.1
30	6.5	10.1	12.9	15.2	7.0	10.8	13.5	8.0	11.4	14.0
20	5.4	8.3	10.6	12.5	5.8	8.8	11.0	6.6	9.4	11.5
18	5.1	7.9	10.1	11.8	5.5	8.4	10.5	6.3	8.9	10.9
16	4.9	7.4	9.5	11.2	5.2	7.9	9.9	5.9	8.4	10.3
14	4.6	7.0	8.9	10.5	4.9	7.4	9.3	5.6	7.9	9.6
12	4.3	6.5	8.3	9.7	4.6	6.9	8.6	5.2	7.3	8.9
10	3.9	5.9	7.6	8.9	4.2	6.3	7.9	4.8	6.7	8.2

Table B-7 - 5/8" Bolts - Maximum Lamella Length [ℓ] (ft)

 $C_{\Delta} = 0.5$ Radius (2) Bolts per Connection (3) Bolts per Connection (4) Bolts per Connection [R] (ft) 2 x 6 2 x 8 2 x 10 2 x 12 2 x 8 2 x 10 2 x 10 2 x 12 2 x 12 2 x 14 1000 80.2 47.4 14.5 46.0 65.4 8.5 66.4 14.5 48.8 67.4 900 43.6 76.1 45.0 63.0 46.3 13.8 62.0 8.1 13.8 64.0 800 13.0 41.1 58.5 71.8 7.7 42.4 59.4 13.0 43.7 60.3 700 40.9 12.2 38.5 54.7 67.1 7.2 39.7 55.6 12.2 56.4 35.7 62.2 600 11.3 50.7 6.7 36.8 51.5 11.3 37.9 52.3 500 10.4 32.6 46.3 56.8 6.1 33.6 47.0 10.4 34.6 47.8 29.2 400 50.8 30.1 9.3 42.8 9.3 41.5 5.5 42.1 31.0 450 9.9 30.9 44.0 53.9 5.9 31.9 44.6 9.9 32.9 45.3 400 9.3 29.2 41.5 50.8 5.5 30.1 42.1 9.3 31.0 42.8 350 27.3 38.8 47.6 28.2 39.4 29.0 40.0 8.7 5.2 8.7 300 25.3 44.1 37.1 8.1 36.0 4.8 26.1 36.5 8.1 26.9 24.3 42.2 25.0 35.5 275 7.8 34.4 4.7 35.0 7.8 25.8 250 7.5 23.2 32.9 40.3 4.5 23.9 33.4 7.5 24.6 33.9 225 22.0 38.2 22.7 7.1 31.2 4.2 31.7 7.1 23.3 32.1 200 6.7 20.7 29.4 36.0 4.0 21.4 29.9 6.7 22.0 30.3 175 6.3 19.4 27.5 33.7 3.8 20.0 28.0 6.3 20.6 28.4 150 5.9 18.0 25.5 31.3 3.5 18.6 25.9 5.9 19.1 26.3 125 5.4 16.5 28.6 17.0 23.7 17.5 23.3 3.3 5.4 24.0 100 4.8 14.8 20.9 25.6 2.9 15.2 21.2 4.8 15.7 21.5 90 4.6 14.0 19.8 24.3 2.8 14.5 20.1 4.6 14.9 20.5 80 4.4 13.2 18.7 22.9 2.7 13.7 19.0 4.4 14.1 19.3 70 4.1 12.4 17.5 21.4 2.5 12.8 17.8 4.1 13.2 18.1 11.5 19.9 60 3.8 16.3 2.4 11.9 16.5 3.8 12.2 16.8 50 3.5 10.5 14.9 18.2 2.2 10.9 15.1 3.5 11.2 15.3 40 3.2 9.5 13.3 16.3 2.0 9.8 13.5 3.2 10.0 13.7 30 2.8 8.2 11.6 14.1 8.5 11.8 2.8 8.7 11.9 1.8 20 2.4 6.8 9.5 11.6 1.5 7.0 9.7 2.4 7.2 9.8 18 2.3 6.5 9.0 11.0 1.5 6.7 9.2 2.3 6.8 9.3 6.1 10.4 6.3 16 2.2 8.5 1.4 8.7 2.2 6.5 8.8 5.7 9.7 1.3 5.9 14 2.0 8.0 8.1 2.0 6.1 8.2 12 1.9 5.3 7.4 9.0 1.3 5.5 7.5 1.9 5.6 7.6 4.9 6.8 10 1.8 8.2 1.2 5.0 6.9 1.8 5.2 7.0

Table B-8 - 3/4" Bolts - Maximum Lamella Length [f] (ft) $C_{\Delta} = 0.5$ (4) Bolts per Radius (2) Bolts per Connection (3) Bolts per Connection Connection [R] (ft) 2 x 14 2 x 8 2 x 10 2 x 12 2 x 10 2 x 12 2 x 14 2 x 12 2 x 14 1000 50.8 31.0 55.9 72.7 86.3 26.3 53.4 70.8 20.4 900 29.4 53.1 69.0 81.8 24.9 50.7 67.2 19.4 48.3 800 27.8 50.0 65.1 77.2 23.5 47.8 63.4 18.3 45.5 700 26.0 46.8 60.9 72.2 22.0 44.8 59.3 17.1 42.6 600 24.1 43.4 56.4 66.9 20.4 41.5 54.9 15.9 39.5 500 22.0 39.6 51.5 61.1 18.7 37.9 50.2 14.5 36.1 400 19.7 35.5 54.7 33.9 44.9 32.3 46.1 16.7 13.0 450 20.9 37.6 48.9 58.0 17.7 36.0 47.6 13.8 34.2 400 19.7 35.5 46.1 54.7 16.7 33.9 44.9 13.0 32.3 350 18.5 33.2 43.1 51.2 15.7 31.8 42.0 12.2 30.2 300 17.1 47.4 14.5 29.4 28.0 30.8 40.0 38.9 11.3 16.4 29.5 38.3 45.4 28.2 275 13.9 37.3 10.9 26.8 250 15.7 28.1 43.3 26.9 25.6 36.5 13.3 35.6 10.4 225 14.9 26.7 34.7 41.1 12.6 25.5 33.8 9.9 24.3 200 14.1 25.2 32.7 38.7 11.9 24.1 31.9 9.3 22.9 22.6 175 13.2 23.6 30.6 36.3 11.2 29.8 8.7 21.5 150 12.2 21.9 28.4 33.6 10.4 20.9 27.6 19.9 8.1 125 11.2 20.0 25.9 30.7 9.5 19.1 25.3 7.4 18.2 100 10.0 17.9 23.2 27.5 8.5 17.1 22.6 6.7 16.3 90 9.5 17.0 22.0 26.1 8.1 16.3 21.5 6.4 15.5 24.6 14.6 80 16.1 20.3 9.0 20.8 7.7 15.4 6.0 70 8.5 15.0 19.5 23.0 7.2 14.4 19.0 5.7 13.7 7.9 14.0 18.0 21.4 6.7 13.3 17.6 12.7 60 5.3 50 7.2 12.8 16.5 19.5 6.1 12.2 16.1 11.6 4.8 40 6.5 11.5 14.8 17.5 11.0 14.4 4.4 10.4 5.5 30 10.0 9.5 9.1 5.7 12.8 15.2 4.8 12.5 3.8 20 4.7 8.2 10.5 12.4 4.0 7.8 10.3 3.2 7.5 4.5 7.8 10.0 11.8 3.8 9.8 3.0 7.1 18 7.5 4.2 7.4 9.4 11.1 7.0 9.2 2.9 6.7 16 3.6 14 4.0 6.9 8.9 10.4 3.4 6.6 8.6 2.7 6.3 12 3.7 6.4 8.2 9.7 3.2 6.1 8.0 2.6 5.9 10 5.9 5.4 3.4 7.5 8.8 2.9 5.6 7.3 2.4

Table B-9 - 7/8" Bolts - Maximum Lamella Length [ℓ] (ft) $C_{\Delta} = 0.5$ (3) Bolts per (4) Bolts per Radius (2) Bolts per Connection Connection Connection [R] (ft) 2 x 10 2 x 12 2 x 12 2 x 14 2 x 14 2 x 14 1000 44.5 64.3 58.9 24.9 79.3 36.1 900 42.2 61.0 75.3 34.2 55.8 23.7 800 39.8 57.6 71.0 32.3 52.7 22.3 700 37.3 53.9 66.4 30.2 49.3 20.9 45.7 600 34.5 49.9 61.5 28.0 19.4 500 31.5 45.6 56.2 25.6 41.7 17.7 400 28.3 40.8 50.3 22.9 37.3 15.9 450 29.9 43.3 53.3 24.3 39.6 16.8 400 28.3 40.8 50.3 22.9 37.3 15.9 350 26.5 38.2 47.1 21.5 35.0 14.9 300 24.5 35.4 43.6 19.9 32.4 13.8 23.5 19.1 13.2 275 33.9 41.8 31.0 250 22.4 39.8 18.2 29.6 12.6 32.3 12.0 225 21.3 30.7 37.8 17.3 28.1 200 20.1 29.0 35.7 16.3 26.5 11.3 18.8 27.1 175 33.4 15.3 24.8 10.6 150 17.4 25.1 30.9 14.2 23.0 9.9 125 15.9 23.0 28.3 13.0 21.0 9.0 100 14.3 20.6 25.3 11.6 18.8 8.1 90 13.6 19.5 24.0 11.1 17.9 7.7 80 12.8 18.4 22.7 10.5 16.9 7.3 70 12.0 17.3 21.2 9.8 15.8 6.9 11.2 16.0 19.7 9.1 14.7 6.4 60 50 10.2 18.0 8.3 13.4 5.8 14.6 40 9.2 13.1 16.1 7.5 12.0 5.3 30 8.0 11.4 14.0 10.5 4.6 6.5 20 6.6 9.4 11.5 5.4 8.6 3.8 6.3 8.9 10.9 5.1 8.2 3.7 18 16 5.9 8.4 10.3 4.9 7.7 3.5 14 5.6 7.9 9.6 4.6 7.2 3.3 12 5.2 7.3 8.9 4.3 6.7 3.0 10 4.8 6.7 8.2 3.9 6.2 2.8

Tabl	e B-10 - 1" B	Bolts - Maxim	um Lamella	Length [l] (ft)
		$C_{\Delta} = 0$).5	
Radius	(2) Rol	ts per Conr	ection	(3) Bolts per
[R] (ft)	(2) 501	to per com	CCCIOII	Connection
[וי] (ויי)	2 x 10	2 x 12	2 x 14	2 x 14
1000	28.7	54.7	71.8	43.7
900	27.3	51.9	68.1	41.5
800	25.8	49.0	64.2	39.1
700	24.1	45.8	60.1	36.6
600	22.4	42.5	55.7	33.9
500	20.4	38.8	50.9	31.0
400	18.3	34.7	45.5	27.8
450	19.4	36.8	48.3	29.4
400	18.3	34.7	45.5	27.8
350	17.2	32.5	42.6	26.0
300	15.9	30.1	39.5	24.1
275	15.3	28.9	37.8	23.1
250	14.6	27.5	36.1	22.0
225	13.8	26.1	34.2	20.9
200	13.1	24.7	32.3	19.8
175	12.2	23.1	30.2	18.5
150	11.4	21.4	28.0	17.2
125	10.4	19.6	25.6	15.7
100	9.3	17.5	22.9	14.1
90	8.9	16.7	21.8	13.4
80	8.4	15.7	20.5	12.6
70	7.9	14.7	19.2	11.8
60	7.3	13.7	17.8	11.0
50	6.7	12.5	16.3	10.1
40	6.0	11.2	14.6	9.0
30	5.3	9.8	12.7	7.9
20	4.4	8.0	10.4	6.5
18	4.2	7.6	9.9	6.2
16	4.0	7.2	9.3	5.8
14	3.7	6.8	8.8	5.5
12	3.5	6.3	8.1	5.1
10	3.2	5.8	7.4	4.7

Appendix C: ASCE 7-10 Tables and Figures

Appendix C features copies of Table 7-2 and Figure 7-3 (complete and author-simplified versions) from the ASCE standard 7-10.

Table 7-2 Exposure Factor, C_e

	Exp	Exposure of Roof"	
Terrain Category	Fully Exposed	Fully Exposed Partially Exposed Sheltered	Sheltered
B (see Section 26.7)	6.0	1.0	1.2
C (see Section 26.7)	6.0	1.0	1.1
D (see Section 26.7)	8.0	6.0	1.0
Above the treeline in windswept mountainous areas.	0.7	0.8	N/A
In Alaska, in areas where trees do not exist within a 2-mile (3-km) radius of the site.	0.7	0.8	N/A

The terrain category and roof exposure condition chosen shall be representative of the anticipated conditions during the life of the structure. An Definitions: Partially Exposed: All roofs except as indicated in the following text. Fully Exposed: Roofs exposed on all sides with no shelter exposure factor shall be determined for each roof of a structure.

Obstructions within a distance of $10h_o$ provide "shelter," where h_o is the height of the obstruction above the roof level. If the only obstructions are a few deciduous trees that are leafless in winter, the "fully exposed" category shall be used. Note that these are heights above the roof. Heights used to establish the Exposure Category in Section 26.7 are heights above the ground. as obstructions.

height of the balanced snow load (h_b), or other obstructions are not in this category. Sheltered: Roofs located tight in among conifers that qualify

afforded by terrain, higher structures, or trees. Roofs that contain several large pieces of mechanical equipment, parapets that extend above the

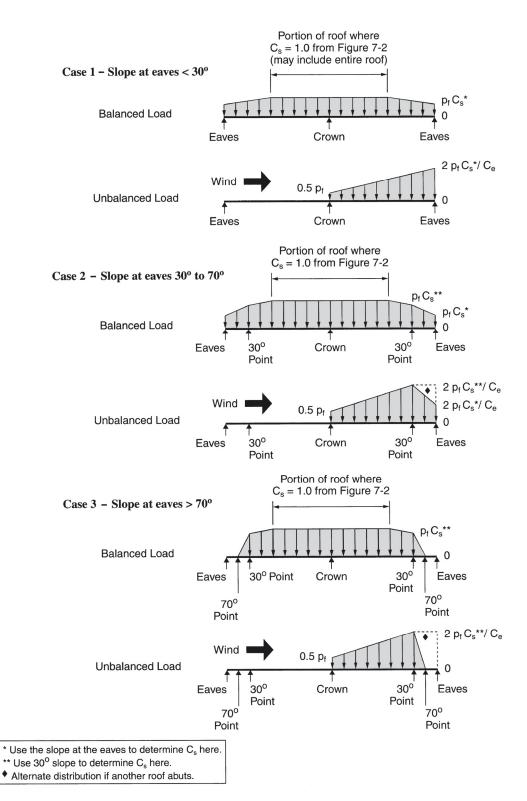
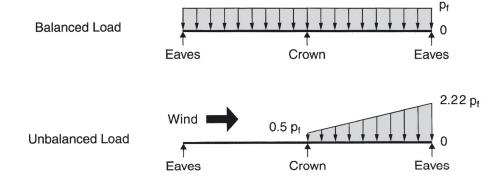
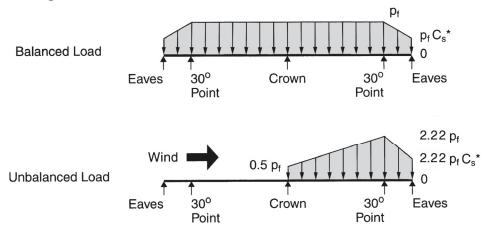


FIGURE 7-3 Balanced and Unbalanced Loads for Curved Roofs

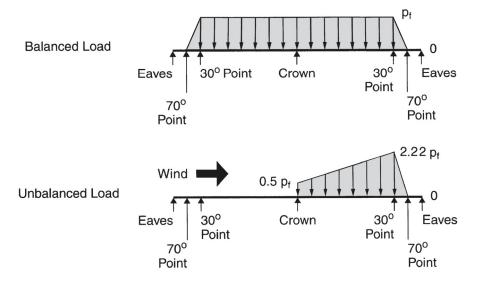
Case 1 – Slope at eaves $< 30^{\circ}$



Case 2 - Slope at eaves 30° to 70°



Case 3 - Slope at eaves $> 70^{\circ}$



SIMPLIFIED FIGURE 7-3 Balanced and Unbalanced Loads for Curved Roofs

Appendix D: Load versus Curvature Graphs

Appendix D displays graphs showing the theoretical behavior of a lamella arch with two different stiffness characteristics. Results are based on a finite element analysis conducted by the author.

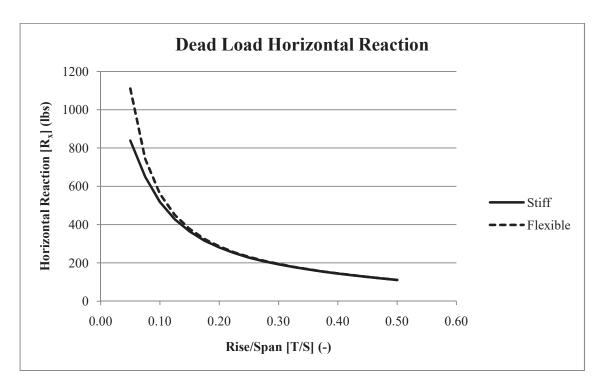


Figure D-1 – Dead Load Horizontal Reaction.

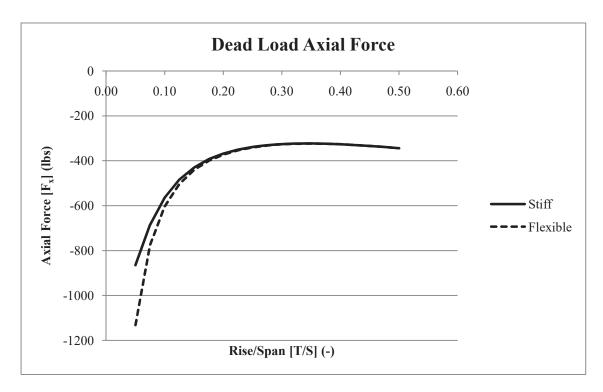


Figure D-2 - Dead Load Axial Force.

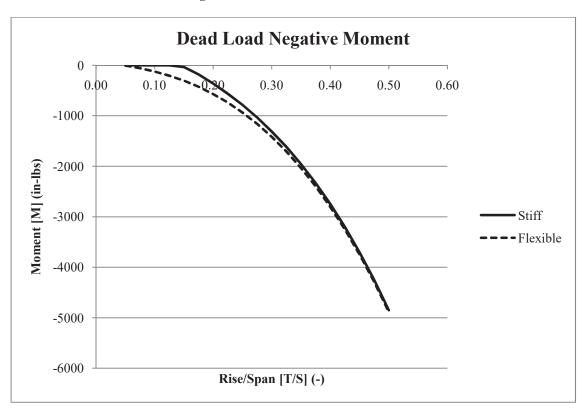


Figure D-3 – Dead Load Negative Moment.

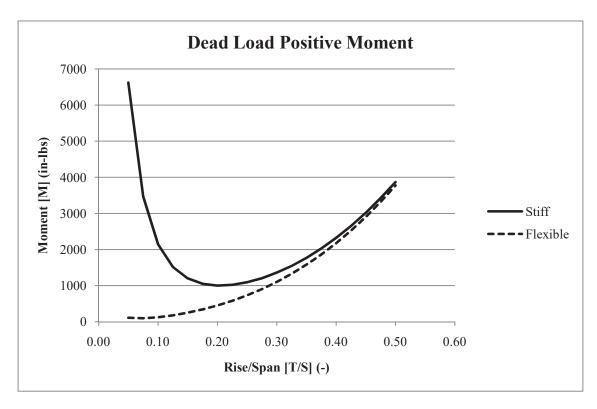


Figure D-4 – Dead Load Negative Moment.

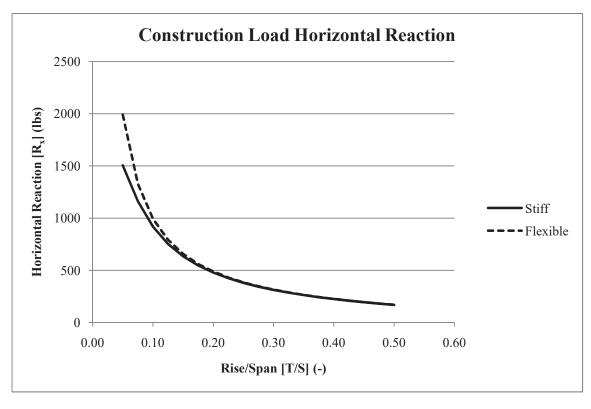


Figure D-5 – Dead Load Negative Moment.

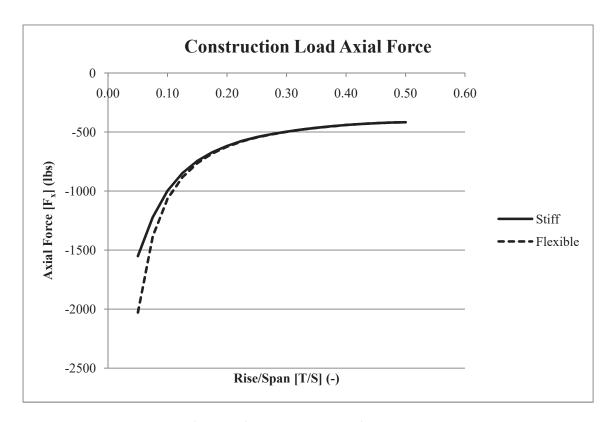


Figure D-6 – Dead Load Negative Moment.

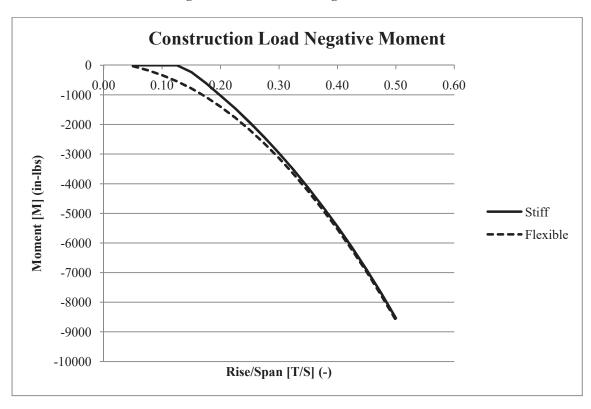


Figure D-7 – Dead Load Negative Moment.

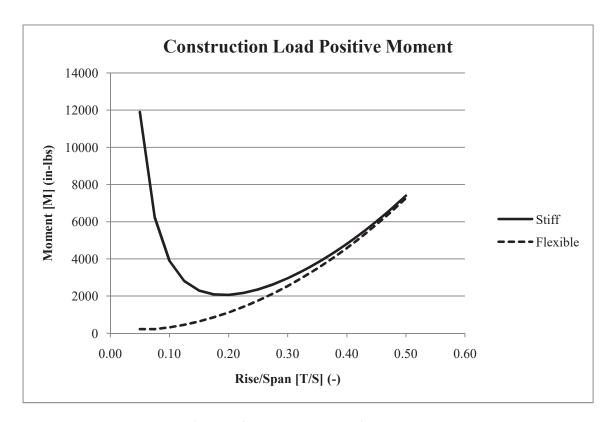


Figure D-8 – Dead Load Negative Moment.

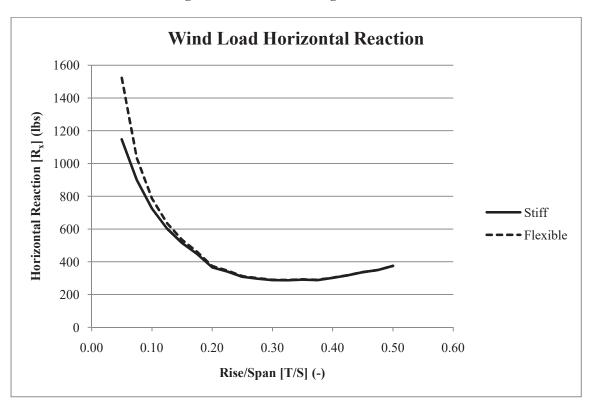


Figure D-9 – Dead Load Negative Moment.

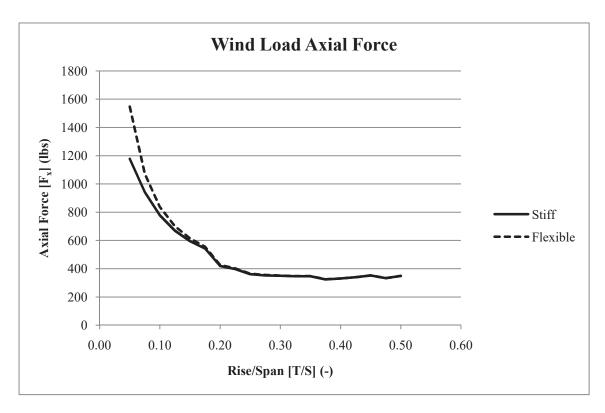


Figure D-10 - Dead Load Negative Moment.

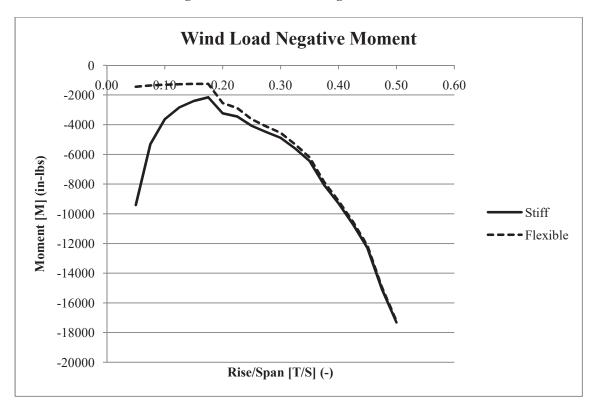


Figure D-11 – Dead Load Negative Moment.

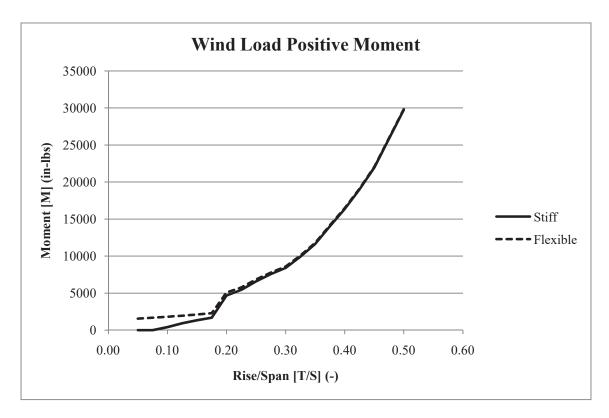


Figure D-12 – Dead Load Negative Moment.

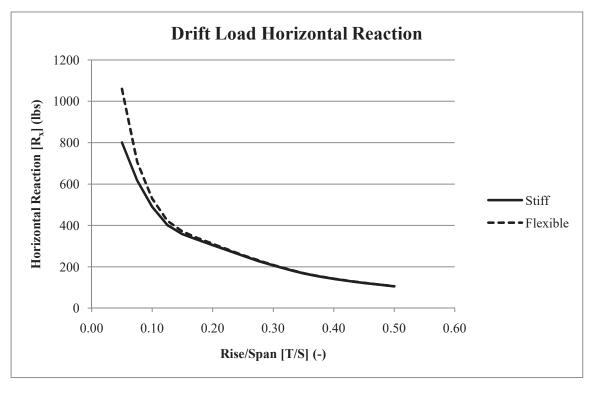


Figure D-13 – Dead Load Negative Moment.

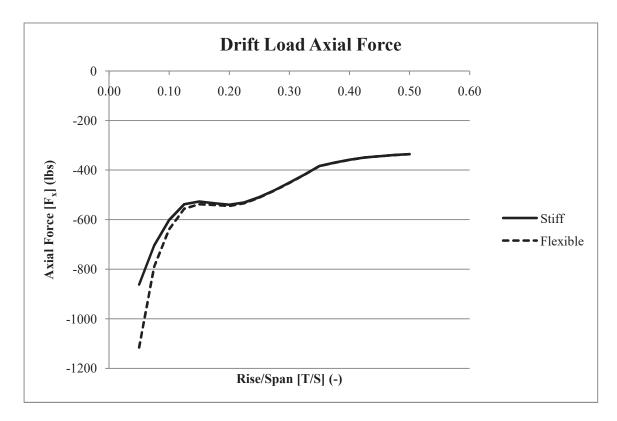


Figure D-14 - Dead Load Negative Moment.

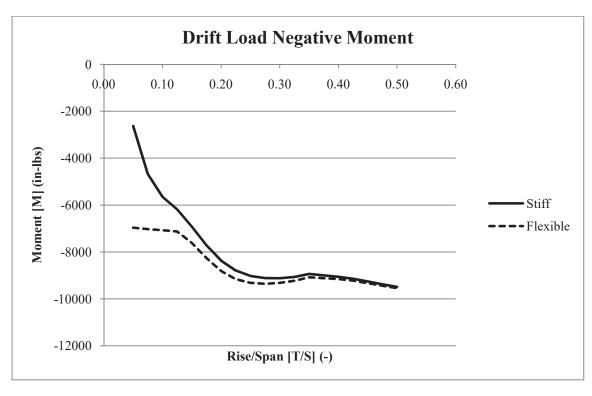


Figure D-15 – Dead Load Negative Moment.

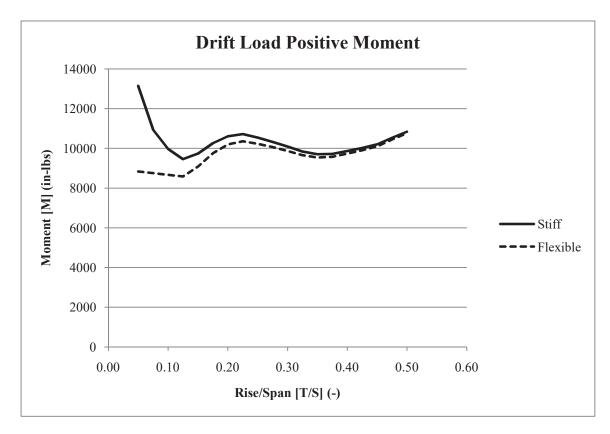


Figure D-16 – Dead Load Negative Moment.

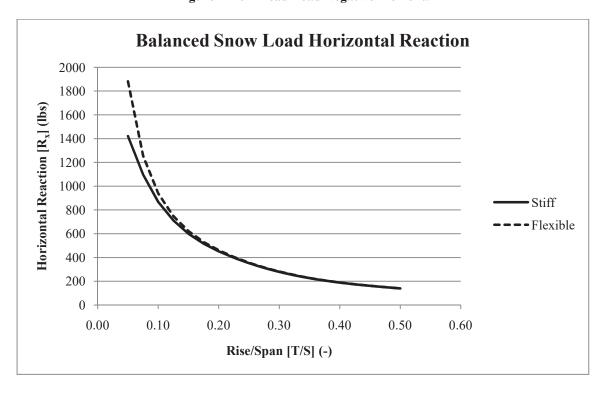


Figure D-17 – Dead Load Negative Moment.

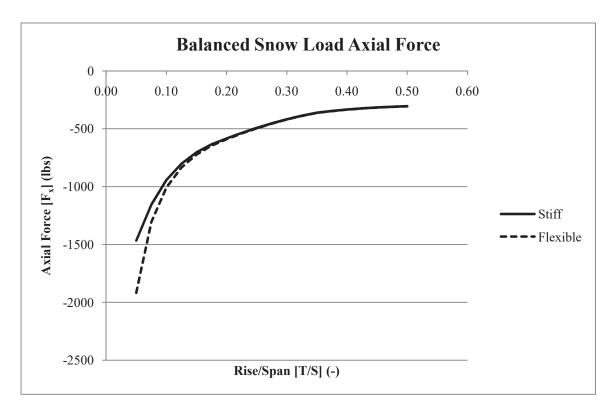


Figure D-18 – Dead Load Negative Moment.

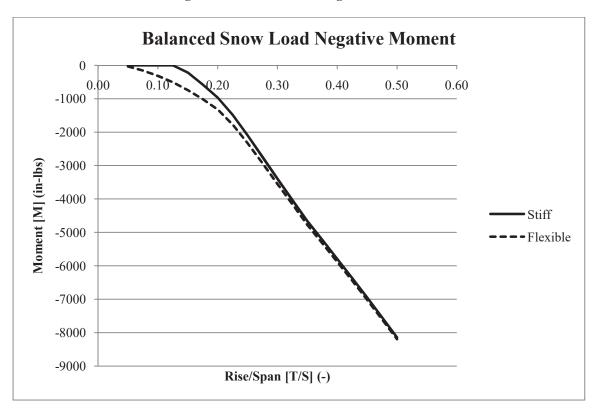


Figure D-19 – Dead Load Negative Moment.

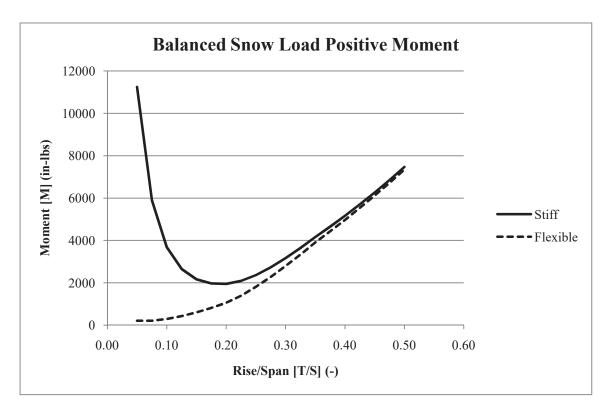


Figure D-20 – Dead Load Negative Moment.

Appendix E: Arched Roof Load Tables

Appendix E contains load tables developed by the author for use in the preliminary design of a lamella roof.

	_				
 | _ | | |
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 | | | |
 | _ | | _ | | |
 | | _ |
|---|---|---|--|--|--
---	---	----------	----------
---	---	---	
--	---	---------------------------------------	---
---	--	---	
--	---	--	
			₊⊠
 | 2630 | 2455 | 2305 | 2230
 | 2305 | 2460 | 2705 |
 | | ₊ω | ql-ui | 2700
 | 4150 | 4045 | 3825 | 3570 | 3430 | 3202
 | 3720 | 3995 |
| | | | Z | in-lb | -485 | -630
 | -1205 | -1550 | -1745 | -1885
 | -2080 | -2315 | -2595 |
 | | M | ql-ui | -485
 | -1220 | -2050 | -2520 | -2755 | -2880 | -3110
 | -3365 | -3655 |
| | | 25 psf | Fa | lbs | 510 | 425
 | 370 | 335 | 315 | 305
 | 295 | 290 | 290 |
 | 40 psf | Fa | sql | 969
 | 575 | 200 | 455 | 415 | 370 | 355
 | 345 | 345 |
| $K_{\rm zt}=1.0$ $K_{\rm d}=0.85$ | | | χ, | lbs | 450 | 350
 | 280 | 230 | 190 | 165
 | 145 | 150 | 160 |
 | | ۳, | sql | 615
 | 475 | 375 | 305 | 250 | 202 | 175
 | 155 | 160 |
| I | | | Ry | lbs | 245 | 245
 | 250 | 255 | 260 | 265
 | 275 | 280 | 290 |
 | | Ry | lbs | 335
 | 335 | 340 | 330 | 315 | 300 | 300
 | 305 | 310 |
| $C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$ | | | ¥ | in-lb | 3665 | 2330
 | 2165 | 2010 | 1880 | 1830
 | 1910 | 2022 | 2700 |
 | | ₹ | in-lb | 4680
 | 3690 | 3570 | 3370 | 3145 | 3030 | 3100
 | 3300 | 3565 |
| = 1.0.
only. | [gd] þ | | M | in-lb | -485 | -510
 | -930 | -1235 | -1415 | -1550
 | -1745 | -1985 | -2525 | d [p _g]
 | | M | in-lb | -485
 | -1020 | -1765 | -2195 | -2415 | -2550 | -2765
 | -3010 | -3295 |
| malized to C _D
gn procedure | Snow Load | 20 psf | Fa | lbs | 510 | 425
 | 370 | 335 | 315 | 305
 | 295 | 290 | 290 | Snow Load
 | 35 psf | L | lps | 630
 | 525 | 455 | 415 | 380 | 340 | 330
 | 320 | 315 |
| All values noi
Use ASD desi | Ground | | χ, | lbs | 450 | 350
 | 280 | 230 | 190 | 165
 | 145 | 150 | 160 | Ground
 | | R _x | lbs | 260
 | 435 | 345 | 280 | 230 | 190 | 160
 | 150 | 160 |
| œ
I | | | R _v | lbs | 245 | 245
 | 250 | 255 | 260 | 265
 | 275 | 280 | 290 |
 | | R | lbs | 305
 | 305 | 310 | 300 | 290 | 275 | 280
 | 285 | 290 |
| | | | ₹ | in-lb | 3665 | 2025
 | 1700 | 1560 | 1460 | 1430
 | 1595 | 1975 | 2700 |
 | | ± | ql-ui | 4160
 | 3235 | 3100 | 2910 | 2725 | 2630 | 2700
 | 2880 | 3135 |
| | | | ω | in-lb | -485 | -400
 | -655 | -915 | -1085 | -1230
 | -1500 | -1985 | -2525 |
 | | M | in-lb | -485
 | -825 | -1485 | -1870 | -2080 | -2215 | -2420
 | -2655 | -2945 |
| | | 15 psf | Fa | lbs | 510 | 425
 | 370 | 335 | 315 | 305
 | 295 | 290 | 290 |
 | 30 psf | Fa | lps | 265
 | 475 | 410 | 375 | 345 | 310 | 300
 | 295 | 290 |
| ă, ă, | | | R _x | lbs | 450 | 350
 | 280 | 230 | 190 | 165
 | 145 | 150 | 160 |
 | | R _x | lbs | 202
 | 390 | 310 | 250 | 202 | 170 | 145
 | 150 | 160 |
| | | | A, | lbs | 245 | 245
 | 250 | 255 | 260 | 265
 | 275 | 280 | 290 |
 | | A, | sql | 275
 | 275 | 280 | 275 | 265 | 265 | 275
 | 280 | 290 |
| l Load
truction Loa
ind Zone | | Radius [R] | (£ | | 26.00 | 18.17
 | 14.50 | 12.50 | 11.33 | 10.64
 | 10.25 | 10.06 | 10.00 |
 | Radius [R] | (£ | | 26.00
 | 18.17 | 14.50 | 12.50 | 11.33 | 10.64 | 10.25
 | 10.06 | 10.00 |
| 10 psf Deac
20 psf Cons
120 mph W | | Rise [T] | (#) | | 2 | 3
 | 4 | 2 | 9 | 7
 | ∞ | 6 | 10 |
 | Rise [T] | (£) | | 2
 | n | 4 | 2 | 9 | 7 | ∞
 | 6 | 10 |
| | All values normalized to $C_0=1.0$. $C_e=0.9$ Use ASD design procedure only. $C_q=1.2$ | Load $ R_v = \frac{R}{R_v} $ All values normalized to $C_0 = 1.0$. $C_e = 0.9$ Use ASD design procedure only. $C_t = 1.2$ $C_t = $ | $ \begin{array}{c ccccccccccccccccccccccccccccccccccc$ | Control Cont | Load R ₁ = 1.0 R ₂ = 0.9 R ₃ = 0.09 R ₃ = 0.00 R ₃ = 0. | Load R _{v=1.0} R _v | Load R. The late of the late | Load R, | Load R, | Load R _i R _i | Load R _i R _i | A | Load R, — — — — — — — — — — — — — — — — — — — | All Values normalized to C _j = 1.0. C _j = 0.9 K _j = 0.85 R _j R _k R | Second Single Paris Par | Park Park | California Paris Paris | Alivablues normalized to C _j = 1.0 C _j = 0.9 C _j = 1.2 C _j = 0.8 R _j R _j | A A A A A A A A A A | Carlot R, R, R, R, R, R, R, R | Alivables normalized to C ₁ = 1.0 Alivables normalized to C ₂ = 1.0 C ₄ = | Ali Vollec in Carlotte Carlot | Part Part | All values normalized to C ₃ = 1.0 C ₄ = 0.9 C ₄ = 0.8 C ₄ | Charlest Charlest | California Part P |

					±	in-lb	8320	6895	6885	6565	6140	5850	5955	6230	6615			±	ql-ui	11455	9645	9725	9310	8715	8290	8410	8745	9265
					Œ	in-lb	-815	-2430	-3745	-4460	-4785	-4915	-5175	-5490	-5850			Œ	in-lb	-1435	-3640	-5440	-6405	-6820	-6955	-7270	-7625	-8050
				70 psf	ъ.	lps	1070	068	770	695	630	555	525	510	200		100 psf	ъ.	lps	1450	1205	1040	935	845	740	695	029	099
	Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$			٣×	lps	955	735	280	465	380	315	265	230	200			٣×	lps	1290	995	785	630	510	420	355	305	265
	<u> </u>				R,	lbs	515	515	520	200	470	440	435	435	440			R,	lps	969	695	200	029	625	575	292	292	570
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			±	in-lb	7280	2980	5940	5655	5285	5035	5140	5395	5730			±	in-lb	10410	8730	8780	8395	7855	7475	7590	7910	8380
		= 1.0. only.	[gd] k		Σ	in-lb	069-	-2025	-3180	-3815	-4110	-4235	-4480	-4780	-5120	[gd] k		Z	in-lb	-1225	-3230	-4875	-5760	-6140	-6275	-6575	-6905	-7315
		All values normalized to C_0 = 1.0. Use ASD design procedure only.	Ground Snow Load [pg]	e0 psf	F _a	lbs	945	785	089	615	260	495	470	455	445	Ground Snow Load [pg]	90 psf	L _e	sql	1325	1100	950	855	770	675	640	615	605
	Notes:	All values noi Use ASD desi	Ground		ď	lbs	840	650	515	415	335	275	235	205	180	Ground		ď	lbs	1180	910	715	575	465	385	325	280	245
		ă, N			R _v	lbs	455	455	460	445	415	390	390	390	400			Υ _ν	lbs	635	635	640	610	570	530	525	525	530
	//	T C			₹	in-lb	6240	2905	4990	4740	4425	4230	4320	4555	4850			₹	in-lb	9360	7810	7830	7480	7000	0999	6775	7070	7495
					Ē	in-lb	-575	-1625	-2615	-3165	-3430	-3555	-3795	-4075	-4385			M	in-lb	-1020	-2830	-4310	-5110	-5465	-5595	-5875	-6200	-6585
	//			50 psf	F _a	lps	820	089	290	535	485	435	410	400	395		80 psf	F _e	lps	1200	995	860	775	700	615	280	260	555
v Load.		ă, x,			٣×	lbs	730	292	445	360	295	240	205	180	160			٣×	lps	1065	820	650	520	425	350	295	255	220
ariable Snov					S,	sql	395	395	400	385	365	345	345	350	355			Α,	sql	575	575	280	555	520	485	480	480	485
Table E-1 - 20 ft Span, Variable Snow Load.		10 psr Dead Load 20 psf Construction Load 120 mph Wind Zone		Radius [R]	(£		26.00	18.17	14.50	12.50	11.33	10.64	10.25	10.06	10.00		Radius [R]	(£		26.00	18.17	14.50	12.50	11.33	10.64	10.25	10.06	10.00
Table E-1	;	10 psr Dead Load 20 psf Construction 120 mph Wind Zone		Rise [T]	(L)		2	3	4	2	9	7	∞	6	10		Rise [T]	(#)		2	3	4	2	9	7	∞	6	10

																						1		1												
					±	in-lb	2962	5035	2909	5115	5035	4870	4735	4665	4740	4915	5180	5205	6085			±	in-lb	8880	7675	7880	8065	2960	7695	7470	7345	7370	7545	7890	8310	8765
					ĮW	in-lb	-1230	-2000	-2815	-3425	-3835	-4085	-4250	-4400	-4585	-4850	-5140	-5525	-5925			M	ql-ui	-2415	-3525	-4685	-5565	-6135	-6460	-6650	-6765	-6915	-7210	-7545	-7945	-8355
				25 psf	Fa	lbs	865	720	630	292	525	495	475	460	450	445	440	435	430		40 psf	Fa	lbs	1175	086	855	770	705	999	625	280	550	535	520	515	515
	Wind Load	$K_{zt}=1.0 \label{eq:Kzt}$ $K_d=0.85 \label{eq:Kzt}$			R _x	lbs	282	625	515	435	375	330	290	260	235	215	220	235	240			×	sql	1075	850	700	290	202	435	380	335	295	265	240	235	240
	힏				R,	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430			æ^	lbs	200	200	202	510	200	485	470	455	450	450	455	460	465
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			± ✓	in-lb	4995	4155	4125	4135	4060	3930	3830	3800	3865	4050	4280	2060	6085			[‡] ∑	in-lb	7910	6795	6940	7080	6985	6755	6550	6450	6495	999	6985	7370	7815
		= 1.0. only.	! [pg]		M	in-lb	-1010	-1500	-2190	-2715	-3070	-3295	-3470	-3625	-3810	-4065	-4375	-4955	-5765	! [p _g]		Ň	ql-ui	-2000	-3015	-4060	-4850	-5370	-5665	-5850	-5970	-6140	-6425	-6745	-7130	-7525
		All values normalized to C_D = 1.0. Use ASD design procedure only.	Ground Snow Load [pg]	20 psf	F _a	lbs	865	720	630	292	525	495	475	460	450	445	440	435	430	Ground Snow Load [pg]	35 psf	T.	sql	1070	890	775	700	640	610	570	530	202	490	480	480	475
	Notes:	All values nor Use ASD desi,	Ground		٣×	lbs	785	625	515	435	375	330	290	260	235	215	220	235	240	Ground		æ	lbs	975	775	635	535	460	400	345	305	270	245	220	235	240
		č I			R _v	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430			ď	lps	455	455	460	465	455	445	430	420	415	415	420	425	435
	//	, i			τ	in-lb	4305	3275	3190	3155	3085	2990	2930	2935	3010	3345	4055	2060	6085			±	ql-ui	6935	5915	6005	6100	6010	5810	5635	2260	5615	5785	6085	6425	6870
					M	in-lb	-785	-1010	-1570	-2000	-2305	-2520	-2690	-2850	-3045	-3515	-4205	-4955	-5765			M	in-lb	-1590	-2505	-3435	-4140	-4600	-4875	-5050	-5175	-5365	-5635	-5945	-6310	-6725
	//			15 psf	Fa	lbs	865	720	630	265	525	495	475	460	450	445	440	435	430		30 psf	T.	lps	096	802	700	630	280	550	515	485	460	450	440	440	435
v Load.		ă, x,			æ,	lbs	785	625	515	435	375	330	290	260	235	215	220	235	240			×	sql	875	695	575	485	415	360	315	275	245	220	220	235	240
ariable Snov		p			R,	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430			~	sql	410	410	415	420	415	405	395	395	400	410	415	425	430
Table E-2 - 30 ft Span, Variable Snow Load.	3	ı Load truction Loa ind Zone		Radius [R]	£)		39.00	30.13	25.00	21.75	19.57	18.06	17.00	16.25	15.73	15.38	15.15	15.04	15.00		Radius [R]	(L		39.00	30.13	25.00	21.75	19.57	18.06	17.00	16.25	15.73	15.38	15.15	15.04	15.00
Table E-2		10 psr Dead Load 20 psf Construction Load 120 mph Wind Zone		Rise [T]	(£)		3	4	2	9	7	8	6	10	11	12	13	14	15		Rise [T]	(£)		3	4	5	9	7	∞	6	10	11	12	13	14	15

				1	1		<u> </u>														l	1														
					₹	in-lb	14710	12980	13520	13955	13825	13415	12995	12700	12625	12910	13300	13965	14570			₹	dl-ni	20545	18305	19185	19845	19715	19135	18520	18060	17880	18275	18710	19625	30105
					Έ	in-lb	-4915	-6590	-8440	-9875	-10765	-11205	-11440	-11540	-11680	-12000	-12355	-12865	-13360			Έ	d-ni	-7435	-9655	-12220	-14195	-15410	-15995	-16230	-16315	-16450	-16830	-17255	-17800	-1926F
				70 psf	. .	sql	1820	1515	1320	1185	1080	1020	945	875	815	785	292	755	750		100 psf	T	sql	2465	2050	1780	1600	1455	1370	1270	1165	1085	1040	1015	1000	005
	Wind Load	$K_{tt} = 1.0$ $K_d = 0.85$			A,	sql	1660	1315	1080	910	775	029	280	202	450	400	365	330	300			R _×	lbs	2250	1780	1460	1230	1045	006	780	089	009	532	485	440	700
	ad				ď	lbs	770	770	775	780	760	735	700	029	655	650	650	655	660			R _v	lbs	1040	1045	1045	1050	1025	980	935	885	855	850	845	820	25.5
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			₽	dl-ni	12765	11210	11635	11990	11860	11505	11155	10915	10870	11125	11495	12080	12615			₹	ql-ui	18595	16530	17295	17880	17755	17230	16675	16275	16125	16485	16910	17740	18480
		= 1.0. only.	J [b,]	la i	Έ	d-ni	-4080	-5570	-7185	-8435	-9215	-9625	-9840	-9950	-10090	-10385	-10750	-11225	-11690	1 [pg]		Έ	in-lb	-6590	-8635	-10960	-12755	-13860	-14395	-14635	-14720	-14860	-15220	-15620	-16140	-16695
		All values normalized to C _D = 1.0. Use ASD design procedure only.	Ground Snow Load [p _e]	eo psł	. т.	sql	1605	1340	1165	1045	955	006	840	775	725	705	685	675	670	Ground Snow Load [pg]	90 psf	L.	lbs	2250	1875	1625	1460	1330	1250	1160	1065	995	955	930	920	910
	Notes:	All values nor Use ASD desi	Ground		S,	lbs	1465	1160	955	802	685	290	515	450	400	355	325	295	270	Ground		Α.	lbs	2055	1625	1335	1120	955	820	710	620	550	490	445	405	370
		e. K			S.	lbs	089	089	685	069	675	650	625	009	585	585	585	290	595			R _v	lbs	950	955	955	096	935	006	855	815	790	785	780	785	790
	//	, a			₹	d-ni	10825	9440	9755	10025	9910	0096	9310	9130	9120	9335	0696	10195	10660			±	in-lb	16655	14755	15410	15920	15790	15320	14835	14490	14375	14700	15105	15850	16525
		S			Σ	dl-ni	-3250	-4545	-5935	-6995	-7670	-8040	-8245	-8355	-8500	-8790	-9150	-9585	-10025			Ē	ql-ui	-5750	-7610	-9700	-11315	-12315	-12800	-13035	-13130	-13270	-13610	-13985	-14500	-15025
	//			50 psf	. т.	sql	1390	1160	1010	910	830	785	730	089	640	620	902	595	590		80 psf	r _e	lbs	2035	1695	1470	1325	1205	1135	1055	970	902	870	820	835	830
v Load.		, a			Α,	lbs	1270	1005	825	695	595	515	445	390	345	310	280	255	240			Α.	lbs	1855	1470	1205	1015	865	745	645	292	200	445	405	365	335
ariable Snov					ď	sql	290	290	262	009	585	570	550	530	515	515	520	525	530			S,	sql	098	860	865	870	820	815	780	745	720	715	715	720	725
Table E-2 - 30 ft Span, Variable Snow Load	3	l Load truction Loa ind Zone		Radius [R]	£		39.00	30.13	25.00	21.75	19.57	18.06	17.00	16.25	15.73	15.38	15.15	15.04	15.00		Radius [R]	Œ		39.00	30.13	25.00	21.75	19.57	18.06	17.00	16.25	15.73	15.38	15.15	15.04	15.00
Table E-2 - 3	9	10 psr Dead Load 20 psf Construction Load 120 mph Wind Zone		Rise [T]			3	4	2	9	7	∞	6	10	11	12	13	14	15		Rise [T]	(L)		3	4	2	9	7	∞	6	10	11	12	13	14	15

able E-3 - 4	10 ft Span, V	able E-3 - 40 ff Span, Variable Snow Load.	v Load.													
						//		Notes:			Snow Load	ad	Wind Load			
L0 pst Dead Load 20 psf Construction L20 mph Wind Zone	L0 psr Dead Load 20 psf Construction Load L20 mph Wind Zone	aq	X.		~ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \		ά. 	All values no Use ASD desi	All values normalized to C _D = 1.0. Use ASD design procedure only.	= 1.0. only.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{zt} = 1.0$ $K_d = 0.85$			
			χ			χ.										
	•							Groun	Ground Snow Load [pg]	ا [ام] ال						
Г	Radius [R]			15 psf					20 psf					25 psf		
(L	£	Ry	R _x	Fa	M	₊ W	Ry	R	^e 4	M	₊ W	R	R _x	Fa	M	±
		lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
4	52.00	485	1110	1205	-1725	2860	485	1110	1205	-2565	7430	485	1110	1205	-3440	0006
2	42.50	490	915	1030	-2305	5180	490	915	1030	-3285	0299	490	915	1030	-4265	8170
9	36.33	490	770	910	-2905	5055	490	770	910	-3995	0629	490	770	910	-5085	8130
7	32.07	495	029	825	-3465	5140	495	029	825	-4680	6765	495	029	825	-5895	8390
∞ 0	29.00	500	585	760	-3930	5200	500	585	760	-5250	0689	500	585	760	-6565	8580
10	25.00	510	525 470	680	-4280	5175	510	470	680	-5955	0990	510	470	680	-7380	8430
11	23.68	515	430	655	-4745	5010	515	430	655	-6170	6645	515	430	655	-7610	8275
12	22.67	520	390	635	-4945	4985	520	390	635	-6345	0959	520	390	635	-7780	8130
13	21.88	525	360	620	-5130	2000	525	360	620	-6530	6525	525	360	620	-7930	8095
14	21.29	530	330	909	-5310	5070	530	330	909	-6695	0629	530	330	909	-8080	8105
15	20.83	535	310	262	-5620	5280	535	310	595	-6995	6755	535	310	595	-8395	8305
16	20.50	545	285	290	-6340	6045	545	285	290	-7310	7050	545	285	290	-8725	8575
17	20.26	550	285	582	-7235	7090	550	285	585	-7705	7370	550	285	585	-9095	8940
18	20.11	260	290	280	-8195	8280	260	290	280	-8195	8280	260	290	580	-9540	9345
19	20.03	292	310	575	-9215	9955	292	310	575	-9215	9955	265	310	575	-10050	9955
20	20.00	575	315	575	-10300	11560	575	315	575	-10300	11560	575	315	575	-10590	11560
								Groun	Ground Snow Load [pg]	ا [اه] ال						
Rise [T]	Radius [R]			30 psf					35 psf					40 psf		
(±)	(£)	Ry	R _x	Е	M	ţΜ	Ry	R _x	Е	M	·Μ	R	R _x	Fa	M	[‡] ⊠
		lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
4	52.00	545	1240	1345	-4340	10585	909	1375	1495	-5235	12175	999	1515	1645	-6130	13765
2	42.50	545	1020	1150	-5240	9675	909	1130	1275	-6220	11185	999	1245	1405	-7200	12690
9	36.33	550	860	1015	-6180	9685	610	955	1125	-7270	11235	029	1050	1240	-8375	12785
7	32.07	550	745	920	-7115	10010	610	825	1020	-8345	11635	670	910	1120	-9575	13270
× c	29.00	555	655	850	-/885	102/0	615	725	940	-9220	11965	6/5	800	1035	-10555	13655
10	25.00	545	520	755	-8800	10110	009	575	835	-9023	11795	655	017	960	-11645	13475
11	23.68	535	465	720	-9050	9910	230	515	800	-10490	11545	640	565	875	-11925	13175
12	22.67	525	420	069	-9220	9710	575	465	092	-10665	11315	625	510	835	-12105	12925
13	21.88	525	380	655	-9355	9665	292	420	725	-10790	11235	610	460	790	-12230	12805
14	21.29	530	350	620	-9465	9625	550	385	685	-10860	11140	009	420	745	-12280	12660
15	20.83	535	320	610	-9790	9855	550	355	999	-11190	11405	292	385	725	-12590	12950
16	20.50	545	295	292	-10135	10095	555	325	655	-11550	11630	009	355	710	-12960	13205
17	20.26	220	285	290	-10520	10510	260	305	645	-11945	12085	009	330	700	-13375	13655
18	20.11	260	290	285	-10970	10970	292	290	640	-12420	12600	610	310	069	-13865	14225
19	20.03	265	310	280	-11460	11450	570	310	635	-12920	13110	615	310	685	-14390	14790
20	20.00	575	315	280	-12020	12090	580	315	630	-13445	13765	620	315	685	-14935	15445

Table E-3 -	Fable E-3 - 40 ft Span, Variable Snow Load.	Variable Sno	w Load.													
9	7					//		Notes:			Snow Load	ad	Wind Load			
LO psr Dead Load 20 psf Construction 120 mph Wind Zone	LD psi Dead Load 20 psf Construction Load 120 mph Wind Zone	ad	, X				× I	All values no Use ASD desi	All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.	= 1.0. only.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{zt} = 1.0$ $K_d = 0.85$			
			٠œٌ			·α̈́										
								Groun	Ground Snow Load [pg]	[g] k						
Rise [T]	Radius [R]			50 psf					e0 psf					70 psf		
(#)	(#)	R _v	R _x	Е	ĮМ	ψ	R_{y}	R _x	Fa	M	Σ	Ry	R _x	F	M	± ∑
		sql	lbs	sql	in-lb	in-lb	lbs	lbs	lps	in-lb	dl-ui	lbs	lbs	lbs	in-lb	in-lb
4	52.00	785	1790	1945	-7925	16945	506	2070	2250	-9720	20125	1025	2345	2550	-11515	23305
2	42.50	785	1470	1660	-9155	15705	902	1700	1915	-11130	18720	1025	1925	2170	-13100	21730
9	36.33	790	1245	1460	-10580	15890	910	1435	1685	-12785	18995	1030	1625	1910	-14990	22095
r «	32.07	795	1075	1325	-12030	16545	915	1240	1525	-14490	19815	1035	1405	1730	-16950	23090
6	26.72	790	835	1135	-14030	17085	902	096	1305	-16835	20500	1025	1090	1475	-19640	23920
10	25.00	770	745	1080	-14510	16835	885	855	1240	-17385	20200	995	970	1405	-20265	23560
11	23.68	750	999	1030	-14805	16445	860	765	1185	-17685	19710	965	865	1335	-20590	22975
12	22.67	730	009	975	-14990	16140	830	069	1120	-17875	19360	935	775	1265	-20760	22575
13	21.88	710	540	925	-15105	15945	802	620	1055	-17975	19080	902	700	1190	-20850	22220
14	21.29	069	490	865	-15120	15715	780	265	066	-17960	18805	875	635	1110	-20800	21900
15	20.83	685	450	845	-15445	16050	780	520	096	-18315	19145	870	585	1080	-21180	22240
16	20.50	685	415	825	-15785	16355	775	475	935	-18665	19500	865	540	1050	-21555	22645
17	20.26	069	385	810	-16230	16800	780	440	920	-19080	19945	865	495	1030	-21980	23090
18	20.11	695	360	800	-16765	17480	780	410	902	-19660	20730	870	460	1015	-22555	23985
19	20.03	200	335	790	-17330	18150	790	385	006	-20270	21510	875	430	1005	-23210	24865
20	20.00	710	315	790	-17915	18795	795	355	895	-20890	22250	880	400	1000	-23870	25705
								Groun	Ground Snow Load [pg]	[gd] k						
Rise [T]	Radius [R]			80 psf					90 psf					100 psf		
(£)	(£	æ [≻]	۳×	ъ.	·W	₽	R,	٣×	L _e	Œ	₹	R,	٣×	F _a	M	₹
		sql	lbs	sql	in-lb	ql-ui	lbs	lbs	lps	ql-ui	ql-ui	lbs	lps	lps	ql-ui	ql-ui
4	52.00	1145	2625	2850	-13310	26485	1265	2900	3150	-15100	29665	1385	3175	3450	-16895	32845
2	42.50	1150	2155	2425	-15075	24745	1270	2380	2680	-17050	27760	1390	2605	2940	-19020	30775
9	36.33	1150	1815	2135	-17200	25200	1270	2010	2360	-19405	28305	1390	2200	2585	-21610	31405
7	32.07	1155	1570	1930	-19410	26360	1275	1735	2135	-21870	29630	1395	1895	2335	-24330	32905
∞ (29.00	1160	1375	1780	-21225	27185	1280	1520	1965	-23895	30570	1400	1665	2150	-26560	33950
ب م	26.72	1140	1215	1645	-22445	2/335	1260	1345	1820	-25245	30750	13/5	14/0	1890	05087-	341/0
1 1	23.68	1075	965	1490	-23495	26265	1180	1065	1645	-26405	29560	1290	1165	1800	-29310	32850
12	22.67	1040	865	1410	-23645	25795	1140	955	1550	-26530	29010	1245	1045	1695	-29440	32230
13	21.88	1000	780	1325	-23725	25360	1100	860	1455	-26600	28500	1195	940	1590	-29470	31635
14	21.29	965	705	1235	-23640	24995	1060	780	1355	-26480	28090	1150	820	1480	-29325	31185
15	20.83	096	650	1195	-24045	25340	1050	715	1310	-26910	28435	1140	780	1430	-29775	31535
16	20.50	955	009	1165	-24440	25795	1045	099	1275	-27330	28940	1130	720	1390	-30215	32085
17	20.26	955	522	1140	-24895	26245	1040	610	1250	-27805	29440	1130	999	1360	-30720	32635
18	20.11	955	515	1125	-25470	27235	1045	292	1230	-28420	30490	1130	615	1340	-31370	33745
19	20.03	096	480	1110	-26145	28225	1050	525	1220	-29085	31585	1135	575	1325	-32050	34940
20	20.00	970	445	1105	-26850	29160	1055	490	1210	-29825	32615	1140	535	1315	-32805	36070

						٩	90	30	20	52	35	35	30	35	75	10	30	50				۰	00	01	15	35	00	10	35	20	30	000	0,	ç
					₹	in-lb	12890	12130	12350	13025	12985	12665	12480	12635	13275	14240	16880	19020			₹	in-lb	20000	19010	19645	20835	20800	20240	19785	19760	20480	21730	23170	00000
					M	dl-ni	-6545	-7295	-9015	-10615	-11595	-12135	-12465	-12905	-13710	-14660	-15910	-16590			M	dl-ni	-11090	-12110	-14605	-16985	-18315	-18950	-19250	-19500	-20350	-21425	-22710	ייייייי
				25 psf	T _a	sql	1545	1345	1095	955	870	815	775	755	735	725	720	720		40 psf	E ₄	sql	2105	1835	1495	1300	1165	1085	995	920	885	865	855	220
	Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$			χ.	lbs	1425	1205	920	740	615	525	460	405	360	355	385	390			٣×	lbs	1945	1645	1250	1005	830	695	290	510	445	395	385	000
	ad ad				R _v	lbs	610	610	615	625	630	640	655	999	089	695	710	715			R	lbs	830	830	835	845	825	800	770	745	745	755	770	775
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			±	dl-ni	10520	9835	9920	10425	10380	10140	10045	10260	10910	12830	16880	19020			±	in-lb	17630	16715	17210	18230	18195	17715	17350	17385	18035	19235	20540	0,770
		= 1.0. only.	J [pg]		×	d-ni	-5025	-5705	-7160	-8515	-9365	-9860	-10255	-10725	-11495	-12500	-14770	-16135	1 [pg]		Z	in-lb	-9575	-10505	-12740	-14865	-16055	-16680	-16985	-17265	-18135	-19170	-20405	3106
		All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.	Ground Snow Load [pg]	20 psf	Fa	lbs	1545	1345	1095	955	870	815	775	755	735	725	720	720	Ground Snow Load [pg]	35 psf	T _a	lbs	1915	1665	1360	1180	1065	066	910	845	815	800	790	700
	Notes:	All values noi Use ASD desi	Ground		×,	lbs	1425	1205	920	740	615	525	460	405	360	355	385	390	Ground		۳,	lbs	1770	1495	1140	915	755	635	540	465	410	360	385	000
		α _χ I			R _v	lbs	610	610	615	625	630	640	655	999	089	695	710	715			R	lbs	755	755	260	770	755	730	705	069	069	200	715	725
	/4	, in the second			±	in-lb	8150	7545	7510	7820	7775	7620	7635	7890	9962	12830	16880	19020			μ	ql-ui	15260	14420	14780	15630	15590	15190	14915	15010	15645	16735	17990	19020
					M	ql-ui	-3510	-4115	-5320	-6420	-7135	-7615	-8055	-8545	-9965	-12245	-14770	-16135			×	in-lb	0908-	-8895	-10875	-12740	-13825	-14405	-14725	-15085	-15920	-16915	-18120	-18875
				15 psf	F.	lbs	1545	1345	1095	955	870	815	775	755	735	725	720	720		30 psf	L.	lbs	1720	1500	1225	1065	096	895	830	770	745	730	725	725
Load.		× × ×			۳,	lps	1425	1205	920	740	615	525	460	405	360	355	385	390			ď	lbs	1590	1345	1025	825	089	570	485	420	370	355	385	390
ariable Snov					A,	sql	610	610	615	625	630	640	655	999	089	695	710	715			A,	sql	089	089	685	695	685	999	655	999	089	695	710	715
Table E4 - 50 ft Span, Variable Snow Load.		10 psr Dead Load 20 psf Construction Load 120 mph Wind Zone		Radius [R]	(L		65.00	55.08	43.06	36.25	32.04	29.32	27.53	26.36	25.63	25.20	25.02	25.00		Radius [R]	£		65.00	55.08	43.06	36.25	32.04	29.32	27.53	26.36	25.63	25.20	25.02	25.00
Table E-4 - :	3	10 pst Dead Load 20 psf Construction I 120 mph Wind Zone		Rise [T]	(ft)		2	9	∞	10	12	14	16	18	20	22	24	25		Rise [T]	(L)		2	9	∞	10	12	14	16	18	20	22	24	35

E-4 - 50	ft Span, Va	Table E-4 - 50 ft Span, Variable Snow Load	v Load.													
-	3			//		/		Notes:			Snow Load	ad	Wind Load			
ead L Instr Win	10 pst Dead Load 20 psf Construction Load 120 mph Wind Zone		x x x		~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~	a di	Ä,	All values no Use ASD desi	All values normalized to C_D = 1.0. Use ASD design procedure only.	= 1.0. only.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{zt} = 1.0$ $K_d = 0.85$			
-								Ground	Ground Snow Load [pg]	[bg] p						
Rise [T] Ra	Radius [R]			50 psf					e0 psf					70 psf		
	(£	A,	χ.	F.	×	₹	R,	٣×	F _a	M	±	R,	~×	F _a	×	Ψ
		sql	lps	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	sql	lbs	ql-ui	in-lb
-	65.00	086	2305	2490	-14145	24735	1130	2660	2875	-17205	29475	1280	3015	3260	-20265	34215
	55.08	086	1945	2170	-15330	23595	1130	2245	2505	-18545	28180	1285	2545	2840	-21765	32765
H	43.06	066	1480	1765	-18330	24505	1140	1705	2035	-22060	29365	1290	1935	2305	-25785	34225
	36.25	995	1185	1530	-21235	26040	1145	1370	1765	-25485	31245	1295	1550	2000	-29735	36450
H	32.04	970	975	1375	-22830	26010	1115	1125	1580	-27345	31220	1260	1275	1790	-31860	36430
	29.32	930	815	1275	-23495	25285	1065	940	1465	-28045	30335	1200	1060	1655	-32630	35380
	27.53	890	069	1165	-23770	24655	1015	795	1335	-28295	29525	1135	895	1505	-32815	34400
	26.36	860	595	1070	-23970	24510	975	089	1220	-28445	29260	1090	770	1370	-32920	34010
	25.63	860	520	1030	-24780	25375	970	595	1170	-29310	30270	1080	675	1310	-33840	35165
	25.20	865	460	1000	-25935	26725	975	530	1135	-30445	31720	1085	595	1275	-34955	36710
	25.02	880	415	066	-27325	28430	985	470	1120	-31940	33690	1095	530	1255	-36555	38950
	25.00	885	390	985	-28060	29255	995	445	1115	-32720	34640	1100	200	1245	-37385	40020
\vdash								Groune	Ground Snow Load [pg]	[⁸ d] p						
Rise [T] Ra	Radius [R]			80 psf					90 psf					100 psf		
	(£)	R _V	Α,	Fa	M	±	R _v	R	Fa	M	± ⊠	R _v	۳,	Fa	M	₹
		sql	lps	lps	in-lb	d-ni	lps	lbs	lps	in-lb	in-lb	lps	sql	sql	ql-ui	in-lb
H	65.00	1430	3370	3645	-23325	38955	1580	3725	4030	-26385	43690	1730	4080	4415	-29440	48430
	55.08	1435	2845	3170	-24985	37355	1585	3150	3202	-28200	41940	1735	3450	3840	-31420	46525
	43.06	1440	2160	2575	-29510	39085	1590	2390	2845	-33240	43945	1740	2615	3120	-36965	48805
	36.25	1450	1730	2235	-33980	41655	1600	1915	2465	-38230	46865	1750	2095	2700	-42480	52070
	32.04	1400	1420	1995	-36380	41640	1545	1570	2200	-40895	46870	1690	1720	2410	-45410	52105
	29.32	1335	1185	1845	-37220	40430	1465	1305	2035	-41810	45480	1600	1430	2225	-46395	50525
	27.53	1260	1000	1675	-37340	39270	1385	1100	1845	-41860	44140	1505	1205	2015	-46385	49010
	26.36	1205	855	1520	-37390	38760	1315	940	1670	-41865	43570	1430	1030	1820	-46335	48375
	25.63	1190	750	1455	-38370	40055	1305	825	1595	-42895	44950	1415	006	1735	-47425	49845
	25.20	1190	099	1410	-39555	41705	1300	730	1545	-44155	46700	1410	795	1685	-48755	51695
	25.02	1205	290	1385	-41165	44210	1310	650	1520	-45780	49470	1420	710	1655	-50410	54735
_	25.00	1210	560	1380	-42050	45405	1320	615	1510	-46710	50790	1425	670	1645	-51375	56175

					_	0	īυ	0	īΟ	5	ίζ	0	ζ.	2	0.	Q	ស៊	0				_	55	5	ñ	5	0	0	ī.	0	Q	7	o.	0	'n
				₹	dl-ni	17610	16375	17590	18455	18515	18155	17840	17785	18215	19020	20180	23385	28560			₹	in-lb	27565	25955	28135	29605	29730	29120	28505	28230	28500	29375	30840	32640	34525
				Σ	dl-ni	-10380	-11625	-13825	-15585	-16740	-17420	-17855	-18215	-18840	-19800	-20910	-22350	-23920			Ē	in-lb	-17290	-18915	-22210	-24855	-26470	-27255	-27700	-27940	-28360	-29380	-30625	-32165	-33735
			25 psf	F.	sql	1875	1505	1285	1150	1060	995	950	920	006	885	875	865	860		40 psf	L	sql	2560	2050	1750	1560	1415	1335	1250	1160	1095	1065	1045	1030	1025
Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$			۳×	sql	1740	1325	1065	890	260	999	290	525	475	430	420	450	465			R _x	sql	2370	1805	1450	1210	1030	882	292	670	262	532	485	450	465
ر 1	a			R,	lbs	730	735	740	745	755	765	775	790	800	815	830	845	860			R	lbs	962	1000	1005	1015	1000	920	940	910	895	895	902	915	930
Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			₹	in-lb	14290	13180	14075	14740	14785	14530	14345	14370	14785	15620	18505	23385	28560			±	in-lb	24245	22760	24620	25890	25975	25440	24930	24745	25070	25860	27290	28915	30770
	= 1.0. only.	[gd] þ		M	in-lb	-8095	-9195	-11025	-12510	-13520	-14140	-14570	-15050	-15685	-16605	-17775	-20050	-23265	[⁸ d] p		M	in-lb	-14985	-16485	-19415	-21765	-23210	-23975	-24420	-24690	-25150	-26190	-27390	-28860	-30375
	Notes. All values normalized to $C_0 = 1.0$. Use ASD design procedure only.	Ground Snow Load [pg]	20 psf	F.	lbs	1875	1505	1285	1150	1060	995	950	920	006	885	875	865	860	Ground Snow Load [pg]	35 psf	ъ°	lbs	2325	1865	1590	1420	1290	1220	1140	1065	1005	980	096	955	945
NO Po	All values no Use ASD des	Groun		۳,	sql	1740	1325	1065	890	200	999	290	525	475	430	420	450	465	Groun		R _x	sql	2155	1640	1320	1100	935	802	700	615	545	490	445	450	465
	κ _χ			Υ _ν	lbs	730	735	740	745	755	292	775	790	800	815	830	845	860			R,	sql	906	910	915	925	910	882	860	840	825	830	840	820	865
				₹	in-lb	10970	9985	10580	11025	11055	10900	10850	11010	12040	14985	18505	23385	28560			₊Μ	in-lb	20925	19565	21105	22175	22245	21780	21360	21265	21640	22420	23735	25190	28560
	μ ω ω			Ē	in-lb	-5820	0629-	-8260	-9465	-10305	-10860	-11375	-11885	-12540	-14400	-17100	-20050	-23265			įМ	in-lb	-12680	-14055	-16620	-18675	-19955	-20695	-21135	-21435	-21995	-22995	-24150	-25550	-27140
//			15 psf	ъ.	sql	1875	1505	1285	1150	1060	995	950	920	006	885	875	865	860		30 psf	^e 4	sql	5002	1675	1435	1280	1165	1100	1035	965	920	895	880	875	870
w Load.	XX XX			A,	sql	1740	1325	1065	890	200	999	290	525	475	430	420	450	465			R _x	sql	1940	1475	1190	066	845	725	635	522	495	445	420	450	465
'ariable Sno	aq			R,	sql	730	735	740	745	755	765	775	790	800	815	830	845	860			R	sql	815	820	825	835	825	802	785	790	800	815	830	845	860
Table E-5 - 60 ff Span, Variable Snow Load	10 psf Dead Load 20 psf Construction Load 120 mph Wind Zone		Radius [R]	(#)		78.00	60.25	20.00	43.50	39.14	36.13	34.00	32.50	31.45	30.75	30.31	30.07	30.00		Radius [R]	(78.00	60.25	20.00	43.50	39.14	36.13	34.00	32.50	31.45	30.75	30.31	30.07	30.00
Table E-5 -	10 psf Dead Load 20 psf Construction I 120 mph Wind Zone		Rise [T]	(£)		9	∞	10	12	14	16	18	20	22	24	56	28	30		Rise [T]	(L)		9	∞	10	12	14	16	18	20	22	24	56	28	30

e E-5 - (50 ft Span, V	Table E-5 - 60 ft Span, Variable Snow Load	w Load.													
200	10 per 0 per 01					//		Notes:			Snow Load	ad	Wind Load			
Cons ph W	10 psi Dead Load 20 psf Construction Load 120 mph Wind Zone	ad	XX XX		~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~	, di	K K I	All values no Use ASD des	All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.	= 1.0. only.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{zt}=1.0 \label{eq:K_zt}$ $K_d=0.85 \label{eq:K_d}$			
								Ground	Ground Snow Load [pg]	d [þg]						
Rise [T]	Radius [R]			50 psf					e0 psf	,				70 psf		
(±)	(£	A,	٣×	F _a	×	±	R _v	Α,	L.	M	±	R	A _x	L _a	M	₊ _
		sql	sql	lbs	ql-ui	in-lb	lps	lbs	lbs	ql-ui	in-lb	lbs	lps	sql	ql-ui	ql-ui
9	78.00	1175	2805	3030	-21895	34200	1355	3240	3200	-26500	40840	1535	3675	3962	-31105	47475
∞	60.25	1180	2135	2425	-23775	32340	1360	2460	2795	-28635	38725	1540	2790	3170	-33495	45115
10	20.00	1185	1715	2070	-27805	35165	1365	1980	2385	-33395	42195	1550	2240	2700	-38985	49225
12	43.50	1195	1430	1840	-31035	37040	1375	1650	2125	-37215	44475	1555	1865	2405	-43395	51905
14	39.14	1170	1215	1670	-32985	37265	1345	1395	1920	-39500	44795	1520	1580	2170	-46015	52330
16	36.13	1135	1040	1570	-33840	36485	1300	1195	1805	-40470	43845	1465	1355	2045	-47100	51210
18	34.00	1095	006	1465	-34270	35655	1245	1035	1680	-40835	42800	1400	1170	1895	-47400	49950
20	32.50	1055	790	1355	-34445	35190	1195	902	1550	-40950	42155	1340	1020	1745	-47455	49120
22	31.45	1030	200	1275	-34830	35355	1165	800	1455	-41305	42210	1305	006	1630	-47775	49065
24	30.75	1030	625	1235	-35785	36400	1165	715	1405	-42320	43430	1295	810	1575	-48855	50460
26	30.31	1035	292	1205	-37105	37950	1165	645	1370	-43585	45060	1300	730	1530	-50130	52170
28	30.07	1045	515	1190	-38775	40090	1180	290	1350	-45390	47540	1310	099	1510	-52000	54990
_	30.00	1060	470	1180	-40460	42035	1190	535	1340	-47180	49780	1320	009	1495	-53905	57520
								Groun	Ground Snow Load [pg]	[gd] b						
Rise [T]	Radius [R]			80 psf					90 psf					100 psf		
£	(£	S,	ď	F _a	Ē	₹	A,	χ.	Ľ.	Ā	±	R,	Α _x	Ľ.	×	±
		sql	lps	lbs	in-lb	in-lb	sql	lbs	lbs	in-lb	ql-ui	lps	lps	sql	ql-ui	ql-ui
	78.00	1715	4105	4435	-35715	54110	1895	4540	4900	-40320	05/09	2080	4975	5370	-44925	67385
8	60.25	1720	3120	3545	-38355	51500	1905	3450	3915	-43215	27890	2085	3775	4290	-48075	64275
_	20.00	1730	2505	3020	-44580	26260	1910	2770	3335	-50170	63290	2090	3030	3655	-55765	70320
12	43.50	1735	2085	2685	-49580	59340	1915	2305	2962	-55760	02.299	2100	2525	3245	-61940	74205
14	39.14	1695	1765	2425	-52535	29860	1870	1950	2675	-59050	67390	2045	2135	2925	-65565	74925
16	36.13	1630	1510	2280	-53730	58575	1795	1670	2515	-60360	65935	1955	1825	2750	06699-	73300
18	34.00	1555	1305	2115	-53980	57100	1710	1440	2330	00909-	64245	1865	1575	2545	-67225	71395
20	32.50	1485	1135	1940	-53965	26080	1630	1255	2135	-60470	63045	1770	1370	2330	-66975	70010
22	31.45	1440	1005	1810	-54245	55920	1575	1105	1990	-60720	62780	1710	1205	2165	-67190	69635
24	30.75	1430	006	1745	-55390	57490	1565	066	1915	-61925	64515	1695	1080	2085	-68460	71545
26	30.31	1430	810	1695	-56745	59275	1560	890	1860	-63355	66385	1690	975	2025	02669-	73495
28	30.07	1440	735	1675	-58615	62440	1570	810	1835	-65275	06869	1700	882	1995	-72010	77340
30	30.00	1450	670	1655	-60625	65265	1580	735	1815	-67350	73005	1710	802	1975	-74070	80750

-31760 -27490 -34300 -37700 -17000 -23230 -24300 -25930 -26775 -29765 -34075 -36560 -31225 -38560 -39110 -39622 -40985 -46925 -49615 -51070 -15270 -19525 -21675 -25095 -28055 -25130 -36345 -42675 -44600 -37900 2330 1970 1740 40 psf 25 psf Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ 2245 1875 1230 1090 850 770 705 645 625 640 655 725 665 655 695 1210 Snow Load $C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$ 35465 -12010 -13535 -17505 -19830 -20740 -21595 -22480 -23895 -25615 -27555 -30425 -34765 -23995 -27320 -30065 -33235 -34070 -34645 -35360 -36675 -38320 -40195 -42640 -45265 -15680 -21845 -46680 -18855-37060 -31975Ground Snow Load [pg] Ground Snow Load [pg] All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. 35 psf 20 psf **F**_a 1**bs** 3565 Rx 1lbs 3365 2565 2065 11725 11290 1135 11005 Notes: **R**_x **Ibs** 2760 1695 1420 850 770 705 645 625 640 640 735 670 640 655 695 710 1220 1240 1250 ď I -15490 -17270 -19790 -26410 -30425 -20495 -23420 -25850 -28765 -29580 -32365 -40910 -42290 -10085 -11835 -13335 -16385 -18370 -22715 -34765 -18555 -27600 -30265 -31065 -33970 -38355 -35960 -8765 -14485-37060 30 psf 15 psf F_a lbs lbs 2925 2320 1960 1730 **F**_a **Ibs** 3250 1750 1250 1250 1355 1040 920 825 740 675 1695 1420 **R**_x **Ibs** 2760 850 770 705 645 625 640 655 640 655 695 710 Table E-6 - 70 ft Span, Variable Snow Load. 20 psf Construction Load Radius [R] Radius [R] 120 mph Wind Zone 36.56 35.42 66.25 57.04 36.56 35.42 35.14 35.01 66.25 57.04 50.75 46.28 43.03 40.63 38.84 37.52 35.88 35.14 46.28 43.03 40.63 38.84 37.52 35.88 Œ 王 15 psf Dead Load Rise [T] Rise [T] Œ Œ

Table E-6 - 70 ft Span, Variable Snow Load.	v Loa	÷	\\	(-			Notes:			Snow Load	7	Wind Load			
15 psf Dead Load 20 psf Construction Load 120 mph Wind Zone	ů.		- 	// _	a di di	X X	All values no Use ASD des	Notes. All values normalized to $C_0 = 1.0$. Use ASD design procedure only.	only.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{rt} = 1.0$ $K_{d} = 0.85$			
				1			Groun	Ground Snow Load [pg]	d [p _g]						
50 psf	50 psf	50 psf		ıl				e0 psf					70 psf		
R _y R _x F _a M	T _e		Σ		₹	α _ν	X	T.	Σ	₹	Α,	××	T _e	Έ	₹
lbs lbs in-lb	lbs		in-lb	-	in-lb	lbs	lbs	lps	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
4250 4505	4505	i	-28860	_	47445	1735	4840	5130	-35105	26505	1945	5435	5755	-41355	65565
3235 3565	3565		-31705		44680	1740	3685	4055	-38285	53425	1950	4135	4550	-44860	62165
1540 2605 3005 -34490	3005		-34490		44620	1750	2965	3420	-41485	53490	1960	3325	3835	-48480	62360
1865 2405	2030		-3505C		79890	1770	2473	3010	-40033	50055	1985	2775	3060	-34040	02007
1620 2210	2210		-45195		50240	1750	1840	2510	-54060	60460	1955	2065	2805	-62920	70685
1420 2100	2100	Н	-46635		49670	1710	1615	2380	-55585	59740	1905	1805	2660	-64625	69810
1480 1260 1990 -47540	1990		-47540		48815	1665	1425	2250	-56520	58615	1850	1595	2515	-65500	68415
1875	1875		-48035		48080	1620	1265	2120	-56960	57565	1795	1415	2360	-65885	67050
1005 1765	1765		-48380		47720	1575	1135	1985	-57205	57125	1740	1265	2205	-66025	66530
910 1700	1700		-49605		48950	1565	1025	1910	-58290	58300	1720	1145	2115	-67130	67655
830 1665	1665		-51380		50270	1565	935	1860	-60085	59820	1720	1040	2060	-68790	69375
760 1635 -53405	1635 -53405	-53405	_		52740	1580	860	1825	-62210	62335	1730	955	2020	-71015	71935
700 1625	1625		-55860		55480	1595	790	1810	-64815	65480	1750	880	1995	-73770	75475
695 1610	1610	+	-58405		58305	1615	730	1795	-67490	68425	1770	810	1975	-76575	78775
1480 710 1610 -59845	1610		-59845		60180	1630	710	1790	-68935	70385	1780	780	1970	-78090	80595
				- 1	•		Groun	Ground Snow Load [pg]	d [pg]	•					
80 psf	80 psf	80 psf				•		90 psf			-	•	100 psf		,
ж. :	ue ;		Σ		₹	~`	œ :	Ľ.	Σ	±Σ ∫	۲ [^] :	œ* :	Ľ.	Σ	-Σ
SOI SOI	SOI	+	QI-III	_	QI-UI	SOI	IDS	SOI	QI-UI	αI-II	SOI	SOI	SOI	QI-UI	ΩI-II
2155 6025 6380 -47600	6380		-47600		70910	2365	5030	7005	-53850	83685	2575	7.205	7630	-60095	92745
3685 4250	4250	Ė	-55480		71230	2380	4045	4665	-62475	80105	2595	4405	5080	-69470	88975
2180 3075 3740 -62445	3740		-62445		76595	2390	3375	4100	-70250	86205	2605	3675	4465	-78055	95815
2195 2635 3390 -68160	3390	_	-68160		80085	2405	2890	3720	-76625	90150	2615	3145	4045	-85090	100215
2160 2285 3105 -71785	3105	_	-71785		80905	2365	2505	3400	-80650	91130	2570	2725	3700	-89515	101350
	2940		-73665		79880	2295	2190	3225	-82705	89950	2490	2380	3505	-91745	100020
2035 1760 2775 -74480	2775		-74480		78215	2220	1930	3040	-83475	88015	2405	2095	3300	-92540	97815
1965 1560 2600 -74815	2600	H	-74815		76535	2140	1710	2845	-83740	86015	2315	1855	3085	-92665	95500
1905 1395 2425 -74845	2425		-74845		75935	2065	1525	2645	-83665	85340	2230	1655	2865	-92490	94745
1880 1260 2320 -75965	2320	H	-75965		77010	2040	1375	2530	-84805	86365	2195	1495	2735	-93640	95720
1880 1145 2255 -77635	2255		-77635		78925	2035	1255	2455	-86545	88480	2190	1360	2655	-95450	98030
1050 2210	2210	Ĥ	-79820	-	81535	2040	1145	2400	-88625	91135	2190	1245	2590	-97560	100735
965 2180	2180		-82725	_	85475	2055	1055	2370	-91680	95470	2205	1145	2560	-100640	105470
890 2160	2160	H	-85660	_	89130	2070	975	2345	-94750	99480	2225	1055	2530	-103835	109830
1930 855 2155 -87245	2155		-87245		08606	2085	935	2340	-96405	101510	2235	1015	2525	-105560	112040

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					₹	in-lb	30030	29395	32035	31985	31680	32420	35405	39680	52060			₹	in-lb	47215	47085	51690	51705	50225	50385	53480	
					Σ	in-lb	-19775	-23520	-28555	-31565	-33415	-35315	-38920	-43745	-49530			M	ql-ui	-32605	-37755	-45125	-49035	-50870	-52095	-52805	
				25 psf	F _a	lbs	2975	2175	1815	1615	1505	1445	1415	1400	1395		40 psf	T _e	sql	3925	2865	2375	2105	1920	1740	1675	
:	Wind Load	$K_{tt} = 1.0$ $K_{d} = 0.85$			α _x	lbs	2760	1865	1405	1115	925	785	710	740	802			ď	lps	3640	2450	1840	1450	1175	970	825	
	힐				Α,	lbs	1140	1160	1180	1200	1230	1265	1305	1350	1395			R,	sql	1500	1520	1540	1510	1465	1420	1435	
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			±	in-lb	24300	23495	25485	25505	25500	26435	29375	37090	52060			μ	in-lb	41485	41185	45140	45135	44045	44395	47455	
		= 1.0. inly.	[þ _g]	0	×	in-lb	-15495	-18795	-23085	-25740	-27730	-29735	-33495	-38420	-48430	[bg]		M	in-lb	-28330	-33010	-39575	-43210	-45010	-46500	-50115	
		All values normalized to C _D = 1.0. Use ASD design procedure only.	Ground Snow Load [pg]	20 psf	F _a	lbs	2960	2165	1800	1610	1505	1445	1415	1400	1395	Ground Snow Load [pg]	35 psf	T.	lbs	3610	2635	2190	1940	1775	1620	1560	
	Notes:	All values nori Use ASD desig	Ground		χ.	lbs	2750	1855	1395	1115	925	785	710	740	805	Ground		۳×	lbs	3345	2255	1695	1340	1085	006	765	
		ď.			R,	lbs	1135	1150	1170	1200	1230	1265	1305	1350	1395			R,	lbs	1380	1400	1420	1400	1360	1325	1345	
	//		Ž		±	in-lb	18570	17620	18935	19115	19345	20760	27045	37090	52060			ψ	in-lb	35760	35290	38590	38560	37865	38410	41430	
1		ω ω			Ē	in-lb	-11280	-14120	-17615	-20025	-22045	-24415	-29710	-38420	-48430			Ē	in-lb	-24050	-28265	-34025	-37390	-39150	-40905	-44425	
\				15 psf	T _e	lbs	2960	2165	1800	1610	1505	1445	1415	1400	1395		30 psf	т _е	lps	3290	2405	2000	1780	1635	1495	1450	
Load.		a a	Á		χ.	lbs	2750	1855	1395	1115	925	785	710	740	802			٣×	lbs	3055	2060	1550	1225	995	825	710	
rriable Snow					₽,	lbs	1135	1150	1170	1200	1230	1265	1305	1350	1395			S,	lps	1260	1280	1300	1285	1260	1265	1305	
o it Span, Va	700	15 psi Deau Load 20 psf Construction Load 120 mph Wind Zone		Radius [R]	£		104.00	72.67	28.00	20.00	45.33	42.57	41.00	40.22	40.00		Radius [R]	£		104.00	72.67	28.00	20.00	45.33	42.57	41.00	
Table E-7 - 80 ft Span, Variable Snow Load.	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	15 psi Dead Load 20 psf Construction 120 mph Wind Zone		Rise [T]	(L		∞	12	16	20	24	28	32	36	40		Rise [T]	(L		∞	12	16	20	24	28	32	

Table E-7 -	80 ft Span, V	Table E-7 - 80 ft Span, Variable Snow Load.	v Load.													
	7					//		Notes:			Snow Load	Б	Wind Load			
15 psr Dead Load 20 psf Construction 120 mph Wind Zone	15 psi Dead Load 20 psf Construction Load 120 mph Wind Zone		The state of the s		~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~	The state of the s	π «	All values noi Use ASD desi	All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.	= 1.0. only.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{tt} = 1.0$ $K_d = 0.85$			
								Ground	Ground Snow Load [pg]	J [p _g]						
Rise [T]	Radius [R]			50 psf					e0 psf					70 psf		
(#)	(£)	Ry	A,	Fa	Ē	± ∑	Ry	R _x	Fa	Ā	⁺ ∑	R _v	a,	ъ.	Σ	₊
		sql	lps	lps	q-ui	d-ni	lbs	lbs	lps	q-ui	ql-ui	lps	lps	lps	in-lb	in-lb
8	104.00	1745	4230	4555	-41160	58675	1985	4815	5190	-49715	70135	2225	5400	5820	-58270	81595
12	72.67	1760	2845	3320	-47245	58880	2000	3240	3780	-56735	02902	2240	3630	4240	-66225	82465
16	58.00	1785	2135	2750	-56225	64795	2025	2430	3125	-67325	77895	2265	2720	3200	-78425	90995
20	20.00	1740	1680	2430	-60680	64855	1965	1905	2755	-72440	78000	2190	2135	3080	-84230	91145
24	45.33	1670	1355	2210	-62590	62825	1875	1535	2495	-74305	75475	2080	1715	2785	-86025	88125
28	42.57	1605	1115	1985	-63375	62360	1790	1260	2230	-74860	74335	1970	1405	2475	-86345	86375
32	41.00	1610	950	1900	-67180	65550	1790	1070	2125	-78560	78010	1970	1190	2355	-89935	90475
36	40.22	1645	820	1855	-72165	71445	1820	920	2070	-83810	84360	1990	1025	2285	-95460	97275
40	40.00	1690	805	1840	-78205	78530	1860	805	2045	-90090	91855	2035	890	2255	-102050	105175
								Ground	Ground Snow Load [pg]	[gd] k						
Rise [T]	Radius [R]			80 psf					90 psf					100 psf		
(#)	(#)	œ^	æ×	ъ.	Σ	₹	Я _V	××	T.	Σ	± ≥	₽,	ď	щ	Έ	[±]
		sql	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
∞	104.00	2465	2990	6455	-66825	93055	2705	6575	7085	-75380	104515	2950	7165	7720	-83935	115975
12	72.67	2485	4025	4695	-75720	94260	2725	4415	5155	-85210	106055	2962	4810	5615	-94700	117850
16	58.00	2505	3015	3880	-89530	104100	2750	3305	4255	-100630	117200	2990	3600	4630	-111730	130305
20	20.00	2420	2360	3405	-96020	104295	2645	2585	3730	-107810	117440	2870	2815	4055	-119600	130590
24	45.33	2285	1895	3075	-97745	100775	2490	2075	3360	-109465	113425	2692	2255	3650	-121185	126080
28	42.57	2155	1550	2720	-97835	98590	2340	1695	2962	-109320	110805	2525	1840	3210	-120805	123025
32	41.00	2145	1310	2580	-101515	102935	2325	1435	2805	-113155	115400	2500	1555	3030	-124795	127865
36	40.22	2165	1130	2500	-107110	110190	2340	1230	2715	-118760	123100	2515	1335	2930	-130410	136015
40	40.00	2210	980	2465	-114015	118735	2380	1070	2675	-125980	132480	2555	1160	2885	-137940	146225

ď.	Table E-8 - 90 ft Span, Variable Snow Load	w Load.	\							,		:			
15 psf Dead Load 20 psf Construction Load 120 mph Wind Zone						ſ	Notes: All values no Use ASD desi	Notes: All values normalized to $C_{\rm D}=1.0$. Use ASD design procedure only.	= 1.0. only.	$C_e = 0.9$ $C_t = 1.2$	p	$\begin{aligned} & \text{Wind Load} \\ & K_{zt} = 1.0 \\ & K_d = 0.85 \end{aligned}$			
		X W		. s	2 2	× Y I				N = 1:TO					
							Groun	Ground Snow Load [pg]	[bg]						
L			15 psf					20 psf					25 psf		
-	R,	R _x	Fa	·ω	±	Ry	χ.	Fa	M	₹	R	A,	Fa	M	→
2	lbs	lbs	lbs	dl-ni	in-lb	lps	lbs	lbs	in-lb	in-lb	lps	lps	lps	in-lb	ql-ui
12	1275	3475	3690	-14125	23650	1275	3475	0698	-19480	30920	1280	3490	3710	-24880	38190
12	1290	2345	2660	-16530	21210	1290	2345	2660	-22105	28240	1295	2355	2675	-27740	35270
13	1305	1765	2175	-20670	23180	1305	1765	2175	-27210	31095	1315	1775	2190	-33765	39010
13	1330	1415	1915	-23920	24055	1330	1415	1915	-31130	32325	1330	1420	1920	-38340	40590
13	1360	1175	1765	-26310	23990	1360	1175	1765	-33500	32080	1360	1175	1770	-40900	40180
13	1390	1000	1675	-28520	24725	1390	1000	1675	-35710	32195	1390	1000	1675	-42895	40010
17	1430	870	1620	-31410	26545	1430	870	1620	-38155	33685	1430	870	1620	-45180	41215
17	1470	795	1590	-37635	35205	1470	795	1590	-42435	37115	1470	795	1590	-49300	44735
Ħ	1515	825	1575	-47345	46650	1515	825	1575	-47615	46650	1515	825	1575	-54640	49395
ij	1535	870	1570	-52685	55450	1535	870	1570	-52685	55450	1535	870	1570	-57760	55450
1	1570	900	1570	-61315	67755	1570	900	1570	-61315	67755	1570	900	1570	-62710	67755
							Groun	Ground Snow Load [pg]	d [p _g]						
			30 psf					35 psf					40 psf		
	R _v	R _x	Fa	M	±Σ	Ry	R _x	Е	M	₊	Ry	R _x	Fa	M	⁺ Σ
	lbs	lps	lps	in-lb	in-lb	lps	lbs	lps	in-lb	in-lb	lps	lps	sql	in-lb	dl-ni
I	1415	3865	4105	-30280	45455	1550	4235	4495	-35680	52725	1685	4610	4890	-41080	26665
	1430	2605	2955	-33375	42295	1565	2855	3240	-39015	49325	1700	3105	3520	-44650	56355
	1450	1960	2420	-40405	46930	1585	2150	2645	-47045	54925	1720	2335	2875	-53680	62920
1	1460	1565	2115	-45550	48940	1595	1710	2310	-52760	57350	1730	1860	2505	-60040	65765
	1435	1285	1945	-48325	48280	1560	1400	2120	-55750	56375	1685	1520	2300	-63180	64475
	1410	1075	1800	-50080	47825	1520	1170	1960	-57415	55640	1635	1265	2115	-64815	63455
	1430	915	1675	-52275	48825	1495	995	1810	-59370	56430	1595	1075	1950	-66465	64035
1	1470	795	1630	-56275	52355	1515	860	1755	-63480	59975	1615	930	1885	-70685	00929
1	1515	825	1610	-61660	57040	1550	825	1730	-68685	65140	1645	825	1850	-75875	73240
ij	1535	870	1605	-64890	60450	1570	870	1725	-72020	68405	1670	870	1845	-79150	76355
15	1570	900	1605	-69970	67755	1605	900	1720	-77230	74040	1705	900	1840	-84490	82465

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---|---|------------|----------------
---|---|--
---------------------------------------|---------|---|---------|---------|---------
---|---------------------------------------|---|
| | | | ±
⊠ | in-lb | 103605
 | 98520 | 110900 | 116245
 | 113685 | 110350
 | 109670 | 114385 | 121855 | 126500 | 133035 |
 | | ₹ | in-lb | 147220
 | 140690 | 158880 | 166720 | 163055 | 157685 | 155590 | 161675 | 170470
 | 176675 | 184970 |
| | | | M | in-lb | -73470
 | -78470 | -93520 | -103910
 | -107830 | -109225
 | -109815 | -113905 | -119970 | -123690 | -129205 |
 | | M | in-lb | -105865
 | -112290 | -133355 | -147785 | -152895 | -153630 | -153505 | -158060 | -164065
 | -168400 | -174645 |
| | | 70 psf | Fa | lps | 7260
 | 5215 | 4245 | 3680
 | 3360 | 3022
 | 2765 | 2645 | 2575 | 2555 | 2535 |
 | 100 psf | L | sql | 0896
 | 6910 | 5615 | 4855 | 4420 | 3995 | 3585 | 3410 | 3305
 | 3280 | 3245 |
| $K_{zt} = 1.0$ $K_d = 0.85$ | | | R _x | lbs | 6840
 | 4600 | 3450 | 2740
 | 2225 | 1845
 | 1550 | 1340 | 1170 | 1100 | 1000 |
 | | R _x | lbs | 0206
 | 6095 | 4570 | 3620 | 2935 | 2420 | 2030 | 1750 | 1525
 | 1430 | 1305 |
| ı | | | R_{y} | lbs | 2500
 | 2515 | 2535 | 2525
 | 2425 | 2310
 | 2215 | 2215 | 2235 | 2255 | 2290 |
 | | Ry | lbs | 3315
 | 3330 | 3350 | 3325 | 3165 | 2990 | 2835 | 2815 | 2825
 | 2845 | 2875 |
| $C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$ | | | [†] ⊠ | in-lb | 89070
 | 84465 | 94910 | 99415
 | 97230 | 94720
 | 94455 | 98625 | 105650 | 109780 | 116180 |
 | | ± | in-lb | 132680
 | 126635 | 142890 | 149895 | 146600 | 141845 | 140135 | 145910 | 154265
 | 159950 | 167580 |
| = 1.0.
only. | [gd] b | | M | ql-ui | -62675
 | -67195 | -80240 | -89290
 | -92880 | -94420
 | -95250 | -99500 | -105270 | -108790 | -114060 | [⁸ d] p
 | | M | dl-ni | -95070
 | -101015 | -120075 | -133160 | -137870 | -138830 | -138940 | -143315 | -149370
 | -153495 | -159495 |
| rmalized to C _D
ign procedure. | d Snow Loa | eo psf | В | lbs | 6470
 | 4650 | 3790 | 3290
 | 3002 | 2745
 | 2495 | 2390 | 2335 | 2320 | 2300 | d Snow Loa
 | 90 psf | Fa | lbs | 8840
 | 6345 | 5160 | 4465 | 4065 | 3685 | 3310 | 3155 | 3060
 | 3035 | 3005 |
| All values no
Use ASD des | Groun | | R _x | lbs | 6095
 | 4100 | 3080 | 2445
 | 1990 | 1650
 | 1390 | 1205 | 1055 | 066 | 006 | Groun
 | | R _x | lbs | 8330
 | 2600 | 4195 | 3330 | 2700 | 2230 | 1870 | 1610 | 1405
 | 1320 | 1200 |
| a*
L | | | R _y | lbs | 2230
 | 2245 | 2265 | 2260
 | 2175 | 2085
 | 2010 | 2015 | 2040 | 2060 | 2095 |
 | | R | lbs | 3040
 | 3055 | 3080 | 3060 | 2920 | 2765 | 2630 | 2615 | 2630
 | 2645 | 2680 |
| | | | ± | in-lb | 74530
 | 70410 | 78915 | 82590
 | 80775 | 79090
 | 79245 | 82860 | 89445 | 93055 | 99325 |
 | | ± | in-lb | 118145
 | 112580 | 126895 | 133070 | 130145 | 126005 | 124880 | 130150 | 138060
 | 143225 | 150195 |
| ⊢ ⊗ | | | M | in-lb | -51875
 | -55925 | 09699- | -74665
 | -78030 | -79615
 | -80685 | -82090 | -90575 | -93885 | -99015 |
 | | M | in-lb | -84270
 | -89745 | -106795 | -118535 | -122850 | -124025 | -124380 | -128575 | -134670
 | -138595 | -144350 |
| | | 50 psf | Fa | lbs | 2680
 | 4085 | 3330 | 2895
 | 2650 | 2430
 | 2220 | 2140 | 2090 | 2080 | 2070 |
 | 80 psf | Fa | lps | 8050
 | 5780 | 4700 | 4070 | 3715 | 3370 | 3040 | 2900 | 2815
 | 2795 | 2770 |
| , a | | | R _x | lps | 5350
 | 3605 | 2705 | 2155
 | 1755 | 1460
 | 1230 | 1065 | 935 | 875 | 006 |
 | | R | lps | 7585
 | 5100 | 3825 | 3035 | 2465 | 2035 | 1710 | 1475 | 1290
 | 1210 | 1100 |
| p | | | R | lbs | 1960
 | 1975 | 1995 | 1995
 | 1930 | 1860
 | 1805 | 1815 | 1845 | 1865 | 1900 |
 | | S, | sql | 2770
 | 2785 | 2805 | 2790 | 2670 | 2540 | 2420 | 2415 | 2430
 | 2450 | 2485 |
| l Load
truction Loa
ind Zone | | Radius [R] | (L) | | 130.56
 | 90.38 | 71.28 | 60.63
 | 54.19 | 50.16
 | 47.64 | 46.13 | 45.31 | 45.11 | 45.00 |
 | Radius [R] | (£ | | 130.56
 | 90.38 | 71.28 | 60.63 | 54.19 | 50.16 | 47.64 | 46.13 | 45.31
 | 45.11 | 45.00 |
| 15 psf Deac
20 psf Cons
120 mph W | | Rise [T] | (#) | | ∞
 | 12 | 16 | 20
 | 24 | 28
 | 32 | 36 | 40 | 42 | 45 |
 | Rise [T] | (£ | | 8
 | 12 | 16 | 20 | 24 | 28 | 32 | 36 | 40
 | 42 | 45 |
| | All values normalized to $C_0 = 1.0$. $C_e = 0.9$ Use ASD design procedure only. $C_q = 1.2$ $C_q = 1$ | $R_{\star} = \frac{R}{R_{\star}} = \frac{All \text{ values normalized to C}_{0} = 1.0}{\text{Use ASD design procedure only}} = \frac{C_{0} = 0.9}{I_{0} = 1.10}$ $R_{\star} = \frac{R_{\star}}{R_{\star}} = \frac{R_{\star}}{R_{\star$ | $R_{s} = \frac{R}{R_{s}} = \frac{All \ values \ normalized \ to \ C_{p} = 1.0.}{R_{s} = 0.9} \qquad \frac{R_{s} = 1.0}{R_{s} = 0.9}$ $R_{s} = \frac{R_{s}}{R_{s}} = \frac{R_{s}}{R_{$ | R ₁ | R _v R _v | R | R _v R _v | R _v R _v | R _v R _v | R ₁ R ₂ R ₃ R ₄ R ₄ | R _V R _V R _V F _S Ni Ni Ni Ni Ni Ni Ni N | R _V R _V R _V F _a Ni Ni Ni Ni Ni Ni Ni N | R | R | R _i R _i R _i F _i M _i M _i R _i | R _i R _i | R | R | R _i R _i | R _i R _i | R, R | R, R, R, R, R, R, R, R, | R | R ₁ R ₁ R ₂ R ₃ R ₄ R ₄ | R | R | R | R ₁ R ₂ R ₃ R ₄ R ₄ | R. R. R. R. R. R. R. R. | R ₁ R ₂ R ₃ R ₄ R ₄ |

					±	in-lb	45310	43925	48115	49920	49560	49245	50100	53035	57400	64980	85750			±	in-lb	71790	70455	2692	81030	209867	78155	78020	81245	86380	93060	101760
					M	in-lb		-35790	-42255	-47285	-50255	-52390	-54730	-58425	-63665	-70165	-77440			Ā	in-lb	-53080	-57330	-67070	-74080	-77755	-79730	-81120	-84975	-89955	-96470	-104340 1
				25 psf	L _e	lps	3740	2865	2415	2145	1985	1880	1820	1780	1755	1745	1745		40 psf	T _e	sql	4935	3775	3170	2805	2585	2400	2220	2120	2070	2050	2040
	Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$			A,	lbs	3475	2500	1950	1595	1340	1155	1015	895	890	920	995			۳,	lbs	4585	3290	2565	2090	1740	1470	1260	1100	975	920	995
	민				χ,	lbs	1425	1440	1465	1475	1505	1535	1575	1610	1655	1700	1745			Α _ν	lbs	1875	1895	1915	1920	1880	1830	1785	1780	1805	1845	1895
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			±	in-lb	36485	35085	38315	39725	39570	39605	40790	43880	29905	64980	85750			±	in-lb	62965	61610	67795	20660	69655	68520	68715	71645	76720	83000	91360
		= 1.0. inly.	[bg]		Σ	in-lb	-25805	-28610	-34085	-38395	-41085	-43490	-45935	-49980	-55110	-63045	-75720	[bg]		Σ	in-lb	-46260	-50150	-58800	-65065	-68585	-70560	-72325	-76125	-80995	-87700	-95375
		All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.	Ground Snow Load [pg]	20 psf	F _a	lbs	3725	2855	2400	2140	1980	1880	1820	1780	1755	1745	1745	Ground Snow Load [pg]	35 psf	ъ.	lps	4535	3470	2920	2585	2385	2220	2060	1975	1935	1915	1915
	Notes:	All values nor Use ASD desig	Ground		ď	lbs	3460	2490	1940	1585	1340	1155	1015	895	890	920	995	Ground		٣×	lbs	4215	3030	2360	1925	1605	1360	1165	1020	006	920	995
		π «			S,	lbs	1420	1435	1455	1475	1505	1535	1575	1610	1655	1700	1745			R _v	lps	1725	1745	1765	1775	1740	1700	1665	1670	1695	1740	1785
	//	, and the second			₹	in-lb	27655	26240	28510	29550	29575	30005	31610	39160	29905	64980	85750			±	in-lb	54140	52770	57920	60290	59550	28880	59405	62195	09029	73310	85750
		~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~			Æ	in-lb	-18985	-21500	-25940	-29505	-32165	-34595	-37450	-41610	-51695	-63045	-75720			M	in-lb	-39440	-42970	-50530	-56175	-59420	-61385	-63530	-67275	-72215	-78930	-86410
				15 psf	L.	lbs	3725	2855	2400	2140	1980	1880	1820	1780	1755	1745	1745		30 psf	T.	lps	4140	3170	2665	2365	2185	2040	1905	1830	1795	1785	1785
w Load.		X X			ď	lbs	3460	2490	1940	1585	1340	1155	1015	895	890	920	995			٣×	lps	3845	2765	2155	1760	1465	1245	1070	940	890	920	995
Variable Sno					S,	lbs	1420	1435	1455	1475	1505	1535	1575	1610	1655	1700	1745			A,	lps	1575	1590	1615	1625	1600	1570	1575	1610	1655	1700	1745
Table E-9 - 100 ft Span, Variable Snow Load	3	15 psr Dead Load 20 psf Construction Load 120 mph Wind Zone		Radius [R]	£		130.00	96.29	78.44	67.82	61.08	26.67	53.76	51.89	50.76	50.17	20.00		Radius [R]	£		130.00	96.29	78.44	67.82	61.08	26.67	53.76	51.89	50.76	50.17	20.00
Table E-9 - 1	2	15 psr Dead Load 20 psf Construction I 120 mph Wind Zone		Rise [T]	(L		10	14	18	22	56	30	34	38	42	46	20		Rise [T]	(L		10	14	18	22	56	30	34	38	42	46	20

					 ±	in-lb	124755	123505	137090	143250	141090	137265	135355	138840	144340	154435	164175				ql-ui	177715	176560	196490	205470	202325	196440	192850	196430	202600	215805	228270
					M	in-lb	-93990	-100410	-116700	-128190	-133020	-134765	-135260	-138485	-143725	-151395	-159555			M	ql-ui	-134905	-143490	-166330	-182300	-188680	-189800	-189500	-192890	-198055	-206395	-215660
				70 psf	Fa	lbs	7320	5590	4680	4120	3785	3480	3175	2995	2900	2845	2815		100 psf	T.	sql	9710	7405	6190	5435	4980	4560	4125	3870	3725	3650	3605
	Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$			A _x	lbs	0089	4880	3790	3080	2550	2145	1825	1590	1400	1245	1110			æ	sql	9020	6465	5020	4070	3360	2820	2390	2075	1825	1620	1450
	ad				R _v	lbs	2780	2795	2820	2810	2710	2600	2490	2460	2465	2500	2545			A [^]	lbs	3685	3700	3725	3700	3545	3370	3200	3135	3125	3150	3195
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			± ∑	in-lb	107100	105820	117290	122510	120680	117540	116190	119640	125020	133975	143370			_	ql-ui	160060	158875	176690	184730	181915	176715	173685	177235	182980	195345	206810
		= 1.0. only.	d [p _g]		M	in-lb	-80355	-86050	-100155	-110155	-114465	-116420	-117180	-120370	-125805	-133060	-140855	[⁸ d] p		M	in-lb	-121265	-129130	-149785	-164265	-170125	-171455	-171420	-174755	-179740	-188065	-196960
		All values normalized to C ₀ = 1.0. Use ASD design procedure only.	Ground Snow Load [pg]	go pst	Fa	lbs	6525	4985	4175	3680	3385	3120	2855	2705	2620	2580	2555	Ground Snow Load [pg]	90 psf	Fa	lbs	8915	0089	2690	4995	4580	4200	3810	3580	3450	3380	3340
	Notes:	All values no Use ASD des	Groun		R _x	lbs	0909	4350	3380	2750	2280	1920	1640	1425	1255	1120	1000	Groun		R×	lbs	8280	5935	4610	3740	3090	2595	2205	1910	1680	1495	1335
		Ι ^χ			R	lbs	2480	2495	2515	2515	2435	2345	2255	2235	2245	2280	2325			R,	lbs	3385	3400	3420	3405	3270	3115	2962	2910	2905	2935	2975
	//				Ψ	in-lb	89445	88140	97495	101770	100270	97815	97025	100445	105700	113520	122565			μ	in-lb	142410	141190	156890	163990	161505	156990	154520	158035	163660	174890	185350
					M	in-lb	-66715	-71690	-83615	-92120	-96085	-98075	-99100	-102670	-107880	-114730	-122270			M	ql-ui	-107630	-114770	-133240	-146225	-151575	-153110	-153340	-156620	-161650	-169730	-178255
				50 psf	Fa	lbs	5730	4380	3675	3240	2985	2760	2540	2415	2345	2315	2300		80 psf	F	lbs	8120	6195	5185	4560	4180	3840	3490	3290	3175	3110	3080
w Load.		x x			R _x	lbs	5325	3820	2975	2420	2010	1695	1450	1265	1115	995	995			R	lbs	7540	5405	4200	3410	2820	2370	2015	1750	1540	1370	1225
Variable Sno		Þ			A,	lbs	2180	2195	2215	2220	2155	2085	2020	2010	2025	2065	2110			S,	sql	3080	3100	3120	3105	2990	2855	2725	2685	2685	2715	2760
00 ft Span,	7	road truction Los ind Zone		Radius [R]	Œ		130.00	96.29	78.44	67.82	61.08	26.67	53.76	51.89	50.76	50.17	50.00		Radius [R]	(≟		130.00	96.29	78.44	67.82	61.08	26.67	53.76	51.89	50.76	50.17	50.00
Table E-9 - 100 ft Span, Variable Snow Load	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	15 psi Dead Load 20 psf Construction Load 120 mph Wind Zone		Rise [T]	(£		10	14	18	22	56	30	34	38	42	46	20		Rise [T]	(£)		10	14	18	22	56	30	34	38	42	46	20

						₊₩	ql-ui	25570	52230	55745	59210	59795	59405	29662	62250	67315	73390	81705	102015			₊⊠	dl-ui	87675	83635	90280	96175	96985	00956	94935	96085	100630	107740	116495	130555
						M	in-lb	-38670	-41860	-48715	-55585	-60420	-63990	-67520	-71250	-77645	-85085	-94060	-107170			M	in-lb	-63365	-67235	-77235	-86865	-93280	-97095	-99780	-103020	-108950	-116110	-125770	-139430
					25 psf	F.	lbs	5165	3925	3275	2895	2645	2500	2370	2310	2275	2260	2250	2260		40 psf	F _e	lbs	6292	2000	4165	3670	3335	3135	2945	2765	2680	2635	2615	2615
	Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$				A _x	lbs	4855	3495	2730	2240	1885	1620	1420	1265	1175	1215	1260	1360			A,	lbs	6200	4460	3475	2845	2390	2040	1765	1540	1370	1225	1260	1360
	ad					R	lbs	1810	1830	1855	1885	1890	1930	1970	2020	2070	2130	2185	2265			Ry	lbs	2305	2325	2350	2385	2370	2330	2295	2265	2285	2325	2380	2460
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$				± ⊠	in-lb	44870	41760	44390	46890	47395	47655	48870	51490	56210	62605	74595	102015			→	in-lb	76975	73165	78770	83850	84590	83535	83280	84805	89525	96085	104895	117975
		= 1.0. only.		d [pg]		M	in-lb	-30435	-33425	-39335	-45185	-49485	-53220	-56770	-61160	-67435	-74745	-84795	-104925	d [p _g]		M	in-lb	-55135	-58780	-67705	-76380	-82325	-85970	-89025	-92400	-98230	-105770	-115200	-128580
		All values normalized to C_D = 1.0. Use ASD design procedure only.		Ground Snow Load [pg]	20 psf	Fa	lbs	2090	3865	3225	2850	2615	2465	2370	2310	2275	2260	2250	2260	Ground Snow Load [pg]	35 psf	Fa	lbs	6120	4640	3865	3415	3105	2925	2755	2590	2520	2480	2470	2475
	Notes:	All values no Use ASD desi		Groun		æ _x	lbs	4785	3445	2690	2205	1865	1615	1420	1265	1175	1215	1260	1360	Groun		A,	lbs	5750	4140	3230	2645	2220	1900	1645	1440	1280	1215	1260	1360
		œ* 				R	lbs	1780	1800	1825	1855	1890	1930	1970	2020	2070	2130	2185	2265			Ry	lbs	2140	2160	2185	2220	2210	2185	2155	2135	2160	2205	2260	2340
	//		۳,			±	in-lb	34170	31290	33045	34785	35440	36115	37835	40980	46955	59420	74595	102015			± Z	in-lb	66275	62700	67255	71530	72190	71470	71620	73530	78420	84425	93300	105390
		v	0			M	in-lb	-22260	-25100	-29955	-34785	-38895	-42450	-46470	-51070	-57700	-70485	-84795	-104925			M	in-lb	-46900	-50320	-58180	-65980	-71370	-74850	-78275	-81785	-87855	-95430	-104630	-117730
					15 psf	Fa	lbs	2090	3865	3225	2850	2615	2465	2370	2310	2275	2260	2250	2260		30 psf	Fa	lbs	5645	4285	3570	3155	2875	2710	2560	2420	2360	2330	2325	2335
ow Load.		Rx	r &			Α×	lbs	4785	3445	2690	2205	1865	1615	1420	1265	1175	1215	1260	1360			R _x	lbs	5305	3815	2980	2440	2055	1760	1525	1340	1190	1215	1260	1360
Variable Sn		þ				R	lbs	1780	1800	1825	1855	1890	1930	1970	2020	2070	2130	2185	2265			Ry	lbs	1975	1995	2020	2050	2050	2035	2020	2020	2070	2130	2185	2265
Table E-10 - 110 ft Span, Variable Snow Load.	7	20 psr Dead Load 20 psr Construction Load 120 mph Wind Zone			Radius [R]	£		156.25	115.04	93.03	79.75	71.17	65.42	61.49	58.80	57.01	55.88	55.25	55.00		Radius [R]	(£		156.25	115.04	93.03	79.75	71.17	65.42	61.49	58.80	57.01	55.88	55.25	55.00
Table E-10 -	7000	20 psf Construction 120 mph Wind Zone			Rise [T]	(L		10	14	18	22	56	30	34	38	42	46	20	55		Rise [T]	(L		10	14	18	22	56	30	34	38	42	46	20	55
									_														_										_		_

																																			_
						₹	in-lb	151885	146450	159340	170100	171370	168000	164880	163745	170035	177670	189430	206060			± ∑	in-lb	216095	209260	228405	244025	246105	240595	235160	232985	239820	247605	263075	281640
						M	in-lb	-112760	-117990	-134415	-150215	-159410	-163820	-166170	-167350	-173275	-180895	-190850	-204655			M	in-lb	-162160	-168740	-191590	-213565	-226025	-230735	-232625	-232790	-238560	-245925	-257170	-272555
					70 psf	F.	lps	9450	7155	5940	5220	4710	4415	4100	3790	3640	3545	3495	3465		100 psf	F.	lps	12305	9310	7720	6775	0609	2690	5260	4820	4600	4455	4375	4320
	Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$				R _x	sql	5888	0889	4962	4055	3395	2880	2475	2150	1900	1695	1525	1360			R _x	lbs	11575	8300	6455	5265	4395	3720	3185	2755	2435	2165	1945	1710
	ad					R,	lbs	3300	3320	3345	3380	3325	3230	3130	3035	3025	3050	3095	3170			R	lbs	4295	4315	4340	4370	4280	4125	3960	3800	3770	3775	3815	3885
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$				₹	dl-ni	130485	125510	136320	145455	146575	143870	141565	141190	146775	154360	164880	180890			±	ql-ui	194695	188325	205385	219380	221100	216265	211515	209880	216560	224295	238525	256395
		= 1.0. only.		d [pg]		Σ	in-lb	-96295	-101070	-115355	-129100	-137205	-141575	-144015	-145535	-151830	-159220	-168740	-182830	[⁸ d] p		M	in-lb	-145690	-151820	-172530	-192445	-203820	-208300	-210475	-210975	-216590	-224245	-235065	-249920
		All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.		Ground Snow Load [pg]	e0 psf	Fa	lbs	0058	6435	5350	4705	4250	3990	3715	3450	3320	3240	3205	3180	Ground Snow Load $[p_g]$	90 psf	Fa	lbs	11355	8590	7125	6255	2630	5265	4875	4475	4280	4150	4085	4035
	Notes:	All values no Use ASD des		Groun		٣×	sql	0662	5740	4470	3655	3060	2600	2240	1945	1725	1540	1385	1360	Groun		A,	lbs	10675	2660	2960	4865	4060	3440	2950	2555	2255	2010	1805	1590
		ě,				Υ _ν	lps	2970	2990	3015	3045	3002	2930	2850	2775	2780	2810	2860	2935			Ry	lps	3962	3985	4010	4040	3960	3825	3680	3545	3520	3535	3575	3650
	//		ď			₹	in-lb	109080	104575	113300	120815	121780	119735	118250	118640	123515	131050	140335	155725			₊Μ	in-lb	173290	167385	182365	194740	196165	192135	188200	186770	193300	200980	213980	231225
			n I			M	in-lb	-79830	-84155	-96292	-107980	-115185	-119335	-121865	-124255	-130390	-137545	-146910	-161130			M	in-lb	-129225	-134905	-153475	-171330	-181615	-186060	-188320	-189160	-194715	-202570	-212955	-227285
	//				50 psf	F _a	lbs	7545	5720	4755	4190	3790	3560	3330	3105	3000	2935	2910	2900		80 psf	Fa	lbs	10405	7870	6535	5740	5170	4840	4490	4135	3960	3850	3790	3745
now Load.		R _x	- ử			A,	lbs	5602	5100	3975	3250	2725	2320	2000	1745	1545	1380	1260	1360			R _x	lbs	08/6	7020	5460	4460	3725	3160	2710	2350	2080	1855	1665	1465
, Variable Si		aq				R,	sql	2635	2655	2685	2715	2685	2630	2575	2520	2530	2565	2620	2692			Ry	sql	0898	3650	3675	3710	3640	3525	3405	3290	3275	3295	3335	3410
Table E-10 - 110 ft Span, Variable Snow Load.	7	20 psi Dedu Load 20 psf Construction Load 120 mph Wind Zone			Radius [R]	(L)		156.25	115.04	93.03	79.75	71.17	65.42	61.49	58.80	57.01	55.88	55.25	55.00		Radius [R]	(#)		156.25	115.04	93.03	79.75	71.17	65.42	61.49	58.80	57.01	55.88	55.25	55.00
Table E-10 .	9	20 psi Dead Load 20 psf Construction 120 mph Wind Zone			Rise [T]	(£		10	14	18	22	26	30	34	38	42	46	20	55		Rise [T]	(£		10	14	18	22	56	30	34	38	42	46	20	55

120 ft Spa	Table E-11 - 120 ft Span, Variable Snow Load	snow Load.													
					//		Notes:			Snow Load	aq	Wind Load			
ŏ	20 psi Dead Load 20 psf Construction Load 120 mph Wind Zone	R×		(« I	All values no Use ASD desi	All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.	= 1.0. only.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{zt}=1.0$ $K_d=0.85$			
		-œ̂		 	, a										
							Groun	Ground Snow Load [pg]	[⁸ d] p						
Radius [R]			15 psf					20 psf					25 psf		
_	Υ _ν	٣×	Fa	_W	μ	R	۳×	Fa	M	μ	R _v	R _x	Fa	M	₹
	lps	sql	lbs	in-lb	in-lb	lps	lbs	lbs	in-lb	in-lb	lps	lbs	lps	in-lb	in-lb
	1950	4765	5130	-28090	39060	1950	4765	5130	-37985	51670	1975	4840	5205	-47910	64285
_	1965	3595	4075	-30650	36445	1965	3595	4075	-40615	48840	2000	3650	4135	-50705	61235
	1990	2885	3475	-36370	39310	1990	2885	3475	-47665	52910	2025	2930	3530	-58955	66515
	2020	2405	3105	-41550	41220	2020	2405	3105	-53940	55620	2060	2445	3160	-66325	70270
	2025	2060	2870	-46040	42005	2055	2060	2870	-58655	26360	2065	2085	2900	-71670	71105
	2095	1800	2715	-49910	42685	2095	1800	2715	-62720	56370	2095	1810	2750	-75530	70730
	2140	1600	2610	-54100	44340	2140	1600	2610	-66575	57375	2140	1600	2615	-79400	71175
_	2185	1430	2540	-58970	47345	2185	1430	2540	-71050	59875	2185	1430	2540	-83350	73235
	2235	1290	2495	-64855	51715	2235	1290	2495	-76915	64350	2235	1290	2495	-88970	77245
_	2290	1295	2470	-76085	64290	2290	1295	2470	-84435	70365	2290	1295	2470	-96655	83200
	2345	1340	2460	-90775	79500	2345	1340	2460	-93105	79500	2345	1340	2460	-105510	91275
	2405	1425	2455	-107025	101435	2405	1425	2455	-107025	101435	2405	1425	2455	-115885	101435
	2470	1475	2465	-124890	124010	2470	1475	2465	-124890	124010	2470	1475	2465	-127560	124010
							Groun	Ground Snow Load [pg]	[b]						
Radius [R]			30 psf					35 psf					40 psf		
	R,	٣×	Ľ.	_W	μ	R,	۳,	e _H	M	μ	R,	R _x	L.	M	μ
	sql	sql	lbs	in-lb	in-lb	lps	lbs	lbs	d-ni	in-lb	lps	sql	lbs	in-lb	in-lb
	2160	5285	2882	-57840	76895	2340	5730	6165	-67765	89505	2520	6175	6645	-77690	102120
	2180	3980	4515	-60830	73630	2360	4315	4890	-70960	86025	2540	4650	5270	-81085	98420
	2205	3195	3850	-70350	80265	2385	3460	4170	-81825	94055	2570	3725	4490	-93300	107850
	2240	2665	3440	-78715	84920	2420	2885	3725	-91105	99570	2600	3105	4005	-103610	114220
	2240	2275	3150	-84690	85845	2415	2460	3400	-97705	100585	2585	2645	3655	-110725	115325
	2220	1965	2985	-88595	85090	2385	2125	3220	-101825	99450	2550	2280	3455	-115060	113810
	2205	1725	2835	-92230	85035	2360	1860	3050	-105055	00686	2515	1995	3265	-118090	112760
	2195	1525	2690	02096-	86595	2335	1640	2885	-108795	99955	2480	1755	3080	-121515	113310
	2235	1360	2595	-101030	90140	2340	1465	2770	-113695	103290	2480	1565	2950	-126370	116930
	2290	1295	2555	-108870	96735	2380	1320	2725	-121090	110265	2510	1410	2895	-133585	123800
	2345	1340	2535	-117920	104495	2425	1340	2700	-130330	117840	2560	1340	2860	-142735	132000
	2405	1425	2535	-128565	114810	2485	1425	2692	-141250	128935	2615	1425	2855	-153930	143065
	2470	1475	2545	-140135	125380	2550	1475	2700	-153045	140355	2680	1475	2855	-165960	155325

| Г | | | | | 10 | 10 | 10
 | 10 | _
 | 0 | 10 | 0 | 10 | 0 | 10 | 0
 | 10 | | | | | 10
 | 10 | 10 | _ | 0 | 10 | 0
 | 10 | 10 | 10 | 10 | 10
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| | | | ±
∑ | dl-ni | 177785 | 172795 | 190595
 | 202115 | 203950
 | 200420 | 196895 | 195830 | 198745 | 205360 | 216985 | 230190
 | 245165 | | | ₹ | dl-ni | 253455
 | 247165 | 273345 | 290010 | 293350 | 287975 | 282010
 | 278845 | 280565 | 289335 | 301965 | 319315
 | 335100 |
| | | | Σ | ql-ui | -137245 | -141850 | -162145
 | -179085 | -189415
 | -194450 | -197415 | -199135 | -202415 | -210490 | -219925 | -231555
 | -243590 | | | M | ql-ui | -196800
 | -202615 | -230990 | -254560 | -268595 | -274335 | -276745
 | -277625 | -280305 | -287645 | -297880 | -311085
 | -324410 |
| | | 70 psf | т _е | lps | 9520 | 7535 | 6400
 | 2200 | 5165
 | 4870 | 4560 | 4250 | 4020 | 3910 | 3840 | 3805
 | 3780 | | 100 psf | ъ. | lps | 12390
 | 0086 | 8315 | 7390 | 0899 | 6285 | 2860
 | 5420 | 2090 | 4930 | 4815 | 4755
 | 7715 |
| $K_{zt} = 1.0$ $K_d = 0.85$ | | | R _x | lps | 8845 | 0999 | 5320
 | 4425 | 3755
 | 3225 | 2800 | 2455 | 2180 | 1955 | 1770 | 1610
 | 1475 | | | R _x | lbs | 11515
 | 8655 | 6915 | 5750 | 4870 | 4170 | 3610
 | 3155 | 2790 | 2500 | 2260 | 2050
 | 1865 |
| | | | R _v | lbs | 3605 | 3625 | 3655
 | 3685 | 3635
 | 3540 | 3440 | 3345 | 3295 | 3310 | 3345 | 3400
 | 3460 | | | R _v | lbs | 4690
 | 4710 | 4735 | 4770 | 4680 | 4525 | 4365
 | 4205 | 4115 | 4110 | 4135 | 4180
 | 7240 |
| $C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$ | | | ± | in-lb | 152565 | 148005 | 163015
 | 172815 | 174285
 | 171250 | 168520 | 168155 | 171475 | 177930 | 188655 | 200485
 | 215220 | | | ± | in-lb | 228235
 | 222375 | 245760 | 260710 | 263550 | 258790 | 253640
 | 251175 | 253290 | 261340 | 273640 | 289605
 | 305060 |
| = 1.0.
only. | d [pg] | | M | ql-ui | -117395 | -121595 | -139195
 | -153925 | -163020
 | -167985 | -170975 | -172970 | -177065 | -184855 | -193940 | -205050
 | -217610 | [gd] p | | M | ql-ui | -176950
 | -182360 | -208045 | -229400 | -242200 | -247565 | -250300
 | -251460 | -254305 | -261760 | -271895 | -284575
 | 077700- |
| rmalized to C _D
gn procedure | d Snow Load | e0 psf | Fa | lbs | 8560 | 0829 | 5765
 | 5135 | 4665
 | 4400 | 4130 | 3860 | 3665 | 3570 | 3515 | 3485
 | 3470 | d Snow Load | 90 psf | Fa | lbs | 11435
 | 9045 | 7680 | 6825 | 6175 | 5815 | 5425
 | 5030 | 4735 | 4590 | 4490 | 4440
 | 7700 |
| All values no
Use ASD desi | Groun | | ۳× | lbs | 7955 | 5985 | 4790
 | 3985 | 3385
 | 2910 | 2530 | 2225 | 1975 | 1775 | 1605 | 1460
 | 1475 | Groun | | ۳× | lbs | 10625
 | 7985 | 6385 | 5310 | 4495 | 3855 | 3340
 | 2920 | 2585 | 2320 | 2095 | 1905
 | 1735 |
| α×
I | | | R _v | lps | 3245 | 3265 | 3290
 | 3325 | 3285
 | 3210 | 3130 | 3022 | 3025 | 3045 | 3085 | 3140
 | 3200 | | | R _v | lps | 4330
 | 4350 | 4375 | 4410 | 4330 | 4200 | 4055
 | 3915 | 3840 | 3845 | 3870 | 3920
 | 3080 |
| <u>a</u> | | | −Σ | ql-ui | 127340 | 123215 | 135430
 | 143515 | 144805
 | 142530 | 140485 | 140480 | 144200 | 150865 | 160330 | 171315
 | 185275 | | | −₩ | ql-ui | 203010
 | 197585 | 218180 | 231410 | 233750 | 229605 | 225265
 | 223500 | 226020 | 233350 | 245310 | 259900
 | 275110 |
| | | | × | ql-ui | -97540 | -101340 | -116250
 | -128770 | -136755
 | -141520 | -144530 | -146960 | -151715 | -159220 | -167950 | -179295
 | -191785 | | | M | in-lb | -157095
 | -162105 | -185095 | -204240 | -215805 | -220910 | -223860
 | -225300 | -228305 | -236125 | -245910 | -258065
 | -270530 |
| | | 50 psf | Fa | lbs | 2600 | 6025 | 5125
 | 4570 | 4160
 | 3930 | 3692 | 3470 | 3310 | 3235 | 3185 | 3170
 | 3160 | | 80 psf | Fa | lbs | 10475
 | 8290 | 7040 | 6260 | 2670 | 5345 | 4995
 | 4640 | 4380 | 4250 | 4165 | 4120
 | 7090 |
| α
α
α | | | R _x | lbs | 2902 | 5315 | 4255
 | 3545 | 3015
 | 2595 | 2260 | 1990 | 1770 | 1595 | 1440 | 1425
 | 1475 | | | R | lps | 9735
 | 7320 | 5855 | 4865 | 4125 | 3540 | 3070
 | 2690 | 2380 | 2140 | 1930 | 1755
 | 1600 |
| | | | S, | sql | 2880 | 2905 | 2930
 | 2960 | 2935
 | 2880 | 2820 | 2770 | 2750 | 2780 | 2820 | 2875
 | 2940 | | | S, | sql | 3962
 | 3985 | 4015 | 4045 | 3985 | 3870 | 3745
 | 3630 | 3570 | 3580 | 3610 | 3660
 | 3720 |
| l Load
truction Loi
ind Zone | | Radius [R] | £) | | 156.00 | 120.50 | 100.00
 | 87.00 | 78.29
 | 72.25 | 00.89 | 65.00 | 62.91 | 61.50 | 60.62 | 60.14
 | 00.09 | | Radius [R] | (£ | | 156.00
 | 120.50 | 100.00 | 87.00 | 78.29 | 72.25 | 00.89
 | 65.00 | 62.91 | 61.50 | 60.62 | 60.14
 | 60.00 |
| 20 psf Deac
20 psf Cons
120 mph W | | F | (£) | | 12 | 16 | 20
 | 24 | 28
 | 32 | 36 | 40 | 44 | 48 | 52 | 99
 | 09 | | Rise [T] | (#) | | 12
 | 16 | 20 | 24 | 28 | 32 | 36
 | 40 | 44 | 48 | 52 | 26
 | 9 |
| | All values normalized to C_D = 1.0. C_e = 0.9 Use ASD design procedure only. C_q = 1.2 C_p = 1.10 C_p = 1.10 | $R_{s} = \frac{1}{R_{s}} = \frac{R}{R_{s}} $ All values normalized to $C_{0} = 1.0$. $C_{e} = 0.9$ Use ASD design procedure only. $C_{1} = 1.2$ $C_{2} = 1.2$ $C_{3} = 1.2$ $C_{4} = 1.10$ $C_{5} = 1.2$ $C_{5} = 1.2$ $C_{7} = 1.2$ $C_{$ | $R_{x} = \frac{1.0}{R_{y}}$ All values normalized to C ₀ = 1.0. C _e = 0.9 $R_{x} = 1.0$ Use ASD design procedure only. $C_{y} = 1.2$ $R_{y} = 0.85$ $E_{y} = $ | All values normalized to C ₀ = 1.0. C _e = 0.9 K _n = 1.0 K _n = 1.0 K _n = 0.85 K _n = | R _x | R ₁ | R _v R _v | R _v R _v | R ₁ R ₂ R ₃ R ₄ R ₅ R ₅ | R, R, R, R, R, R, R, R, | R ₁ R ₂ R ₃ R ₄ R ₄ | Ry Rx Fx | Ry R | R | Ry Rx F ₃ MT MT Ry Rx F ₃ MT Ry Rx Rx Rx Rx Rx Rx Rx | R _i F _i | R, R, R, R, R, R, R, R, | R | R, R, R, R, R, F, M, M, R, | Charle Part Part | R | R ₁ R ₁ R ₂ R ₃ R ₄ R ₄ | R, R | Richard Rich | Recording to Chief Part Part | R. R. R. R. R. R. R. R. | Fr. Fr. | R ₁ R ₁ R ₂ R ₃ R ₄ R ₄ | R ₁ R ₂ R ₃ R ₄ R ₄ | R, R | R | Part Part | Part Part | R. R. R. R. R. R. R. R. |

se normalized to C _p = 1.0. design procedure only. 20 psf 20 psf 1 ps 20 psf 20 psf 1 ps 20 psf 20	30 ft Spaı	Table E-12 - 130 ft Span, Variable Snow Load	now Load.													
Part	g	7						Notes: All values nor	rmalized to $C_{\scriptscriptstyle m D}$, = 1.0.	Snow Lo	pe	$\frac{\text{Wind Load}}{K_{\text{zt}} = 1.0}$			
15 psf N		ı	Î			a di di	ě.	Use ASD des	ign procedure	only.	$C_{\rm t} = 1.2$ $I_{\rm s} = 1.10$		K _d = 0.85			
15 pst 1	-							Ground	d Snow Loa	d [p _g]						
R _k F _b M ^k R _k F _b M ^k R _k F _b M ^k M ^k R _k M ^k R _k M ^k R _k M ^k M ^k R _k H _b Inch lab H _b	<u> </u>			15 psf					20 psf					25 psf		
Hory		ď	٣×	L.	Ē	₹	A,	~×	F.	Σ	₹	R,	~×	F _a	M	±
5585 5960 -32720 46115 2105 5956 -44350 60940 2140 5675 6094 -5595 60940 2140 5675 6094 -5595 60940 2140 5675 -5597 60940 2140 6678 -6580 4480 2142 3380 3395 -3095 4483 2142 3380 3380 60190 63965 2125 3497 -5008 66280 2180 3483 -7078 66280 2210 2870 3580 -7078 66890 2216 2180 -66880 -66880 -7078 66900 2218 2440 3880 -7078 66890 2218 7418 2800 -7078 2800 -7078 2800 -7078 2800 -7078 2800 -7078 2800 -7078 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010 -8010		lbs	lps	lps	in-lb	d-ni	lps	lbs	lps	ql-ui	in-lb	lbs	sql	lps	in-lb	ql-ui
4215 4695 -33475 43290 2125 4215 4695 -3347 4215 4695 -3347 43200 2125 4215 66050 2128 4335 4480 -58050 3380 3387 -46139 2173 2826 2820 2130 2375 -59080 -5808 4930 2174 2820 -4880 -5880 4930 2242 2820 2240 2880 2890 2245 2420 3880 2242 2820 2420 3880 2242 2820 2420 3880 2242 2420 2820 2420 2420 2420 2420 2702 6620 2245 1240 3835 2420 1240 3836 2420 1240 2820 2420 1240 2820 2420 1240 2820 2420 1240 2820 2420 1240 2820 2420 1240 2820 2420 1240 2820 2420 1240 2820 2420 <td></td> <td>2105</td> <td>5295</td> <td>2960</td> <td>-32720</td> <td>46115</td> <td>2105</td> <td>5295</td> <td>2960</td> <td>-44350</td> <td>60940</td> <td>2140</td> <td>5675</td> <td>6045</td> <td>-55975</td> <td>75770</td>		2105	5295	2960	-32720	46115	2105	5295	2960	-44350	60940	2140	5675	6045	-55975	75770
3380 3375 40390 44680 2145 3380 3395 53085 601050 2180 3435 4040 -65820 2282 3330 -64735 2205 2242 2325 65285 65285 65285 2245 2245 2325 2420 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2325 2425 2328 2426 2328 2328 2420 2328 2420 2328 2425 2328 2425 2328 2425 2328 2426 2328 2328 2425 2328 2328 2425 2425 24	_	2125	4215	4695	-35475	43290	2125	4215	4695	-47200	57885	2155	4280	4770	-59050	72480
1,242, 1,243, 1,244, 1		2145	3380	3975	-40390	44680	2145	3380	3975	-53085	60050	2180	3435	4040	-65820	75640
420 3235 5-1460 48835 2205 2420 65285 65	_	2175	2825	3530	-46195	47230	2175	2825	3530	-60190	63905	2210	2870	3585	-74185	80585
1220 3035 55880 49335 2245 2120 3035 7.0785 66090 2245 2135 3075 85990 85990 1880 2395 85990 1880 2395 890150 1880 2395 890150 1880 2395 890150 1880 2395 890150 1880 2395 890150 1880 2395 890150 1880 2395 890150 1880 2395 890150 1880 2395 890150 1880 2395 890150 1890 2395 2445		2205	2420	3235	-51460	48835	2205	2420	3235	-66285	65925	2235	2460	3280	-81265	83015
1880 2900 -60015 50365 2285 1880 2900 -75080 66510 2285 1880 2995 -99150 1685 2805 -99260 1585 2805 -99260 -99260 1885 2805 -99260 -99260 1885 2805 -99260 -99260 1885 2805 -99260 -99260 1885 2805 -99260 -99260 1880 2992 29920 -99260 -99	_	2245	2120	3035	-55880	49335	2245	2120	3035	-70785	06099	2245	2135	3075	-85990	82845
1865 2805 -64785 52550 2330 1685 2805 -79230 67935 2330 1685 2805 -94260 2425 1320 24260 -94260 2425 -70145 56160 2425 1320 -91252 2480		2285	1880	2900	-60015	50365	2285	1880	2900	-75080	66510	2285	1880	2935	-90150	82650
1525 2745 -70145 56160 2375 1525 2745 -84290 70710 2375 1525 2745 -98550 1390 2700 -76775 62860 2425 1390 2700 -90950 75940 2425 1390 2700 -105120 14400 26675 -84832 2480 4440 2665 -102570 117630 2560 1230 2560 117630 2560 1230 2560 117630 2560 1230 2560 117630 2560 2	_	2330	1685	2805	-64785	52550	2330	1685	2805	-79230	67935	2330	1685	2805	-94260	83530
1390 2700 -76775 62860 2425 1390 2700 -90950 75940 2425 1390 2700 -105120 1440 2655 -105170 93635 2540 1440 2665 -108450 23635 2480 1400 2675 -113465 1450 2665 -105170 93635 2540 1440 2665 -108450 23635 2540 1440 2665 -123900 1530 2660 -122570 117630 2600 12530 217630 117630 2600 1330 2660 -134090 1530 R _k F _a M		2375	1525	2745	-70145	56160	2375	1525	2745	-84290	70710	2375	1525	2745	-98550	86250
1400 2675 -89320 76950 2480 1400 2675 -99125 28230 2480 1440 2665 -108450 2665 -108450 2665 -102570 117630 2660 117630 2660 117630 2660 117630 2660 117630 2660 117630 2660 2660 2122570 2660 2122570 2660 2122570 2660 2122570 2660 2122570 2122570 2660 2122570 2660 2122570 2660 2660 2660 2122570 2660 26	_	2425	1390	2700	-76775	62860	2425	1390	2700	-90950	75940	2425	1390	2700	-105120	91130
1440 2665 -105170 93635 2540 1440 2665 -1024570 117630 2660 117630 2660 1225570 117630 2660 117630 2660 117630 2660 117630 2660 117630 2660 117630 2660 117630 2660 117630 2660 2660 2125570 2125570 211630 2660 2125570 211630 2660 2125570 2125570 211630 2660 2125570 211630 211		2480	1400	2675	-89320	76950	2480	1400	2675	-99125	82530	2480	1400	2675	-113465	97590
Mathematical Nation 1530 2600 1530 2600 122570 117630 2600 1530 2600 1330 2600 1330 2600 1330 2600 1330 2600 1330 2600 1330 2600 1330 2600 2250 2		2540	1440	2665	-105170	93635	2540	1440	2665	-108450	93635	2540	1440	2665	-122990	106260
Ground Snow Load [Pg] Rx F _a M¹ R _y R _x F _a M¹ M² R _y R _x F _a M¹ M² M³ M³ <th< td=""><td></td><td>2600</td><td>1530</td><td>2660</td><td>-122570</td><td>117630</td><td>2600</td><td>1530</td><td>2660</td><td>-122570</td><td>117630</td><td>2600</td><td>1530</td><td>2660</td><td>-134090</td><td>117630</td></th<>		2600	1530	2660	-122570	117630	2600	1530	2660	-122570	117630	2600	1530	2660	-134090	117630
Rx F _a M ^r R _x F _a M ^r R _x F _a M ^r M ^r R _x F _a M ^r M ^r R _y R _x F _a M ^r M ^r R _y R _x F _a M ^r M ^r R _y R _x F _a M ^r M ^r R _y R _x F _a M ^r M ^r R _y R _x F _a M ^r M ^r R _y R _x F _a M ^r M ^r R _y R _x R _a M ^r R _y R								Groune	d Snow Loa	d [p _g]						
Rx F _a M¹ M² Rx F _a M³ M³ M³ M³ M³ M³ Rx F _a M³ <	-			30 psf					35 psf					40 psf		
lbs in-lb in-lb lbs lbs in-lb lbs lbs in-lb lbs	•	Α,	٣×	T.	Ē	±	A,	æ,	Ľ.	Ē	±	S,	~×	F.	Ā	₹
6200 6605 -67605 90595 2530 6720 7160 -79235 105420 2725 7715 -90860 4670 5205 -70900 87075 2550 5065 5640 -82745 101670 2745 5455 6075 -94595 3745 4405 -78720 91225 2575 4060 4770 -91625 106815 2770 4375 5140 -104255 2480 3750 -96240 100335 2655 2900 3865 -111215 117800 2820 3120 4155 -12550 2480 3370 -101405 100025 2605 2515 3420 -116820 2770 3875 -12550 2445 3180 -101405 100025 2605 2515 3420 -116820 2790 3875 -13230 2445 3180 -102445 90405 2575 2205 3430 -12030 2770 3875 -13630 </td <td></td> <td>sql</td> <td>sql</td> <td>lps</td> <td>in-lb</td> <td>d-ni</td> <td>lbs</td> <td>lbs</td> <td>lbs</td> <td>in-lb</td> <td>in-lb</td> <td>lbs</td> <td>sql</td> <td>sql</td> <td>in-lb</td> <td>in-lb</td>		sql	sql	lps	in-lb	d-ni	lbs	lbs	lbs	in-lb	in-lb	lbs	sql	sql	in-lb	in-lb
4670 5205 -70900 87075 2550 5665 5640 -82745 101670 2745 5455 6075 -94595 3745 4405 -78720 91225 2575 4060 4770 -91625 106815 2770 4375 5140 -104525 3125 3910 -88185 97260 2605 3885 -111215 113940 2800 3645 4550 -116550 2680 3570 -96240 10033 2625 2900 3865 -111215 117800 2820 3120 4155 -126195 2045 3340 -101405 100025 2615 3610 -116820 1750 2700 3875 -132330 2045 3180 -109295 99405 2575 2205 3430 -124325 11665 2715 2095 3490 -138360 1620 2890 -113440 101955 2530 1745 3100 -124325 2680	┝	2335	6200	9099	-67605	90595	2530	6720	7160	-79235	105420	2725	7245	7715	09806-	120245
3745 4405 -78720 91225 2575 4060 4770 -91625 106815 2770 4375 5140 -104525 3125 3910 -88185 97260 2665 3385 4230 -102330 113940 2800 3645 4550 -116550 2680 3570 -96240 10033 2625 2900 3865 -111215 117800 2820 3120 4155 -126195 2045 3340 -101405 2605 2515 3610 -116820 117310 2790 2700 3875 -132330 2045 3180 -105245 99405 2575 2205 3430 -16230 2700 3875 -133230 1620 2890 -113440 101955 2530 1745 3100 -124325 11665 2715 2095 3490 -139320 1620 2890 -113440 101955 2530 1745 3100 117655 2680		2355	4670	5205	-70900	87075	2550	2905	5640	-82745	101670	2745	5455	6075	-94595	116265
3125 3910 -88185 97260 2605 3385 4230 -102330 113940 2800 3645 4550 -116550 2680 3570 -96240 100335 2625 2900 3865 -111215 117800 2820 3120 4155 -126195 2325 3340 -101405 100025 2605 2515 3610 116230 2790 2700 3875 -132230 2045 3180 -105245 99405 2575 2205 3430 116230 2770 2370 3680 -136350 1815 3030 -109295 99800 2550 1955 3260 -124325 2680 1865 3490 -139360 1620 2890 -113440 101955 2530 1745 3100 117655 2680 1865 3305 -143225 4465 2805 -119290 1560 2540 1575 3000 1134100 121855 2685		2380	3745	4405	-78720	91225	2575	4060	4770	-91625	106815	2770	4375	5140	-104525	122400
2680 3570 -96240 100335 2625 2900 3865 -111215 117800 2820 3120 4155 -126195 2325 3340 -101405 100025 2605 2515 3610 -116820 117310 2790 2700 3875 -132330 2045 3180 -105245 99405 2575 2205 3430 -16230 2750 2370 3680 -136350 1815 3030 -109295 99800 2550 1955 3260 -124325 116665 2715 2095 3490 -139360 1620 2890 -113440 101955 2530 1745 3100 117655 2680 1865 3305 -143225 1465 2805 -119290 106320 2540 1575 3000 -134100 121555 2685 1680 3190 -148990		2410	3125	3910	-88185	97260	2605	3385	4230	-102330	113940	2800	3645	4550	-116550	130620
2325 3340 -101405 100025 2605 2515 3610 -116820 117310 2790 2700 3875 -132230 2045 3180 -105245 99405 2575 2205 3430 -120800 116230 2750 2370 3680 -136350 1815 3030 -109295 99800 2550 1955 3260 -124325 116665 2715 2095 3490 -139360 1620 2890 -113440 101955 2530 1745 3100 -128330 117655 2680 1865 3305 -143225 1465 2805 -119290 106320 2540 1575 3000 -134100 121555 2685 1680 3190 -148990		2430	2680	3570	-96240	100335	2625	2900	3865	-111215	117800	2820	3120	4155	-126195	135265
2045 3180 -105245 99405 2575 2205 3430 -120800 116230 2750 2370 3680 -136350 1815 3030 -109295 99800 2550 1955 3260 -124325 116665 2715 2095 3490 -139360 1620 2890 -113440 101955 2530 1745 3100 -128330 117655 2680 1865 3305 -143225 1465 2805 -119290 106320 2540 1575 3000 -134100 121555 2685 1680 3190 -148990		2420	2325	3340	-101405	100025	2605	2515	3610	-116820	117310	2790	2700	3875	-132230	134595
1815 3030 -109295 99800 2550 1955 3260 -124325 116065 2715 2095 3490 -139360 1620 2890 -113440 101955 2530 1745 3100 -128330 117655 2680 1865 3305 -143225 1465 2805 -119290 106320 2540 1575 3000 -134100 121555 2685 1680 3190 -148990		2400	2045	3180	-105245	99405	2575	2205	3430	-120800	116230	2750	2370	3680	-136350	133055
1620 2890 -113440 101955 2530 1745 3100 -128330 117655 2680 1865 3305 -143225 1465 2805 -119290 106320 2540 1575 3000 -134100 121555 2685 1680 3190 -148990		2385	1815	3030	-109295	00866	2550	1955	3260	-124325	116065	2715	2095	3490	-139360	132335
1465 2805 -119290 106320 2540 1575 3000 -134100 121555 2685 1680 3190 -148990		2375	1620	2890	-113440	101955	2530	1745	3100	-128330	117655	2680	1865	3305	-143225	133355
		2425	1465	2805	-119290	106320	2540	1575	3000	-134100	121555	2685	1680	3190	-148990	137595
		2540	1440	2745	-137535	121695	2625	1440	2925	-152080	137520	2765	1440	3100	-166620	154080
1440 2745 -137535 121695 2625 1440 2925 -1352080 137520 2765 1440 3100 -166620		2600	1530	2745	-148930	132825	2685	1530	2920	-163770	149245	2825	1530	3090	-178615	165665

Table E-12 - 130 ft Span, Variable Snow Load. 20 usf Dead Load	Variable Snow Load.	tow Load.			//			Notes:			Snow Load	g	Wind Load			
ons h W	20 psf Construction Load		, x		S		Ä,	All values no Use ASD des	All values normalized to C _p = 1.0. Use ASD design procedure only.	only.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{zt}=1.0 \label{eq:K_zt}$ $K_d=0.85 \label{eq:K_zt}$			
								Groun	Ground Snow Load [pg]	d [p _g]						
Rise [T]	Radius [R]			50 psf					e0 psf					70 psf		
(£	Œ	Α,	~×	Fa	W	±	R,	۳,	F _a	Σ	±	R _v	۳,	F _a	Ē	±
		sql	lbs	lbs	in-lb	q-ui	lbs	lbs	lbs	in-lb	in-lb	lps	lps	lps	in-lb	in-lb
12	182.04	3115	8290	8830	-114115	149900	3510	9335	9945	-137370	179550	3900	10385	11060	-160625	209205
16	140.03	3135	6240	0569	-118290	145455	3530	7025	7825	-141990	174645	3920	7810	8695	-165685	203835
20	115.63	3160	2000	5870	-130325	153575	3555	5625	9099	-156130	184750	3945	6250	7340	-181930	215925
24	100.02	3190	4165	5195	-144985	164030	3585	4685	5840	-173420	197675	3975	5200	6485	-201855	231325
∞	89.45	3205	3560	4735	-156460	170200	3595	4000	5315	-186860	205130	3980	4440	5895	-217255	240065
32	82.02	3160	3080	4410	-163060	169160	3530	3455	4945	-193890	203730	3900	3835	5475	-225000	238300
9	26.68	3100	2692	4180	-167460	166705	3450	3020	4675	-198570	200360	3800	3345	5175	-229680	234010
40	72.81	3045	2375	3945	-170310	164865	3370	2660	4405	-201285	197400	3700	2940	4860	-232260	229935
4	70.01	2990	2115	3720	-173010	164755	3295	2360	4135	-203220	196750	3605	2610	4550	-233835	229130
48	68.01	2980	1900	3575	-178775	169670	3275	2120	3960	-208565	201745	3570	2340	4345	-238350	233820
52	66.63	3010	1725	3500	-186910	176980	3300	1920	3870	-217000	208735	3585	2120	4240	-247090	240930
99	65.72	3050	1575	3455	-196280	187200	3335	1750	3810	-226740	220315	3620	1930	4165	-257205	253435
0	65.21	3105	1530	3435	-208295	198545	3390	1605	3780	-238660	233175	3675	1765	4125	-269690	267805
								Groun	Ground Snow Load [pg]	[gd] p						
Rise [T]	Radius [R]			80 psf					90 pst					100 psf		
(±)	£)	&^	٣×	L	Σ	± ≥	S,	××	L e	Έ	± ≥	Υ _ν	×	L e	Έ	±
		lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
12	182.04	4290	11430	12170	-183880	238885	4685	12475	13285	-207135	268605	5075	13520	14400	-230390	298325
16	140.03	4310	8595	9570	-189385	233020	4705	9380	10440	-213080	262210	2092	10165	11315	-236775	291400
20	115.63	4335	6875	8070	-207735	247100	4730	7500	8805	-233535	278275	5120	8125	9540	-259340	309450
4	100.02	4365	5720	7130	-230290	264970	4760	6240	7775	-258720	298615	5150	0929	8420	-287155	332260
28	89.45	4370	4880	6475	-247655	274995	4760	5320	7055	-278055	309930	5145	2260	7635	-308450	344860
32	82.02	4270	4210	6010	-256230	272865	4640	4585	6545	-287465	307435	5010	4965	7075	-318695	342005
36	26.68	4150	3670	5675	-260785	267660	4500	3995	6170	-291895	301310	4850	4320	0299	-323100	334960
요	72.81	4030	3225	5320	-263235	262470	4360	3505	5780	-294210	295405	4685	3790	6235	-325185	328480
44	70.01	3915	2855	4962	-264450	261505	4220	3100	2380	-295065	293885	4530	3350	5795	-325680	326260
48	68.01	3865	2560	4730	-268650	262895	4160	2780	5115	-299195	297975	4455	2995	2200	-329740	330050
52	66.63	3875	2315	4605	-277185	273770	4165	2515	4975	-307275	306615	4455	2710	5340	-337665	339455
9	65.72	3905	2110	4515	-287665	286555	4190	2285	4870	-318125	319670	4475	2465	5225	-348585	352790
00	65.21	3955	1930	4470	-300725	302435	4240	2090	4815	-331755	337065	4520	2255	5160	-362785	371690

					±	ql-ui	88170	84615	84890	08606	95275	96460	96245	96300	97340	100345	106175	113130	122395			[‡] ∑	in-lb	139820	135520	137090	147680	155015	156440	155050	153735	153475	155045	159955	168365	177870
					M	ql-ui	-64690	-68065	-72930	-82225	-90620	-96920	-101550	-106085	-110425	-115060	-122615	-131625	-141815			Ā	dl-ni	-105085	-109185	-116015	-129760	-141470	-149865	-155530	-159455	-162665	-166735	-173480	-181915	-192360
				25 psf	Fa	sql	0569	5450	4585	4045	3685	3415	3255	3120	3005	2945	2905	2880	2870		40 psf	T.	lbs	8875	6950	5840	5140	4675	4310	4100	3905	3715	3535	3430	3375	3340
	Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$			R _x	sql	0859	4960	3980	3325	2855	2490	2195	1960	1775	1620	1485	1500	1540			×	sql	8400	6325	5070	4230	3625	3155	2775	2460	2200	1980	1800	1645	1540
	p e				R	lbs	2300	2315	2340	2370	2400	2410	2430	2470	2515	2565	2615	2670	2730			S.	lbs	2930	2950	2970	3000	3035	3025	2990	2950	2915	2885	2895	2930	2975
	Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			±	in-lb	70955	67645	67490	72255	75365	76470	76715	77490	79410	82700	88500	95665	109070			±	in-lb	122605	118550	119690	128715	135105	136450	135300	134255	134610	136810	141525	149955	158725
		= 1.0. only.	[⁸ d] p		M	in-lb	-51225	-54355	-58645	-66585	-73730	-79270	-84070	-88580	-93010	-98685	-106160	-114990	-124970	[bg]		Ā	in-lb	-91620	-95480	-101650	-113870	-124390	-132215	-137535	-141395	-145255	-149500	-156190	-164890	-175515
		All values normalized to C_0 = 1.0. Use ASD design procedure only.	Ground Snow Load [pg]	20 psf	Fa	lps	0589	5370	4515	3980	3625	3380	3210	3090	3002	2945	2905	2880	2870	Ground Snow Load [pg]	35 psf	T.	sql	8235	6450	5425	4775	4345	4015	3820	3645	3475	3315	3225	3175	3150
	Notes:	All values nor Use ASD desi,	Ground		٣×	lbs	6485	4890	3920	3275	2810	2455	2180	1960	1775	1620	1485	1500	1540	Ground		×	lbs	7795	5870	4710	3930	3365	2930	2580	2295	2050	1850	1680	1540	1540
		ŭ I			R,	lbs	2265	2280	2300	2330	2360	2390	2430	2470	2515	2565	2615	2670	2730			æ	lbs	2720	2740	2760	2790	2820	2820	2795	2765	2740	2720	2735	2775	2820
,	//				Ψ	ql-ui	53735	50675	50130	53535	55765	56705	57440	58845	61475	65745	75340	90885	109070			₹	d-ni	105385	101585	102290	109755	115190	116455	115775	115105	115745	118580	123850	131540	140215
					M	ql-ui	-37760	-40695	-44510	-50950	-56845	-62045	09999-	-71175	-76460	-82310	-89705	-103615	-120620			Ē	ql-ui	-78155	-81770	-87290	-97980	-107505	-114570	-119540	-123585	-127840	-132265	-139070	-148255	-158665
	//			15 psf	F _a	lbs	6850	5370	4515	3980	3625	3380	3210	3090	3002	2945	2905	2880	2870		30 psf	T.	lbs	7590	5950	2002	4410	4015	3715	3540	3380	3230	3095	3020	2980	2960
ow Load.		x x			٣×	lbs	6485	4890	3920	3275	2810	2455	2180	1960	1775	1620	1485	1500	1540			×	lbs	7185	5415	4345	3625	3110	2710	2390	2125	1905	1720	1565	1500	1540
Variable Sn					æ [≻]	sql	2265	2280	2300	2330	2360	2390	2430	2470	2515	2565	2615	2670	2730			æ	sql	2510	2525	2550	2580	2610	2615	2600	2580	2565	2565	2615	2670	2730
Table E-13 - 140 ft Span, Variable Snow Load.		ZU psr Dead Load 20 psf Construction Load 120 mph Wind Zone		Radius [R]	(£		210.17	161.13	132.50	114.08	101.50	92.56	90.98	81.25	27.68	75.04	73.12	71.75	70.83		Radius [R]	Œ		210.17	161.13	132.50	114.08	101.50	92.56	90.98	81.25	27.68	75.04	73.12	71.75	70.83
Table E-13 -		20 psr Dead Load 20 psf Construction I 120 mph Wind Zone		Rise [T]	£)		12	16	20	24	28	32	36	40	44	48	52	99	09		Rise [T]	(L		12	16	20	24	28	32	36	40	44	48	52	26	09
-	•		_																							_						_				_

140 ft Span	Table E-13 - 140 ft Span, Variable Snow Load.	now Load.													
Pro Pro C 300 Oc					//		Notes:			Snow Load	<u>ad</u>	Wind Load			
20 psi Dedu Lodu 20 psf Construction Load 120 mph Wind Zone	oad	, X				, R	All values no Use ASD des	All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.	₀ = 1.0. conlγ.	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$		$K_{zt} = 1.0$ $K_d = 0.85$			
		α _y			α×		Ground	Ground Snow Load [p.]	[b]						
Radius [R]			50 psf					e0 psf	20				70 psf		
(#)	æ²	æ	. т	M	[†] ⊠	R _v	A,	. T	M	±	A^	S,	. Т	M	±
	sql	lbs	lbs	dl-ni	in-lb	lbs	lbs	lbs	in-lb	dl-ni	lbs	lbs	sql	dl-ni	in-lb
210.17	3322	9615	10155	-132015	174260	3775	10830	11440	-158945	208805	4195	12045	12720	-185880	243350
161.13	3370	7240	7950	-136600	169460	3795	8150	8950	-164015	203400	4215	0906	9950	-191425	237335
132.50	3395	2800	6675	-144735	171890	3815	6525	7510	-173460	206990	4240	7255	8345	-202185	241490
114.08	3420	4835	5870	-161540	185605	3845	5435	0099	-193320	223535	4265	6040	7330	-225100	261460
101.50	3455	4140	5335	-175765	194840	3875	4655	2990	-210060	234665	4300	5165	9650	-244355	274495
92.56	3435	3595	4905	-185165	196565	3845	4040	5205	-220955	237110	4255	4485	6100	-256755	277655
90.9	3380	3160	4665	-191515	195040	3770	3545	5225	-227500	235030	4160	3930	5790	-263485	275020
81.25	3320	2795	4430	-195575	192685	3690	3135	4955	-231700	231640	4065	3470	5480	-267820	270590
77.68	3265	2495	4200	-198225	191200	3615	2790	4680	-234095	228930	3960	3085	5165	-269960	266655
75.04	3210	2240	3975	-201205	191510	3540	2500	4415	-235925	227975	3870	2760	4855	-271340	265045
73.12	3210	2035	3845	-208060	197225	3530	2265	4260	-242645	234495	3845	2500	4670	-277225	271765
71.75	3240	1860	3770	-216815	205185	3550	2070	4165	-251720	242010	3865	2280	4565	-286620	279340
.83	3280	1705	3725	-226825	216155	3590	1900	4105	-262120	254440	3895	2090	4485	-297410	292725
							Groun	Ground Snow Load [pg]	[bg] pu						
Radius [R]			80 psf					90 psf					100 psf		
(±)	S,	R _x	ъ	M	Ψ	R_{y}	٣×	Fa	W	₊₩	R _v	R _x	Fa	M	₊
	sql	lbs	lps	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	sql	in-lb	in-lb
210.17	4620	13260	14000	-212810	277895	2040	14470	15285	-239740	312445	5460	15685	16565	-266670	346990
161.13	4635	9970	10950	-218840	271275	2060	10885	11950	-246255	305215	5480	11795	12950	-273670	339150
132.50	4660	7980	9185	-230905	276290	2080	8705	10020	-259630	311090	5205	9435	10855	-288355	345890
114.08	4690	6645	8060	-256875	299385	5110	7245	8795	-288655	337315	5530	7850	9525	-320435	375240
101.50	4720	2680	7310	-278655	314320	5140	6195	7965	-312950	354145	2925	6710	8625	-347245	393970
92.56	4665	4925	9699	-292555	318205	5075	5370	7295	-328350	358750	5485	5815	7890	-364150	399295
90.98	4555	4310	6350	-299805	315005	4945	4695	6915	-336240	354995	5335	2080	7475	-372670	394985
81.25	4435	3805	6005	-303940	309545	4805	4140	6530	-340060	348495	5175	4480	7055	-376185	387450
77.68	4310	3380	2650	-305825	304385	4660	3675	6130	-341690	342110	5010	3970	6615	-377555	379835
75.04	4195	3025	5290	-306755	302495	4525	3285	5730	-342170	339945	4850	3545	6170	-377590	377395
73.12	4165	2735	2082	-312285	309035	4480	2970	2200	-347745	346305	4795	3205	5910	-383205	383575
71.75	4175	2495	4960	-321525	317420	4485	2705	5355	-356430	355505	4795	2920	5750	-391685	393590
70.83	4205	2285	4870	-332705	331010	4510	2480	5250	-367995	369295	4815	2670	5635	-403290	407580

			1	_																															_
				₹	in-lb	99370	96075	98280	104650	109225	110580	110275	110045	111020	114175	118620	126205	129820			±	in-lb	158295	154295	158975	169910	177770	179330	178260	176725	175955	176740	180940	188520	193215
				M	dl-ni	-76125	-79140	-85470	-95240	-104165	-110955	-115975	-120750	-125435	-130410	-136920	-146285	-151125			M	dl-ni	-122880	-126560	-135735	-150150	-162585	-171635	-177835	-182220	-185545	-190080	-195515	-204440	-208865
			25 psf	F _a	sql	6925	5620	4825	4305	3950	3680	3510	3370	3240	3175	3130	3100	3090		40 psf	F _a	sql	8835	7160	6140	5470	5010	4645	4420	4225	4035	3850	3705	3645	3615
Wind Load	$K_{zt} = 1.0$ $K_d = 0.85$			××	lbs	6490	5070	4155	3525	3060	2690	2390	2145	1950	1790	1645	1585	1605			×	lbs	8285	6465	5295	4485	3885	3410	3020	2700	2425	2195	2000	1835	1765
pe	I			R,	lbs	2465	2485	2510	2540	2570	2580	2600	2640	2685	2730	2780	2835	2860			æ	lbs	3145	3165	3190	3215	3250	3240	3205	3170	3135	3100	3090	3120	3140
Snow Load	$C_e = 0.9$ $C_t = 1.2$ $I_s = 1.10$			±	ql-ui	79730	26670	78050	83065	86375	87660	87905	88535	90155	93320	38695	105435	109775			±	in-lb	138655	134890	138740	148060	154920	156415	155295	154350	154310	155885	159780	167750	172085
	= 1.0. only.	d [pg]		Œ	ql-ui	-60540	-63335	-68820	-77175	-84770	-90730	-95825	-100670	-105410	-111155	-118120	-127260	-132025	[gd] p		M	in-lb	-107295	-110755	-118980	-131790	-142955	-151410	-157215	-161490	-165490	-170190	-175750	-184465	-189320
	All values normalized to $C_0 = 1.0$. Use ASD design procedure only.	Ground Snow Load [pg]	20 psf	F _a	lps	6820	5535	4750	4235	3885	3640	3460	3335	3240	3175	3130	3100	3090	Ground Snow Load [pg]	35 psf	F _a	lps	8200	6645	2200	2080	4655	4320	4115	3940	3770	3610	3480	3425	3405
Notes:	All values noi Use ASD desi	Ground		۳,	lbs	6400	4995	4095	3470	3010	2655	2375	2145	1950	1790	1645	1585	1605	Ground		×	lbs	2690	0009	4915	4165	3610	3170	2810	2515	2265	2050	1870	1720	1650
	č I			R,	lbs	2430	2450	2470	2495	2525	2560	2600	2640	2685	2730	2780	2835	2860			ď	lps	2920	2940	2960	2990	3025	3020	2995	2970	2945	2920	2920	2955	2970
	, i			±	ql-ui	06009	57265	57920	61480	63880	64880	65545	67020	00269	73745	81470	97280	106105			±	in-lb	119010	115485	118510	126235	132070	133495	132650	131975	132665	135030	138615	146975	150955
	ω ω			M	ql-ui	-44955	-47610	-52335	-59105	-65375	-70940	-75880	-80590	-86120	-92270	-99320	-110450	-118970			Ň	dl-ni	-91710	-94945	-102225	-113430	-123560	-131180	-136595	-140830	-145460	-150300	-155980	-165315	-170225
\\			15 psf	L _a	lbs	6820	5535	4750	4235	3885	3640	3460	3335	3240	3175	3130	3100	3090		30 psf	T.	lbs	7560	6130	5260	4695	4305	4000	3815	3655	3505	3365	3255	3210	3190
ow Load.	x x			٣×	lbs	6400	4995	4095	3470	3010	2655	2375	2145	1950	1790	1645	1585	1605			×	lbs	7090	5535	4535	3845	3335	2930	2600	2330	2100	1905	1740	1600	1605
Variable Sn				S,	sql	2430	2450	2470	2495	2525	2560	2600	2640	2685	2730	2780	2835	2860			a²	sql	2692	2710	2735	2765	2800	2800	2785	2770	2755	2740	2780	2835	2860
Table E-14 - 150 ft Span, Variable Snow Load.	20 psf Dead Load 20 psf Construction Load 120 mph Wind Zone		Radius [R]	£		207.89	165.25	138.84	121.17	108.75	99.72	93.01	96.78	84.14	81.25	79.08	77.49	76.88		Radius [R]	Œ		207.89	165.25	138.84	121.17	108.75	99.72	93.01	87.96	84.14	81.25	79.08	77.49	76.88
Table E-14 -	20 psf Dead Load 20 psf Construction I 120 mph Wind Zone		Rise [T]	(£)		14	18	22	26	30	34	38	42	46	20	54	28	09		Rise [T]	(£)		14	18	22	56	30	34	38	42	46	20	54	58	09

Table E-14 - 150 ft Span, Variable Snow Load. 20 psf Dead Load	n, Variable Snow Load.	Snow Load.		\\				Notes: All values no	Notes: All values normalized to $C_0 = 1.0$.	= 1.0.	<u>Snow Load</u> C,= 0.9	þ	Wind Load $K_{yt} = 1.0$			
20 psf Construction Load R _s R _y R _y	R, R, V	, a s	, a	, a		ī	œ. œ.	Use ASD desi	Use ASD design procedure only.	only.	$C_{\rm f} = 1.2$ $C_{\rm f} = 1.2$ $C_{\rm f} = 1.10$		$K_d = 0.85$			
								Groun	Ground Snow Load [pg]	[⁸ d] p						
Radius [R] 50 psf		50 psf	50 psf						eo psf					70 psf		
(ft) R _y R _x F _a M ⁻ M ⁻	R _x F _a M ⁻	F _a M ⁻	M		±	1	R _y	χ.	F	M	 ≠	R _y	Ж _х	Fa	M	 ±
	lbs lbs in-lb	lbs in-lb	in-lb		ql-ui		lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
3595 9485 10110 -154055	9485 10110 -154055	10110 -154055	-154055	-	197575		4050	10680	11385	-185225	236855	4500	11875	12660	-216395	276225
3615 7395 8190 -158170	7395 8190 -158170	8190 -158170	-158170		193110		4070	8325	9220	-189780	231920	4520	9255	10250	-221395	270735
3640 6055 7015 -169245	6055 7015 -169245	7015 -169245	-169245	_	199435		4090	6815	7895	-202755	239895	4545	7570	8775	-236265	280355
36/0 5120 6245 -186865	5120 6245 -186865	-186865	-186865		213615		4120	2/60	/025	-223580	25/315	45/5	6400	/800	-260300	301020
3700 4435 5715 -201980	4435 5715 -201980	5715 -201980	-201980		223465		4155	4985	6420	-241370	269160	4605	5535	7125	-280760	314860
3680 3890 5285 -212160	3890 5285 -212160	5285 -212160	-212160		225655		4120	4370	5930	-253195	272190	4560	4850	6575	-294225	318720
3630 3440 5030 -219075	3440 5030 -219075	5030 -219075	-219075	-	224195		4050	3860	5635	-260315	270125	4475	4280	6240	-301620	316060
3570 3070 4795 -223670	3070 4795 -223670	4795 -223670	-223670		221475		3970	3440	2365	-265125	266230	4375	3810	5935	-306580	310980
3515 2755 4565 -226815	2755 4565 -226815	4565 -226815	-226815		219245		3895	3080	2092	-268085	262530	4275	3410	2620	-309355	305820
3460 2485 4340 -229860	2485 4340 -229860	4340 -229860	-229860		219170		3820	2780	4825	-270550	262375	4180	3070	5315	-311455	305580
3430 2260 4155 -235050	2260 4155 -235050	4155 -235050	-235050	-	223265		3775	2520	4610	-274585	265590	4120	2785	2060	-314385	307915
3455 2075 4075	2075 4075 -244390	4075 -244390	-244390		230410		3795	2315	4510	-284345	273775	4130	2550	4945	-324300	317140
76.88 3470 1990 4040 -248935 235480	1990 4040 -248935	4040 -248935	-248935		235480		3805	2220	4465	-289010	277745	4140	2445	4890	-329080	320595
								Groun	Ground Snow Load [pg]	d [pg]						
Radius [R] 80 psf		80 psf	80 psf			ш			90 psf					100 psf		
(ft) R_{y} R_{x} F_{a} M^{\dagger} M^{\dagger}	R _x F _a M ⁻	F _a	Σ		₹		æ^	æ	ъ.	Έ	₹	₽,	æ×	T e	Σ	₹
sql	lbs lbs in-lb	lbs in-lb	in-lb		in-lb		lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
	13075 13935 -247565	13935 -247565	-247565	_	315605		5405	14270	15210	-278740	354985	5855	15465	16485	-309910	394370
165.25 4970 10185 11280 -253005 309550	10185 11280 -253005	11280 -253005	-253005		309550		5425	11115	12310	-284615	348360	5875	12045	13340	-316230	387175
4995 8330 9650 -269775	8330 9650 -269775	9650 -269775	-269775		320815		5450	0606	10530	-303285	361275	2900	9850	11405	-336795	401735
5025 7040 8575 -297015	7040 8575 -297015	8575 -297015	-297015		344720		5475	7680	9350	-333730	388420	5930	8320	10130	-370450	432125
108.75 5055 6090 7830 -320150 360555	6090 7830 -320150	7830 -320150	-320150		360555		5510	6640	8535	-359540	406250	2960	7190	9240	-398930	451950
99.72 5000 5330 7215 -335255 365255	5330 7215 -335255	7215 -335255	-335255		365255		5445	5810	7860	-376285	411785	5885	6290	8505	-417315	458315
4895 4700 6850 -343385 361990	4700 6850 -343385 361990	6850 -343385 361990	-343385 361990	361990			5315	5120	7455	-385145	407925	5740	5540	8065	-426905	453855
87.96 4775 4180 6505 -348030 355730	4180 6505 -348030	6505 -348030	-348030		355730		5175	4550	7080	-389485	400480	5580	4915	7650	-430940	445230
84.14 4655 3735 6150 -350625 349310	3735 6150 -350625	6150 -350625	-350625		349310		5035	4065	0899	-391895	393355	5415	4390	7210	-433165	437405
81.25 4535 3360 5800 -352365 348785	3360 5800 -352365	5800 -352365	-352365		348785		4895	3655	6290	-393270	391990	5255	3945	6775	-434175	435195
79.08 4460 3045 5510 -354955 350240	3045 5510 -354955	5510 -354955	-354955		350240		4805	3305	2960	-395525	392565	5145	3570	6415	-436090	434890
4465 2790 5375 -364250	2790 5375 -364250	5375 -364250	-364250		360510		4800	3025	5810	-404695	403875	5140	3265	6245	-445620	447240
2670 5315 -369150	4470 2670 5315 -369150 364305	5315 -369150	-369150		364305		4805	2900	5740	-409220	408020	5140	3125	6160	-449705	451730

Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Notes: Table E-15 - 20 ft Span, Variable Wind Load. 10 psf Dead Load 20 psf Construction Load No Ground Snow Load

		±	in-lb	3998	2025	1410	1210	1220	1360	1595	1915	2460			± ∑	in-lb	3992	2025	1410	1210	1220	1445	2065	2770	3765
		M	in-lb	-405	-285	-295	-355	-680	-1070	-1500	-1985	-2525			M	in-lb	-855	-550	-525	-595	-680	-1070	-1500	-1985	-2570
	115 mph	Fa	lbs	510	425	370	335	315	305	295	290	290		140 mph	Fa	lbs	510	425	370	335	315	302	295	290	290
		α×	lbs	450	350	280	230	190	165	145	150	165			R	lps	450	350	280	230	190	165	140	135	150
		R	lbs	245	245	250	255	260	265	275	280	290			R	lps	245	245	250	255	260	265	275	280	290
		±	in-lb	3665	2025	1410	1210	1220	1360	1595	1915	2310			± ⊠	in-lb	3665	2025	1410	1210	1220	1360	1760	2355	3210
[N]		M	in-lb	-325	-245	-255	-335	-680	-1070	-1500	-1985	-2525	[N]		M	in-lb	099-	-435	-425	-495	-680	-1070	-1500	-1985	-2525
Basic Wind Speed [V]	110 mph	Fa	lbs	510	425	370	335	315	305	295	290	290	Basic Wind Speed [V]	130 mph	Fa	lbs	510	425	370	335	315	305	295	290	290
Basic		R _x	lbs	450	350	280	230	190	165	150	155	165	Basic		R _x	lbs	450	350	280	230	190	165	140	140	155
		R_{y}	lbs	242	245	250	255	260	265	275	280	290			R_{y}	lbs	242	245	250	255	260	265	275	280	290
		₊W	in-lb	3998	2025	1410	1210	1220	1360	1595	1915	2310			₊Μ	in-lb	3998	2025	1410	1210	1220	1360	1595	1975	2700
		M	in-lb	-255	-200	-215	-335	-680	-1070	-1500	-1985	-2525			M	in-lb	-485	-335	-335	-400	-680	-1070	-1500	-1985	-2525
	105 mph	F _a	lbs	510	425	370	335	315	305	295	290	290		120 mph	Fa	lps	510	425	370	335	315	302	295	290	290
		Α,	lbs	450	350	280	230	190	165	150	160	170			Α, ×	lbs	450	350	280	230	190	165	145	150	160
		A,	lbs	245	245	250	255	260	265	275	280	290			A _v	lps	245	245	250	255	260	265	275	280	290
	Radius [R]	(ft)		26.00	18.17	14.50	12.50	11.33	10.64	10.25	10.06	10.00		Radius [R]	(#)		26.00	18.17	14.50	12.50	11.33	10.64	10.25	10.06	10.00
	Rise [T]	(#)		2	3	4	2	9	7	∞	6	10		Rise [T]	(£)		2	33	4	2	9	7	8	6	10

Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Notes: Table E-15 - 20 ft Span, Variable Wind Load. 10 psf Dead Load 20 psf Construction Load No Ground Snow Load

		± ⊠	in-lb	3665	2025	1410	1210	1550	2175	3120	4190	5675			₹	ql-ui	3665	2025	1410	1655	2165	3060	4380	5885	7975
		M	in-lb	-1520	-940	-870	-945	-1060	-1350	-1880	-2475	-3465			Ē	in-lb	-2315	-1410	-1280	-1370	-1520	-1915	-2655	-3465	7070
	170 mph	Fa	sql	510	425	370	332	315	302	295	290	290		200 mph	ъ.	sql	510	425	370	335	315	302	295	290	290
		A,	lps	450	320	280	230	190	165	140	125	135			ď	lbs	450	320	280	230	190	165	145	170	195
		R	lps	245	245	250	255	260	265	275	280	290			S,	lbs	245	245	250	255	260	265	275	280	290
		₹	in-lb	3992	2025	1410	1210	1365	1915	2745	3685	4995			₹	ql-ui	3992	2025	1410	1495	1950	2750	3932	5290	7170
[/] pa		Z	in-lb	-1285	-800	-750	-820	-925	-1180	-1650	-2180	-3055	[N] pa		Σ	ql-ui	-2035	-1245	-1135	-1220	-1360	-1715	-2385	-3120	-4355
Basic Wind Speed [V]	160 mph	F _a	lps	510	425	370	335	315	305	295	290	290	Basic Wind Speed [V]	190 mph	F.	lbs	510	425	370	335	315	305	295	290	290
Basi		R _×	lps	450	350	280	230	190	165	140	125	135	Basi		٣×	lbs	450	320	280	230	190	165	140	150	175
		R	lps	245	245	250	255	260	265	275	280	290			Υ _ν	lbs	245	245	250	255	260	265	275	280	290
		±	in-lb	3998	2025	1410	1210	1220	1675	2395	3210	4355			₹	in-lb	3992	2025	1410	1340	1745	2455	3515	4725	6400
		Z	in-lb	-1060	-670	-635	-705	-800	-1070	-1500	-1985	-2725			Σ	ql-ui	-1770	-1090	-1000	-1080	-1205	-1525	-2125	-2790	-3900
	150 mph	T.	lps	510	425	370	335	315	302	295	290	290		180 mph	ъ.	lps	510	425	370	335	315	302	295	290	290
		R _x	lps	450	350	280	230	190	165	140	125	145			ď	lps	450	350	280	230	190	165	140	135	155
		R	sql	245	245	250	255	260	265	275	280	290			&`	sql	245	245	250	255	260	265	275	280	290
	Radius [R]	(#)		26.00	18.17	14.50	12.50	11.33	10.64	10.25	10.06	10.00		Radius [R]	(£)		26.00	18.17	14.50	12.50	11.33	10.64	10.25	10.06	10.00
	Rise [T]	(£)		2	c	4	2	9	7	∞	6	10		Rise [T]	(£)		2	3	4	2	9	7	∞	6	10

Wind Load $K_{zt}=1.0 \\ K_d=0.85$ All values normalized to C_D = 1.0. Use ASD design procedure only. Notes: Table E-16 - 30 ft Span, Variable Wind Load. 10 psf Dead Load 20 psf Construction Load No Ground Snow Load Rise (f

_							_	_		_					_					_				_			_						
		·Μ	in-lb	4305	2765	2095	1830	1790	1895	2110	2410	2785	3230	3745	4605	5540			₊ W	in-lb	4305	2765	2095	1830	1790	2085	2395	2980	3835	4690	2692	2060	8490
		M	in-lb	-580	-480	-440	-565	-850	-1285	-1775	-2305	-2885	-3515	-4205	-4955	-5765			M	in-lb	-1120	-850	-730	-945	-1030	-1285	-1775	-2305	-2885	-3515	-4205	-4955	-5825
	115 mph	Fa	lbs	865	720	630	265	525	495	475	460	450	445	440	435	430		140 mph	Fa	lbs	865	720	630	292	525	495	475	460	450	445	440	435	430
		Α _x	lbs	785	625	515	435	375	330	290	260	235	220	225	235	245			Α×	lbs	785	625	515	435	375	330	290	260	235	215	200	215	225
		Ry	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430			Ry	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430
		±	in-lb	4305	2765	2095	1830	1790	1895	2110	2410	2785	3230	3745	4330	5015			±	in-lb	4305	2765	2095	1830	1790	1895	2110	2540	3265	3990	4845	6020	7240
[N]		M	in-lb	-490	-415	-390	-500	-850	-1285	-1775	-2305	-2885	-3515	-4205	-4955	-5765	[N]		M	in-lb	068-	-695	-605	-785	-860	-1285	-1775	-2305	-2885	-3515	-4205	-4955	-5765
Basic Wind Speed [V]	110 mph	Fa	lbs	865	720	630	292	525	495	475	460	450	445	440	435	430	Basic Wind Speed	130 mph	Fa	lbs	988	720	630	292	525	495	475	460	450	445	440	435	430
Basic		R _x	lbs	282	625	515	435	375	330	290	260	235	225	230	240	250	Basic		R _x	lbs	785	625	515	435	375	330	290	260	235	215	210	225	230
		R_{y}	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430			R _y	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430
		Ψ	in-lb	4305	2765	2095	1830	1790	1895	2110	2410	2785	3230	3745	4330	4975			Ψ	in-lb	4305	2765	2095	1830	1790	1895	2110	2410	2785	3345	4055	2060	6085
		M	in-lb	-405	-355	-340	-440	-850	-1285	-1775	-2305	-2885	-3515	-4205	-4955	-5765			M	in-lb	089-	-545	-490	-635	-850	-1285	-1775	-2305	-2885	-3515	-4205	-4955	-5765
	105 mph	T.	lbs	865	720	630	292	525	495	475	460	450	445	440	435	430		120 mph	F _a	lbs	865	720	630	292	525	495	475	460	450	445	440	435	430
		Α, ×	lbs	785	625	515	435	375	330	290	260	235	225	235	245	250			R _×	lbs	785	625	515	435	375	330	290	260	235	215	220	235	240
		R,	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430			R _v	lbs	365	370	370	375	380	385	390	395	400	410	415	425	430
	Radius [R]	£		39.00	30.13	25.00	21.75	19.57	18.06	17.00	16.25	15.73	15.38	15.15	15.04	15.00		Radius [R]	Œ		39.00	30.13	25.00	21.75	19.57	18.06	17.00	16.25	15.73	15.38	15.15	15.04	15.00
	Rise [T]	(ft)		3	4	2	9	7	∞	6	10	11	12	13	14	15		Rise [T]			3	4	2	9	7	∞	6	10	11	12	13	14	15
-			_				_		_		_		_		_	_	_			_	_	_			_		_		_				

2030 2470 3130 3615 4505 5800 7100 10660 12815 3440 4375 5075 8140 9975 18015 -10780 -1405 -1165 -1500 -1605 -1945 -2160 -2600 -3395 -4105 -4970 -6405 -2080 -1695 -2165 -2305 -2760 -3065 -3675 -4780 -5775 0969--8945 200 mph 170 mph 460 720 630 565 525 495 720 495 440 **F**_a lbs 475 450 445 440 435 430 525 475 450 445 435 430 460 435 375 330 290 260 235 215 1195 180 200 R_y lbs 365 370 370 3370 385 380 385 390 4400 4410 Ry 365 365 370 370 370 370 385 380 385 385 395 4410 4410 4425 4425 430 415 425 430 Wind Load $K_{zt} = 1.0$ $K_{d} = 0.85$ 1830 7320 2180 3180 3962 5100 6245 9375 2095 4565 3095 3935 -3610 -1700 -1210 -2305 -2980 -5650 -6815 -1510 -1935 -2060 -2745 -4295 -6255 -9705 -1010-1305 -1400 -1895 -4380 -1845 -2470 -3295 -5190 -1640 -8055 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 190 mpł 435 720 630 565 475 450 445 440 435 430 430 525 495 460 Notes: 515 435 375 330 290 260 260 235 215 195 785 200 R_x R_y 1lbs 365 340 370 370 385 385 390 395 400 410 415 425 430 365 370 370 375 385 395 400 410 415 425 430 ď I 14455 1830 2410 3455 5440 2095 2275 2775 3520 4075 6535 8005 2775 6610 8175 2095 2765 5080 -1025 -1115 -1475 -2305 -3515 -5120 -1330 -1825 -2445 -3830 -5595 -8685 -1775 -1620 -2200 -4630 -4205 -865 150 mph 180 mph 865 720 630 440 F_a lbs lbs 865 865 720 630 630 555 555 475 460 450 445 440 430 430 565 525 495 460 445 495 475 450 430 R_y lbs 365 370 370 375 380 385 390 395 400 410 415 425 430 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 16.25 15.15 39.00 21.75 18.06 15.73 25.00 17.00 15.73 15.38 15.15 15.04 15.00 25.00 17.00 30.13 21.75 19.57 18.06 16.25 £ £ 10 psf Dead Load Rise [T] Rise [T] (ft) £ 10 11 12 13 14 11 12 13 14 15 9

Table E-16 - 30 ft Span, Variable Wind Load.

-10640

5925

2430

3455 3920 4310 5070

-1085 -1570 -2115 -2692 -3995 -4720 -5500 -6340 -7235 -8195 -9215 -10300-1045 -1520 -1655 -2115 -2692 -3320 -3995 -4720 -5500 -6340 -7235 -8195 -9235 -685 -720 Έ 115 mph 140 mpł 1205 260 089 635 825 655 620 595 585 575 655 620 605 595 590 585 580 575 575 760 680 635 605 590 580 525 470 430 390 330 330 310 290 915 770 670 585 430 330 330 330 330 285 265 265 285 285 295 300 315 525 585 320 515 520 525 530 535 545 545 560 560 565 495 500 505 510 Wind Load $K_{zt} = 1.0$ $K_{d} = 0.85$ 11845 2675 3345 3345 4305 2430 4315 6130 3670 5045 3790 -10300 -6340 -1010 -3995 -5500 -2115 -3320 -3995 -4720 -5500 -7235 -8195 -9215 -1570 -2695 -3320 -4720 -6340 -7235 -9215 -1085 -2692 -885 -8195 -615 -685 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 590 575 655 620 595 620 580 575 260 590 585 580 680 635 605 г_а Notes: 525 470 430 390 360 330 310 295 300 320 320 1110915770670585 525 470 430 390 330 330 330 285 275 285 286 286 305 770 670 325 R_× R_× 510 495 500 520 525 530 535 545 550 560 565 495 505 515 520 525 530 535 545 550 560 565 575 490 515 575 ď I 2375 2675 4305 5520 2430 2470 2970 3790 5135 6045 9955 2970 2675 3345 3790 7815 8705 4305 8280 -4720 -2692 -9215 -2115 -3320 -6340 -7235 -8195 -10300 -1570 -5500 -6340 -7235 -1085 -2695 -3995 -5500 -9215 -1085 -2115 -3320 -3995 -4720 -8195 -545 -755 -655 105 mph 120 mph 089 825 655 620 605 595 590 580 585 585 575 635 605 590 655 620 580 575 575 760 680 635 1110 915 770 915 770 670 585 Table E-17 - 40 ft Span, Variable Wind Load. 585 525 470 470 330 330 330 335 335 330 525 430 330 330 330 310 285 285 285 290 310 670 ⊼_× a 510 515 520 525 530 530 535 545 550 560 495 515 520 525 530 535 545 550 560 565 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 25.00 29.00 21.29 20.50 20.11 26.72 23.68 21.88 20.83 20.50 20.03 36.33 32.07 26.72 23.68 22.67 21.88 20.83 20.26 20.03 20.00 52.00 36.33 32.07 29.00 25.00 22.67 21.29 20.11 Œ E 10 psf Dead Load Rise [T] Rise [T] £ £ 10 18 13 14 15 16 17 17 19 20 12 141516 17 10 11

2430

15325 18065 21190 24750

5250 6040 7270 8285 9155

-20310 -12515 -10290 -14565 -17460 -1595 -2370 -3065 -3405 -3700 -4285 -4960 -6240 -7320 -8580 -10040-2840 -2475 -2265 -3390 -3635 -4340 -4805 -5220 -6040 0669--8775 -12025 -14020 200 mph 1205 260 635 585 089 760 680 635 620 580 655 620 605 595 430 390 360 330 770 670 585 525 470 430 390 330 330 300 325 355 375 410 525 265 250 260 285 285 495 500 505 510 515 520 520 525 535 550 560 565 500 505 510 515 515 520 520 530 530 535 545 545 550 565 Wind Load 13260 22240 26465 $K_{zt} = 1.0$ $K_{d} = 0.85$ 19040 11335 13775 16235 5200 6740 9595 5440 6540 8225 7895 7450 9680 -11045 -10810 -12625 -15725 -18290 -7570 -12855 -5500 -2210 -3890 -9245 -1395 -2690 -2985 -3320 -3995 -4720 -6445 -8860 -2530 -2030 -3030 -4310 -4685 -6275 -7885 -1695 -2065 -3255 -5425 All values normalized to C_D = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] F_a lbs 1205 1030 260 715 680 655 635 620 605 Notes: 670 585 525 470 430 390 360 330 310 285 265 250 255 915 770 670 670 585 525 470 430 330 330 3310 285 285 315 335 365 260 515 520 525 530 545550560 495 505 535 490 500 505 510 515 520 520 525 530 535 545 550 560 565 575 ď I 10120 14495 19860 27415 16075 23635 2430 4535 8360 4240 5850 7345 8640 12300 3995 5875 0889 4870 6655 4995 -16375 -1450 -1215 -2692 -5500 -7235 -1960 -3465 -8255 -11295 -14075 -2335 -3995 -6340 -8195 -9775 in-lb -2745 -2235 -2690 -4180 -5600 -7040 -1785 -3320 -4720 -2895 -3845 -4835 -9655 150 mph 180 mph F_a lbs 1205 1030 910 825 760 655 585 580 575 680 655 635 620 605 585 580 580 575 575 620 635 670 525 470 430 330 330 3310 310 285 265 250 270 770 670 585 525 470 330 330 330 330 285 285 285 285 335 515 520 525 525 530 535 550 560 515 520 525 530 535 545 550 560 565 575 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 20.50 32.07 26.72 23.68 21.88 20.26 36.33 29.00 25.00 22.67 21.88 20.83 20.26 20.11 42.50 36.33 29.00 25.00 22.67 21.29 20.83 20.50 20.11 20.03 32.07 26.72 23.68 20.00 £ Œ 10 psf Dead Load Rise [T] Rise [T] £ £ 19 10 12 13 14 15 16 17 18 19 20 12 14 15 16 18

2430

Table E-17 - 40 ft Span, Variable Wind Load.

Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ All values normalized to C_D = 1.0. Use ASD design procedure only. Notes: Table E-18 - 50 ft Span, Variable Wind Load. 10 psf Dead Load 20 psf Construction Load No Ground Snow Load

			₹	in-lb	4945	3835	3010	3060	3575	4415	5515	0969	0906	11665	15380	17330			₹	d-ni	4945	3835	3010	4035	5180	6320	7815	10735	13975	17990	23520	26490
			M	in-lb	-1100	-1030	-1040	-1915	-3145	-4540	-6130	-7930	-9962	-12245	-14770	-16135			M	dl-ni	-1865	-1675	-1495	-2265	-3145	-4540	-6130	-7930	-9962	-12245	-15155	-16970
		115 mph	Fa	sql	1545	1345	1095	955	870	815	775	755	735	725	720	720		140 mph	F _a	sql	1545	1345	1095	955	870	815	775	755	735	725	720	720
			æ _x	lps	1425	1205	920	740	615	525	460	405	360	365	390	400			۳×	sql	1425	1205	920	740	615	525	460	405	360	320	350	355
			R	lps	610	610	615	625	630	640	655	999	089	695	710	715			R,	sql	610	610	615	625	630	640	655	999	089	695	710	715
			± ⊠	in-lb	4945	3835	3010	3060	3575	4415	5515	6870	8460	10550	13945	15715			±	in-lb	4945	3835	3010	3455	4430	5390	0999	9135	11890	15310	20075	22610
	Σ		M	in-lb	-970	-920	066-	-1915	-3145	-4540	-6130	-7930	-9962	-12245	-14770	-16135	Σ		M	in-lb	-1540	-1400	-1280	-1915	-3145	-4540	-6130	-7930	-9965	-12245	-14770	-16135
	Basic Wind Speed [V]	110 mph	Fa	lps	1545	1345	1095	955	870	815	775	755	735	725	720	720	Basic Wind Speed [V]	130 mph	T _e	sql	1545	1345	1095	955	870	815	775	755	735	725	720	720
	Basic \		Α×	lbs	1425	1205	920	740	615	525	460	405	365	375	400	405	Basic \		۳×	lbs	1425	1205	920	740	615	525	460	405	360	340	365	375
			R	lps	610	610	615	625	930	640	655	999	089	695	710	715			A,	lps	610	610	615	625	630	640	655	999	089	695	710	715
Ý			±	in-lb	4945	3835	3010	3060	3575	4415	5515	0289	8460	10295	12575	14170			±	q-ui	4945	3835	3010	3060	3735	4530	5275	7655	9962	12830	16880	19020
			M	in-lb	-845	-815	-945	-1915	-3145	-4540	-6130	-7930	-9965	-12245	-14770	-16135			Ē	in-lb	-1240	-1150	-1090	-1915	-3145	-4540	-6130	-7930	-9965	-12245	-14770	-16135
		105 mph	Fa	lps	1545	1345	1095	955	870	815	775	755	735	725	720	720		120 mph	T _e	sql	1545	1345	1095	955	870	815	775	755	735	725	720	720
Ý			۳×	lps	1425	1205	920	740	615	525	460	405	370	380	405	410			۳×	sql	1425	1205	920	740	615	525	460	405	360	355	385	390
			R,	sql	610	610	615	625	630	640	655	999	089	695	710	715			R,	sql	610	610	615	625	630	640	655	999	089	695	710	715
		Radius [R]	(£)		65.00	55.08	43.06	36.25	32.04	29.32	27.53	26.36	25.63	25.20	25.02	25.00		Radius [R]	£		65.00	55.08	43.06	36.25	32.04	29.32	27.53	26.36	25.63	25.20	25.02	25.00
		Rise [T]			2	9	∞	10	12	14	16	18	20	22	24	25		Rise [T]			2	9	∞	10	12	14	16	18	20	22	24	25
	Ь									_		_				_	_					_						_				_

Wind Load $K_{zt} = 1.0$ $K_{d} = 0.85$ All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Notes: Table E-18 - 50 ft Span, Variable Wind Load. 10 psf Dead Load 20 psf Construction Load No Ground Snow Load

		± ∑	in-lb	4945	3832	3010	6025	7790	9535	11820	16235	21140	27220	35480	39940			±	ql-ui	4945	3832	3030	8400	10905	13390	16625	22800	29700	38230	49815	29095
		M	in-lb	-2985	-2620	-2270	-3485	-4455	-5315	-6495	-9220	-11955	-15420	-21120	-23845			M	in-lb	-4335	-3755	-3190	-4955	-6290	-7490	-9150	-12955	-16785	-21545	-29445	-33230
	170 mph	Fa	lbs	1545	1345	1095	955	870	815	775	755	735	725	720	720		200 mph	Fa	sql	1545	1345	1095	955	870	815	775	755	735	725	720	720
		R _x	lbs	1425	1205	920	740	615	525	460	405	360	320	350	380			۳,	lps	1425	1205	920	740	615	525	460	405	395	450	200	540
		R	lbs	610	610	615	625	630	640	655	999	089	695	710	715			R	lps	610	610	615	625	630	640	655	999	089	695	710	715
		± ⊠	in-lb	4945	3835	3010	5320	6865	8385	10400	14280	18600	23945	31220	35145			±	in-lb	4945	3835	3010	7565	9810	12035	14935	20490	26690	34360	44780	50400
[N]		M	in-lb	-2585	-2285	-1995	-3045	-3910	-4670	-6130	-8110	-10520	-13605	-18645	-21055	[]		M	in-lb	-3860	-3355	-2865	-4435	-5645	-6730	-8220	-11645	-15085	-19375	-26520	-29935
Basic Wind Speed [V]	160 mph	F	lbs	1545	1345	1095	955	870	815	775	755	735	725	720	720	Basic Wind Speed [V]	190 mph	F _a	lps	1545	1345	1095	955	870	815	775	755	735	725	720	720
Basic		R _x	lbs	1425	1205	920	740	615	525	460	405	360	320	310	330	Basic		ď	lbs	1425	1205	920	740	615	525	460	405	360	400	450	485
		R	lbs	610	610	615	625	630	640	655	999	089	695	710	715			R,	lps	610	610	615	625	630	640	655	999	089	695	710	715
		Ψ	in-lb	4945	3835	3010	4655	5995	7320	9065	12450	16210	20865	27220	30660			μ	ql-ui	4945	3835	3010	6775	8770	10750	13330	18305	23840	30690	40000	45025
		M	in-lb	-2215	-1970	-1740	-2640	-3400	-4540	-6130	-7930	-9965	-12245	-16320	-18430			M	in-lb	-3410	-2980	-2560	-3945	-5030	-6005	-7335	-10400	-13475	-17340	-23745	-26805
	150 mph	ъ	lbs	1545	1345	1095	955	870	815	775	755	735	725	720	720		180 mph	T.	lbs	1545	1345	1095	955	870	815	775	755	735	725	720	720
		R _x	lbs	1425	1205	920	740	615	525	460	405	360	320	330	340			ď	lps	1425	1205	920	740	615	525	460	405	360	355	395	430
		R _v	lbs	610	610	615	625	630	640	655	999	089	695	710	715			A,	lps	610	610	615	625	630	640	655	999	089	695	710	715
	Radius [R]	(£)		65.00	55.08	43.06	36.25	32.04	29.32	27.53	26.36	25.63	25.20	25.02	25.00		Radius [R]	£)		65.00	22.08	43.06	36.25	32.04	29.32	27.53	26.36	25.63	25.20	25.02	25.00
	Rise [T]	(ft)		2	9	∞	10	12	14	16	18	20	22	24	25		Rise [T]	(£)		2	9	8	10	12	14	16	18	20	22	24	25

-24850

33940

-23385

28560

-23265

30.00

5865 8675 10185 12905 16865 20990 25920

3555 6905

-11950 -14400 -17100 -20050 -11950 -17100 -20535 -1455 -1785 -2960 -4375 -5960 -7735 -9730 -14400 -23265 -2105 -2160 -3170 -4375 -5960 -7735 -9730 -1365 Έ 115 mph 140 mph 1505 1150 1505 1285 1150 1060 1875 1285 950 875 1060 920 900 995 950 885 875 865 860 995 920 885 865 860 1740 1325 1065 **R**_x **Ibs** 1325 1065 890 760 665 590 525 475 430 890 760 665 590 525 475 430 430 475 390 405 420 Ry lbs | 1bs | 730 | 735 | 740 | 745 | 755 | 775 | 775 | 775 | 775 | 770 | 800 | 800 | 815 830 845 860 730 735 740 745 755 765 775 790 800 815 830 845 860 Wind Load 17870 $K_{zt} = 1.0$ $K_d = 0.85$ 10095 15225 14360 6875 12330 19330 23610 10985 5015 7400 3555 4615 8370 3795 5890 3900 3555 -11950 -17100 -20050 -14400 -17100 -20050 -23265 -2960 -14400 -1790 -5960 -9730 -11950-1235 -1725 -2960 -4375 -5960 -7735 -9730 -1990 -4375 -7735 -1980 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 110 mph 130 mph 1150 1875 1285 950 1505 006 1875 1285 1060 995 950 920 885 920 885 865 860 900 875 865 860 т_т Б г_а Notes: 1740 1325 1065 890 760 665 590 525 475 430 440 470 480 Rx lbs R_× 730 735 740 745 755 775 790 800 830 845 860 735 740 745 755 765 775 790 800 815 830 845 860 815 765 ď I 10095 14250 14985 12055 17435 12040 23385 4615 6875 21300 3795 4235 6225 7275 9210 3795 3555 3900 5620 8370 3555 4950 -11950 -17100 -14400 -1110 -4375 -7735 -14400 -20050 -23265 -1500 -2960 -5960 -9730 -11950 -17100 -20050 -1725 -2960 -5960 -9730 -1850 -4375 -1140 -1625 -7735 105 mph 120 mph 1150 1060 1505 1150 1875 1285 1285 1060 950 900 875 995 950 920 900 885 875 865 860 lbs 995 920 860 ஈ <u>த</u> **R**_x **Ibs** 1740 1740 1325 1065 890 760 665 590 525 475 1065 890 760 665 590 525 475 475 420 420 450 440 450 475 490 ~<u>, 영</u> 730 735 740 745 755 765 775 790 800 815 830 845 860 735 740 745 755 765 775 790 800 815 830 845 860 No Ground Snow Load Radius [R] Radius [R] 34.00 36.13 31.45 30.31 60.25 43.50 32.50 30.75 30.07 78.00 50.00 43.50 39.14 36.13 32.50 30.75 30.07 30.00 50.00 39.14 34.00 31.45 30.31 £ Œ 10 psf Dead Load Rise [T] (ft) Rise [T] Œ 6 8 8 10 11 11 11 11 11 11 11 18 20 22 24 26 26 28 8 110 112 114 116 118 22 22 22 24 26 26 26 30 30

3555 3900 4615 6875 8375 10950 13625 16830 21315 26030

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Table E-19 - 60 ft Span, Variable Wind Load.

20 psf Construction Load

Wind Load $K_{zt}=1.0 \\ K_d=0.85$ All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Notes: Table E-19 - 60 ft Span, Variable Wind Load. 10 psf Dead Load 20 psf Construction Load No Ground Snow Load Rise (f

						Basic	Basic Wind Speed [V]	[N]						
		150 mph					160 mph					170 mph		
A _x		F.	.W	±Σ	R _v	R _x	Fa	M	 ±	Ry	R _x	Fa	ĮW.	⁺ Σ
lbs		lps	in-lb	q-ui	lps	lbs	lps	in-lb	in-lb	lbs	lps	lps	in-lb	ql-ui
1740		1875	-2795	5215	730	1740	1875	-3245	5215	730	1740	1875	-3730	5215
1325	2	1505	-2440	3795	735	1325	1505	-2810	3795	735	1325	1505	-3200	3795
1065	2	1285	-2340	3555	740	1065	1285	-2635	3555	740	1065	1285	-2980	3555
890		1150	-3690	0229	745	890	1150	-4245	7740	745	890	1150	-4845	8775
209	0	1060	-4375	7995	755	260	1060	-4865	9155	755	260	1060	-5540	10395
999	2	995	-5960	10040	292	999	995	-6340	11510	765	999	995	-7215	13080
590	0	950	-7735	11810	775	290	950	-7735	13545	775	290	950	-8420	15395
525	5	920	-9730	14970	790	525	920	-9730	17170	790	525	920	-10580	19520
475	2	006	-11950	19555	800	475	006	-12660	22435	800	475	006	-14390	25495
430	0	885	-14400	24340	815	430	885	-15730	27920	815	430	885	-17870	31735
390	0	875	-17100	30060	830	390	875	-19505	34480	830	390	875	-22100	39185
385	35	865	-22500	37695	845	360	865	-25700	43230	845	415	865	-29110	49115
400	0	860	-27590	45985	860	415	860	-31510	52720	860	475	860	-35680	29900
						Basic	Basic Wind Speed	[N]						
		180 mph					190 mph					200 mph		
	۳×	Fa	_W	ψ	Ry	R _x	Fa	M	 ±	Ry	R _x	Fa	_M	⁺ ≥
_	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	sql	in-lb	in-lb
17	1740	1875	-4240	5215	730	1740	1875	-4780	5215	730	1740	1875	-5350	5215
13	1325	1505	-3615	3795	735	1325	1505	-4050	3795	735	1325	1505	-4515	4080
1(1065	1285	-3350	3875	740	1065	1285	-3740	4330	740	1065	1285	-4150	4810
∞	890	1150	-5480	9870	745	890	1150	-6155	11030	745	890	1150	0989-	12250
7	260	1060	-6255	11710	755	200	1060	-7010	13100	755	209	1060	-7810	14565
9	999	995	-8140	14745	765	999	995	-9120	16505	765	999	995	-10150	18360
S	290	950	-9500	17370	775	290	950	-10645	19455	775	290	950	-11850	21655
2	525	920	-11935	22005	790	525	920	-13365	24635	790	525	920	-14875	27425
4	475	006	-16220	28740	800	475	006	-18155	32175	800	475	006	-20195	35790
7	430	885	-20140	35775	815	440	885	-22540	40050	815	495	885	-25065	44555
4	435	875	-24850	44175	830	490	875	-27800	49450	830	550	875	-30915	55015
4	475	865	-32720	55365	845	532	865	-36545	61965	845	265	865	-40570	68925
23	535	860	-40105	67510	860	909	860	-44780	75555	860	675	860	-49710	84035

Table E-20 - 70 ff Span, Variable Wind Load.

15 psf Dead Load

20 psf Construction Load

Refreshed to Company of the All Values normalized to Company of the ASD design procedure only.

 $\begin{aligned} & \textbf{Wind Load} \\ & K_{zt} = 1.0 \\ & K_d = 0.85 \end{aligned}$

							Basic	Basic Wind Speed [V]	[<u>N</u>]						
Radius [R]			105 mph					110 mph					115 mph		
Œ	&^	œ×	T.	Έ	₹	₽,	٣×	T _e	Σ	₹	A,	٣×	ъ.	Σ	₹
	sql	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
105.08	066	2760	2925	-1255	7970	066	2760	2925	-1460	0262	066	2760	2925	-1670	7970
99.28	995	2105	2320	-1365	5470	995	2105	2320	-1530	5470	995	2105	2320	-1700	5470
66.25	1005	1695	1960	-1845	4775	1005	1695	1960	-1930	4775	1005	1695	1960	-2015	4775
57.04	1015	1420	1730	-3120	4920	1015	1420	1730	-3120	4920	1015	1420	1730	-3120	4920
50.75	1025	1220	1575	-4750	5275	1025	1220	1575	-4750	5575	1025	1220	1575	-4750	5575
46.28	1040	1065	1470	-6580	9099	1040	1065	1470	-6580	9099	1040	1065	1470	-6580	9099
43.03	1055	950	1390	-8620	7940	1055	950	1390	-8620	7940	1055	920	1390	-8620	7940
40.63	1070	850	1340	-10900	9260	1070	850	1340	-10900	9260	1070	850	1340	-10900	9260
38.84	1085	770	1300	-13430	11455	1085	770	1300	-13430	11455	1085	770	1300	-13430	11455
37.52	1105	705	1270	-16240	13615	1105	705	1270	-16240	13615	1105	705	1270	-16240	13615
36.56	1125	029	1250	-19330	16045	1125	645	1250	-19330	16045	1125	645	1250	-19330	16045
35.88	1145	999	1235	-22715	18750	1145	650	1235	-22715	18750	1145	640	1235	-22715	18750
35.42	1165	089	1230	-26410	21740	1165	999	1230	-26410	21740	1165	655	1230	-26410	21835
35.14	1185	695	1225	-30425	25015	1185	685	1225	-30425	25015	1185	029	1225	-30425	26075
35.01	1210	730	1220	-34765	28580	1210	720	1220	-34765	29062	1210	710	1220	-34765	32195
35.00	1220	740	1220	-37060	30480	1220	730	1220	-37060	31660	1220	720	1220	-37060	35060
							Basic	Basic Wind Speed [V]	[N] p						
Radius [R]			120 mph					130 mph					140 mph		
£	æ [^]	٣×	F _a	Σ	₹	R,	٣×	T.	×	±	R _v	٣×	F _a	M	±
	sql	lps	lbs	in-lb	in-lb	lps	lbs	lps	in-lb	in-lb	lbs	lbs	lps	in-lb	in-lb
105.08	066	2760	2925	-1890	7970	066	2760	2925	-2370	0262	066	2760	2925	-2895	7970
80.56	995	2105	2320	-1880	5470	995	2105	2320	-2265	5470	995	2105	2320	-2690	5470
66.25	1005	1695	1960	-2110	4775	1005	1695	1960	-2305	4775	1005	1695	1960	-2640	4775
57.04	1015	1420	1730	-3120	4920	1015	1420	1730	-3200	4920	1015	1420	1730	-3405	4920
50.75	1025	1220	1575	-4750	5222	1025	1220	1575	-4750	0999	1025	1220	1575	-4750	7815
46.28	1040	1065	1470	-6580	9099	1040	1065	1470	-6580	2992	1040	1065	1470	-6580	9040
43.03	1055	920	1390	-8620	2962	1055	950	1390	-8620	9540	1055	950	1390	-8620	11235
40.63	1070	820	1340	-10900	9260	1070	850	1340	-10900	11175	1070	850	1340	-10900	13190
38.84	1085	770	1300	-13430	11455	1085	770	1300	-13430	12965	1085	770	1300	-13430	15330
37.52	1105	202	1270	-16240	13615	1105	705	1270	-16240	15795	1105	705	1270	-16240	18685
36.56	1125	645	1250	-19330	16625	1125	645	1250	-19330	19985	1125	645	1250	-19330	23625
35.88	1145	625	1235	-22715	20055	1145	009	1235	-22715	24085	1145	290	1235	-22715	28475
35.42	1165	640	1230	-26410	24065	1165	615	1230	-26410	28875	1165	585	1230	-26410	34135
35.14	1185	655	1225	-30425	28735	1185	630	1225	-30425	34450	1185	009	1225	-30425	40720
35.01	1210	695	1220	-34765	35465	1210	675	1220	-34765	42415	1210	645	1220	-35190	49925
35.00	1220	710	1220	-37060	38615	1220	685	1220	-37060	46180	1220	099	1220	-38130	54350

116300 88070 11820 20125 23470 28635 43590 16600 19415 24090 33185 40510 61630 73870 6085 -10900 -15430 -13430 -16240 -24675 -29715 -10165 -13170 -21795 -28750 -34630 -63570 -69315 -6240 -7155 -9300 -20375-35605 -49605 -5500 -5400 -8875 -17935-41605-49825 -3945 -45480 -5925 170 mph 200 mph 1235 1225 1270 1470 1340 1235 1225 1220 1390 2320 1960 1575 1390 1300 1250 1420 1420 1065 2760 1220 1065 950 850 1695 1220 1695 850 770 705 590 555 645 705 645 590 685 740 785 950 505 770 1040 1070 1105 1055 1070 1055 1085 Wind Load 17430 36340 104375 $K_{zt} = 1.0$ $K_d = 0.85$ 12090 17665 20580 25105 54645 25495 55290 79020 31710 38225 66615 21615 29770 45870 10400 72505 6100 14920 4920 -10900 -13430 -16240 -22715 -43745 -11810 -13835 -19550 -31090 -44825 -57215 -26410 -31375 -25805 -62390 -19330-40100 -5305 -37430 -3645 -3880 -5455 -6580 -8620 -4950 -4895 -7945 -9105 -16085-3480 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 190 mph 1270 1235 1225 1220 1220 1270 1300 1250 1235 1225 г_а 1420 1420 1065 950 850 770 705 645 Notes: 2105 1695 1220 1065 950 850 770 705 645 590 530 2760 2105 1695 1220 545 545 605 660 695 009 R_x lbs 1015 1070 1145 1015 1040 1105 1145 1220 1025 1055 1085 1125 1165 1025 1055 1085 ď I 15350 33185 22735 32380 49280 70430 10515 21790 27535 57990 15550 26535 40880 4920 13055 17865 63125 5440 13325 19280 9065 5470 4915 -22715 -26410 -30425 -12325 -17415 -23005 -55820 -10900 -13430 -16240 -40345 -4430 -10520 -14325 -27790 -33465 -40085 -51185 -3150 -3635 -6580 -8620 -7070 -8105 -3045 -4750 -19330-4720 -4435 150 mph 180 mpł 1270 1250 1270 1250 1235 1225 1220 1220 1300 1235 1420 2760 1420 1065 1220 1065 1695 1695 1220 Table E-20 - 70 ft Span, Variable Wind Load. 950 850 770 705 590 550 565 620 630 950 850 770 705 645 590 615 645 545 535 580 a_x S lbs 1040 1070 1105 1145 1105 1055 1085 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 46.28 40.63 35.14 40.63 37.52 35.14 35.00 66.25 57.04 50.75 43.03 38.84 36.56 35.88 35.42 35.00 80.56 66.25 57.04 50.75 46.28 43.03 38.84 36.56 35.88 £ Rise [T] Rise [T] 112 114 116 118 22 22 22 24 26 26 28 30 £ 32 E 35

15 psf Dead Load

Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Notes: Table E-21 - 80 ft Span, Variable Wind Load. 15 psf Dead Load 20 psf Construction Load No Ground Snow Load Ris

													<u> </u>	Ι											
		₊ω	in-lb	6725	5470	6865	9630	13490	18400	24545	33660	47285			₊Σ	in-lb	6725	5470	10380	14550	18670	26450	38385	52545	73200
		M	in-lb	-2180	-3085	-6385	-10680	-15925	-22235	-29710	-38420	-48430			Σ	in-lb	-3480	-3725	-6385	-10680	-15925	-22235	-29710	-38420	-50375
	115 mph	Fa	sql	2960	2165	1800	1610	1505	1445	1415	1400	1395		140 mph	Fa	lbs	2960	2165	1800	1610	1505	1445	1415	1400	1395
		R _x	lps	2750	1855	1395	1115	925	785	725	755	815			R _x	lbs	2750	1855	1395	1115	925	785	675	029	745
		R _v	lps	1135	1150	1170	1200	1230	1265	1305	1350	1395			R _v	lbs	1135	1150	1170	1200	1230	1265	1305	1350	1395
		M	in-lb	6725	5470	6865	9630	13490	18400	24385	31480	42710			μ	in-lb	6725	5470	8830	12360	15805	22370	32485	44480	62225
[V]		M	in-lb	-1950	-2980	-6385	-10680	-15925	-22235	-29710	-38420	-48430	[2]		M	in-lb	-2920	-3435	-6385	-10680	-15925	-22235	-29710	-38420	-48430
Basic Wind Speed [V]	110 mph	T _e	lps	2960	2165	1800	1610	1505	1445	1415	1400	1395	Basic Wind Speed [V]	130 mph	Fa	lbs	2960	2165	1800	1610	1505	1445	1415	1400	1395
Basic		R _x	lbs	2750	1855	1395	1115	925	785	740	770	830	Basic		R _x	lbs	2750	1855	1395	1115	925	785	675	710	775
		R	lps	1135	1150	1170	1200	1230	1265	1305	1350	1395			R	lbs	1135	1150	1170	1200	1230	1265	1305	1350	1395
		₹	in-lb	6725	5470	6865	9630	13490	18400	24385	31480	39745			±	in-lb	6725	5470	7395	10335	13490	18590	27045	37090	52060
		M	in-lb	-1735	-2900	-6385	-10680	-15925	-22235	-29710	-38420	-48430			M	in-lb	-2415	-3200	-6385	-10680	-15925	-22235	-29710	-38420	-48430
	105 mph	Fa	lps	2960	2165	1800	1610	1505	1445	1415	1400	1395		120 mph	Fa	lbs	2960	2165	1800	1610	1505	1445	1415	1400	1395
		A,	sql	2750	1855	1395	1115	925	785	755	785	840			A,	lbs	2750	1855	1395	1115	925	785	710	740	805
		S,	sql	1135	1150	1170	1200	1230	1265	1305	1350	1395			A,	lbs	1135	1150	1170	1200	1230	1265	1305	1350	1395
	Radius [R]	#)		104.00	72.67	28.00	20.00	45.33	42.57	41.00	40.22	40.00		Radius [R]	£		104.00	72.67	28.00	20.00	45.33	42.57	41.00	40.22	40.00
	Rise [T]	(ft)		8	12	16	20	24	28	32	36	40		Rise [T]			∞	12	16	20	24	28	32	36	40
_					_		_		_		_		_			_	_				_				_

Wind Load Notes: Table E-21 - 80 ft Span, Variable Wind Load. 15 p 20 p No

					/	/		Notes:			Wind Load					
15 pst Dead Load 20 psf Construction	15 pst Dead Load 20 psf Construction Load No Ground Spow Load	ad			<u>~</u>		C	All values nor Use ASD desi	All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only.	= 1.0. only.	$K_{zt} = 1.0$ $K_d = 0.85$					
			××			\$ 	× Ľ									
								Basic	Basic Wind Speed [V]	N						
Rise [T]	Radius [R]			150 mph					160 mph					170 mph		
(#)	(£	Ry	a,	F _a	Ē		R _v	R _x	F _a	Σ	Ψ	R,	æ _x	F _a	M	±
		lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
8	104.00	1135	2750	2960	-4085	6725	1135	2750	2960	-4725	6725	1135	2750	2960	-5410	6725
12	72.67	1150	1855	2165	-4035	5470	1150	1855	2165	-4475	5470	1150	1855	2165	-5020	2990
16	28.00	1170	1395	1800	-6385	12045	1170	1395	1800	-7160	13820	1170	1395	1800	-8175	15715
20	20.00	1200	1115	1610	-10680	16900	1200	1115	1610	-10680	19415	1200	1115	1610	-11970	22090
24	45.33	1230	925	1505	-15925	21750	1230	925	1505	-15925	25045	1230	925	1505	-15925	28550
28	42.57	1265	785	1445	-22235	30830	1265	785	1445	-22235	35515	1265	785	1445	-22235	40500
32	41.00	1305	675	1415	-29710	44720	1305	675	1415	-29710	51495	1305	675	1415	-33155	58705
36	40.22	1350	635	1400	-38420	61205	1350	595	1400	-40330	70465	1350	290	1400	-45760	80320
40	40.00	1395	710	1395	-53345	84990	1395	675	1395	-58810	97595	1395	640	1395	-66680	111010
								Basic	Basic Wind Speed [V]	[<u>N</u>]						
Rise [T]	Radius [R]			180 mph					190 mph					200 mph		
(#)	(#)	æ^	æ×	щ.	Σ	₹	æ [^]	æ×	тe	Σ	± ∑	æ [^]	æ×	щ	Σ	<u></u>
		lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb	lbs	lbs	lbs	in-lb	in-lb
∞	104.00	1135	2750	2960	-6140	6725	1135	2750	2960	-6910	6725	1135	2750	2960	-7720	6925
12	72.67	1150	1855	2165	-5595	6745	1150	1855	2165	-6210	7545	1150	1855	2165	-6875	8390
16	58.00	1170	1395	1800	-9255	17725	1170	1395	1800	-10395	19845	1170	1395	1800	-11600	22080
20	20.00	1200	1115	1610	-13535	24930	1200	1115	1610	-15185	27960	1200	1115	1610	-16925	31150
24	45.33	1230	925	1505	-17385	32265	1230	925	1505	-19515	36195	1230	925	1505	-21760	40340
28	42.57	1265	785	1445	-24540	45785	1265	785	1445	-27535	51380	1265	785	1445	-30695	57270
32	41.00	1305	675	1415	-37335	66350	1305	675	1415	-41775	74430	1305	675	1415	-46530	82955
36	40.22	1350	610	1400	-51515	90775	1350	695	1400	-57605	101825	1350	785	1400	-64020	113475
40	40.00	1395	725	1395	-75025	125250	1395	820	1395	-83850	140455	1395	925	1395	-93150	156480

-21870 -15670 -53230 -21870 -37635 -47345 -52685 -10485 -15670 -29165 -37635 -47345 -3260 -6185 -10485-29165 -61315 -4380 -4245 -6225 Σ Σ 115 mph 140 mph lbs 760 745 800 _× 정 Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ ≥ -15670 -29165 -47345 -10485 -15670 -21870 -37635 -47345 -52685 -21870 -61315 -3125 -10485-37635 -52685 -3735 -29165 -2435 -6185 -3670 -6185 Έ All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 110 mph 130 mph lbs Notes: 825 860 760 785 835 865 ъ В R_× ď I -15670 -29165 -47345 -10485 -21870 -37635 -47345 -52685 -6185 -21870 -37635 -52685 -3405 -15670 -29165 -2160 -61315 -6185 -2990 -10485-3030 Σ 105 mph 120 mpł 1575 lbs lbs 845 875 870 900 ~ <u>청</u> R_x Sq 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 54.19 60.63 54.19 130.56 50.16 47.64 45.31 45.00 130.56 46.13 71.28 60.63 46.13 45.11 90.38 71.28 47.64 45.11 Œ £ 15 psf Dead Load Rise [T] Rise [T] £ £ 112 116 116 22 22 28 28 28 33 36 40 112 112 20 24 24 28 33 33 40 42

 -64405

-61315

-61315

45.00

Table E-22 - 90 ft Span, Variable Wind Load.

-120895

-108835

-97390

45.00

-21870 -70560 -30490 -80220 -17025 -43030 -57355 -24045 -45075 -98590 -6835 -6205 -7370 -11925 -31955 -86565 -8605 -8760 -16870 -60395 200 mph 170 mph F_a 2345 1765 1000 870 760 670 675 935 1065 R_x 1290 1305 Wind Load 50990 72165 $K_{zt} = 1.0$ $K_d = 0.85$ -37915 -15670 -21870 -50555 -62225 -8260 -15135 -21580 -27355 -40465 -72185 -88740 -76360 -5505 -10485 -29165 -54225 -6962 -7760 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 190 mph г_а Notes: 760 670 720 760 790 830 950 ъ В ъ В ď I -21870 -37635 -36090 -64565 -15670 -47345 -56265 0969--7805 -19240 -24380 -79395 -4855 -6585 -10485 -29165 -68255 -5145 -13485-48450 150 mph 180 mph 1590 1570 lbs F_a 760 760 760 735 840 ж <u>Б</u> No Ground Snow Load Radius [R] Radius [R] 60.63 50.16 46.13 54.19 90.38 45.11 130.56 71.28 50.16 47.64 71.28 54.19 47.64 45.31 45.00 90.38 60.63 46.13 45.31 45.11 £ € 15 psf Dead Load Rise [T] Rise [T] £ £ 112 116 116 22 22 22 23 33 33 33 44 40 45 112 112 20 24 24 28 33 33 40 42

Table E-22 - 90 ft Span, Variable Wind Load.

20 psf Construction Load

-80240

-75720

-75720

50.00

-25020 -41610 -41610 -51695 -18390 -63045 -12765 -18390 -51695 -63045 -3375 -4545 -8045 -12765-25020 -32740 -75720 -5185 -8045 -32740Έ Σ 115 mph 140 mph lbs 2490 800 830 _× 정 Wind Load 21535 26060 $K_{zt} = 1.0$ $K_d = 0.85$ ≥ -18390 -51695 -12765 -18390 -25020 -41610 -51695 -63045 -32740 -41610 -75720 -12765-25020 -63045 -32740 -3055 -8045 -4415 -5125 -8045 -4385 Έ All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 110 mph 130 mph г_а lbs Notes: 880 955 R_× ъ. В ď I -32740 -41610 -51695 -12765 -25020 -41610 -51695 -63045 -8045 -12765 -18390 -25020 -63045 -4720 -18390 -32740 -75720 -3710-8045 -2750 -4230 Σ 105 mph 120 mpł lbs 2490 950 980 890 920 995 ~ <u>청</u> R_x Sq 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 53.76 67.82 51.89 50.76 130.00 26.67 50.76 78.44 67.82 61.08 26.67 51.89 50.17 50.00 78.44 61.08 53.76 Œ £ 15 psf Dead Load Rise [T] Rise [T] £ £ 114 118 22 22 22 26 26 33 34 42 42 46 114 122 22 22 22 26 33 33 34 42 42 46 50

Table E-23 - 100 ft Span, Variable Wind Load.

-152655

-137430

-122990

50.00

 -25160 -47530 -109330-11120 -111645 -29180 -86645 -20680 -61940 -79860 -10395 -35555 -46965 -7620 -9395 -14825 -33250 -111110 -2094509699-200 mph 170 mph F_a **R**_x **Ibs** 3460 2490 ж <u>Б</u> Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -18390 -25020 -41885 -54595 -70415 -10490 -26195 -31905 -42145 -77960 -100480 -32740 -96450 -6805 -13005 -18795-60130-8895 -9975 -9400 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 160 mph 190 mph г_а г_а Notes: 955 1085 ъ В R_x ď I -41610 -25020 -23360 -37575 -53645 -69730 -6040 -18390 -32740 -51695 -63045 -9930 -28440 -89885 -8425 -85105 -8900 -8485 -6030 -12765 -16755Σ 150 mph 180 mph lbs F_a 800 780 ж <u>Б</u> R_x 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 51.89 50.76 96.29 67.82 61.08 26.67 50.76 130.00 78.44 61.08 53.76 78.44 53.76 50.00 96.29 67.82 26.67 51.89 £ € 15 psf Dead Load Rise [T] Rise [T] £ £ 14 18 22 22 22 26 30 33 42 42 46 14 18 22 22 26 30 30 34 42 42

Table E-23 - 100 ft Span, Variable Wind Load.

-107365

-104925

-104925

55.00

-57700 -36430 -57700 -84795 -20365 -27785 -46375 -104925-14060 -20365 -46375 -3870 -5120 -8745 -14060-36430 -70485 -84795 -6420 -9190 -27785 -70485 Σ 115 mph 140 mph R_x lbs Wind Load 17420 22575 $K_{zt} = 1.0$ $K_d = 0.85$ -20365 -36430 -46375 -57700 -104925 -14060 -20365 -36430 -46375 -57700 -84795 -14060-70485 -84795 -5145 -5810 -8745 -27785 -70485 -3485 -4915 -8745 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 110 mph 130 mph г_а Notes: R_× -36430 -57700 -104925 -20365 -36430 -57700 -84795 -8745 -14060 -20365 -46375 -70485 -84795 -8745 -14060 -46375 -4725 -5330 -70485 -3125 -27785 105 mph 120 mpł **F**_a **Ibs** 5090 lbs ~<u>, 영</u> 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 115.04 79.75 93.03 71.17 61.49 61.49 55.25 115.04 58.80 55.88 55.25 93.03 65.42 58.80 57.01 55.88 79.75 65.42 57.01 Œ Œ 20 psf Dead Load Rise [T] Rise [T] £ £ 118 118 122 22 22 22 23 33 33 42 42 46 46 55 14 18 22 22 26 26 33 34 42 42 42 55 55

Table E-24 - 110 ft Span, Variable Wind Load.

Wind Load $K_{zt} = 1.0 \label{eq:Kzt}$ $K_d = 0.85 \label{eq:Kd}$ All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Notes: Table E-24 - 110 ft Span, Variable Wind Load. 20 psf Dead Load 20 psf Construction Load No Ground Snow Load

_	_																1														
		₹	in-lb	10185	10860	12290	31475	40785	50170	60550	77940	103545	130830	163925	221230			₊⊠	in-lb	13760	15210	17535	44475	57650	71115	86080	110865	147150	185910	232860	312490
		Σ	in-lb	-9310	-9190	-10945	-15935	-21825	-27785	-36430	-46375	-58650	-74660	-94205	-133640			M	in-lb	-13230	-12540	-13430	-22640	-30960	-38050	-45995	-59150	-82425	-104785	-132130	-187120
	170 mph	L e	lps	2090	3865	3225	2850	2615	2465	2370	2310	2275	2260	2250	2260		200 mph	Fa	lbs	2090	3865	3225	2850	2615	2465	2370	2310	2275	2260	2250	2260
		æ	lps	4785	3445	2690	2205	1865	1615	1420	1265	1130	1020	980	1120			χ,	lbs	4785	3445	2690	2205	1865	1615	1420	1265	1130	1020	1125	1315
		æ^	lps	1780	1800	1825	1855	1890	1930	1970	2020	2070	2130	2185	2265			R	lps	1780	1800	1825	1855	1890	1930	1970	2020	2070	2130	2185	2265
		₹	ql-ui	9820	9565	10735	27610	35775	43940	52960	68150	90580	114455	143430	194100			₹	in-lb	12505	13680	15695	39905	51710	63755	77110	99295	131830	166555	208640	280425
[v]		Σ	q-ui	-8150	-8190	-10325	-14060	-20365	-27785	-36430	-46375	-57700	-70485	-84795	-119780	[X]		M	in-lb	-11855	-11365	-12305	-20285	-27750	-34095	-41195	-52990	-74035	-94200	-118805	-168330
basic Wind speed [V]	160 mph	ъ.	lps	2090	3865	3225	2850	2615	2465	2370	2310	2275	2260	2250	2260	Basic Wind Speed [V]	190 mph	г	lbs	2090	3865	3225	2850	2615	2465	2370	2310	2275	2260	2250	2260
Dasic		æ×	lbs	4785	3445	2690	2205	1865	1615	1420	1265	1130	1020	1040	1175	Basic		A,	lbs	4785	3445	2690	2205	1865	1615	1420	1265	1130	1020	995	1165
		₽^	lps	1780	1800	1825	1855	1890	1930	1970	2020	2070	2130	2185	2265			R _v	lbs	1780	1800	1825	1855	1890	1930	1970	2020	2070	2130	2185	2265
		[†] ≥	ql-ui	9820	8440	10000	23980	31070	38090	45825	58955	78400	99075	124175	168610			ψ	in-lb	11310	12230	13945	35570	46095	56775	00989	88320	117295	148195	185660	250005
		Σ	in-lb	-7080	-7255	-9740	-14060	-20365	-27785	-36430	-46375	-57700	-70485	-84795	-113365			M	in-lb	-10545	-10245	-11605	-18050	-24705	-30345	-36645	-47240	-66125	-84160	-106165	-150505
	150 mph	ъ.	lps	2090	3865	3225	2850	2615	2465	2370	2310	2275	2260	2250	2260		180 mph	Fa	lbs	2090	3865	3225	2850	2615	2465	2370	2310	2275	2260	2250	2260
		æ×	lps	4785	3445	2690	2205	1865	1615	1420	1265	1130	1060	1100	1225			A,	lbs	4785	3445	2690	2205	1865	1615	1420	1265	1130	1020	925	1060
		œ	lps	1780	1800	1825	1855	1890	1930	1970	2020	2070	2130	2185	2265			A,	lps	1780	1800	1825	1855	1890	1930	1970	2020	2070	2130	2185	2265
	Rise [T] Radius [R]	(£		156.25	115.04	93.03	79.75	71.17	65.42	61.49	58.80	57.01	55.88	55.25	55.00		Radius [R]	(#)		156.25	115.04	93.03	79.75	71.17	65.42	61.49	58.80	57.01	55.88	55.25	55.00
	Rise [T]	(ft)		10	14	18	22	26	30	34	38	42	46	20	55		Rise [T]	(#)		10	14	18	22	56	30	34	38	42	46	20	55

-107025 -128675-107025 -124890-31570 -51110 -11080 -23655 -31570 -51110 -62885 -11425 -16860 -40695 -4755 -6810 -16860-40695 -76085 -90775 -7200 -8280 -23655 -62885 -76085 -90775 Έ 115 mph 140 mpt Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -124890 -23655 -31570 -40695 -51110 -62885 -107025 -11080 -16860 -31570 -40695 -51110 -90775 -107025 -124890-11080 -76085 -90775 -16860-7650 -23655 -62885 -76085 -4325 -6165 -6575 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 110 mph 130 mpł т_а г_а Notes: R_× ď I -62885 -107025 -124890 -16860 -31570 -51110 -90775 -107025 -124890-11080 -23655 -31570 -40695 -7065 -11080 -40695 -62885 -51110 -76085 -90775 -76085 -3920 -6350 -16860-5205 -23655 105 mph 120 mph lbs ஈ <u>த</u> R_x Sq R_x No Ground Snow Load Radius [R] Radius [R] 100.00 00.89 100.00 61.50 60.14 78.29 62.91 61.50 60.62 87.00 72.25 87.00 72.25 65.00 60.00 78.29 68.00 65.00 62.91 60.62 60.00 £ Œ Rise [T] Rise [T] £ £ 116 20 20 24 28 28 33 36 40 116 116 22 22 22 33 33 33 44 44 44 44 48 48 55 56 60

Table E-25 - 120 ft Span, Variable Wind Load.

20 psf Construction Load 20 psf Dead Load

-140095 -132050 -112780 -226905 -31570 -162095-52270 -90425 -184955 -23655 -40695 -64145 in-lb -15110 -16285 -27380 -33070 -66040 -11165 -13520 -19295 -51110 -80355 -99880 -44555 -15080 200 mph 170 mph **R**_x **Ibs** 4765 ж <u>Б</u> Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -125965 -143720 -204135 -23655 -31570 -40695 -51110 -62885 -76085 -118235 -13585 -15140 -24540 -29630 -46825 -59165 -81070 -101390 -166370 -10005 -90775 -16890 -39940 -13705All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 190 mph ъ lbs г_а R_x lbs Notes: R_x ď I -112560 -182530 -112110 -31570 -40695 -51110 -76085 -135945 in-lb -12140 -41660 -52645 -72320 -148730 -12080 -23655 -62885 -12400 -14305 -21845 -26365 -90580 -8955 -90775 -8315 -16860 -35555 150 mph 180 mph 2455 2465 lbs R_x lbs ж <u>Б</u> 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 100.00 120.50 100.00 65.00 156.00 120.50 87.00 72.25 61.50 78.29 68.00 62.91 60.62 78.29 68.00 62.91 60.62 60.14 87.00 72.25 65.00 61.50 60.14 60.00 Œ Œ 20 psf Dead Load Rise [T] Rise [T] (ft) £ 116 116 20 22 22 33 33 34 44 44 44 48 55 56 116 120 220 224 228 233 332 336 440 444 444 448 60 60

Table E-25 - 120 ft Span, Variable Wind Load.

-122570

-122570

-122570

65.21

-105170 -105170 -23410 -62045 -89320 -122570 -31245 -50495 -12120 -16655 -31245 -40250 -50495 -74965 -89320 -5515 -7150 -10865-16655-40250 -74965 -8405 -9030 -23410-62045Σ 115 mph 140 mpt lbs **R**_x **Ibs** 5595 Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -122570 -23410 -31245 -62045 -89320 -105170 -11390 -16655 -31245 -40250 -50495 -74965 -89320 -105170 -40250 -74965 -62045 -10865 -16655-50495 -8170 -23410-5005 -6870 -7180 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 110 mph 130 mpł г_а г_а Notes: R_× R_× a ď I -23410 -62045 -105170 -122570 -31245 -50495 -62045 -74965 -105170 -10865 -31245 -50495 -89320 -7480 -10865 -16655 -40250 -89320 -40250 -74965 -4520 -6600 -16655-6045 -23410105 mph 120 mph lbs F_a R_x Sq R_x 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 100.02 115.63 100.02 76.68 140.03 115.63 140.03 89.45 70.01 66.63 82.02 76.68 68.01 65.72 82.02 72.81 68.01 65.72 89.45 72.81 70.01 66.63 65.21 Œ £ 20 psf Dead Load Rise [T] Rise [T] £ £ 116 20 20 24 28 28 33 36 40 56 116 116 22 22 22 33 33 33 44 44 44 44 48 48 55 56 60

₹

Table E-26 - 130 ft Span, Variable Wind Load.

-134660 -109865-164560 -213885 -33400 -77970 -117340 -152725 -56370 -81695 -12880 -25110 -40250 -62045 -17510 -21140 -35615 -65930 -18235 -50495 -95975 -18735 -47310-14625 200 mph 170 mph 2825 2420 Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -121070 -74965 -11500 -23410 -31245 -50495 -62045 -105170 -135525 -15940 -17130 -20120 -31925 -50530 -59075 -73200 -98515 -147970 -192395 -89320 -13735 -17370 -40250 -15880-42420 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 190 mph г_а г_а 1390 **R**_x **Ibs** 5595 Notes: R_x ď I -108175 -172010 -31245 -50495 -74965 -105170 -128485 -15610 -19150 -28425 -44990 -52570 -65145 -132230 -10205 -12900 -23410 -40250 -62045 -14230 -14340 -37785 -87895 -89320 -9720 -16655 150 mph 180 mph lbs Table E-26 - 130 ft Span, Variable Wind Load. **R**_x **Ibs** 5595 ж <u>Б</u> 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 115.63 140.03 100.02 29.92 115.63 82.02 72.81 68.01 100.02 89.45 70.01 66.63 89.45 76.68 70.01 66.63 65.72 82.02 72.81 68.01 65.72 65.21 £ Œ 20 psf Dead Load Rise [T] Rise [T] (ft) £ 116 116 20 22 22 33 33 34 44 44 44 48 55 56 116 120 220 224 228 233 332 336 440 444 444 448 60 60

-61340 -103615 -103615 -120620-120620 -23190 -30970 -49975 -88105 -10215 -12885 -16580 -39875 -74030 -88105 -6340 -7550 -10860 -16455-39875 -74030 -9715 -23190-30970 -49975 -61340Έ Σ 140 mph 115 mph lbs 4890 3275 4890 Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -120620 -10645 -23190 -30970 -49975 -61340 -88105 -103615 -12030 -16455 -39875 -49975 -74030 -88105 -103615 -120620 -39875 -74030 -30970 -16455-23190 -61340-5745 -8285 -8955 -7195 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 110 mph 130 mpł г_а г Б Notes: R_× R_× ď I -120620 -103615 -23190 -61340 -88105 -103615 -16455 -49975 -74030 -120620 -10645 -30970 -49975 -7935 -11235 -39875 -88105 -39875 -74030 -30970 -5175 -6880 -164550969--23190-61340105 mph 120 mph lbs ஈ <u>த</u> 4890 4890 R_x Sq R_x 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 132.50 101.50 132.50 114.08 73.12 161.13 101.50 75.04 210.17 92.56 90.98 77.68 210.17 92.56 86.06 81.25 71.75 114.08 81.25 75.04 70.83 77.68 73.12 70.83 Œ £ 20 psf Dead Load Rise [T] Rise [T] £ £ 116 20 20 22 28 32 33 36 40 44 44 48 116 116 22 22 22 33 33 33 44 44 44 44 48 48 55 56 60

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Table E-27 - 140 ft Span, Variable Wind Load.

-131420 -191210 -81410 -158690-31870 -74030 -113130 -136375 -21410 -22975 -14745 -27100 -41895 -61340 -20160 -38380 -59335 -69820 -99225 -16070 -19465 -49975 -93300 -45195-14635170 mph 200 mph 2810 **R**_x **Ibs** 6485 Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -117855 -120620 -171940 -13135 -23750 -30970 -49975 -61340 -74030 -103615 -19535 -21695 -34420 -40510 -53210 -62590 -72955 -88925 -142680 -88105 -12890-14790 -18440 -39875 -18255All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 190 mpk ъ lbs г_а Notes: R_x
lbs R_x ď I -105165 -153665 -11620 -30970 -49975 -74030 -103615 -120620 -20550 -64935 -127495-13805 -23190 -61340 -16505 -17755 -30660 -47395 -55730 -79160 -39875 -88105 -11245-17480 -16450 -36070 150 mph 180 mph 2905 2880 2870 lbs **R**_x **Ibs** 6485 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 101.50 101.50 92.56 132.50 114.08 161.13 132.50 114.08 161.13 75.04 81.25 75.04 71.75 210.17 90.98 77.68 210.17 86.06 77.68 73.12 92.56 81.25 71.75 70.83 Œ Œ 20 psf Dead Load Rise [T] Rise [T] (ft) £ 116 116 20 22 22 33 33 34 44 44 44 48 55 56 116 120 220 224 228 233 332 336 440 444 444 448 60 60

Table E-27 - 140 ft Span, Variable Wind Load.

-110450 -110450 -118970-66835 -118970 -13240 -26720 -34990 -54985 -94515 -12235 -15525 -19505 -44395 -54985 -79990 -7430 -9650 -19505-44395 -79990 -11125 -26720 -34990 -66835 -94515 Έ Σ 115 mph 140 mpt lbs 4995 4995 Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -94515 -118970 -13240 -26720 -34990 -54985 -66835 -110450 -11020 -14535 -19505 -34990 -44395 -54985 -94515 -110450 -118970-79990 -79990 -19505-44395 -26720 -66835 -6775 -9260 -9225 All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 110 mph 130 mph г_а г Б 1950 Notes: R_× R_× a ď I -118970 -110450 -66835 -94515 -110450 -54985 -94515 -118970 -13240 -26720 -34990 -44395 -54985 -10085 -13620 -19505 -34990 -44395 -8840 -79990 -8110 -79990 -6150 -19505-26720 -66835 Σ 105 mph 120 mph lbs ஈ <u>த</u> 4995 R_x Sq R_x 2525 2560 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 108.75 165.25 165.25 138.84 121.17 93.01 207.89 138.84 121.17 108.75 84.14 79.08 99.72 87.96 81.25 77.49 99.72 87.96 81.25 77.49 93.01 84.14 79.08 76.88 £ Œ 20 psf Dead Load Rise [T] Rise [T] £ £ 18 22 22 26 26 33 33 34 42 42 46 50 50 58 18 22 22 26 26 33 33 44 42 46 50 50 60

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Table E-28 - 150 ft Span, Variable Wind Load.

-141880 -36730 -100490-108335-184900-120595 -131845-25040 -169385 -17435 -31560 -55575 -66835 **in-lb** -23010 -26910 -44660 -67625 -78750 -16515-18930 -22840 -47785 -79990 -23650 -52050 -89880 Σ 170 mph 200 mph R_x lbs 6400 4995 4095 1465 Wind Load $K_{zt} = 1.0$ $K_d = 0.85$ -127250 -118970 -166260 -15585 -27665 -34990 -44395 -54985 -79990 -110450 -22875 -25420 -40060 -60655 -70605 -80555 -97100 -152095 -94515 -21660 -66835 -21465 -46665 -14600All values normalized to $C_{\rm D}$ = 1.0. Use ASD design procedure only. Basic Wind Speed [V] Basic Wind Speed [V] 190 mph ъ lbs г_а Notes: R_x
lbs R_x ď I -113370 -148575 -13850 -110450 -118970 -18545 -71710 -86440 -135910-12805 -16585 -20545 -26720 -34990 -44395 -66835 -79990 -19390 -20820 -24095 -35690 -54040 -62885 -94515 -54985 -41560150 mph 180 mph lbs **R**_x **Ibs** 6400 1465 ж <u>Б</u> 20 psf Construction Load No Ground Snow Load Radius [R] Radius [R] 138.84 108.75 165.25 121.17 108.75 99.72 165.25 138.84 87.96 81.25 121.17 84.14 79.08 93.01 84.14 79.08 77.49 99.72 93.01 87.96 81.25 77.49 76.88 Œ £ 20 psf Dead Load Rise [T] Rise [T] (ft) £ 18 18 22 22 26 33 33 34 42 46 46 46 50 50 58 118 22 22 26 26 33 33 34 42 44 46 50 50 60

Table E-28 - 150 ft Span, Variable Wind Load.

Appendix F: Connection Tables

Appendix F displays tables showing design connection strength values of bolts for varying connection skew angles. Linear interpolation is allowed to find values for skews between those shown.

				- 19° Skew	O			
		Lamella	Shear Coni	nection Stre		s/bolt)		
Wood				Bolt Di	ameter			
Specific	1/4"	5/16"	3/8"	1/2"	5/8"	3/4"	7/8"	1"
Gravity [G]	-		-	,		-	-	
0.31	135	165	210	300	335	365	395	425
0.35	150	180	230	355	400	435	470	505
0.36	155	185	235	370	415	455	490	525
0.37	155	190	240	385	430	475	510	545
0.38	160	190	245	400	450	490	530	570
0.39	165	195	250	410	465	510	550	590
0.40	165	200	255	420	485	530	575	615
0.41	170	200	255	430	500	550	595	635
0.42	170	205	260	435	520	570	615	660
0.43	175	210	265	445	535	590	635	680
0.44	180	215	270	450	555	610	660	705
0.45	180	215	275	460	575	630	680	725
0.46	185	220	280	465	595	650	700	750
0.47	185	225	285	470	610	670	725	775
0.48	190	225	290	480	630	690	745	800
0.49	190	230	295	485	650	710	770	825
0.50	195	235	300	495	670	735	795	850
0.51	200	240	300	500	690	755	815	870
0.52	200	240	305	510	710	775	840	895
0.53	205	245	310	515	730	800	865	920
0.54	205	250	315	525	750	820	885	950
0.55	210	250	320	530	770	845	910	975
0.56	210	255	325	535	790	865	935	1000
0.57	215	260	330	545	805	890	960	1025
0.58	220	260	330	550	815	910	985	1050
0.67	240	290	370	610	905	1125	1215	1295
0.68	245	295	375	620	915	1145	1240	1325
0.71	250	305	385	640	945	1220	1320	1410
0.73	260	310	390	650	965	1270	1375	1470

				2 - 20° Skew	Ü			
		Lamella	Shear Con	nection Stre		s/bolt)		
Wood		ı		Bolt Di	ameter			
Specific	1/4"	5/16"	3/8"	1/2"	5/8"	3/4"	7/8"	1"
Gravity [G]	-			-				
0.31	135	165	210	305	345	375	405	435
0.35	150	180	230	365	410	450	485	520
0.36	155	185	235	380	425	470	505	540
0.37	155	190	240	395	445	485	525	565
0.38	160	190	245	405	460	505	545	585
0.39	165	195	250	410	480	525	570	605
0.40	165	200	255	420	500	545	590	630
0.41	170	200	255	430	515	565	610	655
0.42	170	205	260	435	535	585	635	675
0.43	175	210	265	445	555	605	655	700
0.44	180	215	270	450	570	625	675	725
0.45	180	215	275	460	590	645	700	750
0.46	185	220	280	465	610	670	720	770
0.47	185	225	285	470	630	690	745	795
0.48	190	225	290	480	650	710	770	820
0.49	190	230	295	485	670	735	790	845
0.50	195	235	300	495	690	755	815	870
0.51	200	240	300	500	710	775	840	895
0.52	200	240	305	510	730	800	865	925
0.53	205	245	310	515	750	820	890	950
0.54	205	250	315	525	770	845	910	975
0.55	210	250	320	530	785	865	935	1000
0.56	210	255	325	535	795	890	960	1030
0.57	215	260	330	545	805	915	985	1055
0.58	220	260	330	550	815	935	1010	1080
0.67	240	290	370	610	905	1155	1250	1335
0.68	245	295	375	620	915	1180	1275	1365
0.71	250	305	385	640	945	1255	1355	1450
0.73	260	310	390	650	965	1310	1415	1510

				3 - 21° Skew	Ü			
		Lamella	Shear Con	nection Stre		s/bolt)		
Wood				Bolt Di	ameter		1	T
Specific	1/4"	5/16"	3/8"	1/2"	5/8"	3/4"	7/8"	1"
Gravity [G]				,				_
0.31	135	165	210	315	355	390	420	450
0.35	150	180	230	375	420	465	500	535
0.36	155	185	235	390	440	480	520	555
0.37	155	190	240	395	460	500	540	580
0.38	160	190	245	405	475	520	565	605
0.39	165	195	250	410	495	540	585	625
0.40	165	200	255	420	515	560	610	650
0.41	170	200	255	430	530	585	630	675
0.42	170	205	260	435	550	605	650	695
0.43	175	210	265	445	570	625	675	720
0.44	180	215	270	450	590	645	700	745
0.45	180	215	275	460	610	665	720	770
0.46	185	220	280	465	630	690	745	795
0.47	185	225	285	470	650	710	770	820
0.48	190	225	290	480	670	735	790	845
0.49	190	230	295	485	690	755	815	875
0.50	195	235	300	495	710	780	840	900
0.51	200	240	300	500	730	800	865	925
0.52	200	240	305	510	750	825	890	950
0.53	205	245	310	515	765	845	915	980
0.54	205	250	315	525	775	870	940	1005
0.55	210	250	320	530	785	895	965	1030
0.56	210	255	325	535	795	915	990	1060
0.57	215	260	330	545	805	940	1015	1085
0.58	220	260	330	550	815	965	1045	1115
0.67	240	290	370	610	905	1190	1285	1375
0.68	245	295	375	620	915	1215	1315	1405
0.71	250	305	385	640	945	1295	1400	1495
0.73	260	310	390	650	965	1325	1455	1560

				1 - 22° Skew	0			
		Lamella	Shear Con	nection Stre		s/bolt)		
Wood		ı		Bolt Di	ameter			
Specific	1/4"	5/16"	3/8"	1/2"	5/8"	3/4"	7/8"	1"
Gravity [G]	-			-				
0.31	135	165	210	325	365	400	435	465
0.35	150	180	230	380	435	480	515	555
0.36	155	185	235	390	455	500	540	575
0.37	155	190	240	395	475	520	560	600
0.38	160	190	245	405	490	540	585	625
0.39	165	195	250	410	510	560	605	645
0.40	165	200	255	420	530	580	630	670
0.41	170	200	255	430	550	600	650	695
0.42	170	205	260	435	570	625	675	720
0.43	175	210	265	445	590	645	695	745
0.44	180	215	270	450	610	665	720	770
0.45	180	215	275	460	630	690	745	795
0.46	185	220	280	465	650	710	770	825
0.47	185	225	285	470	670	735	795	850
0.48	190	225	290	480	690	760	820	875
0.49	190	230	295	485	710	780	845	900
0.50	195	235	300	495	730	805	870	930
0.51	200	240	300	500	740	825	895	955
0.52	200	240	305	510	755	850	920	985
0.53	205	245	310	515	765	875	945	1010
0.54	205	250	315	525	775	900	970	1040
0.55	210	250	320	530	785	925	1000	1065
0.56	210	255	325	535	795	950	1025	1095
0.57	215	260	330	545	805	975	1050	1125
0.58	220	260	330	550	815	1000	1080	1150
0.67	240	290	370	610	905	1230	1330	1420
0.68	245	295	375	620	915	1255	1360	1450
0.71	250	305	385	640	945	1300	1445	1545
0.73	260	310	390	650	965	1325	1505	1610

			Table F-5	- 22.5° Ske	w Angle			
		Lamella	Shear Con		ngth [Z] (lbs	s/bolt)		
Wood		1		Bolt Di	ameter		1	T
Specific	1/4"	5/16"	3/8"	1/2"	5/8"	3/4"	7/8"	1"
Gravity [G]		-		,				
0.31	135	165	210	330	370	410	440	470
0.35	150	180	230	380	445	485	525	560
0.36	155	185	235	390	465	505	550	585
0.37	155	190	240	395	480	530	570	610
0.38	160	190	245	405	500	550	595	635
0.39	165	195	250	410	520	570	615	660
0.40	165	200	255	420	540	590	640	685
0.41	170	200	255	430	560	615	660	710
0.42	170	205	260	435	580	635	685	735
0.43	175	210	265	445	600	655	710	760
0.44	180	215	270	450	620	680	735	785
0.45	180	215	275	460	640	700	760	810
0.46	185	220	280	465	660	725	785	835
0.47	185	225	285	470	680	745	810	865
0.48	190	225	290	480	705	770	835	890
0.49	190	230	295	485	720	795	860	915
0.50	195	235	300	495	730	820	885	945
0.51	200	240	300	500	740	840	910	970
0.52	200	240	305	510	755	865	935	1000
0.53	205	245	310	515	765	890	960	1030
0.54	205	250	315	525	775	915	990	1055
0.55	210	250	320	530	785	940	1015	1085
0.56	210	255	325	535	795	965	1040	1115
0.57	215	260	330	545	805	990	1070	1145
0.58	220	260	330	550	815	1015	1095	1170
0.67	240	290	370	610	905	1245	1350	1445
0.68	245	295	375	620	915	1260	1380	1475
0.71	250	305	385	640	945	1300	1470	1575
0.73	260	310	390	650	965	1325	1530	1635